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INTERNAL MIXER INVESTIGATION FOR JT8D ENGINE JET NOISE REDUCTION

Volume II- Appendicies A, B, C, and D

A.B. Packman and D.C. Eiler



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16. Abstract				
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METRIC CONVERSION FACTORS

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APPENDIX D Hot/Cold Flow Model Tests to Determine Static Performance of 1/7-Scale JT8D Mixer Exhaust Nozzles.

APPENDIX A

TEST CONDITIONS FOR ACOUSTIC AND EXIT PROFILE TESTS

The operating condition for each test point of the configurations tested under the contract are included in this Appendix. The data are listed in the following order:

Configuration Name	Configuration Number
Basic Long Flowpath Mixer	1A
Reference Exhaust System	2Au \
Reference Exhaust System (tested at ideally mixed conditions)	2Am
Shallow Scalloped Long Flowpath Mixer	3A
Scalloped Long Flowpath Mixer	4A
Cutback Scalloped Long Flowpath Mixer	5A
Cutback Unscalloped Long Flowpath Mixer	6A
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Severe Cutback Short Flowpath Mixer with Engine Flow Simulatio	n 10A
Severe Cutback Short Flowpath Mixer	11A

NOTE: Unless noted otherwise, the long flowpath mixers were tested with a 16 inch engine extension, and the short flowpath mixer was tested without an engine extension.

The following nomenclature is used:

BPR -	Bypass ratio, fan flowrate divided by primary flowrate
Ca .	Speed of sound in the test chamber, ft/sec.
F_n -	Full scale JT8D engine thrust on an FAA day, lbs
P." -	Static pressure in the test chamber,
$ \begin{array}{ccc} C_{a} & - \\ F_{n} & - \\ P_{a} & - \\ P_{t}/P_{a} & - \\ \end{array} $	pressure
RH _a -	Relative humidity in the test chamber
Ta -	Temperature in the test chamber, °F
T _t ·	Nozzle total temperature, °R
$\begin{array}{ccc} RH_{a} & \cdot \\ T_{a} & \cdot \\ T_{t} & \cdot \\ V & \cdot \end{array}$	Jet velocity; ideally expanded to test chamber pressure, ft/sec.
V/Ca -	Ratio of jet velocity to test chamber speed of sound
w -	Flowrate, lbm/sec
Primary -	Primary or inner exhaust stream
Fan -	Fan or outer exhaust stream
Ideal Mix	The mass flow average value, e.g., Vmix = W _{pri} V _{pri} + W _{fan} V _{fan}
	W _{pri} + W _{fan}

The full scale JT8D engine thrust, F_n , was determined as follows:

- A. Reference System: The V_{pri}/C_a of each test point was related to the FAA day JT8D engine thrust by the curve of V_{pri}/C_avs. F_n, Figure A1, developed from full scale engine test data*. Since for low bypass exhausts the jet noise is essentially independent of small variations in the secondary flow conditions, this procedure allows the model data to be scaled using primary velocity to predict the full size engine noise.
- B. Mixer Nozzles: The $V_{\rm mix}/C_a$ of each test point was related to the FAA day JT8D engine thrust by the $V_{\rm mix}/C_a$ vs. F_n curve of Figure A-1, which was determined by calculating the $V_{\rm mix}/C_a$ for the engine operating line conditions by the formula defined in the no nenclature listed above*. For exhausts with internal mixer nozzles, the jet noise is a function of the calculated mixed velocity. This procedure allows small variations in bypass ratio and the resultant effect on mixed velocity to be accounted for correctly, thus, allowing the mixer nozzle data to be scaled properly to predict the effect of a mixer on JT8D engine noise.

The above procedure for relating JT8D engine noise with thrust is based on observations that jet noise is a direct function of the characteristic jet velocity divided by the ambient (test chamber) speed of sound, as shown in ref. 2. Since the model nozzles were tested at pressure, temperature, and density conditions simulating the JT8D engine, density effects on noise could be neglected.

*For these calculations, it was assumed that the installation of a mixer on the JT8D engine would not change the engine thrust-airflow relationship established for the baseline configuration.

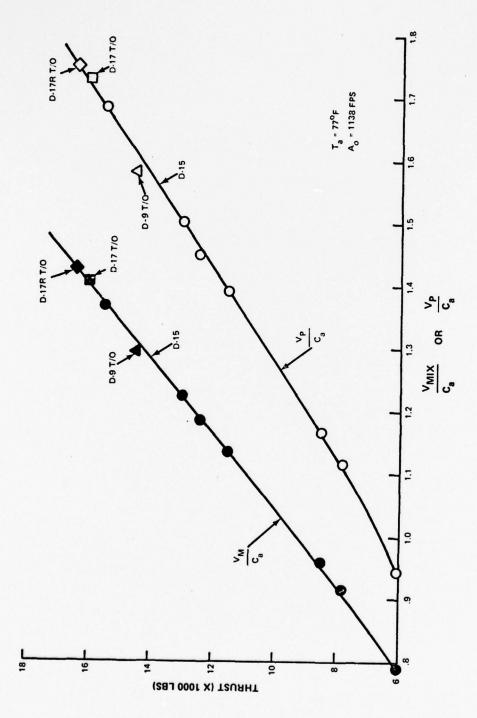


Figure A.1 Thrust vs 17_{Ca} for Reference and Mixed JT8D-15

TABLE 1A

BASIC LONG FLOWPATH MIXER; WITHOUT ENGINE SECONDARY FLOW SIMULATION; CONFIGURATION 1A

	F _n (5)	16500	16150	15650	13250	12600	11550	0006	8500	0069
	RH	30	30	31	29	28	38	27	36	36
Chambe	P ₃ (PSI)	14.8	14.8	14.8	14.8	14.8	14.8	14.8	14.8	14.8
Test	T ₃ P ₃ RH ₃ (°F) (PSI)	52	52	52	53	53	51	52	53	51
	V _{mix}	1.43	14.1	1.38	1.24	1.20	1.14	1.99	96.	98.
×	V _{mix} (FPS)	1,588.5	1,567.9	1,528.9	1,381.8	1,338.8	1,264.1	1,101.5	1,060.4	957.2
Ideal M	T _T mix Vmix (PPS)	1,141.9	1,129.1	1,100.5	1,035.6	1,015.3	963.2	939.8	914.0	891.7
	RA	1.101	1.112	1.143	1.188	1.203	1.445	1.201	1.153	1.261
	W (lb/sec)	3.423	3.425	3.433	3.225	3.153	3.390	2.646	2.557	2.402
Fan	$(^{\circ}R)$ (FPS) (1087.0				
	TT (°R)					667.4				
	티콥	2.009	1.995	1.963	1.789	1.741	1.740	1.483	1.446	1.368
	W (lb/sec)	3.109	3.081	3.005	2.715	2.622	2.345	2.203	2.218	1.905
,	V (FPS)	1969.	1947.	1808.	1694.	1635.	1546.	1347.	1288.	1158.
Primary	PT TT (°R)	1616.7	1606.7	1569.7	1462.2	1434.3	1410.4	1316.8	1249.9	1239.3
	티	2.153	2.135	2.080	1.858	1.798	1.696	1.527	1.504	1.390
	Condition	5310	5309	5308	5306	5305	5304	5303	5302	5301

TABLE 2A_u

REFERENCE EXHAUST SYSTEM WITHOUT ENGINE SECONDARY FLOW SIMULATION; CONFIGURATION $2A_{\mathbf{u}}$

	(B)	16500	16250	15700	14900	13400	12700	11900	8550	7850	6200
	RH _a	19	09	09	65	59	58	55	62	89	89
Chambe	P ₃	14.7	14.7	14.7	14.7	14.7	14.7	14.7	14.7	14.7	14.7
Tes	T ₃ P ₃ RH ₃ (°F) (PSI)	53	53	54	53	53	53	55	53	50	20
:	ات ا <mark>قا</mark>	1.76	1.74	1.70	1.64	1.53	1.48	1.42	1.17	1.12	96
×	Vmix (FPS)	1623	1602	1549	1508	1414	1365	1310	1086	1032	893
Ideal M	T _C mix Vmix (FPS)	1187.7	1179.3	1148.2	1120.2	1077.9	1048.4	1023.9	929.1	920.1	8.928
	BPR	.862	.874	924	.920	956	1.009	1.012	1.283	1.300	1.400
	W (lb/sec)	3.042	3.036	3.086	3.027	2.922	2.960	2.864	2.768	2.653	2.393
Fan	V (FPS)	1234	1221	1184	1169	1115	1089	1043	616	872	892
	PT TT V Pa (°R) (FPS)	700.4	7.769	8.789	681.1	673.9	664.5	647.1	628.7	617.1	599.2
	두 로	2.012	1.985	1.915	1.806	1.790	1.755	1.694	1.516	1.464	1.350
	W (lb/sec)	3.530	3.472	3,339	3.290	3.058	2.934	2.829	2.157	2.041	1.709
,	V W (FPS) (lb/sec)	1958	1935	1886	1819	1700	1644	1580	1301	1240	8901
Priman	PT TT (°R)				-		_	_	1314.6	_	-
	다리	2.150	2.113	2.057	1.993	1.862	1.807	1.743	1.486	1.434	1.313
	Condition	5510	5509	\$208	5507	5506	\$505\$	5504	5503	5502	5501

TABLE 24m

REFERENCE EXHAUST SYSTEM (TESTED AT IDEALLY MIXED CONDITIONS); WITHOUT ENGINE SECONDARY FLOW SIMULATION; CONFIGURATION 2Am

Į.		Primary	,		4	1	Fan	.		Ideal M	×i		Tes	t Chamb	er.	
Pa (°R) (FPS) (Ib/sec)	(*R) (FPS)	(FPS)	=1			(%) 	Pa (R) (FPS) ((lb/sec)	P	T mix vmix (PPS)	Vmix (FPS)	C ₃	(°F)	(°F) (PSI)	KH _a	(b)
1156.9		1608.	1	3.423	2.077	1166.8			0.853	1161.5	1615.8	1.45	09	14.3	36	16800
1154.8		1587.0	_	3.370	2.036	1154.2			0.854	1154.5	1590.7	1.42	09	14.3		16300
1.982 1126.9 1547.6		1547.6		3.367	1.983	1131.4	1551.8	2.802	0.832	1128.9	1549.5	1.39	09	14.4	34	15800
1091.0		1463.2		3.156	1.881	1090.8			0.874	1090.9	1469.4	1.32	52	14.4		14600
		1388.6		3.138	1.796	1035.4			0.841	1037.2	1388.7	1.27	41	14.4		13800
1020.0		1345.7		3.029	1.763	1016.4			0.878	1018.3	1350.7	1.23	41	14.4		13100
982.1	-	1269.4		2.915	1.689	993.9			0.874	987.6	1279.8	1.17	41	14.4		12100
925.8		1095.2		2.567	1494	917.6			988.0	922.0	1095.8	1.00	41	14.4		9200
900.2	150000	1034.4		2.447	1.446	901.2			0.878	2006	1039.0	95	42	14.4		8300

TABLE 3A

SHALLOW SCALLOPED LONG FLOWPATH MIXER; WITHOUT ENGINE SECONDARY FLOW SIMULATION; CONFIGURATION 3A

	F. (6)	650	16500	009	00	950	750	90	850	200	200
			16	15	15	13	12	12	00	00	9
H	RH	4	4	43	4	9	40	39	37	36	47
t Chambe	$\begin{pmatrix} T_a & P_a & RH_a \\ (^6F) & (PSI) \end{pmatrix}$	14.6	14.6	14.6	14.6	14.6	14.6	14.6	14.6	14.6	14.6
Tes			45								
	$\frac{V_{mix}}{C_a}$	1.44	1.43	1.38	1.34	1.28	1.21	1.17	86.	.94	18.
.≚	Vmix (FPS)	1591	1573	1523	1478	1411	1338	1286	1085	1837	889
Ideal M	T _{T mix} V _{mix} (PPS)	1150.9	1139.8	1111.6	1085.2	1054.5	1024.8	1004.1	923.6	926.1	863.2
	BPR		1.068	1.090	1.094	1.121	1.167	1.199	1.379	1.247	1.529
	W (lb/sec)	3.291	3.278	3.232	3.219	3.131	3.027	2.999	2.790	2.664	2.346
Fan	V (FPS)		1229	1196	1170	1134	1085	1042	931	885	782
	(°R)	705.5	699.4	686.2	678.5	6.089	665.4	647.0	635.2	622.9	609.7
	PT TT (°R)	2.014	1.997	1.946	1.900	1.818	1.744	1.693	1.524	1.473	1.356
		4	3.069	2.966	2.943	2.794	2.594	2.502	2.024	2.136	1.536
	V (FPS)	1957	1940	1877	1804	1722	1634	1578	1297.	1227.	1052
Primary	PI TT V W Pa (°R) (FPS) (lb/sec)	1617.2	1610.2								
	디즈	2.133	2.109	2.037	1.962	1.887	1.784	1.719	1.477	1.422	1.305
	Condition	5711	5710	\$709	5708	5707	\$706	5705	5704	5703	5712

TABLE 4A

DEEP SCALLOPED LONG FLOWPATH MIXER; WITHOUT ENGINE SECONDARY FLOW SIMULATION; CONFIGURATION 4A

	Fn	(lb)	16800	16650	16100	15100	13800	13100	12400	8850	8350	6350
her	RH ₃		2	63	62	62	63	63	49	65	19	99
t Cham	۳۵	(PSI)	14.5	14.5	14.5	14.5	14.5	14.5	14.5	14.5	14.5	14.5
Tes	Ta	(%)	47	47	47	47	47	47	48	48	48	84
	, xim	ړ	1.45	4.1	1.41	1.35	1.27	1.23	1.19	86.0	0.95	0.82
Mix	V mix	(FPS)	1601.3	1589.1	1557.9	1488.9	1399.2	1361.1	1316.9	1081.2	1046.7	904.5
Ideal	TT mix	(°R) (FPS)	1161.4	1159.8	1133.5	1098.2	1058.9	1046.0	1028.7	924.7	918.4	880.8
	BPR	1	0.958	0.959	0.980	1.004	1.044	1.025	1.118	1.281	1.256	1.415
	*	(lp/sec)	3.104	3.072	3.095	3.044	2.952	5.909	2.891	2.658	2.547	2.334
Fan	>	(FPS)	1230	1219	1213	1167	1111	1089	1053	806	880	191
	T_{T}	(°R)	7.769	694.8	696.5	681.3	668.3	664.8	647.7	616.2	617.1	613.8
		밉	2.001	1.982	1.963	1.885	1.792	1.755	1.711	1.513	1.472	1.366
	*	(lp/sec)	3.239	3.203	3.158	3.031	2.826	2.838	2.586	2.075	2.038	1.649
Primary	>	(FPS)	1957	1944	9681	1812	1700	1640	1612	1303	1256	1065
Prir	$T_{\rm T}$	(R)	1605.7	1605.7	1561.7	1516.7	1466.7	1436.7	1454.6	1319.8	1296.9	1258.6
	PT	먑	2.145	2.1.2	2.085	1.986	1.863	1.800	1.747	1.486	1.451	1.314
	Condition	Number	6120	6119	6118	6117	9119	6115	6114	6113	6112	11119

TABLE 5A

CUTBACK SCALLOPED LONG FLOWPATH MIXER; WITHOUT ENGINE SECONDARY FLOW SIMULATION; CONFIGURATION 5A

	Fn	(lb)	16650	16650	15800	14950	14750	13400	12600	9200	8500	6500
ē												
Chamb	₽.	(PSI)	14.9	15.0	15.0	15.0	14.9	14.9	14.9	14.9	14.9	14.9
Test	⊢ °	(%F)	51 14.9 18	51	51	51	29	29	50	32	32	34
	, Xim	C _a	44.	1.44	1.39	1.34	1.30	1.25	1.20	00.1	96.0	0.83
4ix	/ mix	(FPS)	1598.9	1596.8	1545.0	1483.4	1407.9	1356.5	1303.0	1086.5	1039.3	900.4
Ideal N	T _T mix	(°R) (FPS)	1159.2									
	BPR		0.984	1.00.1	1.000	1.013	1.041	1.059	1.070	1.273	1.282	1.40's
		(lp/sec)										
an	^	(FPS)	1232	1241	1206	1159	1112	1085	1046	816	882	782
4	T_{T}	اگا	2.018 694.2 1232	9.102	9.689	6.579	664.0	657.8	650.7	625.7	619.4	603.0
	$^{P}_{\mathrm{T}}$	[조]	2.018	2.023	1.96.1	1.878	1.801	1.756	1.695	1.517	1.471	1.361
	*	(lb/sec)	3.282	3.266	3.210	3.080	2.829	2.714	3.714	2.135	2.043	1.701
1a y	-	(FPS)	1960	1953	1884	1812	1716	1644	1578	1301	1241	1067
Prin	T	(R)	0961 19191 0	1608.7	1557.7	1524.7	1480.7	1433.9	1408.1	1318.5	1291.4	1270.6
	PT	Pa	2.140	2.136	2.067	1.976	1.872	1.806	1.739	:.485	1.440	1.310
	Condition	Number	6510	6059	*8059	6507	9059	6 205	6504	6503	6502	1059

TABLE 6A

CUTBACK NON-SCALLOPED LONG FLOWPATH MIXER; WITHOUT SECONDARY FLOW SIMULATION; CONFIGURATION 6A

	Fn	<u>a</u>	17000	16800	16100	15000	13800	13100	12400	0006	8350	6350
)er	RH	' !	22	;;	22	21	77	51	21	21	17	50
t Chaml	~ e	(PSI)	14.9	14.9	14.9	14.9	14.9	14.9	14.9	14.9	14.9	14.9
Tes	⊢ _e	(F)	39	38	39	39	30	39	36	39	39	39
	V mix	ا"ر	1.46	1.45	1.41	1.34	1.27	1.23	1.19	66'0	0.95	0.82
Mix	Vmix	(FPS)	1601.1									
Ideal	T _{T mix}	(°R)	1162.6									
	BPR	1	0.978	0.087	1.020	1.007	1.076	1.102	1.121	1.327	1.276	1.426
	*	(lp/sec)	3.197	3.215	3.226	3.084	3.045	3.042	2.979	2.766	2.592	2.408
an	>	(FPS)	1230	1229	1207	1157	1112	1085	1052	416	877	178
ī	$T_{\rm T}$	(°R) (FPS)	697.2	698.0	6.169	678.1	668.1	8.799	649.6	624.0	619.4	9.109
	$^{P}_{T}$	립	1.999	1.997	1.965	1.870	1.791	1.754	1.708	1.521	1.464	1.360
	*	(lp/sec)	3.269									
ıary	^	(FPS)	1964	154	1888	1781	1694	1634	1588	1307	1251	1067
Prin	T_T	<u></u>	1617.7	1604.7	1570.7	1477.7	1466.7	1426.3	1419.1	1333.7	1308.7	1264.2
	P _T	옵	2.142	2.130	2.073	1.975	1.858	1.797	1.743	1.483	1.441	1.314
	Condition	Number	6310	6309	6308	6307	9069	6305	6304	6303	6302	1069

TABLE 7A

DEEP SCALLOPED LONG FLOWPATH MIXER; WITH ENGINE SECONDARY FLOW SIMULATION; CONFIGURATION 7A

	F	(lb)	16550	16800	16300	15800	14250	13600	12900	9350	8700	6650
	1		٥	6	∞	œ	07	7	7	7	S	4
Cham	~ °	(PSI)	14.5	14.5	14.5	14.5	14.5	14.6	14.6	14.6	14.7	14.7
Test	T	(°F) (PSI)	6	46	25	20	61	61	61	18	35	35
	' . <u>×</u>	"	1 4	1.45	1.42	1.39	1.30	1.26	57	1.01	760	3.84
×	/ mix	(FPS)	593.6	602.3	530.8	496.9	397.7	348.2	306.4	085.7	061.5	916.2 (
Ideal M			1171.5 1593.6 14	182.1	111.9	114.4	366.3	36.8	122.5	1 122.1	108.3	373.1
	1	ı	0.966									
		1	0.9	0.9	1.0	0.9	1.0	0:	0.	<u></u>	1.3	4
		(lb/sec)		3.068	3.093	2.932	2.835	2.874	2.806	2.725	2.628	2.361
an an	>	(FPS)	697.5 1224.4	1215.0	1193.9	1172.1	1108.3	6.1801	1052.7.	935.9	908.3	800.1
<u>L</u>	T_{T}	(R)	697.5	0.169	682.7	687.7	6.829	664.8	6989	627.8	620.5	599.0
	L	[조	2.000									
		(lb/sec)			3.052							
ary	^	(FPS)	1950.3	1980.3	1872.0	1804.5	8.1691	1635.7	1585.3	1290.4	1264.9	1086.4
Prim	T_{T}	ا گ	1629.4 1950.3	1661.4	1546.7	1518.4	1460.0	1438.5	1424.0	1324.4	1290.1	1275.1
1		Pa										
	Condition	Number	7510	7509	7508*	7507	7506	7505	7504	7503	7502	1051

TABLE 8A

CUTBACK SCALLOPED LONG FLOWPATH MIXER; WITH 7.6" ENGINE EXTENSION; WITHOUT ENGINE SECONDARY FLOW SIMULATION; CONFIGURATION 8A

	E.	e	16800	16800	16000	15300	13950	13250	12400	8200	6750
ber	RH ^a	1	38						20		
t Cham	~.	<u>S</u>	14.5	14.5	14.5	14.5	14.5	14.5	14.5	14.5	14.5
Tes	⊢ <u>"</u>	(F) (PSI)	46	46	46	45	43	42	42	42	43
	N X	ا"ی	1.45	1.45	1.40	1.36	1.28	1.24	1.19	96.0	0.85
Mix	Vmix	(FPS)	1600.8	1602.9	1547.2	1494.1	1408.1	1359.2	1310.2	1058.0	930.3
Ideal	T _T mix	(°R) (FPS)	1158.7	1170.9	1129.8	1108.3	1073.5	1053.6	1028.4	956.1	947.3
	BPR	1	0.944	0.934	0.964	096.0	896.0	0.987	0.50	1.063	0.958
	≩	(lb/sec)	3.054								
Fan	>	(°R) (FPS)	1230.3	1225.4	1201.3	1165.2	1108.6	1079.8	1044.4	880.3	7.677
	T	E	693.0	696.5	685.2	678.4	667.7	660.3	648.8	620.3	607.3
	4		2.015								
	≩	(lp/sec)	3.236	3.215	3.125	3.037	2.875	1.77.2	5.699	2.105	1.797
lary	>	(FPS)	1950.5								
Prin	Τ,	3	1598.4	16140	1558.4	1521.0	1466.3	1441.7	1404.3	1313.0	1273.1
	4	Pa	2.141	2.133	2.060	1.980	1.859	1.788	1.735	1.435	1.315
	Condition	Number	7610	6092	8092	1607	9092	7605	7604	7602	1092

TABLE 10A

SEVERE CUTBACK SHORT FLOWPATH MIXER; WITH ENGINE SECONDARY FLOW SIMULATION; CONFIGURATION 10A

		(lp)	_	16500	15600	15000	13400	12750	11900	8700	7850	0009
per	RH	(PSI)	12	21	53	21	5	5	5	20	20	20
t Cham	4	(PSI)	14.5	14.5	14.5	14.5	14.5	14.5	14.5	14.5	14.5	14.5
Tes	Ta	(°F)	2	2	\$	63	63	63	63	9	65	9
		ر _a		1.43	1.38	1.34	1.25	1.21	1.16	0.97	0.92	62.0
Mix	V _{mix}	(FPS)	1610.4	1599.8	1551.8	1498.6	1405.5	1355.8	1300.6	1094.7	1031.9	891.9
Ideal	Trmix	(%R)	1173.5	1173.1	1141.5	1111.9	1072.9	1041.0	1015.8	947.8	924.1	871.1
	BPR	1	0.903	0.899	0.938	0.938	0.962	1.014	7.041	1.187	1.189	1.360
	*	(lb/sec)	2.922	2.883	2.916	2.843	2.729	2.765	2.714	2.474	2.358	2.221
an	^	(FPS)	1231.8	1220.5	1197.2	1158.9	1101.7	1080.4	1042.7	916.6	865.2	773.5
	$T_{\rm T}$	(°R)	0.669	697.4	1.789	677.1	0.799	657.5	641.9	631.1	616.3	6965
	PT	Pa	2.008	1.982	1.948	1.880	1.776	1.748	1.693	1.509	1.45	1.357
		(lp/sec)	3.235	3.206	3.108	3.031	2.837	2.725	2.608	2.084	1.982	1.633
пагу	>	(^R) (FPS)	1952.3	1940.8	1884.5	1817.2	1697.8	1635.1	1569.0	1306.0	1230.1	1052.9
Prin	$\mathbf{I}_{\mathbf{I}}$	8	1602.0	1600.7	1567.7	1519.7	1463.3	1429.9	1398.7	1323.7	1290.1	1244.1
	PT	B	2.144	2.123	2.059	1.995	1.860	1.798	1.734	1.486	1.430	1.309
	Condition	Number	7810	7809	₹808	7807	9082	7805	7804	7803	7802	1087

TABLE 11A

SEVERE CUTBACK SHORT FLOWPATH MIXER; WITHOUT ENGINE SECONDARY FLOW SIMULATION; CONFIGURATION 11A

		(g)	-	17000	C0191	15300	13800	13100
er	RH	•	81	18	18	81	18	61
Chamb	P.	(PSI)	14.6	14.6	14.6	14.6	14.6	14.6
Test	T,	(°F) (PSI)	62	19	59	38	57	99
	, air	ۍ	.46	1.46	1.41	1.36	1.27	1.23
Mix	V mix	(FPS)	1635.5	1629.4	1572.2	1515.3	1417.5	1374.5
Ideal	T	(°R) (FPS)	1204.0	1205.1	1156.0	1129.1	1080.2	1061.5
	BPR	1	0.836	0.850	0.895	868.0	0.952	0.958
	×	(lb/sec)	2.864	2.851	2.929	2.841	2.779	2.728
		(FPS)						
	T_{T}	(R)	704.0	704.1	692.3	683.3	669.3	663.3
	P	Pa			1.974			
	A	(lp/sec)	3.424	3.356	3.272	3.165	2.921	2.846
yary	^	(FPS)	1968.6	1966.6	1893.3	1821.8	1705.8	1648.3
Prin	T_{T}	(R) (FPS)	1622.0	1631.0	1571.0	1529.4	1471.4	1443.0
		2						
	Condition	Number	7910	4064	€8064	1907	9062	2062

APPENDIX B

ACOUSTIC DATA, MEASURED MODEL AND SCALED TO JT8D ENGINE

This appendix contains the one-third octave band acoustic data obtained on the contract configurations. Two types of data are presented,

a.) Measured Model Data: The as-measured data were corrected for microphone and cable frequency response calibrations, and are presented for the microphone distance of 15 ft. used in the test. Since the test chamber is anechoic, the data are free-field and are not contaminated by sound wave reflections.

Atmospheric absorption corrections were used to analytically correct data from test chamber conditions to FAA day (77°F, 70% relative humidity). These corrections were obtained from the formulas of ref. (3). The SPL values from 100 to 80,000 Hz were integrated to provide Overall Sound Pressure Levels (OSPL) for each angle.

b.) Data Scaled To Predict Full Size JT8D Engine: In order to scale model data to predict full size engine noise, it is necessary first to eliminate the effects of atmospheric absorption on the measured noise spectra. This is done by determining the reduction of SPL levels due to the atmospheric absorption present during the model test, and adding these values to the measured levels, thus producing the true source noise spectra. Since the data were corrected to reflect the atmospheric absorption that would occur on an FAA day to present the model data on a consistent basis in step a.), the additional correction to obtain the true source spectra was added to the FAA day model data. Then the data were scaled to JT8D engine size (7 times linear model dimensions) by reducing the frequency values by a factor of 7.0 and by adding the factor 20 log(7) to the SPL level of each third-octave band. The scaled data then were extrapolated to 1200 ft. linear distance by subtracting 20 $\log \frac{1200}{15 \sin \theta}$ to account for spherical divergence. Atmospheric attenuation corrections necessary to present the full scale data on an FAA day were applied per ref. 3. The Overall Sound Pressure Level (OASPL) and Perceived Noise Levels (PNL) for each angle are listed at the bottom of the SPL values for each angle. Also listed are the PNL values for 400, 2000, 4000, and 6000 foot linear distances.

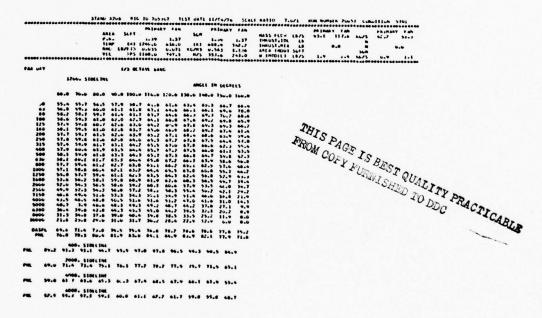
The heading information at the top of the data sheets contains pertinent nozzle information in both U.S. customary units as well as the International System of Units (S.I.). However, a more complete listing of conditions is presented in Appendix A. The identification number for each data point is found in the top right hand corner of each data table listed after condition.

To facilitate use of the acoustic data, the sheets have been assembled by increasing condition numbers. Thus, to locate the data for a particular configuration and operating conditions, first find the condition number from the tables of Appendix A.

a.) Model Data Measured At 15 Ft. Radius

							RIMARI	-		PR 1	MARY	***********			1 140			
									SUR				10/1	0.0			M	
				P			. 30	1.37		4.5	1.37	IMAUST, 164		•.•		***		
				11.00		. 154			141		14.7	Imust, ata			0.0	•		
				-		1) 0.			.CM)		1.130	APEA INUCI						
	000000		20300	ALL		PS 11m	0.0	747.1	N/S	353.0				1.4		**		
****	*****	*****	*****	******	****	*****	*****	*****	*****		*****	W (MOLL)				46/3	U.V	1.1
					V	MITAN		MUDE L	Jt I	-	DATA LD.	LOT RADIUS					- 1814	
**																376	- 141.04	.,
W							-10	I COM HOM	-	15 IR	2164113							
	••	10	••	**	100	110	140	130	140	130	140							-
				AND D														
030	•.•	0.0	0.0				٠.٠				4.6							14
	0.0	0.0	0.6	6.6		0.0	0.6			6.6	6.5							:
44		0.0		0.0	0.0	6.0	0.0		6.0		4.0							•
123	72.0	14.4	75.4	10.0	14.7		77.4				44.2							1-1
140	73.0	74.1		10.7	74.0		77.7		•).•									100
~	71.4	73.3	15.3	75.7	****	?>	****											
~	14.4	75.0	P .3	77.6	17.0		*1.4				*1.0							1.0
	****	77.5		70.6	70.4					•7.0	**.0							100
	P. 1			14.7	60. I			87.0		**.1	*1.4							100
-			*1.1					44.7		*1.0								11.
-	.1.)			4).1		**.*	•1.			*7.4								44.
~~			42.7		***			*1.4			*7.1							
								*1.0			*7.4							
						67.1		•/										
	80.7	.7.6						•2.5										
.60	80.4							*1.*			44.0							***
.~	81.4		84.0		**.5	80.2		***										
.1>	41.7	37.7			****	44.5		vi										
.00	.1.9		****	02.7		**.*	*				45.0							
				45.4			***			**.)	**							.1.
. %				45.3			****				43.7							
.~	80.7		*3.2					14.1		****	•2.•							
		01. 1		64.1							41.3							
	77.0	84.2		43.1	44.5			**.*		11.3	74.7							
0	70.1	19.3		W.1	63.5			12.3			77.4							1.4
v.0	70. 3	11.1		P1.4	41.4	63.4				17.2	57.5							
	15.2	70.7	70.1							75.7	70.7							
1.3	11.1	74.7	76.3	77.4	74.4			79.3			****							1
	71	72.0	74.7	79.4	77.7	70.0	79.1			71.1								1
		76.4	77.4	73.0	74.0	73.0	74.3		72.4									**
	60.7	47.4		70.5	71.7	71.0	11.6			67.3								**
	.7.		4.60															•
			0.0	4.0	0.0		4.6				0.6							
								-										
																	LAFOL .	

b.) Model Data Scaled To Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

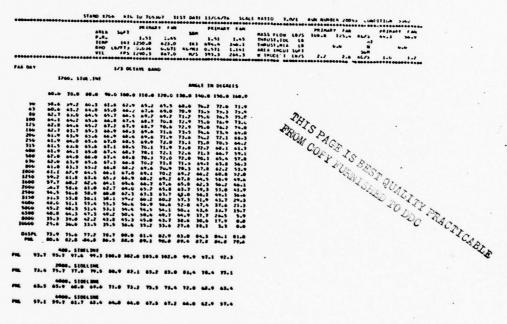


Configuration 1A; Condition 5302

a.) Model Data Measured At 15 Ft. Radius

Mail								-			201	MAY FA	•			PRIMARY	-		PRIRMI	
									TEN SYLE	500			MAS		10/5	0.0		46/1		
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## 179 CTAN BAND MORE JET BOISE DATA 15.0FT RADIUS RAW CORRECTED SPL - INCOME ### COR							11230													
### 175 CLIAN BAND MORE ATT 15.0FT SADIUS ### COMMICTING SPL - (Micro Micro Mi														. (400)	100			344		
1/5 CLIAN DAND DODE: JET NOTICE DATA 13.0FT DADIUS RAW (CRACCITY SPL - INCAN	-						1340	••	147.0	~	393.	700.		# COLT 1	10/5	2.1	2.0		4.0	
### ### ### ### ### ### ### ### ### ##							*****	*****	•	****		*****	••••••	*****	-	********	*****			•••
## 151 61 70 60 60 60 60 60 60 60							~													
Mile Silv No. 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0							~ 144	-	-	261	-0176			Plus		RAW CO	MALCIE	U SPL	- INLE	
NAT																				
\$\frac{1}{2}\$ \$\begin{array}{cccccccccccccccccccccccccccccccccccc			74		-	160														
13						•••	•••					100								44
83	16		0.0	0.0	0.0	6.0	0.0		0.0	0.0										
10		4.6		0.0																
80		4.6	0.0	0.0	4.6															
12				77.5																
60 Tel. 7 7 6 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7					70.3															
60 71.5 77.6 77.2 80.3 81.7 81.7 81.7 81.7 81.7 81.7 81.7 81.7				19.0	P	70.7	10.7	80.9												
10				79.2	80.5	61.5			84.7											
No.				80.1	41.7		41.7		44.4	91.0	91.5	99.3								
Mar.							47.0			*5.4	**.4	102.7								•
M				85.4		67.6	44.4	.1.5	94.5		162.7	104.7								
10							41.0		**.	100.2	101.9	105.0								:
60								*3.5	94.7	100.4	104.5	44.1								:
25																				·
88											101.0	100.5								i
Mar.												97.8								- 1
90 90 2 80 2 81 2 81 2 89 3 80 2 89 2 89 2 89 2 89 2 89 2 89 2 89 2											**.*	96.6								i
15																				i
No.																				
66 Bart B																				ī
No. 50-46 - 60-7 - 67-76 - 69-7 - 51-1 - 57-7 - 60-10 - 69-10																				i
66 Sec. Sec. 1 87.1 6.2 6.2 70.7 77.5 70.6 70.5 71.1 87.2 6.2 11.1 87.2 6.2 6.2 70.7 77.5 70.6 70.5 71.1 87.2 6.2 6.2 70.7 77.5 70.6 70.5 71.1 87.2 6.2 6.2 70.7 70.7 70.7 70.7 70.7 70.7 70.7 70																				
10 11.0 15.1 16.2 16.2 16.2 16.2 17.1 17.1 17.1 17.2 17.2 17.2 17.2 17																				
30 0.77 0.05 0.05 0.77 0.05 0.70 0.75 0.75																				
0																				
0																				
16 19.7 01.7 01.7 00.3 09.9 02.1 00.3 00.9 09.3 01.0 17.2 17.2 17.3 17.3 17.3 17.3 17.3 17.3 17.3 17.3																				
35 37.7 79.1 81.3 82.7 86.3 85.2 86.6 85.2 87.6 71.6 71.5 4.4 5.7 71.6 71.6 71.6 71.6 71.6 71.6 71.6 71																				
06 15.7 17.3 70.2 00.0 02.5 03.5 00.0 02.5 00.5 17.5 27.5 00.5 17.5 17.5 07.6 07.7 17.5 17.5 17.5 17.5 17.5 17.5 17.5 1																				
0 71.1 70.0 70.0 17.2 79.7 00.0 01.5 00.0 77.0 70.0 71.6 71.6 0 70.2 71.0 72.5 75.0 70.0 70.0 70.0 70.1 75.1 75.3 72.3 0 60.4 67.1 70.0 71.0 73.3 71.0 70.7 71.0 72.7 72.5 72.5		75.7	77.3																	
.0 70.2 71.6 73.5 75.0 70.0 70.0 70.7 70.0 75.1 73.3 77.3 .0 66.4 67.1 70.0 71.0 73.3 71.0 76.7 73.2 72.5 71.4 71.7		71.1	74.4																	•
·0 44.4 67.1 70.0 71.0 73.3 71.0 76.7 73.2 77.5 71.0 71.7			71.6	73.5	75.0	70.0														10
						73.3	11.0													
			6.6	0.6	0.0	0.0		0.0												

b.) Model Data Scaled To Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

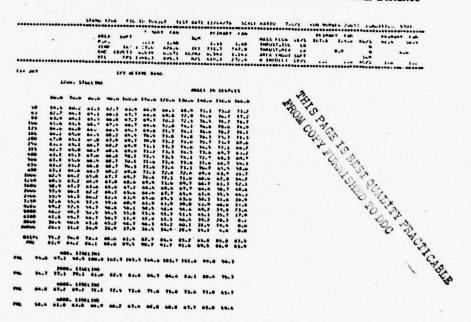


Cox.figuration 1A; Condition 5303

a.) Model Data Measured At 15 Ft. Radius

				444	1 500		-	,	Len	-			155 FLOW					PE IRAL	
								1.40	-	1.5	1.4		AUST. DOL		•	•.6		6.6	
				11.00		1 1317				731.	1 347.		AUST . AT				- :		
				AML		3 0.0		71		0. 54			164 (MOD)	1041					••
				VEL		5 1344		1.00	4/5	410.	212.		(MODEL)	LO/S	2.2	2.4	46/3	1.0	1.
							*****	•	******		******	***	*******	*****	******	******	******	*******	
					1/1	DC TAME	-												
													TAD In?		RAW C	DRRICE	IL MI		.,
							B)(02100	-		4318430								
-41	.0	Pe	0.0	43	100		120	134	100	120	144								Pu
										•									
30		0.0	4.0	0.0	0.0				0.0		0.0								
		6.0	4.6	0.6	6.6		0.0	0.0	6.0	0.0	6.0								
*	6.6	0.0	0.0	0.0	0.0			0.0	0.0	0.0	0.0								
25	70.4	70.0	70.4	N	10.4				83.5	49.1	94.5								
2	16.6	17.0	14.2	14.1	80.6	.1.4			.7.	*0.8	**.0								::
	70.7		80.0	07.4	79.7					**.1	96.7								
~		**.1	41.1	62.0							**.5				-				- 17
13			***	61.0	83.4	**.7				**.*									- 11
•	**.1	85.0	***		88.7	90.0				100.6									
		.*	*		47.1	11.3	*/. }		**.										-
30			44.4			*1.7	99.7		101.7	105.4	107.5								1.
no	80.7	67.4				*1.7		****	102.0	100.7	100.1								
			**.1		41.1				101.8	103.0	107.5								
25		.7.7				91.0		***	101.2	107.0									
									100.7										
-			44.7	40.2		+1.0	**.		**.7										
30			44.4	40.1	*2.2	94.4				*7.1	*1.4								
45		87.0	89.2	90.6	92.4			**.4			92.0								
•••				*1.1	42.0	**.*	*1.2		97.0		47.7								
10			**			****		97. >		**.1	91.0								•
*	47.2								94.2		90.0								**
				w		*4.7		**.*			90.1								*
::		****	04.1			* 3. *	**. *	***											- 11
::	M.*		**	44.5		*7.0	44.1	4.1		90. 9	07.0								**
			84.5				*3.7	*).1	*1.3	84.2	84.4								**
::			***	***		****		***	07.6		41.7								ii
::	74.7			4.1		••. :	90.1				19.5								
		10.1		22.3		44.7			84.3		10.7								
.0	14.7	70.0	70.0				03.0		.7.6	70.7	14.7								1-
	****	74.0	74.0			10.1	74.5	70.7		20.,	73.>								
	.1.2	64.6	11.1	1/.4		75.7			73.7	?:·:	74-1								
•.		0.0	0.0	0.0		•.•				1	73.5								
				0.00	-														

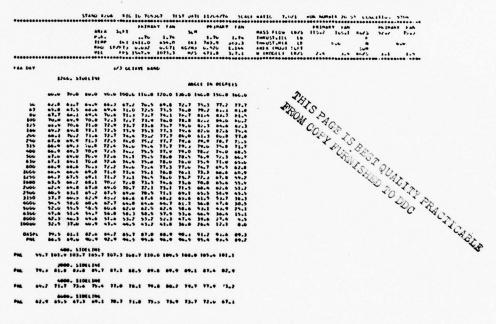
b.) Model Data Scaled To Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

				4844			INT	-			-	AM				IMMAY			PRIRARY	-
				P. A.			.70		Sum				MASS PLO				0.0	4	0.0	
				11.00		1 141		1.74		743.			THEUST. I				-			-
				AND				0.071	46/73				ARLA IND			•		340		
				WIL		1 194	1	673. 5	~	471.	1 12		# 4800fs	1 14/		1.4	1.4		1.1	1.7
	*****	*****	*****		*****	*****	*****		*****	•		****				****			****	••••
					1/3	CLIAM	BAND	wood r	161	10174	DATA	15.0	I RADIUS		•		MELLIE	U SPE	- IMUDE	
AMD I DT L H																				
4 2	***	N	••	16	100	110	120	RUPHDM 130	100	130	100									•
			-	**	•••	***	170	230	100	120	100									14 -1
450				0.0	0.0		6.0	6.0	0.0											
			6.6				6.6	0.0		6.6										
	0.0		6.6	0.6	6.6	6.6		0.0	0.6		0,0									
ILL	70.0	80.5	67.3	65.4	43.2						98.2									110
143			84.7	15.1																
100		07.7		63.0	07.0						100.0									444
100				***						100.8										
***				**.)			**.			107.5										
114	40.3	**.3			****					104.0										441
~	w	87.7		*1.0	****	94.3	90.1	100.1		106.5										
410	97.7	94.0	92.3	93.7		***		102.6												1.
b n0		94.5		w.1	95.2			102.0												443
	.2.0	*1.*		w				103.5												120
.45	90.4	92.5	93.4	***	**			103.0												
	w.7	9/.1						103.7												12
.00	w.1	41.7	+2.0	**.7				103.7												1,,
.26	2.00	*1.0	*1.0	***	**. 3			163.5												144
-15	w	41.0	2	.7	**.*		161.0	101.1	163.5	101.5	100.2									
		*2.1		45.0	**.*			163.1												1.1
	*1.1		*1.4	***	***			107.5												1/1
	41.1	*7.7		m.0	***			107.7												
		*1.4	**.*		**.*			101.0												1.1
	**.*	96.7			**.5			101.1												14
		.1				*7.5		100.2	**											
		****			**.7		*7.		****											* **
	00.7	00.7	49.7		43.3		**.													***
1.5	84.3				*1. *															1
	42.5	84.4		***	46.2		47.4													117
0.0	4.64				.7.4		90.													
	77.1	70.0		4.54	24.5		80.4	85.3	83.4											-
4.6	72.1	1			41.7		42.1	t1.t	86.2	74.0										1
	0.0		0.0	0.0	0.0		0.0	0.0												
																			-	4.3

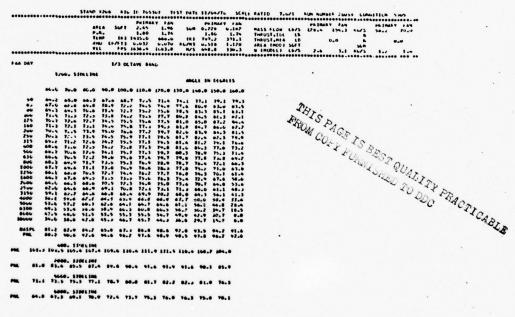
b.) Model Data Scaled To Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

••••							-			PEI		•			PRIMA			PR IRAGE	
				484					100		0.0			10/5	0.0				
							. 00	1.74		1.0		THE	UST. 8 DE						
				***		11 141				701.) m.	1 1100	UST . REA						
				8 1		1)		6.676	REM?								2.0		
				460		PS 103	•••	1043.0	~s	***			HODEL)	10/5	2.6	2.4	14/1	1.4	1.0
		*****				•••••	•••••	****	*****	***		******	•	*****	******	******	*****	*******	
-					.,,			MUDEL	361		-	POLL BW	DINI			CORRICT	IU are	- IMU	
tam!	•0	10	-	**	100	4.0	120		100		DIGREES								-
			•		•••	•••	120			134	100								44 -4
. 650	6.6			0.0	0.0						0.0								
			6.0							:::									
	4.6	0.0				0.0					0.0								
. 100	74.4				84.4				96.5		***								
.125	41.1								*7.0	v5.4	**.*								***
	***		85.0								M.1.0								117
	03.7							97.4		bu2.4									1:5
		83.4		.7.0				****											
. 345			47.7			**.3		1.1	142.3		114-1								***
-	04.5		91.7			94. 1		101.7	164.1	114.4	110. 1								441
. ***	.1.0	*1.1	42.4	11.2	w.1	97.0		1.501	140.5	112.5	111.9								1/2
	****	*>	**	49.1	45.7	47.4	14.	100.0	140.0	111.9	111.3								177
	****				*1.6	97.5	144.2	164.0	110.4	114.0	115.4								100
				****	*7.4	**.4	167.1	149.7	110.2	113.9	113.7								1/1
	• 1.4		***		*1.6	**.1	167.5	165.0	100.4	112.2	111.6								177
1.00	****				47.0	**. 3	102.4	165.0	100.1	110.4	111.7								17-
	***					**.*	102.1	105.0	107.3	107.7	104.1								443
	**.1				*1.*	**.	101.1	103.3	100.5	105.0	106.1								124
	47.3				**.0	140.1	103.	109.1	105.5	100.5	104.9								140
.00	****							14.0											4.3
	****		w			100.1	161.	104.2	101.0	103.3	lul								1
.36	***					100.4	100.0	.03.0	101.0	105-1	106.4								
	w				****	100.5	102.	163.5	101.2	105.3	**.*								
2.5	*1.0					100.1	107.	103.1	163.0	101.4	**.3								1.
	****		*1.2					100.3	105.0	100.2	****								
	***							101.4	101.0										130
3.0			*1.5		*2.4	****		****			91.0								
1.5								•											
	14.7							****											110
0.0			47.4								83.4								110
1.0	70.0																		111.
	70.4									***	03.5								
· .	4.4									***	6.6								100
1000				-						3.0									•
																			430.
-	9 9																	.afet .	•
				144.7															

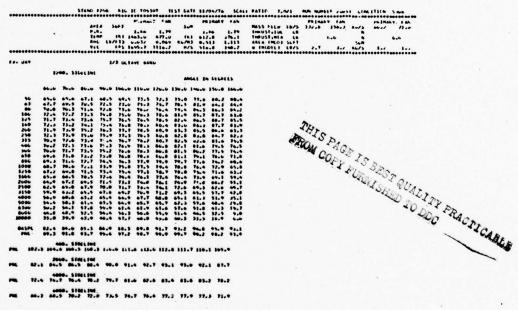
b.) Model Data Scaled To Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

				20000	US COURS		IMARY				**	7 64				PRIMARY	***		PG . PART	
				444	340				500		3000	1	MASS PLE		8/8					
					12.000	1.		1.70		1.0		1. 19	THOUST,	104						
				110		1001		.n.6		017.		D.1	I HE USE ,			•.				
					10/11					0.51		.113						-		
				***	***	1000	., ,	110.2	4/5	110.	. ,	4.7	W INCOLL		0/5	1.7	3.2	46/5	1.0	1.0
		*****	******	••••••	******	*****	•	*****	*****	*****	••••		••••••	••••	••••	•	***	*****		****
MD						~		-	***	win !	-	15.6					atte	U SPL	- (muse	
							-	-												
***	-	24		**	100	216		2 30	140		1.00	••								4-1
										•••										•• ••
1 50		0.0	0.0							6.6										
41		6.0	6.4	v.L		4.4	4.6		6.0	6.6	٠.	•								ī
-10			0.6			6.6	0.0													
•••			••••	***			11.0				100.									111
125		**		w.7																
100		*1.1	****	****						**.*										
~					****				**.*											110
113		47.4							100.3											44*
		*1.3		24.4		****		****	103.0	107.0		•								
w.		****	***	90.1	***		****	107.1	109.3	111.3	1117.	•								1.0
			.	90.4	*	****		101.0	104.7	*****	****									
-		**. 1						100.	*****			:								
		*5.5			**.*		10	104.0				•								1 1
		***			**.1		141.7	14.7.2	111.6	***	114									:;:
	****	**.1	***	*7.5	**	4.00	163.7	160.4	104.5	112.0	114.	•								1
.00		**.			**. 1		163.4	100.0	148.7	114.2	114.	-								***
. >				.1.5	**.6	166.6	1.4.1	100.0	167.6	160. 1	169.	•								1/2
			w	41.3	¥8.4 1	41.1	100.0	144. 1	100.0	100.7	147.	•								1.5
· co				41.4	**.1			164.2	100.0	145.7	Ive.	•								10.
•••		***	95.4	•7.2	90.7	161.5	100.1	164.2	169.1	14.0	101.	•								1/0
. 10		**.*	**.*		**.1 1	61.3	10.1	14.4	104.6	100.2	107.	•								10
.00		**.1	97.1	•1.6	***		101.0	144.7	104.3	103.0	Ivi.									143
		73.6	m		***	61.7		164.1	104.0	167.0	**.	!								11.
•••	****			**-1	***			163.4	103.1	101.6	***	:								111
	**		***	***			144.	107.1	100.4	500.2	***	:								14
0	***	***	***	**.*	**			****	99.2											:
1.5	44.5		*1.1	*1.1		****			***											***
			44.7	*1.5		**.*				97.7										100
																				442
1.4	P			***			w./			***										1
		71.7		.1.0					65.3											Lit
w.									0.0											
																				-

b.) Model Data Scaled To Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

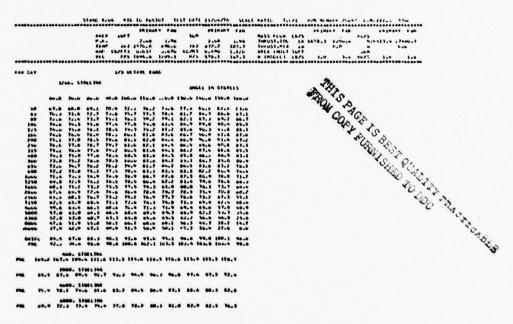


a.) Model Data Measured At 15 Ft. Radius

							-	*		M 10	-		PRIMARY			PE ! NA.	
				444			Salve.	4 2515	100	· comm		MASS PLOW	•.•	0.0	4/1		
							.00	1.04	2000	1.00		THEUST, FOR					
				11.00		1 1570				• 72		THEUST . FLA	6				
				791	LEVE	3 189		205.1	****	570.		M INCULL			-		
													1.6		44/2	1.4	
					1/1	UL TANK	-	-			DATA 15.0	SULGAR THE	RAW CO	RECTE	. 104	- (800)	
-										-	200000 12000				10.00.	0.0000	
	-						RIL				Mentil						Pus
	**	20		*	100	110	150	100	146	156	100						11 -1
***				0.0							0.6						
41						•.•											. :
-											0.6						
100		**.			01.5		**.				102-1						11
149		.7.1		01.3		89.7			94.1	**.1	104.5						.1:
100				84.1	.7.4		40.0			141.1							
200		00.7			41.0	*1.1				105.0							10
		.7.0				*2.6				147.5							10
145	40.5									W0.0							17
~	42.8	***								114.0							
··	****									110.0							
• *	.1.									110.0							
										110.0							• •
										110.3							::
-29										110.0							
										117.2							11
										115.7							.;
										114							- 12
-										112.0							1.5
										****							10
- 20		*7.1	***	1.00.1	104.4	104.5	101.5	106.4	104.1	110.1	107.4						
	**	**	**	100.1	14.1	104.7	167.4	100.1	160.1	108.0	104.3						1.
										.01.>							
										100.7							
										104.4							

1.0		*1.0								161.2							
	.									**							•
	•7.7		***							***							
						**.5											- 11
***	70.0								¥1.2		****						111
1.4.																	
	0.0		•••	•.•	•••	•••	•••		•	••••							
																UAT DE	
											STATE OF THE PARTY						
: 1		414.2		1.2.0	110.7	110.7	119.0	· Ich	127.2	120.3	170.4						

b.) Model Data Scaled To Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

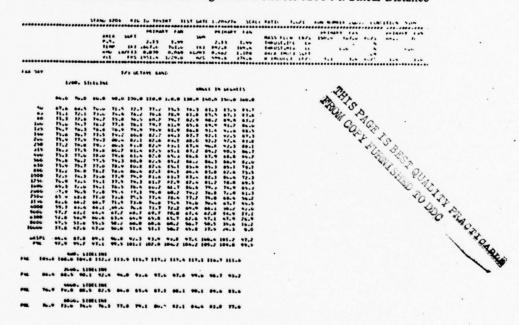


1

a.) Model Data Measured At 15 Ft. Radius

							-	-		- N 1	-	AM			PRIMARY			PRIMARY	
									100				MASS FLOW	13/5		6.0	46/5	6.6	
								1.00		2.1			THEUST, IDE						
				TEN		1001		701.0	141		9 300		THRUST, 41		6.				
									16/41				ARLA IRUDI				. 104		
				AFF	•••	2 1001	1	254.0	4/5	394.			W INCOM! I	10/5	3.4	1.4			1
						******	*****		***	•••••	*****	•••••		*****	*********	****	*****		
						~							CUIDAR TH						
-						w		-		40126	-	15.0	SOLDER 140			BECTE	o m	- IMLU	.,
							-				D4 684 65								A San
1		76		90	144	116	DL	130	140	130	140	•							-
							-		• • •	•••									84 -1
->0		0.6							0.0		0.0								
	0.0		6.0	0.0	6.6		0.6			0.0	0.0								
100		45.7		w.u	00.0														11
143		47.4			**.	**.					101.2								
700	**:	~·!								107.7									
~~	47.4		**.*		*1.5		**.		101.6	100.0	100.								
,15	W. I	**. 1			****	*1.5	***	***	163.0	100.7	111.2								
400	*1.0		****			****		101.4	100.5	110.0	114.5								10
-	**		**			100.4	101.	10	110.7	117.7	110.0								
. 36	97.7			**.7	****	101.6	102.0	100.7		*****									
004				100.3	10.1.4		1.4		*****										
	***			100.7	107.4	1.4.	1.2.1	****	****	170.7	*****								
.43		100.0	.00.	141.0	14.1.7	104.7	167.9	111.0	1.7.7		110.1								
	Iw.I	41.0	141.7	1.4.1	143.5	145.1	140.4	111.5	144.5	120.4	110.0								**
	**.5	**.*	100.0	16.	101.0	144.9	140.1	111.4	113.3	111.1	119.7								1:
. 20	10.0	**.>	100.1	141.7	143.3	M>.3	140.0	111.9	110.1	117-4	117.1								ii
. 15	06.4	**. 6	**.	161.4	40 3.4	144.0	140.0	111-6	117.0										17
	*1.7	40.1	**.>	141.5	163.3	140.0	100.1	110.7	117.7	114.0	111.3								in
		40.4		1.1.1	143.3	141.1	140.5	100.0	111.3	111.0	110.0								14
	•1.2	w.,		1.101	16 ** 1	104.4	100.0	104.5	116.5	111.7	140.0								4.74
	**.1		70.7	161.1	101.1	103.5	104.1	164.1	110.1	110.7	107.1								1.0
	****		****	14.0	104.0	107.3	107.0	100.5	100.4	100.7	101.6								1.
				100.5	107.4	104.4	107.7	107.0	100.1	101.4	100.0								
	*/.5	****		W			106.6	107.3	107.0	100.0	105-1								
	1.	*1.4	**.*	**						102.4									
			94.3		***	100.5	104.	102.5	104.0	100.0	::::								
	.7.7		*2.5		***		166.4	144.4		70.0									•••
			49.4					***			97.0								1.
			87.0							**.1									44
	10.7								92.1		w. 2								***
40.	0.0	0.0	0.0	0.0		0.0													
											3.00								
																		CAPEL .	144

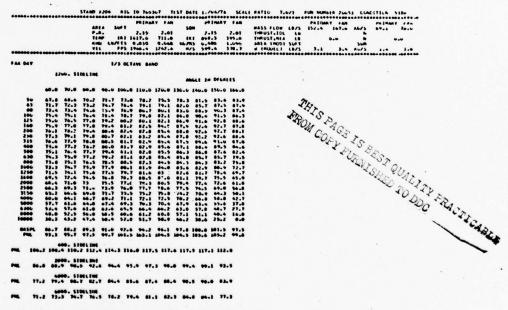
b.) Model Data Scaled To Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							I MAR T			PR 31	-	**********		PRIMARY			PR IMARY	
				AREA					100			MASS FLOW	18/5			46/3		
							. 15	2.01		2.1		THRUST, 10						
				1600		1 1611		711.0	(#)			Denust, ne					•	
				8100		3		•.•		0.40						104		
				VEL			•••	14.4		,00.	370.7	u (mcDft)	10/5	3.1	1.4	46/5	1.4	1.4
-					-				*****	•			•••••	• • • • • • • • • • • • • • • • • • • •	*****	*****	********	***
					1/1		-	-		-		SUIGAR THE					- 1804	
-									•••									
	a sate						416	AGPHON	-		21 1830							-
MI		70		*	-	110	120	1 30	100	110	100							14 -1
								Tar v										
643	•.•	•.•			6.0	0.0	•.6				0.0							
000	0.0					6.0	•.4				0.0							•
100	44.4					44.5					104.1							٠
125	65.0	40.1		W.2	**	****				**.;								
144	49.4				***	****	-			102.0								113
200					*1.*	92.7	*			164.7								117
150	47.9									104.5								173
343										111.0								126
***	94.2									114.9								179
100	W	45.2	**.*	97.2	**.3	107.5	104.4	100.0	111.7	117.7	110.4							1.11
**	**.1																	1 11
*	.7.7			100.0														134
				100.9														100
.25	**.*	100-3	101.6	105.1	164.0	104.9	100.7	111.4	114.1	120.6	114.6							13>
.00				102.3														120
.00				.02.3														IM
-15				101.9														
	****			101.7														135
-	\$4.6			161.2														
.10	90.0			101.1														
.00	94.7			Ivi.i														15.
6.0		97.5		101.1														124
4.5	99.4			100.0														127
	94.5	94.3		100.6														1/4
4.0	45.4	****		99.0														10
3.6	91.7		**	10.2	144.4	142.3	14.2	144.1	164.7	105.0	**.*							140
14.5	40.2									101.5								111
	44.3									**.*								100
	67.0									+7.3								110
1.0	62.0						**.			47.7								117
	74.4				****	***	92.2											115
w.	6.6			•.•	6.0	6.6	6.0	• • • •	0.6		6.6							
																	-	1
																	ores .	
194			112.2	113.0														

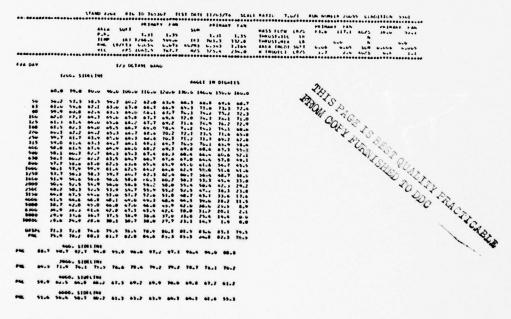
b.) Model Data Scaled To Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							I BARE			PRI	MARY F	44	**********		PRIMAR			PRIMARI	
				-					540				MASS FLUW	10/5	0.0	0.0	44/5	0.0	
				•.•.			. > 1	1.0		1.1	1 1.	**	THRUST . I DL			-			-
				IIAP		1 140		****	161				THRUST , MEA		- 3	0.0			
				-		3 0.		6.673	rew)				AREA INCOL				-		
				711		\$ 144	.,	767.7	~>	375.			W IMEDILI	10/1	1.7	7.4		0.0	1.1
	***		•••••	******	*****	*****			•••••	•••••	•••••	*****	•••••••	•••••	*******	*****	*****	*******	*****
													RADIUS						
-						w		-		m124	DATA	15.001	RADIUS			MARCIE	u ser	- 1864	
							-10												
		Pu		**	100	114	130	1 1	14		140								-14
						•••				1 10									14 -1
no .		0.0		0.6		4.6				4.0	6.6								
			0.0	6.0		0.0	•.•												
	.0	0.0		4.4	4.4	4.6					6.6								•
w 76.		11.0	14.2	D.2	75.0	74.2	77.5												1
25 11.		75.4	14.4	D.1	70.4	70.1	11.2												15
		70.3	75.0	70.2	13.0	74.6	70.1				*1.*								15
w 7,		14.0	76.8	77.7	70	.0.6													10
n 15.		17.0	70.5	70.0	80.U	.1.5				**.*									11
	••	14.1	W					**.*		*7.2									11
		.1.4			85.2		**.*			100.7	161.7								11
L 00									**.*	107.6	101.1								
40 60		****	**.*		47.3	**.*		**	100.1	101.7	103.3								11
-		****	**.*	07.0	.7.7	••••	*1.*	***	100.1										11
12 01		****	•1.•	****	64.1	*1.1	****	**-1		104.0									
			***			90.0		93.1		**.>									111
			45.3			***	¥/-9			.7.3									44
N 0/		**.*	45.5	**.*	**.1	*0.7		m. 1		92.1	••								
15			45.0		.7.	90.4		*1.2											
w			7	85.5	07.4	44.7	*1.4			.7.5									
		02.7	41.4	45.2			90.1			45.1	84.6								
10 00		.1.7									70.2								10
uo 14.			04.1					.7.4			74.0								-
10		74.1		62.3				80.6			74-1								
.5 11.		10.0	80.7								71.4								16
to.		17.3	10.0			64.9		W		75.0									100
1		13.0	17.7	70.3	P 3		.1.5												10
77.		14.0	10.1	17.4	74.3	14.5	w				47.4								100
. 71.		14.0	14.0	D.0	70.0	77.6	70.4			71.0	47.7								"
		11.,	13.5	h	15.2	70.5	10.0			71.0									- 7
.0 .1.		•••	71.4	11.5	13.5	17	76				11.0								
47.				64.6	71.7	71.4	77.,				17.9								
			• • • •							71.0									
· ·							6.0	***	•.0		6.6								
																			_

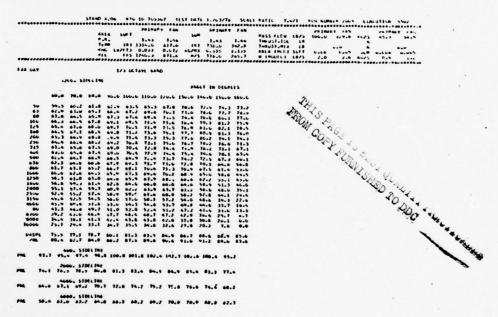
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

						~	-	*		P# 1	-				PR PAGE				
				4484					340				MASS FLOW	10/5	0.4		-		
								1.4		1.4			THEUST . TUL				•		
				10.00		1 1314		417.0	(4)				IMBUST.PEA						
				4	10/01			0.017	SEA!			155	MIA 18001	1441			-		
				***		1500		.71.0	WS				. tacher	100	2.0	7.0	14/3	6.4	1.2
		*****			******	••••	****	*****	•••••	****	*****	•••••	**********	*****	*******	*****	*****	*******	***
***						w		-	***	40174	DATA	15.0	+ 1 AADIUS		-		w 30L	- Incut	
4664																			
1 101	••	1	•	**	144	110		110	140			•							-
	-		-			•••	•••	• •			100								De -1
440				4.6	4.6	4.4			6.6										-
	0.4	0.0	0.0	0.0		6.0					0.0								٠.
				4.4			6.4												•
146	17.0	15.1	11.1	70.5	70.1	19.5	.0.1				93.3								
605	***	14.1	14.1	P.1	****														105
100	11.0	60.0	14.1	14.0	70.1	14.0	41.4				**.*								100
74	11.4	74.1	19.0		+7.0						**.*								***
. >0	77.1			61.5					*5.5										***
112	04.0	04.0	43.2	4.2			***	*1.4		101.7	104.6								116
-								**.*	101.2	105.1	.07.7								115
MOU	40.5				84.0	*1.4		****	100.0	W1.5	147.6								1.1
	****		64. 1		11.0	*1.1	***	100.:	105.5		161.1								17.
- 40		**. >					****	100.7	14.4	144.7	144.7								
		84.2	.00	*1.5		95.7		101.0	100.4	109.2	147.5								4.7
		.0		.1.4	*3 *	*>.*	*	101.5	165.7	107.0	105.0								1
.00			**.*			*5.3		100.0	161.9	14.7	W1.1								444
.40			**.5	w	**.*	43.2		166.1	167.5	141.7	**								
- 200	**.1			***	*2.5	**.*		**.*	100.6	**.*	10. 1								
4>			44.2		44.2		**.,	**	44.1	*4.>	*1.0								. 1.
				•	*1.4	**.7			*7.2										
. 54	45.0		***	***	+1.6			**.*											
	****		****		**. 7	***		45.0											11-
	2.7		****		****			*											
			****		07.5	*1.1		*1.7											
	*1.1		* 7		****	**.:		**.		***	77.4								110
	79.1							***											
								43.7											100
1.5	75.0	77.0	19.0			03.7		67.1											100
		74.0	70.1	74.3		*1.7				7									
	14.4				70.2	74.4	77.0												•
,	44.7	71.0		N.7	15.5	10.0	,,,,												1.0
		****		71.1	71.6	74.2	7												**
14.				6.0															••
								•••		***									

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

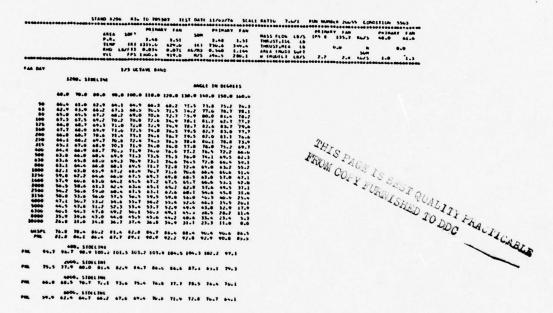


Configuration 2Au; Condition 5503

a.) Model Data Measured At 15 Ft. Radius

							11444	*		Pa 1	-							-	
				-					-				155 +660		4.6		4		
								1.51		1.4			AUST. FOR				•		
				11 10				~~.0	141				aust ,=14						
				-	100			0.671	scm)				IF A ERCOI				**		
				***		· • •	••	.1	4/5	140.			1400111		7.7	2.0		1.0	1.,
											******	••••••	********	*****	********	*****	*****	********	*****
					1/1	IL TAN	-	-		-		13.411	Levius		-			- 1944	
-																			-
	PALO	-									***								**
-	••	76	*	**	100	116	150	1 30	14	110	100								80 -8
910			0.6	0.6	6.0			6.6	0.6		6.6								
			0.0	0.0	0.0														
-							4.6												
100	Pe. 1	10.2	70.5	P.4	19.0			85.0			**.0								
153	70.0		80.5	N.)	41.3					92.0	94.3								100
100	PO. 2	01.2	80.0	80.5	10.1		84.0			93.0									-
100	70.5	70.4	.1.0	W.1	43.2						100.0								4 44
ne					84.8				***										
313		65.0	M.3			44.1			**.0										
***		65.1	47.6	10.0					105-1										251
100	47.3	67.5	***	***	*0.4				165.2										844
=	77.5		W	***	92.3	**.			100.0										134
===		*1.0	11.5		99.1	**.			100.0										14
.75			92.2	W	w. 7	90.7		103.4	100.2	110.	104.2								841
	44.0		91.4		W. I				103.6										124
				92.1	*1.*	***			104.1										12
. 50	47.1		90.4	92.2		***			102.7										120
. 13	87.4		10.7	92.0	91.0	94.3			164.7										1.3
.00	87.4		10.5	91.7	93.4	90.1													111
	87.3	07.0	89.7	40.4	42.4	**.1	*7.0				90.5								111
. 30		67.4	40.4	10.4	92.2	94.3	4.1	94.7	94.4	41.4	86.6								4.5
	85.0	84.5	88.4	87.6	*1.3		93.0	95.2	92.7	89.2	05.1								110
•••		43.7	47.5		40.0	42.3	*3.0				62.3								441
2.5		**.>		47.5	84.7	.1.0	45.2				19.0								1 61
	62.4				87.6	·		90.2		41.4	77.0								8.85
•.•	00.5	11.0	*3.4		84.1	.7.	44.7			01.3	75.7								100
	70.0		65.5		6>.0	84.4			84.7	14.7	76.6								1 -
1.5	77.6	70.5				**.	**.4		07.7	70.0	74. 7								861
•••	n.4	17.3	10.1			• > . :	**.0			79.1	77.								10
	71.4	17.2	77.4	70.0	74.4	41.7	82.6		86.6	86.3									163
	44.7	70.0	74.0	73.1	77.4	76.4	77.1				***								14
	***	0.0	0.0	72.5	2.5					01.3									
			***	3.0		3.0	4.0	•.•		3.0									
																		-	

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

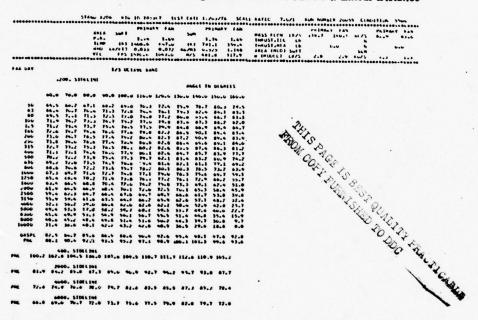


Configuration 2Au; Condition 5504

a.) Model Data Measured At 15 Ft. Radius

1.00 1.00								1-40			C# 10	MARY FAR	************		-				
### (81										104			MASS FLOW	10/5			46/1		
### 18 18 18 18 18 18 18 1													THRUST . TOL			•••	:	4.0	•.•
### ### ### ### ### ### ### ### ### ##																0.0			
### ### ### ### ### ### ### ### ### ##													AREA IMUDI	1041			-		
					ALC		PS 13 M		4,600	W.S	41.	. 317.0			2.0	2.0	867		1.1
### DETAILS DATE 10-15 10-						******	•••••			*****	*****	• • • • • • • • • •	************	*****	*******				
Middle Mart										1122		array are							
	-					4/1	OCIAM	-	MODEL		10174	DATA 15.	SET BADRUS		RAW C			- 1044	
Mart Sec. No. Sec. S																			••
			-																-
### ## ## ## ## ## ## ## ## ## ## ## ##		-		••	**	100	110	130	1 300	1.00	136	100							11 -1
	-																		••••
March Marc																			
125 74-8 above 11-1 21-7 22-7																			. :
1	123																		
10																			144
2																			111
10	720																		447
March Marc										100.	105.0	107.1							111
Mart	•									103.4	107.7	110.3							14
10	***	42.6					***		107.4	101.0									170
10	-	**.*	**.1																1.0
10	-								16.			113.7							170
1.		*>.*				**	101.0	100.0	114		*****	*****							434
10		**	***	w.1	***	140.0	142.4	1.4.				*****							4.1
1.0					**.0	100.0	102.0	103.	110.		****								11
10		**.*	**		10.1	100.1	10	101.	144 4	*****		*****							432
10		**.*	**	**.	*1.*	146.6	142.4		1	****	117.								1 10
10		*7.0	*>.0	94.2	**.*	**.	10/.5	105-1	100. 2	109.4		107.7							124
10	.00	*1.0	**.0		47.1	**. 1	104.4	100.4	107.7	144.4	104.0	101.4							147
2.50 Val. 92.5 No.2 Vo. 90.1 100.0 100.1 100.0 101.0 1		47.0	**.6	91.1			161.0	100-1	105.4	164-6	141.4								445
10	. 14		93.5		77.9	94.1	100.9	40.1	10	101.0	101.0								1.0
1	.00			**.	**.5		100.4	14.4	102.0	141.0	90.4								
1-3					w.1	**.5	19.1	101.1	101.0	99.2	***.2								1.1
10			90.4	94.2	*1.5	45.4	41.4		90.0	94.4									1.0
1					W.5		**.7		***	94.7									
10 844 804 804 812 814 112 824 814 812 824 814 812 814 813 814 814 814 814 814 814 814 814 814 814																			417
1.5 81.4 84.5 84.6 41.6 84.6 44.6 42.6 4						*1.5		***	94.2		87.4								110
100								97.4											
164 184 184 154																			411.
1.00 78.0 78.0 80.1 81.7 82.7 80.0 85.2 85.2 85.0 86.0 86.0 86.0 86.0 10.0 10.0 10.0 10.0 10.0 10.0 10.0 1						43.0	17.1												110.
100 13-1 13-3 17-3 18-3 80-0 02-0 03-0 00-0 00-0 00-0 07-0 07-0 07-0 07								45.5											100
do tob but but but but but but but but but bu										80.7									I.
가장 보기 때문에 가장 가장 하는데 보다 내가 되었다. 그렇게 되었다면 하는데									4.4	6.6									les.
		104-6																	
L 100.0 107.5 100.0 107.7 111.0 113.0 110.7 117.0 123.0 175.7 176.5																			

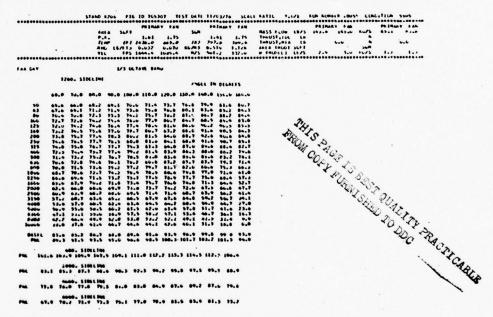
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							I MAR Y	FAM		PR 14	MRY FAM		PR I HAR Y			M IMM	
				-					340			MASS FLO	•	6.6	W/5		6.4
						1 143	•1	1.75	(#1	797.		THRUST .			- :		
				810		13 0.4		.070	MA)			4814 INC	•		140	•	••
				VIL		1 1044			W.S			w tecot	2.0	3-0		1.3	1.3
	*****	*****	*****						****			*********		****			••••
					1/3	DETAVE	-	400t L	JET 1	101 M 108	ATA 15.		-			- (mad	
-																	
		•		-		116	126	1 30	100		ICREES.						14-1
MI I	••	70	*	••	160	116	176	1 30	140	136	100						
156	6.0	6.0			6.6	6.6		0.0	6.6	0.0	6.6						
•		0.0			0.0	0.6	0.0	0.0	0.0	•.0							•
			6.6	0.0	6.0	0.6	0.0		4.6	6.6	6.6						
160	70.0		04.1	****	***	65.0	67.3	**.*		**.	**.*						***
143		**.*	****	m.,	84.7	47.7	M	90.8	****	144.6	100.2						
166	m.1	07.1		67.4			*4.			104.4							111
200		***	47.4		44.4		*1.1			104.3							4.0
115	44.1	.7.4	44.4			*1.7				100.0							17
we	96.1		97.4	93.7	95.0					112.0							1.
100	92.9	**	*3.7	***	93.4					115.0							4.0
-30	**.	**.	45.7							117.0							
-	94.7	**.1	90.1		**	100.3	101.7	109.0	115.6	.10.0	146.7						
.00	90.0	14.0	47.4							110.4							
-45			**.6							110.5							
	11.1									110.6							
.00		*1.4	.7							117.6							A 34
. >0	*>		47.0							115.0							1.5
. 13		**.	*1							117.4							
.00		**.*	97.1							105.7							10
		**.*	94.6							162.9							14
		***	93.1							100.0	99.3						47
		*3.0								***							
4.5		*1.*	*1.*			**.4					96.4						17
4.0			92.5			98.4	**.4	***	90.9	***	67.4						
	****		41.4	92.5	****	**.2	**.3	94.6	****	*1.1	84.4						
3.6							**.*				87.4						
1.5	63.4																
6.6	40.0										04.0						11
4.6		87.0	**.1				****				45.7						***
	11.0										***						10
41.0	70										67.5						••
w.			0.0	0.0	0.0		0.0	0.0	0.0	•.•	4.6						
																wheel .	
***	107.1	100.0	107.4			114.7		144.	4470	4 /	W >. 0						

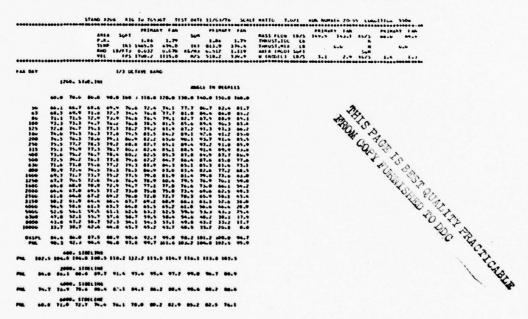
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

AREA 1807 P. 10. 1.06 L.79 10.0 1.06 L.79 10.0 10.1 1.7 10.0 10.0	10 00000000000000000000000000000000000
Time Int 1405.00 150.0	
AND LEAF 19 COLD 2 COUNTY AND MOST 1 11 COLD 1 COLD 2 COLD	
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LT OCT SAME	
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	LOLLI
1021 40 79 60 40 10 10 10 10 10 10 10 10 10 10 10 10 10	***
### ### ### ### ### ### ### ### ### ##	14-1
### ### ### ### ### ### ### ### ### ##	
### Park Park	•
March Marc	
25	
100 88.7 84.7 84.8 86.6 42.1 85.3 90.1 92.2 44.9 100.2 104.9	111
No.	111
13	111
	12
00 01.0 97.3 96.3 97.7 96.7 96.7 96.1 101.4 109.7 111.7 116.3 175.9 90.0 97.5 96.3 97.8 99.3 190.1 105.4 107.4 116.3 117.0 117.0 116.3 175.9 90.0 97.5 96.8 96.8 97.5 97.1 107.1 105.0 111.0 117.0 118.0 117.0 90.0 97.5 96.8 96.8 96.8 97.1 97.1 107.1 105.0 111.0 117.0 118.0 117.0 90.0 97.5 97.5 97.5 97.5 97.5 97.5 97.5 97.5	1.0
30	10
00	130
00	
25 91.3 91.4 92.5 100.6 100.6 100.7 100.4 107.5 104.3 107.6 110.7 110.7 117.7 100.6	11.
46	
00	1 14
50	
130 Mail Via Make Mail Wale Mail Mail 1804 1804 1804 1804 1804 1804 1804 1804	10
00	8 34
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36. Whi Wish Wish Wish Wish Wash 1606 18956 1806 1875 1806 1807. Whi Wish Wish Wish Wish Wish Wash Wash Wash Wash Wash Wash Wash Wa	
\$\text{\text{\chi}} \text{\text{\chi}} \text{\chi} \text{\text{\chi}} \text{\text{\chi}} \text{\text{\chi}} \text{\text{\chi}} \text{\text{\chi}} \text{\text{\chi}} \text{\text{\chi}} \text{\text{\chi}} \text{\text{\chi}} \text{\text{\chi}} \text{\text{\chi}} \text{\text{\chi}} \text{\text{\chi}} \	1/
10 11:0 10:0 10:0 17:2 17:2 17:2 10:1 10:1 10:1 10:1 10:1 10:1 10:1 10	10
1-5 * 11-0 * 17-0 * 10-10 * 10	10
	17.
100 82.9 50.7 52.6 52.7 50.3 51.5, 50.6 52.9 50.2 52.0 50.5 50.2 50.3 50.3 50.3 50.3 50.3 50.3 50.3 50.3	10
100 80.2 87.0 90.0 72.3 10.1 95.7 10.2 95.0 90.5 90.6 81.2 10.5 10.0 10.2 10.2 10.2 10.2 10.2 10.2 10.2	10
1.5 Ba.5 Bh.9 By.6 Wa.7 W.2 Ya.2 Wa.6 Wa.5 W.2 Sy.2 Ba.2 Ba.2 Ba.2 Ba.2 Ba.3 Bh.9 By.6 W.2 Sy.5 By.7 By.7 By.7 By.7 By.7 By.7 By.7 By.7	111
1.0 80.0 87.4 85.1 80.3 88.5 80.0 90.0 91.2 90.4 90.1 87.5 1.0 180.3 80.3 87.4 80.3 87.7 80.8 80.3 87.8 87.7 90.4 88.6 1.0 75.4 77.4 77.5 81.0 83.4 87.4 85.4 88.2 97.1 97.4	111
1.0 78.3 00.7 02.4 04.3 05.7 06.8 08.3 07.8 40.2 40.6 08.6 1.0 75.4 77.4 79.5 01.0 03.4 04.3 05.4 09.2 40.1 91.1 07.4	447
1.0 75.4 77.4 79.5 al.0 al.0 al.0 al.0 al.0 al.0 al.0 al.1 al.1 al.0	111
	100
70. T.D T.T T.T T.D T.D T.D T.D C.D C.D C.D C.D C.D C.D C.D	& site
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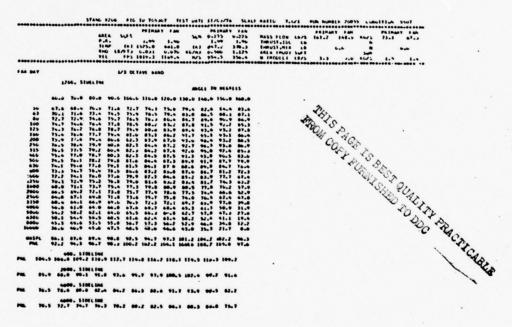
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							IAMET	FAR		-	TANK	PAN			PRIMARY	*		PE IRMET	**
				4844					sen				MASS FLOW	10/5	0.0	6.6	4645	4.0	0.4
				P.R.			**	1.90		1.0	•	1.90	THEUST. SO						
				11-		1323		61.6	181			70.3	THRUST . ME		6.				
					COVETS			4.6 PG		0. W		-125	MEA INCO				244		
		*		AFF		1010				****		20.4	. INCOLL!		3.3	3.0	44/5	1.5	1.0
****	****	*****	*****	******	******	****	****	*****		*****		•••••	********	****	*********	***	*****		
													FT RADIUS						
-							-	-					A I MEDIO?		RAW COM	MICH	0 20	- 1404	
							-	ROPMON	-	18	01 686								PUel
1	40	70	•	**	100		120	134			144								4-1
200																			
620	6.0	0.0		0.0	0.0	0.0						•							
103		6.6	6.0		6.6		0.6												
-	6.6		0.0	0.0	0.0	0.0	0.0				•.								
100				87.0		87.8													114
153	43.7	47.3	.7.1	4.4		40.5	84.1			**.*									113
100							10.4												4.7
			90.1	**.		.5.1			162.4										151
720	60.1	90.4			97.0														120
		***			97.3														140
=					*7. *														1 2
					100.0														1.5
-	47.6				Iul.4														1.3
-40					102.6														12
					103.9														437
-42					104.0														1 47
-06	**.7	**.*	161.2	102.2	104.3 1	100.6	110.0	110.0	121.1	171.6	117.	•							
. 20	90.7	**.5	100.0	102.2	104.4	167.6		11	120.6	.19.3	11>.	•							10:
-13		***	166.5	101.0	103. 4 1	1.7.2	110.4	114.7	114.5	116.0	11 3.	•							
-					10.0 1														1 30
					103.5 .														
					163.7 1														134
•••					102.0														1.0
		****			101.0														140
2.5	*				101.6														175
•••	*::				****														1.3
***			***		****														171
			*2.0																110
								**.7											11.
4.7			\$7.9				***												1.5
3.0							*1												***
			04.2		43.4														110
0.0			0.0		0.0														

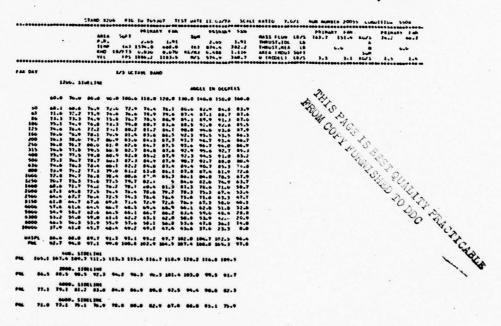
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							-	-			MAT F			PEIRALY			PR IRANT	
				-					100			MASS PL	DW 18/5	0.0		46/1	0.0	0.0
				P.A.		2.	.05	1.91		2.01	. 1.1	neust.	104 48					
				1100		1 15M			(8)	874.	. 147	.2 THRUST	HEA LB	6.				
				RIED		3 64		6.670	40	6.48			1108 100			100		
				ALL	**	1 1000		101.5	NI	574.	, see.	7 # (#006	LI LO/S	3.3	3.4	aws.	1.>	1.4
	****			*****	*****	-	*****		****	•	****	**********	******	-	****	*****	-	
-					1/3	OC I AW	DAME	WOOL !		MIN (DATA	S.OFT MONS		RAW CO	RECTE	0 20	- INUDE	
~	***	76		-	140						DECAL ES							PU
•	••		••	•	100	110	150	130	100	130	100							14-1
-20	0.0	0.6	0.6	0.6	0.0		•.				0.0							
-							-			6.6								
-	6.6	6.6	6.0	0.0	6.6		6.6			6.0	0.0							•
		45.3	07.3	47.7			40.											
2>		07.2	07.3	67.7		94.7	49.			100.4								***
46	80.7	90		89.1						107.7								***
**	87.5			10.4	92.0	42.7			102.0									- 12
10		44.3			93.0		**.		104.0									
19	94.0	94.4	97.3	93.3		94.7			100.1									12
		**.*	45.3						112.3									
-	**	**.1	44.3	+7-1					115.1									ii
34	97.0	90.0		**.7	100.5	107.4	100.	114.3	117.4	120.7								43
	47.3	**.3	96.7	100.5														10
~			**.7	141.4	163.0	105.7	140.1	1 116.2	121.0	121.4	117.0							1 4
25	**.1	100.4	101.3	10	100.0	104.4	110.	117.0	127.2	W1.)	410.2							
	**.7	100.0	101.7	4.504	164.5	lue	110.4	117.7	122.4	122.1	110.7							4 .
"				102.4														
30	**.4	100.0	101.1	105.0	100.0	107.7	4. 4.4	116.0	120.7	.19.0	.44.1							
15				102.4														
00		**.1	100.7	16.5	100.7	107.0	111.2	114.5	117.4	115.0	111.0							
•	90.1			141.7														
				101.6														8 36
w	**.*			101.1														100
	**.7			100.														14
:									100.0									
	¥2.4		**.7						104.0									100
	40.1		****						167.4									12
	67.4		***	9.7	97.4	.7.5			100.7	***								474
		40.0			70.1	****	***											4.10
	M.,			40.0	92.6	*3.7	***											***
			05.7		84.2	vo. 7	7.											:::
	bu.4				****	40.7	***											***
4.	7.5			7.0	*	0.0	***											
-																		
																	CAPUS .	444

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

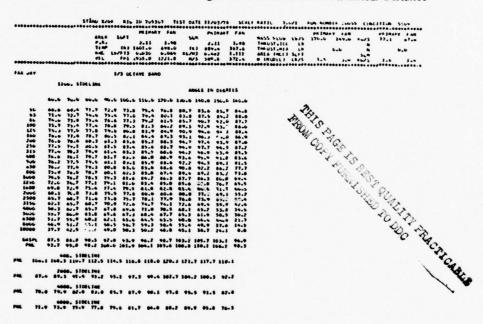


Configuration 2Au; Condition 5509

a.) Model Data Measured At 15 Ft. Radius

							-			PRI	RARY FA					818 700 00000000			
				ARL					500			MAS		40/5	0.6	0.0	46/5	6.6	4.4
							1.11	1.90		7.1	1 1.0		UST . 14				:	•••	
				104		41 10		440.0	141			. THE	UST.REA			0.0	-		
				RIFE		**		6.669		0.4		1 ARE	A INCOI				24.0	•	
				***		Pb 10		1551.0	M/S	100.	. 372.		HUDLLI	10/5	3.>	3.6		1.0	1.4
****						•••••	-	*****	*****	*****	*****			*****		*****		********	****
-								-	387	WIN.	DATA 1	5.0FT RA	DIUS		RAW	CORRICTO	10 20	- IMOGE	4.1
	-																		
12041		70		**	100	1.0	120				DECREES								Puel
	7000	-	-	•	•••		40	134	100	136	100								te-Li
					0.0				0.0		0.0								
		v.6	6.6								0.0								•.
140	6.0	0.0		0.0	0.0														٠.
100	04.0		87.7		44.3														
153			87.4		11.1					101.5									115.
-				4.70					****										\$ \$ \$ 7
				11.2	94.4			97.5	143.4	167.4	100.0								4.0
750					+3.4			100.3	:05.7	110.1	112-1								14:
315		****			****	*7.	* **.	164.1	140-8		113.4								124
-	***					101.	162.	147.4	112.7	114.4	114.4								121
***	****	**		47.4	*4.7	141-1	100-	. 100. 1											1-1
-000	***		***	144.2	161.6	16.34	1144												100
***	₩	••.•	w.,	101.2	142	14.0-1	167-												120
-60	****	****	100.3	14.6	103.7	100.	110.	117.0	1 1.0	121.0	110.4								1.77
-63	****	100.7	101.0		103.5		1111.	110.6	142.0	121.4	110.5						,		144
.00	100.0		102.4		109.1	107.0	111.0	119.6	17 4.4	127.0	119.6								1.4.
- 24		100.1	101.0	163.2	105. 3	107.	1111.0	110.0	153.0	155.0	110.0								12
-13		100.0	161.0	163.6	105.6	100.	115.	110.1	122-1	120.6	110.5								10
	****	100.5	101.0	165.7	10.0	100.	117.	117.6	150.4	110.3	115.0								130.
	144.1		101.0	103.7	103.3	100.	117.	115.0	110.1	115.0	111.0								In.
		144.5	141.4	102.0	107.7	100.		113.0	114.7	111.0	164.1								113.
.00			100.0	100.5	10-1	100.0		112.1	114.5	111.03	100.0								
	***	70.6	***	141.7	104	164	164	116.7		100.4	103.5								
7.0		90.1	**.1	100.4	103.1	105	107	100.7	100.1	.04.5	100.6								Biv.
4.0	+3.*	**.*	97.0	**.*	104.4	lus.	100.	100.0	lua.	164.5	***								1.7.
		***	**.1	14.2	100.2	141.1	161.	104.4	104.4	107.0									
>		**.*	**.*			101-	102	104.2	101.1	90.4									1.1.
1.5	80.6	10.0	43.7			**.	144	100.4	161.1	94.4	49.4								4
4.6			+2.1	+1.7	11.3				**.7	**.	40.7								170.
4.0	7			90.5		**.	****	****	97.1	**.									110.
3.0	04.3	84.1			W. 3	.1.0	****		95.7		41.7								
	74.5	.1.3		03.0			*1.1		14.4										110.
						6.4			6.6										""
																			٠.
																		-	142.

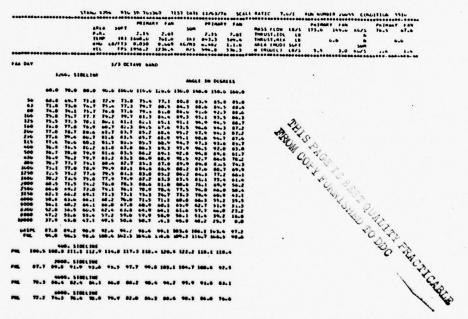
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

Fig. 12.0								-	-			THE T	PAM			PRIMARY	-		PRIMARY	-
Table 181 160 161 16										100				MSS FLOW	10/5	0.0	4.4	RE/S		
### LEATH 3 0.000 0.																				
### ### ### ### ### ### ### ### ### ##																0.			6.	
1/2 (CLAW DAMS DOCK JET NOISE DATA 15.0°T RADIUS BAN CHRISTING PL - (RCG1) RITE 5746 R																				
1/2 OCC ANY DATE DATE DATE 1/2 OCC DATE DA							2 147		714.4	~'s	300.	•)	. .,	# (#CDEL)	18/5	3.5		MW'S	1.0	1.4
											•					**********	*****	•••••	•••••	****
	7.50					1/1	UC 14W	-	-	#11	M 100	DATA	15.0	FT RADIUS			MELTE	0 101	- (840)	
AND 1 00 70 60 70 100 110 120 100 100 100 100 100 100 10																				
10									BOSHOR				5							PUW
### ### ### ### ### ### ### ### ### ##	D4 1		10	••	**	100	***	120	130	144	110	100								Bt -1
	640		0.0	6.0	0.0	0.0	•.•		0.0	0.0	0.0									
### ### ### ### ### ### ### ### ### ##	m 3			0.0		0.0		0.4			v.0									
12	-										0.0	0.0								
																				4.3
100 100																				847
196 84.7 84.6 84.1 84.1 84.2 8																				***
135 11.5 11.5 12.7 12.7 12.7 12.7 12.8 10.4 10.5 1																				144
200 201																				
100 100																				
120 40.5 VI.0 40.7 VI.0 40.7 10.0 10.1 10.2 10.7 10.7 10.7 10.7 10.7 10.7 10.7 10.7						****	161.5	463.6	161.4	113.2	110.0	110.								
Mar. West																				
10		***					103.7	101.1		1110.5										
10					101.7	14.4.0	164.4	107.0		170.7	144.1									
A	.75	**.	140.4	102.2	101.2	105. 1	147.4	110.		177.										
100 100																				
1,000 10 10 10 10 10 10		160.6	161.0	162.2	14.1.5	145.5	100.1	112.1	110.1	121.4	122.2									
13 180-3 180-6 180-4 181-7 181-6 180-6 181-7 181-6 181-7 181-6 181-7	.10	166.1	101.0	107.3	143.7	109.9	140.0	112.4		122.7	120.0	114.								
100 100	.15	100.3	100.0	104.2	163.7	105.0	100.0	112.1	117.0	121.4	110.0	115.								
100 100 101 102 103																				133
18	.00																			
160 Wei 1 Wei 1001 1 1 101 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	. 36	**	166.0	102.2	163.1	10>.4	100.4	111.4	114.1	115.2	117.0	104.								137
2-3	.40	47.7	**.5	101.4	107.1	105.1	100.1	110.	412.9	1.3.2	107.4	104.								1 .00
100 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0		90.3	**.4	100.3	102.4	104.4	101.1	140.4	111.4	111.7	101.2	101.	•							1. 1
8.0 V.0 V.0 V.0 V.0 V.0 V.0 V.0 V.0 V.0 V																				
\$\text{500}\$ \$\text{\$0.3}\$ \$\text{\$0.12}\$ \$\text{\$0.12}\$ \$\text{\$0.11}\$ \$\text{\$10.12}\$ \$\text																				4.0
1.5 84.6 91.6 90.0 92.6 97.7 90.7 100.6 101.0 101.0 97.6 90.1 101.0 101.																				134
000 8500 8722 8928 9127 9346 9525 9627 9725 9826 9628 9126 8725 9426 8721 8726 9726 9726 9726 9727 9727 9727 100 8726 8127 8021 8727 8027 8027 9026 9727 9827 9727 9727 100 8726 8127 8021 8727 8027 8027 9826 9725 9725 9725 9726																				
0.0 02.5 04.6 07.1 07.6 46.6 42.5 43.4 45.1 47.2 45.7 42.7 0.0 19.0 07.1 05.7 06.2 46.2 46.6 47.3 43.3 46.5 45.5 45.6 11.																				
Price alliv Soil SS-7 48-2 1000 113 133 1005 135 1300																				
	w.																			
								•••		•••		•••	2							

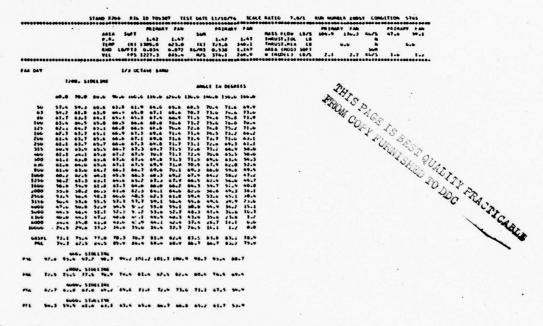
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							-	-		PAI	RARY F	*				PRIMANI			PRIMARY	
				446					344				MASS 1		16.4		6.6	46/5		
								1.47		4.0			IMPUSI		4.0				-	-
				1:00		1 136		621.0	181	129.			THAUST							
				RMU		3 0.0		0.072	EC'M)				-				1000	-		
				ALL		\$ 1221		865.4	4/5					Deti	ters	7.1	1.1	86/3	1.0	1
*****	*****	*****	*****	*****	******	*****	*****	•••••	*****	•••••	****	••••	*****	*****	*****	********	*****	******	********	***
-					4/3	G. 1 . VI	-	MODE L	36.4	MO125	LATA	15.0	FT RADIO	us		MAD CL	RAILE	6 SPL	- (mile)	
L Same	•0	70		***	100	110	120	130	140	130	Tob									-
	-		••	•						130										44 -1
-	6.6	6.0	6.6	0.0	0.0		0.0		0.0											_
-			0.0		0.0	0.0	0.0			6.6										•
	6.6		6.6		0.0		0.0			0.0										:
100	74.3	75.5	11.1	10.4	77.7	79.1			84.4											
1.5	74.4	**.1	70.0	78.2	70.9	81.4	60.1		87.2	89.7							14.			100
100	76.7	14.3	70.5	78.:	77.4	70.4		82.6	87.7	*1.7	95.2									14
100	70.5	77.4	19.3	80.5	84.4	02.6			*1.4	*>.>	47.0									
**	*7.0	70.1	84.4	1.10		8 3. 7	85.7		42.4	97.0	97.7									111
115	74.4	:1.1	82.0	1.40			80.1		*5.0		100.7									110
*	7		45.3	84.2			*1.1			162.0										
***					86.7	84.2	*1.4		**.											417
. 10			67.3			70.7	*3.3													
100			84.4			₩.7	*3.1		100.5											117
	45.1		84.7			.1.7	**.2		166.0											
			67.2			*1.7			**.2											
	84.1	85.4	07.4		40.6	*2.6	***			*7.4										110
	4.4		87.3		49.4	•2.1	94.7		•7.3	*5.5										44>
	84.7			m.:	44.7	92.0	**.		****	**.3										
	83.4	87.0	****		90.0	*2.	***		****	***	***									110
	45.2	80.4	87.4	4.0	87.6	92.4	W. 1		*13.5	90.2	****									110
	4.9	84.3	47.4		84.4	92.1	***		12.4	49.5	43.4									113
.00	43.4	83.4	84.9		89.0	91.4	93.1		91.3	80.2	02.7									
			84.3	87.4	80.5	*1.1	92.3		49.4	84.7	81.1									111
1.5	42.4	84.1	85.8		87.4	90.0	91.0		87.9	84.7	78.0									110
	91.7		84.4			89.3	96.6		84.4	82.7	74.4				2					100
	80.1	81.9	4.1		85. 5	87.4	80.1		84.1	80.4										4-7
	70.7		82.2		84.5	84.0	80.8		42.0	70.4	71.0									14
	77.0	19.2				84.4	0>.2		10.4	70.3	49.1									100
		77.1	PO. 1	80.2	81.0	82.5	03.2		70.4	74.5	41.7									1.
	72.2	14.6	74.2	77.0	70.3	79.7	80.2	70.8	75.0	73.0	69.2									164
	49.1	71.2	73.2	74.3	75.0	74.0	70.0		73-1	71.2										**
	45.4	67.6	69.6	70.5	71.0	12.9	73.3	72.5	70.7	49.5	44.5									
w.			0.0	0.0					0.0											

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

				-			I MAR Y			-	-		**********					• • • • • • • • • • • • • • • • • • • •	
				F.4			1000		344					40/5			44/3		
				110			**	1.24		1.		. 34	100.UST.164	44		•.•			•
				Beel		1 432		. 20.0	11.1				Imaust, mt				- 2		
				74.6	***			6.671	46/45			1 34		Sut 1				•	
				*****				*31.0	W/3	145		3.0			2.4				
						******	*****	•••••	•••••	****	******	*****	- (M(D) ()	*****	*******				::

w					• • •		••••	-	211	D(124	DATA	15.61	T RADIUS					- 144	
										20 20								-	••
m/:	-	Po		**	100	110	140	-unio			-	•							
		-	••			***	140	1 30	140	120	100								
*																			
.,	6.6	6.6			6.6	6.6	0.0												
•	0				3.0		0.0	0.0	0.0										
~	77.0				74.6	84.7	07.6	•••	0.0										3
*	***				80.5	02.0		4.0	****										
•	77.4	.0.7			70.0		0/.0	44.3											1.
-	11.0	10.7			42.1	.1.0	83.7		*7.4										1.
•	74.0		4.10		03.1		47.0		**										**
1>		87.3	61.5			61.1	49.5		****	****	102.2								
*									140.6	100.0	103.7								
10							*1.6				100.0								
•				89.7		92.1	**.	94.4	142.2	WA.	164.6								1.
~				44.6		42.1	***	***	142.2	105.	100.7								
*	4.1				94.7		40.4				100.4								
		07.5		49.7			***		161.2										
	83.7			4.0	+1.7	*1.7	+++1		100.2		93.5								
•				4.1	.1.3	*3.0	4.5		**.)										
	85.º				91.7				***	**									
15			44.3	w		**.3	**. 7	10.4	97.5	**.*									
~				**.*	44.4	**.7	**.*		**.5	*1.4	40.7								
2		88.6				**.*	90.7	90.9	**	97.4	07. 7								110
2		.7.7		w.,			45.7	90.7	94.5	.1.7									
~		07.4			41.2		**	95.2	*1.4	70.7	45.1								
•				47.4			94.2	**.1	47.1	89.3	43.4								111
	61.4	85.0				45.0			90.0	87.4									443
•	****		80.7	67.0	84.2	41.2	45.1			85.3	74.1								***
•		•3.•							84.0	4.50	75.9								***
•	70.4	64.6	62.7		80.4		84.7		84.7		73.4								100
•	b. 1	70.7			05.3	***	87.6		02.4		71.7								100
•	73.0		70.1	02.1		••			06.2	70.3	69.>								100
	74.5	12.0	14.4	76.0			42.0	06.0	77.8	79.1	71. 3								***
•	44.7		71.4	72.1	11.0	77.0	70.4	17.0	74.9	73.3	10.0								100
		6.6	0.6	72.6	70.0	14.0	1>	74.2	11.0	71.5	74.3								**
							•.•	4.0		•.•									*?

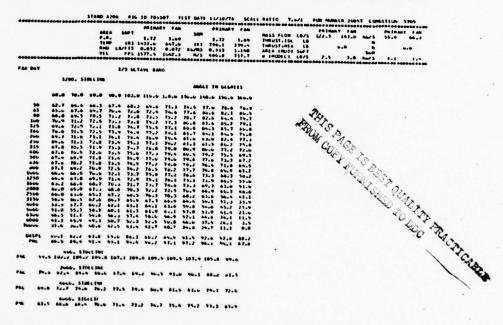
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

						MINA				LINA	* ***	1 4410 7.	******	******	*******	*****	*****	****
				4.	700 L				OM .					****	130.7		bulw	447
				-		1427.6	434			.40	1.32	THE UST . LO		****		4	w.,	
					VIII	6.033	0.67		4) 73		1.14	IMMUST . RE				-		0.0
******	*****	****	- V		***	1276.0	. 11.		15 30	3.3	403.0	M COUNTY	100			340		
							****	•••••	*****		-	# tarper !			******	16/5		
-					/3 WL	-	MD											
	12	w. s	10E L 100															
									DEL! 10									
					100.	· 110.6	120.0	170.0	140.0	150.0	100.0							
50	50					7 45.5	.7.0	49.0	71.0		71.4				S PAG COFY			
**							70.1			70.4				~				
100		4.2	****						74.5	77.0	70.0			I'W.				
125		44.4			**		77.2	74.0	70.0	70.4	77.0			11	C -			
100			**.*		44.		71.7	75.2	76.8	77.7			-	D	D			
706	•3.•	45.4	67.1	40.2	47.	. 71.2	73.4	77.4		75.5				VOA	4/	2.		
256	47.0				69.	9 71.5	75.4	74.7		71.5				- W	On 4	CY		
400	4.,	•>	***	44.3			13.3	74.5	73. 7						URE	. 4.	50	
100		44.2	67.0				73.5	74.3		47.5					* *	Div	50	~
434		42.3	67.2	=::			73.0	73.4								· Un	~	3 27
800	4.0			61.7	40.1	. 71.0	12.0	****	**-	64-1	34.5					-4	Wy	4
1430	41.6		**.*	.7.1							49.5						15	1-
1000	2.1		**.!			4 49.3		44.5	44.5								-	28
2000	57.0	22.5	*1.7	41.1	**		44.1		47.0	>>.								4
2300	>>.6	30.0	30.0		04.4			44 -	30.7		30.7				S PAG COFY			
3150	24.0	>>.1		54.2	60.		40.4		33.0	47.7	77.4							
5000	****	74.4	34.8	**.*	37.		37.6	47.0	44.4		17.5							
4 144	*1	**.*	34.0	20.4	33.4		44.4	30.1	43.2	32.0	12.4							
8000	14.1		****			* ****			\$7.4	25.6	1.0							
·www	23.4	31.2	30.4	14.7	17.	34.3	****	34.5	30.0	10.7								
										3. 3								
POL	74.7	77.0	10.5	74.4		67.7		05.3	65.7		41.0							
	••••	••••				**.1	***	90.7			70.6							
	••	e. 110																
	1 47.4	70.0	100.	101.	. 103	101.	7 142.											
										• •••	•							
4 70.		e. 11	W C 146															
-				• • • •	• •,	.,			79.5	17.	,							
	400	0. 110																
4 4.	67		10.9	71.	. "	. 74.	7 74-1		10.5		,							
										•••	•							
4 17.	41-7	0. 510																
***						.5					•							

a.) Model Data Measured At 15 Ft. Radius

							ME I MAD	* ***			-	A.b.					•••••	
				444		*1			34			PASS 1100	10/1					
				P.A			1.72	1.07		1.1		60 THEUST.16		•	0.0	44/5		*
				-				0-7.u	1.1	140.	1 174					- 7		
				-				0.472										
				***		10	77.5	147.1	./5									
***	*****	•	*****	*****	*****	****	*****	******	*****		******	.7 # (#12011.)	10/3		, ,,,	4-/3		
																	••••••	*****
					1/1	OC 1a		-		-	DATA	IS.UFT AADTUS					220	
-														***		. 201	- (m)	
		72.1						LAUPHUS	-		DILALIS							24
w	••	Pu-			100	116	170	130	140	450	100							*
4.2																		61 -1
40		0.6			6.0				0.0	6.0	•							
**									0.0		6.4							•
••	0.6		6.6	0.0	6.6													
	10.0				63.7													•
43	N.1		4.4	11.0														
••	04.5	89.4	84.1						91.2									**
**				45.5					97.7	101-4	104.4							
*	63.4				.7.6		***		99.7	101.7	104 1							4.00
13				M.7	84.7	*1.		90.1	162.2	104 0	103 2							
	47.5		*1.1															10
-		41.3	71.9	47.5	*1.2	**.		167.1	107.0									1.0
10																		
00	94.5		*1.0															***
•	*1.0	*1.0	****	W	***		1 14.1 - 4											1//
13																		1.1
••	***	73.5	****	**.5	97.6	99.1	14.2 - 1	11.4 4	14.7 7									1
																		144
10	**.*	9,.0	90.1	92.4	97.4		100	163.0			104.5							1.1
15	*1.0		**.1	93.5	97.7	99.4	102.4	14.0	107.4	104.3	104.7							
60		**.2				160.2	102.4	100.3	:	102.5	100.7							442
						***	1. 3	103.7			****							100
*		94.4				-		162.1			**.							141
							101	102.5	101.5	****	****							
	89.1	91.7	93.2	****	**		1.4	101.6			43.0							10
	49.0		92.4					****			*1.3							10
			*1.0	91.3				****										114
			W-/					****	****	**.0	• • • •							11.
		81.9		91.1	92.4	***		****	****									114
		84.1						*2.1	****	**.*	• • • •							
		84-1		47.0			***		****									
					00.7				****	85.0								111
								04.1			19.3							100
		14.6	77.1		80.4	***		80.7			70.7							100
•.			0.0					-0.7			70.0							1.1
				2.0			0.0	•.•										

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

						-	I RAB T	*			-							PAIRANY	
					341				348			8455		40/5	0.0		41/3	w	
						. 1.		1.N		1.7	1.	- 1-00	ST. 164						
					401	1 1005			101		170.		ST.814						
				-	40/011		. 12	w.w. Po	46/43	w. >1	. 1.1.		INUCI	241			**		
				WEL		1011		-01./	NS				CHIL	11/5	1.0	1.0	/>	1	1.
****	*****	*****	*****	******	******	*****	****	*****	*****	*****	*****	********	*****	•••••	********	*****		••••••	••••
					W1 0		-	-	A1 0				aus.				. :	- 1714	
-																			
		-									*****								.1
-	-	10	•	**	100	110	130	1 10	-	130	100								
200	4.0					6.0		6.6		6.6									
		4.6		6.6					6.6	0.0									1
-											6.6								
-	77.0			M.7			.1.			**.7	.1.1								
	Du. 3	****		****		.1.5					**.1								٠.
•			4						****										
.00				40.7	44.1				**. *										
36	••••	****					**.6		166.3										
13	• • • •	90.0							163.4										
~	*1.0	****	****	97.7	***				100.0										**
-	**.*	****		w	**														:
	*1.7	**.1							111.2										::
.00	94.7	**.		***	97.9														
.25			**.1	*7.1					110.3										- 63
		**. 1							400.9										
									107.7										1.
. 500	41.50	**.*	**.*						107.0										
.15	94.3	94.0	95.3	.7	90.0	1.1	161.4	164.4	103.6	14.2.6	100.1								
*		4.2			7	14.5	104.2	102.0	105.1	144.1	101.0								8.
		*1.0		w.5	20.3 1														
. 30		93.7			40.3														
									107.0										.,
•••		*2.*		W.6					101.7										
	*		***	*>.3					100.0										10
•••	**.*		•2.0						****										• •
	***	**.		***					**.*		86.0								::
	*:	47.4		71.1		**.1	97.4			**.*	41.9								**
	47.5						***			47.3									**
				84.2		**.*				41.4	.1.5								**
		19.4								41.7	80.4								14
6.0	73.6	75.0	70.2		01.5														10
٠.			0.6						•.•	•.•									

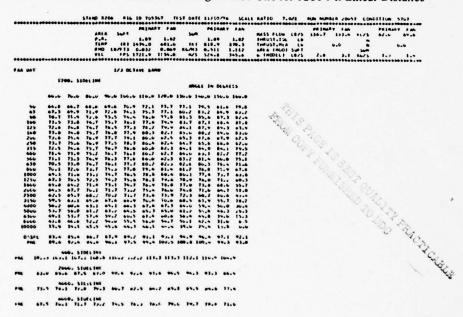
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

											-						100000000000000000000000000000000000000		T PAN
				44					16				4455 FLOW	L0/1	170.9	1.0.5		37.4	47.3
				• •			1.78	1.74			. 76	1.74	THRUST.IL						
				111			1445.0					370.0	Deust, mta		•				
				76			6.632	1003.4		3 0.		330.0					34-74		
•	****	***	*****	*****		****	******	1007.2		•			# (MCD()	18/3	7.6	3.0	46/5		1.
44 D41						3 OL	-	me.											
		12	De. 51	M . IM															
									-	41 1	. 44	***							
			70.0	00.0		100.	. 110.0	170.0	130.0	140.0	150.0	100.4							
34				67.2			10.7	72.4	75.0	90.7	80.0	10.2							
• 1				10.0	71 .4				70.0			4.10							
-				"	22	71.			74.6										
125		77.0	77.0	73.3	74.0	74.			•7.1	****									
100			12.7	7	n.0	70.			04.3	•>.•	• 7. 1								
/00		71. 1	****	74.7		77.	10.7		***	**.*	***								
234		71.4	73.1		73.3	10.													
111		4.70	72.0	11.4	Po.6	70.		84.5	07.2	.1	74.6				1				
***		4.0	74.0	73.5	14.6	70.	. 70.5		04.1		70.1				400				
304			11.7	77.0	PO.A		1 10.4			19.2	75.0				20 1				
. 34			71.2	77.0	74.5	70.				70.7					40. V				
1000		67.6	10.4	24-1	71.0	73.			74.)	10.4	74.1				11	40			
1234		**		71.7	77.7	74.			76.7	73.7	74.0				40	49-			
4000					74.7		. ;;;;		72.0	71.4	45.4				4	, G			
2000		.1.0			10.7	71.			74.7							O. 8			
2500			64.2			Pv			76. 1	47.4	50.1				,	1			
3134	•	10.0				.7.		44.7			53.4	35.7				*	15		
4066		m.7								. 50. 4	.7.4					N'	. 0		
5000		34.1		54.6			41.7		34.1	>1.4	****					- 0	0 4	Ġ.	
4300		•7.•	22.5	****	21.0		* **.*				34.4					,	W. 1	60	
luvi (c)		W.,							4.0	19.0	20.7						16.3	3	
				****			* ****	-2.0			1 10 1						10	. 0	
LASE		41.>		84.8				*1.5	*1.4	****	**.						1	13 6	1
~			.,	*7.7	m.,	**	* *1.7	**.*	**.*	**.*	****	64.4						60	2 x
																			4.
-	In.				1 14.		111	. , ,,,,	, 160.	, 100	101	. •			Alta San Co.			10	1.0
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-		• • • • • • • • • • • • • • • • • • • •	* ***				v. • • 2	42.		* **.	., .,								0
•	71.7	/4.		• "	, m.			., .,.		• • • • •									
100	45.1		DO. 51				10												

a.) Model Data Measured At 15 Ft. Radius

				22.50			IMARY	***		**1	-	44			1-4			M . MA- 1	
									***				MASS PL	10/3		6.0			v
				1				4.02					IMUSI.						
						1 1414		1.0		.10.			Immust.			0.0	•		0.4
				TIL		. 0.0			WCW?				APLA IM				***		
•••••	•••••	*****	*****			17/1		1 M.u					W (M(E)				4 /3		!:
					1/2	414		-			LATA		AAI-105		***			- 1-41	
-																			
							MIC	- OPHON	AMEL	15 14	M6+1+5								
M/ 1	••	70	•0	+0	100	110	110	1 30	1-0	150	1 40								
150	0.0						٠.٠												
• 3							6.6		4.6	4.6	6.6								
-	0.6		0.6	6.6	0.6		6.6		0.0	0.6									
•	W.)	****		*>.7	***	***	.7.4	***	47.4	**.1	**. 1								
25	67-6	***	45.6	**-7			•1.•	41. 1		40.0									
*	t	11.0	84.1	44 -1	44.7	****					101.6								•
~			**.6		47. 9			**.?											• •
15	.7.3		**	w.v	****			100.7											
			11.3		¥3.7			100.0											
~	*1.7		**					104.9											•
14	****	**.7	****																1
		44.7			**.*														- 11
			97.3		**.5														- 1
19	***		**.5		100.7														
••	**.*		**.*		100.4														1.
00	99.7	97.0	97.7	90.5	100.0	104.4	105.3	100. >	140.4	11/.7									4.
>0			**. 3		100.2														
13		****	97.2		166.1														
4	94.3		*7.2		100.7														
00			**.*		**.*														
30	*1.0		**.*	*1.*				100.2											. 1.
00	97.5	**		**.				10>-6											• •
::	*1.*			.7.4				164.7											
::		97.7	**.*	w				103.5											٠.
	BY							100.											44
	67.0	w	*7.4	-:	*3.4			98.1											
:5	80.4		*1.1	92.0	****	****	97.3			****									::
		86.7	**.		92.4	**.*	97.5			**.*									- ;;
		****					92.5												- 83
			43.2	82.3	04.7	87.7	89.4												1:
		77.4	19.0	80.9		84.5	45.2												
4.	0.6	0.0		0.0	0.6	6.0	0.6			0.0									•

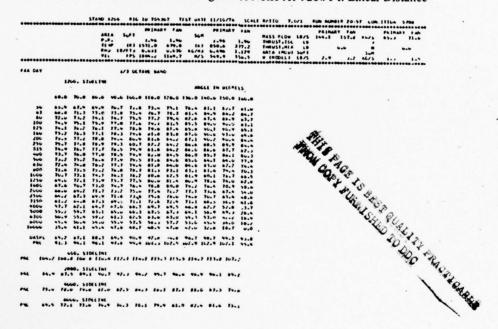
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

											HART FAR			PPIMAP	* 144		F6 1 544 T	
									34.00			MASS F44.0	10/5			44/5	u.+	3
								1.00		1.*		THEUST . TEE		-				
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				410				0.6 Pc		U. 4W		ARTA (MUL)					-	
				Vet		. 100		1100.7	M/5	>44.		W (ALDEL)	10/>				1.1	1.
•••••	*****	*****	•••••	*****		*****	•••••		•••••	•	*** *** ***		******	******	******	*****	********	••••
												CONTRACTOR OF THE PARTY OF THE						
-								, work		10134	15.0	PT NADIUS			*****	4 201	- INCIA	
							-11	-										
1 344	••	10		-	100	110	170		146	120	100							
10000	75			-		•••		•	•									41 -
-						6.0				0.0	0.6							
(-)	0.0		0.0		0.0					4.4	4.4							٠
-		6.6	6.0	4.6	6.0													•
140			44.7	.1.0	.7.0	.1.					160.4							
10		01.2				**.1			*>.*									•
100			.7.7															:.
~~						*1.0	*3.0	****	166.4	103.7	104.5							- 11
·								****										10
112								162.6										
~			**.*			**.5	100.	105.0	110.1	111.0	119.5							10
			**.					104.7										
***	.7.4				**.1	101.0	100.	w.,	114.4	117.0	414.3							47
***			**.4		100.3	101.7	104.	109.4	115.5	110.5	110.3							
	.*	**	**.7	w	101.3	103.0	100.	116.0	110.1	110.7	114-1							11
	****	****	100.4	100.0	107.5	103.0	107.0	111.0	115.7	110.	415.4							43.
	****		100.0	100.4	107.4	100.0		110.7	114.7	110.0	110.7							
.~		****	****	100.1	101.9	104.7	107.	110.4	117.7	110.3	114.5							11
.15	****	****	****	100.4	102.0	104.7	101.	100.0	*****		1.3.2							
	**							104.7										1.
.00								100.0										10
. 10	***		90.0					100.0										40
	94.2		*1.7					107.0										14
	93.7		97.>					100.0										
4.5	93.0		97.0		144.1	103.5	100.	105.6	104.9	147.0	94.7							14:
			94.2					14.0										17
0.0	10.1		**.*					162.0										1
1.6								100.9										12
1.5	.7.7		93.0					99.2			07.2							11
			41.1		.4.4	***		97.3			45.4							-
	41.2		84.4				95.0		*1.1	90.9	84.0							111
3.0			85.4		64.7						85.1							- 111
											84.0							10
w.	4.6	6.0	0.6	0.0	6.6	6.0		•.•	0.0		0.0							
																		1
																	DAPEL .	14

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

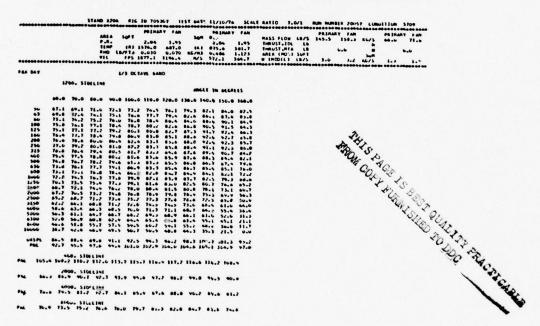


Flow Simulation; Configuration 3A; Condition 5709

a.) Model Data Measured At 15 Ft. Radius

### 13 200								WILT.					***********		Pa 1 64				
### (474) Color with a 11 Plant 12 12 12 12 12 12 12 1										SUR			MASS FLOW	10/1					
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### (45/4) 5.001 Lubb Actiful Chee 11/2 Ana 11/2 1										(4)									
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					441					-/5	\$77.	1 100.7	# 4miz.ts 1		1.0				
### ### ### ### ### ### ### ### ### ##					******	*****	*****	*****	******	*****	*****	********	**********	*****	*****				::
### ### ### ### ### ### ### ### ### ##							~												
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0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0		tate																	
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1						•••	•••				430	100							
1	,>O				0.0	4.0					4.4	4.6							
5	• >																		
5	-				0.0														
3	-0						84.1												
18.1 18.1 18.1 18.2 18.2 18.3	43							w		*7.4	144.4	147.6							• •
\$ 10.0 \$1.0 \$1.0 \$1.0 \$1.0 \$1.0 \$1.0 \$1.	~								97.1	94.0	107.4	104.4							
Sec.	60							**.1	****	104.1	100.0	107.5							
Section Sect	20			w.,	*1.4	*7.4	+3.0	**		144-4	14.4 4								
1. 1. 1. 1. 1. 1. 1. 1.	15		₩.		•3.0	94.0	90.3	**.	102.9	107-3	111.4	114-1							
91. 91.0 91.0 101.0 101.1 101.0 101.	00	****	****	**.*	***	90.0	40.5	101-1	104.2	111.3									
70. Va.0 Va.0 Va.0 Va.0 Va.0 Va.0 Va.0 Va	•	**	***	**.*	47.3	40.7	100.7	102.4	. 107.6	113.6	111.5	115.7							
10	10	40.5	40.0	70.0	**.	100.2	102-1	10.	.10.4										
1		**	44.0	70.7	100.4	101.4	144.7	145-4		117.1	13								
10.0 10.1 10.2 10.2 10.3	•	****	****	100.6	101.3	102.5	105.1	101.1	112.5	116.1	156.0	117.3							
		****		101.6	105.0	14.6	163.6	100.4	115.1	1.7.4	150.5	117.2							
1 1 1 1 1 1 1 1 1 1	=	****	101.4	102.2	105.0	163.9	105.9	104.1	1112.5	110.0	120.4	417.0							::
10 10 10 10 10 10 10 10	**		100.8	102.4	102.3	163.7	165.7	1:0.0	1115.5	115.5	114.3	110.7							1.
0 11-0 4-0 100-2 101-3 102-3 102-3 102-3 102-3 11-2 11-2 11-2 11-2 11-2 11-2 11-2 1	13	=:			102.1	167-1		100.	115.1	114.5	110.2	115.0							12
10 10 10 10 10 10 10 10				100.5	101.7	103.6	100.1	100.1	111.0	113.5		113.1							1
1	=	.7.1	**	****	101.3	103.0	100.3	104.3		114.3	114.0	110.7							
	=		90.		101.0		100.2		110.4	111.6	113.5	107.5							10
	~	***			140	10.2			104.0	110.7	140.0	103.5							.7
\$ 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.	=	20.4	47.2	***	166.5	107.5	105.0	167	104.5	104.6	100.0	103.7							10
9 93.4 93.7 97.3 97.5 107.5 108.5 108.7 108.5 108.7 108.5 108.7 108.5 108.7 108.5 108.7 108.5 108.7 108.5 108.7 108.5 108.7 108.5 108.7 108.5 108.7 108.5 108.7 108.5 108.7 10		94.2			99.9	163.1	104 4	164	100.	100.7		101.0							12
10.7 Vector State 10.5 16.7 18.4 16.7 18.4 16.7 18.4 16.5 18.5 18.5 18.5 18.5 18.5 18.5 18.5 18		93.4	95.7	97.5	44.4	101.5	100.1	105.	101.0	100.0	105.)	**.0							440
10 00.3 03.0 09.0 07.3 09.3 101.5 107.6 10			**.*	94.4		140.2	142.4	144		103.7		70.7							1.
0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0		96.3	93.0		97.5	99.1	101.4	102.	102.0	1011									10
\$ 60.0 80.0 97.7 96.3 96.3 96.3 96.3 97.2 97.0 87.0 97.0 97.0 97.0 97.0 97.0 97.0 97.0 9	.5		.1.0	94.4	94.3	90.1	100.1	101-4	101.3	100.		****							1.
6 64.4 87.4 90.1 91.5 97.9 97.8 90.2 90.2 90.2 97.4 97.5 97.7 97.7 97.7 97.7 97.7 97.7 97.7		80.7		92.7		94.3	10.4	w. 1	94.4			***							
0 01.1 04.7 04.9 09.0 09.7 97.1 97.4 97.5 97.5 97.5 97.5 98.6 97.7 98.6 01.0 01.0 01.0 01.0 01.0 01.0 01.0 01		84.4	87.4	96.1	91.5	*7.*	99.0	**.	90.4										11.
5 72.7 80.6 83.6 84.6 85.7 85.6 80.1 80.1 80.6 80.2 83.5 83.5 11.6 80.8 80.8 80.8 80.8 80.8 80.8 80.8 80					89.0	90.7	92.1	*1.4											
. 9.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.		77.7		83.4	4.7	47.1		wo. 1	90.1	80-4									111
DAPUL 4 14	•.	0.0		0.0	0.0	0.0	0.0	0.0	0.0		6.4	6.6							116
						1000	-	1000	-										•
																		-	100
				201															

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

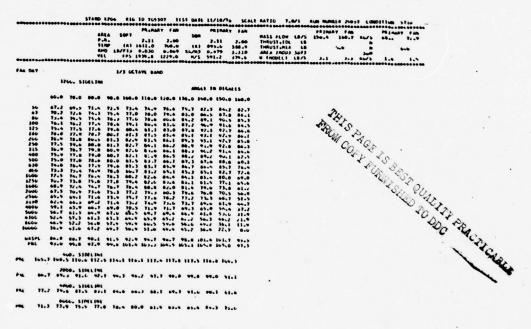


Flow Simulation; Configuration 3A; Condition 5710

a.) Model Data Measured At 15 Ft. Radius

				444			-						1000000	P# 1844				
							.11	4.00	700					4.6			*	
				11.0		. 101		Nw.6		7.1		THRUST-16			2.0	•		
					LOVE				14 /M			110,120,001						
				714		3 193				>91.		-		2				
••••	•••••	*****	•••••								370.0				•••••			
					1/3	OC TAN	-	-			Ala 15.	CUE CAP T 40					- 1944	
***							-											
LAILE							MIC	A OPHEN	-		264115							
ant I	•0	70		90	100		150		100	134	100							
650	4.6	0.0	6.0	0.0	0.0				0.0		0.0				100			
•	6.6	6.6	6.0	0.0	6.6	u.6	w.0		6.0	0.0	4.6							
40	6.6	0.0	0.0	0.0	0.0	0.0	0.0		0.0	0.0	9.0							
100											tuz-u							487
145		****			***	*1.0				100.7								
100										m2.0								
250		87.7 8v.3																4.4
315		91.1			****													
446		****			**.*													1.0
>					***	100.0	164.1	147.4	114.									
•30		***			100.0		103.1											131
-	40.1	**.		141.4	101.9	104.6	164.7	****	*****	126.1								in
-00		100.4	100.1	141.4	107.9	144.4	100.2	111.0		1.0.4	117.4							1
-3>					104.5													110
-40	100.4	161.8	102.5	14.9	104.4	100.3	109.1	113.0	117.0	170.9	110.1							433
-90	100.1	101.1	101.0	164.7	104.0	106.0	104.3	112.7	110.3	120.0	117.1							in
-36	**.2	14.3	101.1	162.4	104.2	104.4	164.6	112.7	115.4	114.4	115.4							. 11
-15		99.0	106.9	14.1	102.0	104.5	109.7	117.1	114.2	117.2	113.4							4 37 .
-00		**.4	100.4	805.0	103.0	106.7	100.7	111.0	113.3	115.4	110.4							4 34 .
	47.5	**.4	**	101.1	163.6	100.0	104.3	110.0	117.4	111.1	107.4							
- 36	97.2	78.5	**.7	101.5	163.5	16 0.0	104-1	110.4	111.0	111.3	105.0							in
-00					103-5													
					165.4													140
2.5					107.5													441.
		94.0			102.0													140.
3.0		94.1	**.0		99.0													1.0
1.3		92.1																121
		90.1																110
0.0		87.4	90.4								87.7							1.7
3.0		84.4	87.2	5.00	90.9	92.6	94.0		***	92.7	84.2							114
4.0		41.1			47.4	49.3					85.5							110
₩.					6.0				6.0		6.6							

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



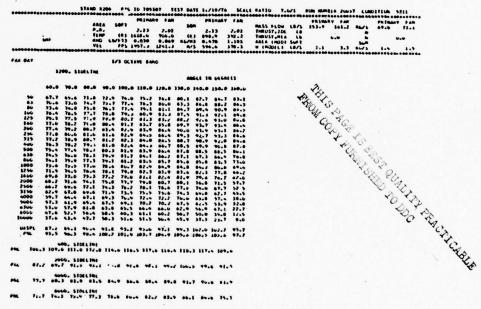
Flow Simulation; Configuration 3A; Condition 5711

a.) Model Data Measured At 15 Ft. Radius

							1				-						
				444					200				MASS FLUM	6.6	 ALT:		
				11.0			.13	2.01		é.1		or	1MRU51-114		 :		
				AMU		13 -			86/83				MIN TOUM				
				VIL		13 145		291.3	P/ 5				. INCOLL		 200		

											the backs			 	 		
					1/3	CLIAN	-	MLOIL.	111	MIM I	-	15.00	T RALIUS	-	 U 301	- (#164	
MeO.																	
								A UP HUR									***
en! I		70	••	**	100	1.0		134	1-	136	100						
636	0.6	0.0	0.0		0.0			0.0									
	4.0		0.0		6.6		0.0				6.6						
-	6.0	0.0									0.0						
100	04.3						*1.1	93.7			142.2						443
125		64.2	44.4		41.3	*1.4	w. 5	**.1	40.0	101.1	101.1						
100		90.9						93.3	90.7	103.1	100.1						
100		10.4			+1.1			97.0									
730		44.0			*1.1			****									
363				***				us.7									
•				97.6				100.0									1 1
100	**.1							100.3									
• 34								111.3									
•••		****		101.1	105.5	10 3- 5	103.4	117.0	110.0	Du	110.0						
.00								113.0									13
								113.4									13
								113.1									12
.*								113.6									15
.15								117.0									13
								114.2									
								111.3									1 .
. 10								110.9									434
.00	94.7	10.0	100.2	101.7	163.6	100.0	100.1	110.3	111.2	110.0	104.4						14
	95.7	**.1	**.	101.5	161.5	166.2	160.5	100.6	104.7	106.5	107.5						120
2.5	**.2	97.5	99.4	160.9	103.0	105.7	167.7	160.0	104.1	100.9	100.4						47
								167.9									
		**.3	97.2	44.5				100.1									141
5.6			***					104.1									443
1.5			93.3					102.6									4.0
			*1.7					100.4									170
	***	**.						90.1									4.1
3.0		****	87.7			*>.1		9.2									***
			***		40.3				*1.0								•
		****		•.•				•.•	4.0	•.•	0.0						
																-	143

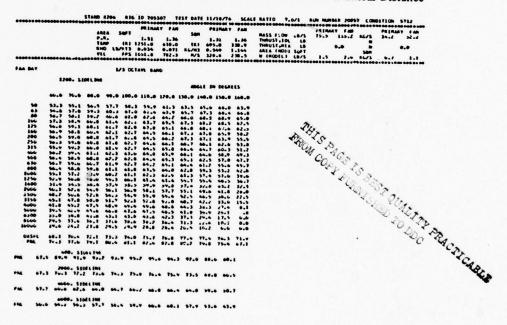
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

				***	1 14						MARY PAR				PRIMAD			POINAL	
							-54		360			MASS (LO/S	v.v		44/1		
				1.00		1 125		1.30		4.5		THEUS	1.104						
				-				0.071		***.		I HOLUS				0.0			
				711		5 105				0.34		AREA					344		
****	*****							107.3	4/5	>10.			DILI	10/>	1.5				4.1
									•••••	*****	*********	•••••••	****		******	******	*****		*****
										2002200									
-								-	*	W176	DA14 45	LOS RADIO	us.		RAD (BRECTE	0 304	- 184	444
1 14	-	70		**	100	110	176	130			DIGREES								***
			-			•••		130	140	150	100								84 -4
100		6.0																	07000
43					4.6		6.6		6.0		6.6								
-		0.0		•.•	6.6		6.0		0.0	4.4	6.6								ī
	.7.4	71.4	73.0	70.0	74.1	75.1	77.1												
25	70.3	77.0	15.4	70.5	13.5	77.0	70.3		07.1		***								100
••	12.2	14.0	74.0	h.1	79.1	75.2	70.1		***		**. >								141
•	14.3	73.3	75.4	D.1	70.0	70.3				****	60.4								
34	71.5	79.4	74.1	27.4	77. 0	19.7			87.0		*1.1								1-1
45	n	70.0	77.0	P	79.7	.1.7					*1.0								14
	11.3	10.0	84.7	81.4			***		90.2	*3.4	*4.7								100
•	79.2	70.0		41.9			44.7				*7.4								
-			04.3				88.2		*3.5		*5.0								
	74.5	41.2				45.7	67.6		*3.4	***	93.7								414
	20. 7	84.9		45.4					*3.5	*5.7	*3.7								221
	10.3		84.4	01.4	84.4				*1.1	*3.4	00.7								
-	19.2			41.4		44.4	89.4	****	*2.4	*1.0	67.0								410
	29.4			41.7		er . •	87.0				85.6								110
*	70.5		63.2	84.2			89.2	90.8			03.7								100
15				84.1	85.1		44.0		90.2	67.7	62.2								100
		07.0	5.40		03.4		4.		**.1	84.2	41.2								100
			04.7		4.7		****			4>.1	74.4								100
	79.0		42.2	81.4	M. 4		67.4		***		74.1								44
	70.1	80.3		W.9		93.7		87.4	84.3	62.3	76.7								107
	77.5	10.4		82.4				83.1		**.*	74.7								100
	77.3	70.7	84.2	01.2	67.2	84.0	84.7	43.4		70.0	72.0								100
		77.0	Po. 4		01.2	62.9		2.4	79.5	77.2	70.7								1.4
	74.0	74-1	77.4	70.0	70.7		*1.7				40.>								
		15.4	74.0	77.9	70.7	20.0		10.5	77.3	73.4	65.6								1.4
		73.7	13.3	70.4	77.5	70.3	79.0	74.0		71.0	M.L								1-
			73.7	M.7	75.7	70.0	77.7	D.)	74.1	76.0	42.4								
	41.3		71.1	71.4	73.0	73.0	m.;	72.7	49.4	4	.1								**
			47.0		49.9	70.4	71.0				43.5								-
			44.4		44.7	47.3	67.6	47.4	67.2	***	44.0								
		0.0	0.0	•.•	0.0	•		***	4.5	63.6	62.7								-
0.71	-								•.•	•.•	•.•								

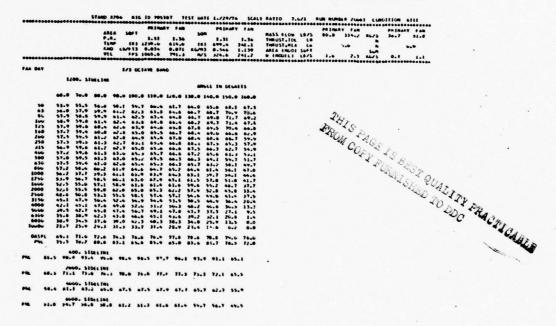
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

								,		201	-				PAIRARY			PA I MALY	
									SUR					10/5	0.0	0.4		v.u	
							. **	1.30		4.1		1. 10	Testust . ILL	10					
				1.00		11 137		.14.0	(4)			.1.1	THEUST , MIA		**				
				TIL		3 6.0		6.671	46/43			. 1 10	ARLA IMULI				34.M		
	***					\$ 100		741.3	4/5			-1.4	. IMMELL		1.0	1.1	1445		
						*****		*****	•••••	•••••	*****	*****	**********	••••	********	*****	*****	*******	***
						~ ***							#1 04DIUS						
-							-	-		401.25		15.6			-		a 24.	- IMM	
							-11		-										
1		74		**	14	110	126	136	14	136									
				-															64 -1
**				4.6		4.6			0.0	4.4									
263		0.0	0.6	0.0		0.0													•
-	0.0			4.6					6.6										
14		72.0	74.1	n	79.4	74.0	77.7		01.7	05.7									
153	71.1	75.0	P3.4	D.0	70.0	77.7	Po. 0					•							1
***	13.5	15.0	74.0	D.4	77.3	77.0	77.4	7.1			¥1.								10
tub		73.7	79.2	76.7	70.6	70.0				97.7		,							
230	13.9	75.5	10.0	70.0	74.4		82.1			*1.0		•							
***		17.3	10.2	P		.5.5			41.6	**.*									11
-		10.7	.1.5	4.5			87.1				101.								
136				W.7							100.								**
106		01.7				44.7	84.1			**.0									11
	W. 5		11.*							**.0									
		41.6				.7.0			94.5	**.									
			*2.4						94.1	**.*									
			63.2	M.)		.7.4													1.
	00.5	07.1	01.4	m.;		44.2	**.			.1.3									
		.2.3	03.6	6.1			10.0			*6.1									
		82.4	03.5	43.0	47.	44.7	***		96.1	67.0									1.4
.00		81.8	83.3		87.5	44.1													
.10			82.4		07.2		m.4			43.4									
.00		81.0	42.3	41.7						84.6									
		80.1	61.6	83.2		84.3	67.7			87.5									100
1.5			84.3	07.1	W. 1	91.2			84.1	41.2									
-0		77.4	PO.6	01.6	84.4		85.4			79.0									10
	74.0	74.2	70.0	P. 7	42.5		63-2			74.0	Pa.								10
	73.0	79.3	N . 9	70.4	01.3		42.6			74.7									100
1.5		73.5	7.0	17.0		79.9			76.5	73.0									12
		71.9	P4.2	D.4	70.7	70.1	70.5	17.2	14.9	71.7									
		10.1	72.0	73.6	70.7	75.9	N. 1	75.4	73.1	70.4	44.								**
		67.5	44.3	70.0	74.7	70.0	73.5		71.0	74.2	71.1								-
0.6		7		67.5	10.0	70.7	70.5		70.4	47.6	71.0								•
.			•.•	0.0	4.0	•.•			0.0										

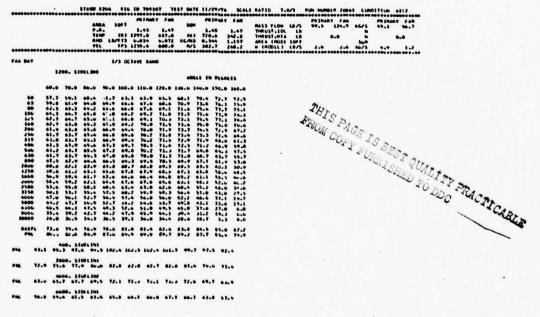
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

											-					PA IRA			PF IMAL!	
				4014					Sun				MASS #	LUM	10/5	4.0	0.0	44/4	0.0	0
								1.47		1.0		. 47	THAUS	1.14						
				1100		1 124		.17.0	141				EMBUSI					•		
				-		3 6.0		6.612	*CMI			157	AREA I					**		
		*****		***		3 1431	•••••	*****	s		7 /4		-		16/3		7.6	46/3		!.
						~					0414									
-					.,,							15.0	T PADIC	18		•**	Contra		- 194	
							410	AUPHUR	-	15 10										
-	•-	1			***	110	14	1 36	100		166									14 -1
-		•.•		0.6	0.0		6.0		6.0	6.0	6.0									
	4.6	0.6	0.0					6.0	6.0	0.0	0.0									
Ne0	0.0	0.0		0.6		0.0				4.6	4.6									
luu	71.0	15.6	17.0	70.4	70.0	79.1					97									
25	74.3	70.0	70.0	78.0		41.5			87.4	**.*	93.7									
-	77.2	77.4	70.0	P	76.7	74.4	w			45.0										
*	77.4	79.1		80.7	63.6	62.9	**.1		*1.7		**. 7									
43	84.2			03.0	84.3	43.7	80.0													
	82.5	81.7		80.7	88.6	47.4	91.1		****	100.1	103.3									
44			43.4		44.2	**.	*1.4		100.0	107.7	106.7									
J w	87.6		07.3	44.4	84.7		*1.1		101.6											**
-	45.4			84.5	90.3	91.0	93.0		100.9											::
	85.5	85.9			90.8	94.1	94.1		100.5											::
23	4.7		67.2	80.4	91.0	92.1			100.0											**
••			87.1	80.4	91.3	92.4	90.1	97.3	99,)	**.>										
•	84.2		87.3	80.7	91.5	92.0	4.1		98.0											i
>0			87.7	84.2		92.0	94.9	97.0	97.4	45.7	*1.*									- 11
15		84.5		09.3		93.6	**.*		***	**.>	*1.0									
•	05.5	84.7		80.4	92.5	93.6	99.1		**.*											
••	45.7	**.)		80.2	42.2				**.*	45.7										
10						.5.0	**.2		*3.7	•1.4										
-0	43.0	84.5		10.4		.2.2	93.7		92.7	**.*										
:	83.4			87.9	*1.1	91.5			91.2											
::		13.1		***	*2.*	*3.5	97.6		**	87.4										
		81.4		m.:	87.0	*4.0	49.0		64.2	***										8 40
	70.7	80.3		53.7	00.7	87.6	87.7		84.5	81.3	77.0									
.,	74.7	70.5		62.3	84.1	87.6	84.2			79.5	73.9									1:
	75.4	77.1			4.5	84.0			80.4	77.0	72.5									
	73.2	74.0		70.2	02.3				70.4	74.2	72.7									10.
	70.6	72.1	N.3	D.0	79.4	70.4	79.0		75.7	75.3	74.1									
	47.4	44.9	71.3	72.0	70.2		P		73.4	74.4										
·-			0.0	0.0			0.0			0.0										7
											140000									

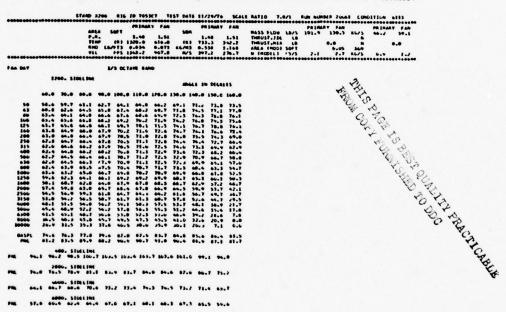
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							-	-		-		- 44			PR 1 TAR				
				-		•			34.0				MASS FLUE	10/1			44/3		
							.40	1.31		1.4		. 11	THEUSE . IU						
				1100		11 130			(4)	711.	1 1.	2.6	THOUST . P.						
				-				4.073	ELAS	0.75	. 1.	100	MALA INUL	1 544 1			240	•	•
		NOCENO.		AFF		* 12	1.2	wo1.0	WS	141.	2 21		-		2.1	4.1	44/5	U.V	
	*****	*****	*****	*****	*****	*****	•••••	*****	•••••	•••••	*****	•••••	***********		*******		*****	********	••••
													** *****						
-					.,,							13.0	A MALIUS			Metch	- >-	- 1014	
							-	ACPHON	-										
-41	•	To		**	160	116	120	134	144	110	100	•							1.
***	6.0	6.6	0.6	6.6	6.6	6.6													
43			•.•		0.0														
*	•.•	6.6	0.0				0.6												
00	77.0	70.3	70.2	N.0	19.2	19.0	01.3												
2	72.1	74.7	79.3	70.1			01.2			.1.)									
=	77.1	70.1	P. 4	01.3	14.1	70.0													1.
~	70.3	19.0					****		92.7										
-		41.5		62.0	43.7	***	44.1		**.2		101. 3								
•		***	4>.9	67.4	84.0		*7.7		**.7										••
-			84.2		49.0		92.4		101.5										1.
×	44.1			84.6	VG. 7		**.		107.2										40
~				89.4			**.6		162.1										٠.
66	84.4			49.4		9,.1	95.4		161.6	144	14.								
23	85.0	84.9	44.0	89.4			***		101.0										**
*	02.7		44.4	m.4			92.1		106.1										::
	5.40	87.0	84.2	89.7	92.5		99.1		99.2		**.*								**
30	85.7	07.3		90.1	+2.9		**.2	90.1	**.5	97.0	93.3								**
45		87.3	88.7			94.0	96.1	97.7	47.5		92.3								**
"		87.5			95.4	90.1			94.8	94.0	91.2								41
•	80.0	87.3		10.2			99.9	**.	***		90.0								- 11
36	84.3	87.0				94.0	95.4				89.1								**
			.7.9	.,		*1.5	95.1			45.3									44
		05.0	67.3							.1.0	.7.0								11
••	***	85.7		10.0	*3.7	*1.7	*3.*												
::	**.	•1.•	****	67.0	*1.0	*1.3				87.3	02.7								
	14.0	***			89.3	**.*	***			85.1	79.6								13
:	77.0	79.7		85.2	67.1		80.2				17.0								10
::	74.4	70.1		42.2	45.7		87.4			79.5	75.6								10
:	74.3	75.0	78.2	70.7	13.7	82.9	43.3			77.4	74.5								10
::	71.5	73.4	75.3	77.4	04.3	79.4				77.4									
	44.1	49.4	12.5	73.4	17.5	74.8	77.2			70.2									14
					-														•
200						•••	-		•••	3.0	•••								•
																		-	

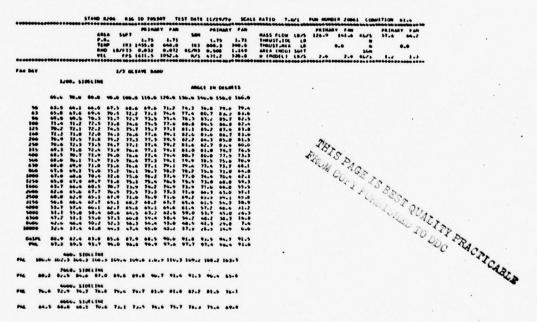
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							I I MAR !	*		**1	MAY FA	•		PRIMARY	-			
				-					140			MASS FLUE	10/5			44/3		
				P.A.			. 75	4.71		4.7	. 1.7	I THOUSE, IN	1.0					
				***		. 144	9.6		141	860.	3 300.0	I IMAUSE, ME		0.			0.	
				-		11 0.		0.072				· AREA INCL	1441			Sug		
				44.6	•	PS 161	1.2	.w.	N	441.	320.		10/	4.4	1.4	4.55	1	
****	****	*****	*****	*****	*****		•••••	****	•••••	•••••	*****	***********	****	********	*****	*****		****
					143	-						CULDAR T TO.C		*** ***				
-									•••					RAD CC		O SPL	- IMICE	
	-						-	-	-		DIGALES							
I ma		70			100		130	130	140	130	140							4
			• • •				6-6				4.6							•
-			0.0		0.0	0.6					6.6							·
=	77.4		43.7		0.0	•.•	• • •				0.0							
	79.3	****			84.1	****					**.6							110
100					43.7	43.4					101.3							44-
7.4					47.4	***	w. i			142.5								
.74	***		45.0															,,,
	****				**.				**									
	44.3				93.4	****		101.0			110.4							
=	***		91.7		94.1	99.3		101.9										150
-								104.4										150
-	92.9		93.7		97.7			104.9										
		93.9	99.4		94.2			109.										140
			93.4		98.0			100.1										140
-40			95.2		90.0		101.	105.7	104.0		*****							1.7
	97.5		94.4					103.4										120
- 20			94.0					105.2										1.0
			w			100.1	102.1	104.7	100.7	104.3	100.1							1.4
			94.1					104.3										1/2
-00		92.4	93.9					194.7										122
								103.1										1
			93.3					102.4										10
	49.4	91.4	94.0		90.4			101.4			99.0							121
2.5					90.0			100.5			97.8							120
			91.4		97.3					94.7	*1.0							117
-		80.7			***				94.4									117.
13.0			00.4		**.					92.4	47.1							114
11.5							95.0			****	49-1							110
	42.1				97.7		93.1			87.2								113
	79.8				90.7					94.7	43-2							110
13.0	77.0	79.0			87.4		07.6				84.0							1.7
10.0	72.7	79.7	70.4		84.2													104
		0.0	0.6															
											-							
																	DAPOL -	1 34
						_												

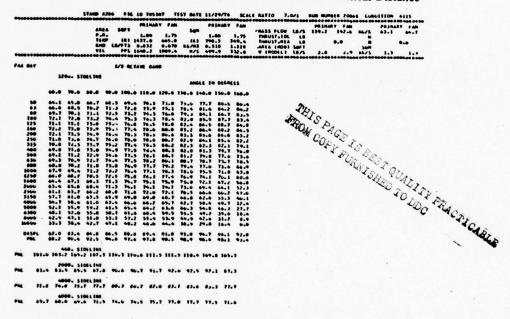
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

				-	144				-	-41			MASS FLUE		PRIMAN		444	v.e	
								1.75	-	1.0		25	THRUST. IC		•.•	•	:	•	•
				11.00		1 4437							THEUST-ME						
						. 0.0			86/M3				MALA IMOL				-		-
				411		3 1000							. INCUEL!		2.0	7.4		4.7	
****	****	*****	*****		****	*****	****		*****			****	-	-	*******	*****	*****	********	****
					1/3	CL TANK	-	-		-	ATA	15.00	T RADIUS		-			- !	
-																			
	fatt					100000					DI GAL I S								PL
and I	••	70		••	100	110	150	1 36	144	100	100								
-10	4.0	0.6						•.•	6.0										
9	•.•	0.0		0.0	0.0		6.6				4.6								
	•.•	0.0	0.6	0.0	0.6	0.0	6.6		6.6		0.0								
143	10.4	2.1	84.1	M.1	14.4	67.0	84.5				**.								***
146					44.3	8	44.5			**.*									. !!
200	07.7				44.2		10.7			103.3									***
776		5.40	****	44.4						165.5									11
113		24.4		89.4		92.4	**.			107.9									:;
-	19.3	94.3		92.4	94.2	95.7		107.0											10
**	94.4	.1.0			-		90.5	14.9	100.3	114.1	114.3								11
34		91.9	94.7	W.0		90.7	161.1	105.0	110.9	115.3	114.6								
-								104.2											
-	**.6	**.1						167.1											
.75		**.*	W.5	97.9				107.2											13
	**.7	*>.*		47.4				107.0											
	**	**.7						166.7											12
.10	***	**.	99.1	94.7				104.5											10
	*3.1	***						103.0											
	***	*3.4						100.9											14
	92.4	93.2						104.1											
-	91.2	93.0						103.0											1.
0.0	90.5	92.2		95.4				142.7											
2.5						100.0	101.1	161.7	106.7	**.	95.0								12
	89.3	90.7	92.5					100.0			93.0								14
								**.4											
		88.6	90.1	W.4	95.0	**.*	97.4				86.3								44
1.5		84-7		*1.1	93.3	93.0	W.1				84.5								111
•••		0>.6	07.7	M.7	*3.5	•3.4	***				****								111
•.•	80.7			67.1	*1.0	.1.0	*1.1												84.
b.0	17.0			***	00.Z		87.0				45.4								10
···	70.0	74.4		*1.0	***	64.7	0.6				***								
		***		0.0				•.•	•.•	•.•									

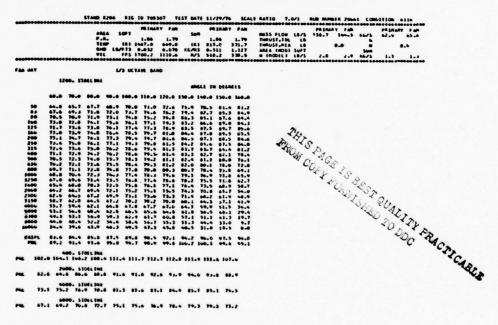
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							1 MAR A			**!	-	-						**	
				-					200			200	MASS FLOW		0.0		46/5		
								1.77		1.0		. 10	THEUST. ILL	10			•		
				1100		1 1461		****				1.7	THRUST, MEA			6.0			••
				RHU		. 0.0		0.070	M6/M3			141	ARLA IMUDI				344		
				ALL		1114		110.4		110.		•.,	A (MODEL)		2.0		***	•••	4.
								*****				•••••		*****		******			••••
					1/1				41.	-		14.0	FI RADIUS					- (814	
-						~·			•••			••••							•
	Pat .						MIC	A UPHILAD	-										
-		70		90	100	110	170	130	140		100								44 -4
***	0.0	•.•			6.0	0.6	0.6		0.6										
	0.0			0.0	4.6	6.0	6.4		6.0										
-	0.6			0.0	0.0	0.0	0.0		0.0		0.0								
100	79.0		4.0		45.1	45.7	87.2				100.0								
153				85.3	67.5		84.1				101.2								111
144			85.7	**.*	05.2		•7.2				103.0								**
700	•3.•			67.5		****	*1		99.2										1 .
270			07.3	**.1	**.*	**.*		76.3											13
*11	90.1	91.1	***	93.3	93.1	*7.4		103.0											::
=	*1.0	***	*1.1	w.4	13.4	17.6		103.9											**
	93.7	94.7	93.4					107.0											
	****	93.3	9.3					107-3											- 15
=	93.0	10.0	76.4					100.3											- 13
3	***	**.	97.7					100.4											1.
	94.3	97.2	97.0					100.3											14
-	95.4	94.4	97.0					100.1											12
	94.4	***	94.3					107.0											14
-12	94.1	95-1	94.3					107.2											12
-		95.1	94.1	97.9	100.0	102.3	105.6	107.1	107.0	107.7	107.6								120
	43.4	94.5	93.7	97.5	106.7	102.2	144.5	104.2	105.9	104.2	104.4								14
- 36	*3.5	94.2		97.5	100.7	162.1	104.1	105.4	109.1	165.1	102.5								824
-00	45.2		95.2	97.1	100.6	101.7	103.6	104.9	104.4	104.0	100.1	0							14
			***					104.1											13:
2.5	41.5	92.4	.7					103.6											17
	₩.2	.1.5	*3.5					102.0											10
••			92.1			**.													**
	87.4		*1.4	***	97.1		90.1			**.7									111
1.5	85.7	87.6	₩.3	92.4	***	**.*	97.			**.									- 11
••	84.2	1-10			****	**.7	95.1												***
4.6		04.0			*2.0		97.			**.*									110
3.0	79.0	77.7	43.0	62.1			**.			88.5									10
	73.5	***	*	4.1	4.1	4.1	**.												
-			0.0	9.0				•••	•	•.•	•.•								
																		-	14

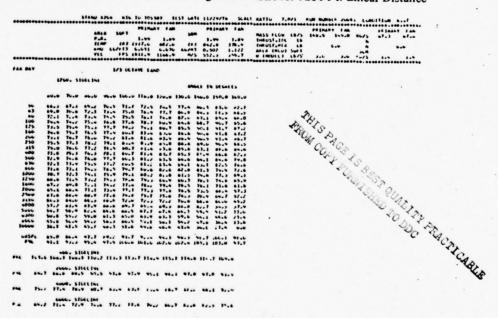
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							-					-							** ! ***	
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						-														
-					•••			-		-12		13.00	1 PADIUS			***		-	- Inch	
								-												
-	-	74			100	110	176		100	130	100									
	-	-	-																	44 -1
010		0.0		0.0		0.0														
400		6.0	6.0	0.6						6.0	6.6									
-		6.0	6.0	0.0	6.0					6.0										:
100										**.7										
43	64.6		84.4							**.6										***
	86.7		67.4							101.7	104.9									**
100					90.4			1 95.0	100.0	100.1	100.4									1.
***	84.5			w.7					107.0	107.4	110.7									1.
115			90.4					0.1'4 0												1.0
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•		4.1		w			fot.	1 101.	117.1	116.6	116.6									
36					**.1	100.0	100.	100.0	:	110.3										
•	90.4				100.7	101.0	101.	164.1	111.5	119	110.4									
*		.1.0	₩.1	**.*	161.7	103.5	105.	110.5	114.	110.7	110.7									
.25			**.4	100.7	107.5	a 1.7	100.	1 16.0	1.0.6		110.2									
=	***	**.		100.7	103-1	100.1	166.	1 110.7	114.7	110.0	110									
=	90.2			100.3	102.7	103.0	100.	110.2	113.5	111.2	117.1									8 41
		•1.0	**.	100.0	165.0		167.	100.0	115-1	111.0	119.4									
	***	97.4	3.4		107.0	104.0	167.	1 104.4	110.7	111.0	111.3									17
=	93.0	**.	97.4				107.	104.7	110.0	115.5	110.0									100
-	***		97.4		102.0			107.0	100.0	110.5	107.4									10
	*1.0		.7.0		102.4	101.4	100.	107.0	100.0	100.0	105.									
			***	-	162-1			164.2												1.0
	92.0							165.7												141
			93.2	97.7	101.1	102.1	101.	104.4	101.0	107.0										130
		92.3	94.1		***	100.0	104-	107.4	101.0	100.0	93.5									041
		.1.0			**.0			100.0		90.7	93.4									120
	07.2		92.3			90.5			***		*1.0									111
	09.7	4.00		92.0	***					**.*										- 11
				10.3			93.			*1.1	89.4									115
			85.7	67.0	*1.6		95.0	1.1		*1.4	64.4									144
-	77.00		4.5				00.0			49.4										110
٠.							0.0	0.0	0.6											
																			-	440

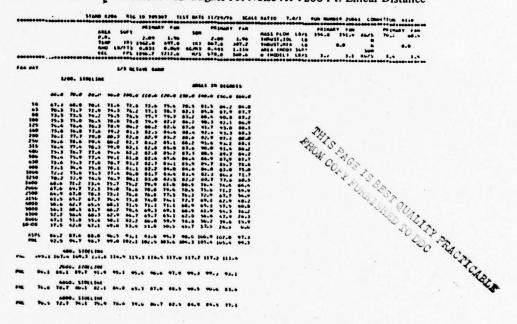
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

2.85.65		10 0.0 0.0	•a	7.5 1:40 0:11		el isc		411.4 4141	4/3	17.	1.110	mais fille leatistelli leatistelli anta (m.c a (m.c/c))	1 10		••••••		1	
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2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	***	10 0.0 0.0	•a	911 911 911 91	1/3	c(law		# 54 E	4/3	17.	1.110	# 14.1	1 5.01	••••	::	::::.	!:`	::
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2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	***	10 0.0 0.0	•a	**	"	4.6	410	# W1					••••••	440 (
2.85.5	•••			4.0	***	4.6	410					1 44: 145		440				
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7 97587	•••			4.0	-													
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-				11.0		***	**	**.1	144.1	lun.L	411.							
			.*		**			1.4.1	Ive. 7	111.4	114.1							- 11
	41.50				47.6			11 7	111.0	111.1	110.3							
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*	**.L	*6.*	40.0	w	14.1	# 1.0	114.4	110.0	115.7	114.7	111.1							i
			97.5	14	141.7	16.7.8	14.1	110.0		171.0	419.							
~ "	***	70.7	**.	100.7	100.9	160.0	100.2	110.1	110.0	1.1.1	114.0							
<i>(</i>)	***	**	14.4	1-1-6	100.0	103.0	1.0.0	117.5	110.4	1.1.0	114.5							
	**.	tro.e	101.1	101.0	14.4	14.0	100.0	417.3	111.3	171.3	114.4							
•	****	**.*	106.3	101.7	100.1	105.7	100	11	114.1	1,0.0	110.7							
* .		**	**.1	141.3	100.1	141.1	100.4	111.0	114.7	110.5	117.1							
10 4	••••	**. 7	**. 7	101.7	14.1	144.0	100.0	111.7	117.5	111.1	115.4							
		**. 7	**.	101.1	104.7	145.7		111.0	114.0	11:	114.0							
		47.6	**.	100.0	100.0	144.6	100.7	1.4.1	111.5	111.0	110.1							
. :		****	****	14.4	100.1	105.5	100.1	100.5	110.5	111.0	100. 3							
				100.1		****	101.1	100.4	100.0		104.5							
	-:-					.45.6	107.1	101.7	100.5	100./	104.7							
		****	**			1.1	100.0	100.0	100.4		107.6							••
					101.4	14 1		100.5			100.4							
					100.4			100.0			****							
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			84.6		****			47.7										**
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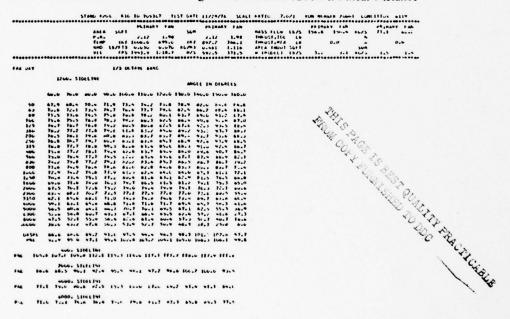
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							d www .			b# 1				P4 1 mm			100-1	
									***				10/2	0.0		41/3	U.0	
							-97	1.4	1000	7.1		UST . 104						
				***				4.00	141			UST LA						
				-		. 0.		0.070				. 14001				>		
				411		5 100		710.7	~		171.5	MODIL!		1.7	7.1		4.5	
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~									-			 						
							MIC	AUPHEN	-		DECREES							
11	•	Tu	80		100	110	176	1 14	144	496	100							
170			•.•	0.0	•.•				0.0	•.0	•.•							
•	•.•	•.•									4.6							
40		0.6	0.0		0.0	6.0			6.0	6.6	0.6							
100	62.4	44.6	87.4		w.,				****	**.2								:::
42	4.7	· 7		****	***	*1.3				107.3								**
-	4.4	47.5			97.1	97.1			107.7									::
~			·		*	****			104.5									16
115		90.2	91.0		-	**.												17
			**.*					100.1										
-	**.*	***	**.					107.1										- 11
			90.4					116.7										
-	97.4							111.0										- 63
40	**.*							111.1										
. 25								111.										
-								111.0										
-	100.0	100.1	100.8	102.1	100.0	105.0	100.	117.7	117.1	121.0	114.2							
. 20	99.1	99.7	100.2	101.0	104.7	100.2	100.	1 114.5	415.8	119.3	147.4							
-89		**.2	100.0	101.4	144.5	100 .2	100.	111.0	114.5	110.1	115.0							
		90.0	**.7	101.5	104.7	100.4	109.		111.0	110.4	113.4							
.00	97.6							110.0										
. 30	47.4	**.*		101.1	100.0	100.2	108.	1 110.1	111.5	117.4	108.7							4.
		.7.7						104.4										12
								100.9										44
	94.5	***						1 107.8										42
	*3.4	*5.1	•7.0					107.1										
•.•		•3.•	99.1					105.2										14
	96.0	*3.3	95.1					104.5										12
	**.*		w					102.2										12
	07.5	67.7	*2.4					100.6										11
3.0	42.4		77.4		****						91.7							**
::	79.1				90.7													- 11
	***	•.•	7.3			•				***	0.0							•••
	3.0			•••	3.0	•••	•	• • • •		•.•								
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501			112.0	4.4.7			130-	. 123.4										

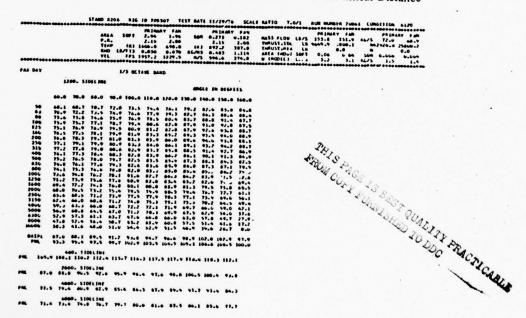
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

				461							#48 Y				F-140.7			
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						: 144		174.5	4/>		. 11			40/1	1.1			
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440					•••			-		w121	LATA	13.0	CLIGA4 14					
-	*	14		**	166	.16	-11	-				1						
					•••	***	·M	130	140	130	100							
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					*													
-				4.0		6.6												:
106					10.5	****												
143	14.1	44.1	***		***	*1.5					103.7							
101		****									164.6							
-	.7.1		**.*					****	****		100.1							44.
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.0	43.3	**		101.1	400.0	402.4		144.4	11.w A									1. 7
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	-7.1	**	**	**.>	104.4	141.4		144-1	144.1									1.0.
																		1/2.
::		41.0		**.*	101.1	141.1	142-4	100.0	142.1									
	47.4	*6.3	*,	**.,	**.5	**.	14.4	161.0	160.4	104.4								
			W		97.0													121.
::				***	**.*	94.3												
:-	7.3				91.7	*1.0		*3.6	43.7	**.7	11.4							114.
••		0.0		0.0		0.0		6.3	•.•		0.0							111.
																		0.4

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



Flow Simulation; Configuration 6A; Condition 6301

a.) Model Data Measured At 15 Ft. Radius

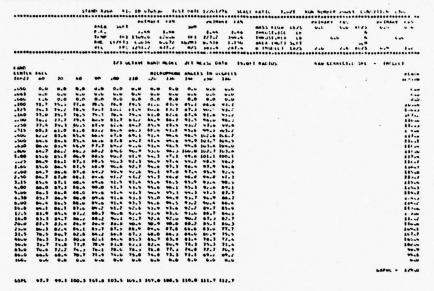
								***		1			••••					
				444		• •			34					0.0	0.4		** 1 ***	
							. 11	1. 10		1.1	1 1.10	1 120 41	 	***		46/5	u."	
				11.00	• •	41 170	1.0	c 0	10.1	lur.								
				-		13	434	41.4.0	46/43	0.34	4 1.141					24.0		0."
		Acres 1		44.		es ne		104.4	4/5	3/1.	1 211.0	. IniDel		1.7				
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							200000	ren ones										•••••
					.,,	WIAN		MITTE	311	M 131	LATA AS	culus 14.						
							0.00											
	**	Pt.		**	144	114	410	W. CAHCO										***
		•		-	100	•••	140	130	100	1>4	100							41 -1
me			0.0		0.0													••
130			4.0				6.0		4.0									
-	0.0	0.0							6.6									
100	44.7	71.7			15.5		77.0											
145	70.9	75.0	79.5		76.1		77.0		63.7	****	****							44.4
100	12.0	12.2		72.4		76.4	77.0		*1.0		***							100
700	7:.2	74.1			76.1	74.0			-1.0	47.5	*1.0							100
: 30	74.6		77.4		24. 4			03.3	• • • •	****								414
.40	70.7	27.5	70.1			42.7		.7.7	****	*1.7	****							8
-	70.4	79.7	81.5	04.7	41.7	45.3			71	• 7 3	**							4 4
***	w.1	44.7	41.2					wo.2	****	****	100.							4
• 30								91.7	****	****	40.0							11.
800			02.4		45.1			*1.5		****	****							
			82.9		41.0	41.7		+2.2		****	••••							
.25	.00				84.2		-				***							11
		84.6						*7.1			89.0							4
	80.9	62.0	43.4	85.1	44.4		-	97.6	**.3									411.
. >6				43.3	60.9	40.7	-		*1.4		****							111.
.13			84.3		87.4				M		03.3							Hiv.
.00			84.5			84.7		41.4			84.0							410.
	65.5			6>.>	47.1			40.1										
. ,0		.2.6			17.6					45.5								110.
	80.3			45.1			44.1	89.4	67.3		.0.4							14.
	14.4			m.,		.7.		80.1			74.3							100.
	10.7	80.0	4.54	04.6	6	67. 5		87.2			77.7							100.
				63.5		84.5	47.6	84.6			75.5							tur.
				*1.0		80.4	65.2	84.1	.1.0		72.0							1
	1	77.5	14.5		67.3		7			74.2								ter.
			77.0	14.5	H		8/.1	86.2	77.0	73.4	48.5							403.
			7>.7		70.1	70.0	P.7	17.0	73.2	71.0	44.1							tor.
			12.0	7		70.7			12.2	49.1								***
		47.3	**.1	10.0	72.7	77.0	73.0	71.5	49.0	.1.0	43.3							•
			.			44.3	64.6	47.7										*
	0.0				6.6	6.6	5.6		0.0	0.0	0.0							*:
																		•.
																	-	1.4.4

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

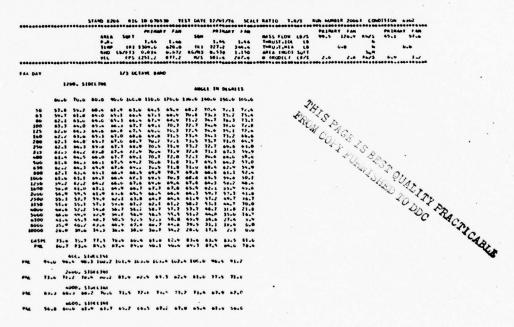
			27000							-		************	•••••						
					144			,	-					62.0	110.1				144
				*:		1.31	1.30			. 31	1. 10	THE UST . TOL		••••		"MY	37.0		****
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					***		150.0		As c.		1.141	MATA (MUUI	Suff			344		-	
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and a section						100000000						•••••		******	******	*****	*****	****	*****
				4/	3 44 1/		10												
			w 151																
	••																		
								•	mett 1		115								
		70.4	80.0	₩	144.6	110.6	1:0.6	134.0	146.0	156.6	104.0								
.,	20.4	27.7	***	>0.>	>*		42.1	44.1	45.4	67.4	47.4								
		***		****	***		****	**.	****	44.4	44.4								
100		10.4			***	***	44.7		*** ,	70.4	41.0								
123	30.0	44.1		42.7	***	****		****											
100	44.7	34.9	41.1	41.1	44.	****	66.7		•4.7	**.0	•1.4			2.					
200	20.1	60.2		41.1		***	47.3		• 7 • 8	****	****			YA.					
296	14.1	60.4	41.9	47.7	44.1	44.1	.7.1	40 0		• > • •			2	11	^				
313	71.1	••••			64.L	44.6	44.4	47. 4					4	0.	S' .				
4110	37.6						47-4	41.	44.4	A1. 7			•	Via.	10				
500														14	. 4	3			
*30	****			41.4	44.4	66.2	44-7							. (500	24			
														,	U.L.	1 1			
tuce	97.6	30.0	4.04		.3.7	44-4	44-0								4	4	,		
1230															*	Lin	A		
3000																60	H	~	
1346										44.5	11.0					47	. 3	2	
3.20		.,	4.8		****	>4.5	>4.0	35.0	30.0	44.1	:					- 10	1.	4	2
***	44.7				****	36.4	23.2	>1.5	44.7								X. C.		27.
2000	41.4			71.0	****		****		****	32.0							4	in	24.
4346	17.4	40.4			47.	30.6	****	****	36.4	70.0								0%	4.
4000											*							-	2 12
Horo	21.5	15.7	17.1	11.0	34.1	11.7	20.4	22.	1	14.5	4.0								40 4F
											4.6								RUALLAT PRACE
		71.7	11.2	74.0	74.1	77.7	74.1	70.0	70.0	77.0	1.								- A
-																			

Flow Simulation; Configuration 6A; Condition 6302

a.) Model Data Measured At 15 Ft. Radius



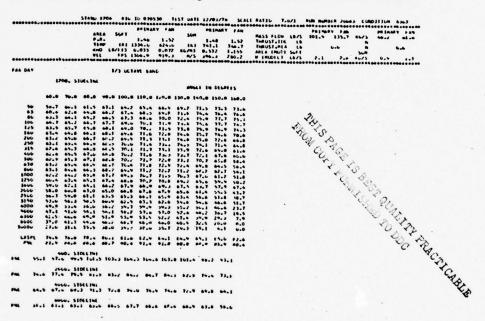
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							1444									-			
				4-14					54.00	De 15		-		10/1			44/5		
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-										-		3.01 RA	0105		-40 60		6 -01	. Inch	
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.003	0.0			0.0	U. 0														٠.
-	4.6	0.0	0.0		6.0			6.6											٠.
100	73.0	10.2	70.4	P.)	14.4	.0.4													
143	74.4	Bu. 1	19.0	N.0	61.3					*1.1									
100	11.0	40.0	17.1		Tv.5		.1.1			42.7									107
con	11.4					1.00			94.3	47.4	44.7								
. 50	70.1	eu.?				45.2				94.7									1.5
:15		41.4		***					44.7										
44				11.5					**.4										111
300	*>.*	**.*		.7.0					141.1	165.1	105. 0								
**	17.4					45.0			101.0										110
***	***		47.>		w. 5				101.3										115
	86.2				*1.2				101.7										410.
.25		47.1				*1.4	****	***.	100.5	100.5	94.3								
	00.0		**.*	~	41.7				**.6										
. 10	***	****		٠.,	.5.1														447.
.17				90.7	14.3			**.7											
	.7.0			***	92.5	**.5		***		**.6									
		44.6		w	****					**.2									
. 30	44.0	44.4	**.1	*	47.4	**.*	**.*			*1.1									
	65.0		89.1	91.0	*2.0	94.5													11>.
0.3		87.2		10.5	12.4	94.0	**.0		****	*2.1									445.
7.5	41.0	44.7		w.i	12.1	*1.7	W				67.7								.4>-
	44.5					****	F.:			10.7	84.7								11.
	82.7			80.2	90.1		97.3												
		01.4		07.2	84.0	90.1	*1.0			14.4	19.4								
1.5	79.4			83.7	80.1		64.				17.3								110.
	77.4			41.5	83.0	84.5	67.1			04.2	74.4								
0.0	74.0	76.4	77.0	w.2	03.1	83.6	.1.			74.9	73.2								100.
J.0 .	71.4	13.4		11.3	19.3	Po. 6	84.3				72.6								16.
	47.1	69.6	72.0	73.2	75.2		74.2				74.1								100.
w.	6.0	0.6	0.0	0.0	0.0	0.0					•.•								•••
							- 7		100										
																		-	

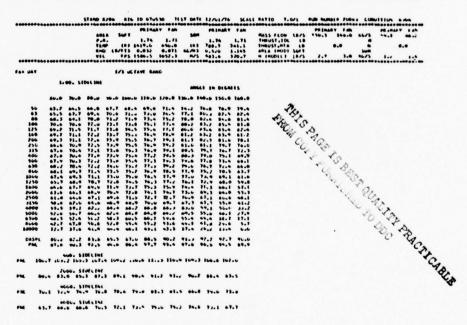
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

						-						44			** 1 ***				
									100					16/5		0.0	A./:		
						1.		4.74		1.70			1mt ust . 1:						
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				411				L>4.1	4/5				. INCOLL			1.1	41/3	1./	1
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					1/1	CIAM	-	-		C126 (12.01	PALTIES		•••	Lester		- 14	
	****						-												
-	**	74		**				12	1	194									41-1
			••	-		•••		• • •	•										
uno	4.0	0.0								0.0									
41				0.0						4.6	6.6								
-		0.0	4.6								6.6								
-	11.2			11.1		84.2				93.6	**.*								
125	74.0			4.3		67.7													•
				**.*						.7.7									
200			84.0	84.6					47.0										
***									**.7										• • •
				0.00		.1.0			102.6										1.
•••		••••	*	*1.*					107.0										**
***		••••	*1.*	**.*					160.0										::
**	*1.*	****	*1.7	#:;					100.0										
		*1.7	*7.*	**					100.7										•
	*/.1		**	**.*					107.0										1.
		*1.4	94.7	95.5					100.0										1.0
.06			94.1	W		**.1	104.3	104.9	105.0	105.1	104.0								
. **					47.5	**.7	162.0	104.0	165.7	101.4	lui. 6								
.15	*1.5				11.7														١.
.00			4.7	**.4					103.6										44.
.00	*7.1	*1.1	w	** . 2					162.7										
. 30	.2.1		**.*						105-1										44.
	w. •	43.5							161.6										1:
	10.1		***	**.2					101.4										- 11
4.2	~	*3.3	****	5.7					166.1		99-1								**
	00.0	*1.4	**.	**.7		**.			**.		91.5								• • • •
>.6		64.3	*1.5	*3.4		.7.6	**.		\$7.0										111
	**			97.6		**.7													
	43.2					*3.7	****												
			45.2				11.4												***
1.0	17.3	19.2				7	87.4		05.2		79.0								10
	73.0	75.4	70.4	79.5							76.0								10
w.			0.0	0.0					•	4.0	6.6								•
																		-	

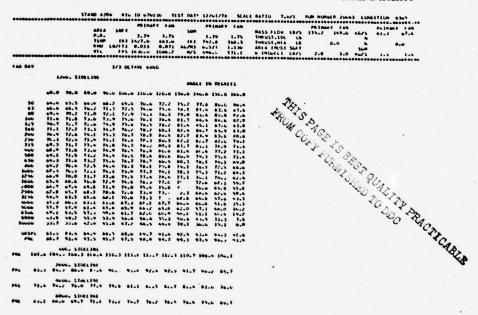
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

								* **				• • • • • • • • • • • • • • • • • • • •				-	** 184*	
				441					34.4				1000				W-**	
								1.05		1.7	. 1.15	THRUST-114	4.0				•••	
				110				***		Mr.		1m US1 . M (4			1.0	- 7		
				900				U.4.71		6.57				4.05				
				4.1			w.u	1.00.2	*/5		1.14. 4						4	٠.
••••	•••••		• • • • • •	*****	*****	•••••	*****		•••••	•••••		***********		******	******	*****		:
-					1/3	CLIAV	I DAM	-	311	PCIN I	DATA 15.	GOT RADIUS			CLAPICE.	4	- 14	
																		-
-	-	10		*				-			Ltuntts							
	-		••	-	100	110	150	1 11	146	100	100							et -
. 46				0.0	6.6				565.00									
(4)	0.0	0.0			4.0													
**	9.6																	
-	70.0				64.7					6.6								
					****						100.0						2	
			**.			**:												
~			*>.						•1.6	**.	107.							
*			84.7					*11.7	**.*	103.4	100.0							
ñ						₩	***	****	166.5	165.2	107-5							
ž	44.1		vi.		***	***	•	****	163.6	167.4	111.5							
-			•/		****	****	****	101.7	106.6	111.0	114.4							
-			4.5		****	****	****	162.6	100.0	111.5	117.4							
2	*1.*					**	101.1	165.1	100.0	110.7	110-1							
=	*1.*		-		***	****	100.4	105.4	110.5	114.0	115-0							
7	9,,7		m.;	-	****	100.5	167.5	104.1	116.7	113.0	114-3							
	*1.2		**		****	100.7	103.3	100.0	104.4	111.7	147.3							
=	*/.:				****	100.5	101.2	100.3	108.4	110.1	110.							
~	97.7		93.1		****	100.0	101.7	106.1	101.2	107.5	108-0							
			.		**	100.1	101.1	105.9	100.0	100.0	100.0							
	***		45.7		****	101.4	104.	105.7	105.7	104.0	104.6							
	•1.			****		101.7		105.4	105.0	103.4	103.4							
5	***		***		****	101.5	104.6	104.6	104.6	102.0	w1.v							
=	·2.1		**.			101.3	103.6	104.4	103.5	107.5	100.0							4.
	*1.4			*7.7				103.6		107.3	100-4							
			93.1	****		101.4	107.	103.0	102.7	101.4	.06.6							
	*1.6	.2.7				104	102.0	103.0		101.2	***							
:	44.2		*3.2				100.1	166.6	.01.4		****							
			92.4					**.		41.6	****							
::	44.2		vi.3		***	v7.0	***		**.3		49.6							
			49.2			*>			****	*1.*								••
						97.7		77.0		**:	43.4							
	70.3	00.2																
::			79.3			84.7				****	74.0							
٥.	0.0		0.6			6.6			6.0		0.0							1.
-					3.0	3.0	•.•	•.•	0.0	0.0	0.0							
																	marte .	
		-	167.9	100.>	1.1.9	413 4	114.7											

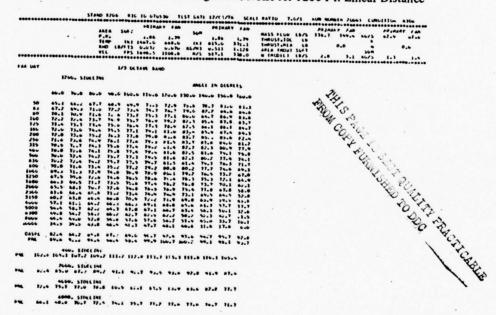
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

										P0 1	-	-		PR 194			*******	*****
				40.		41	ra (co.		348					 0.0			I'm IMAL	
				111				1.7.		1.0			Imbust.ILL		0.0	2003		
						146				815.		1.1				•		
				a me	10/	11	417	w to				20			"."			
				V: L		PS 164	•••	1.01	4/5	*17.	1 130			7.0		**		
				•••••	*****	******	*****	*****	*****	*****		******	*********	 		16/5		1.
							9 15 150							 			•••••	
044						DETAN	-	MCIL	341	40174	LATA	15.011	RALIUS				- 140	
																	- 140	
444		14		***			-10	acomica			LICALLS							
-			••	70	***	110	100	1 34	1	140	100							
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100	10.4																	
125	41.2						47.4				100.1							•
						60.7			**.1		1-1-1							**
-							A1.		**.	**.5	103.1							!!
50						84.7	****	****	**.1	100.0	100.4							11
		88.4	64.			***	*5.5	**. 1	101.0	100.0	104.4							- 10
14	49.7	91.1	97.			*3.1	****	**.*	161.0	100.0	115.1							1:
-			***		-	**.	10.0	107.6	107.4	117.0	113.4							12.
-			**.		:	:::	****	103.0	104.4	114.)	114.4							1.1
-	94.1	93.2				**.7	,	100.1	110.0	1115.4	114.4							
						101.		100. 1	111.5	115.7	113.4							1.5
.25			***		-	161.1		107. 1		113.5	115.4							1.5
	90.5		****		**. 1	101.5			110.6	.,	113.5							
.00	43.7	95.1	94.1	97.4														
	*1.1	**.1	90.7	*7.7	**.	101.0	104.6	107.1	170.0	104.3	116.4							1
45		95.2	90.3	91.0	19.4	162.3				167.7	100.7							170
*		***																44.
•		*5.0																10
		*>																100
•																		100
				****	100.3	104.5	100-1	144.3	141.0		10							144
																		100
				41.,	**. 1	161.9	161.1	101-2	167.4	102-1								100
				****	78.7	100.5	101.4	M-1-A	101-0									17.
				**.	97.6	**.1	140.4				91.4							
:		49.5		73.7	.,	97.9	99.2	90.>	97. 5									130
	.7	47.7		*1.0		**.*	*7.1	**	95-1	*1.4								
•	64.2	***	47.2		45.0			*3. 3	97.2	90.4								410.
.0	79.1																	
••								80.1		84.2								110
••	•.•			0.0	6.6	6.6	4.0	0.0		6.0								Lue.

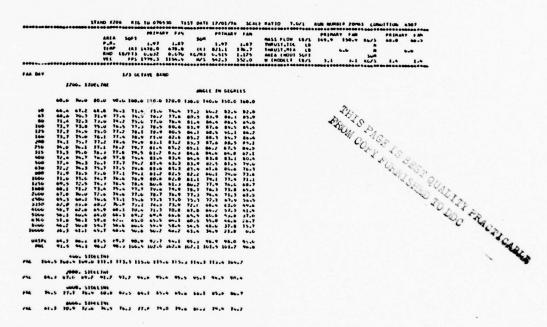
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							IPART	***		** 10					***			
						• •			-			MASS FLUE	10/5	0.0		44.05	v	
				F.A.				1.47			1 1.67	THE UST . 10L						
								4 10.6	141		176.7	THE :- 57 . 41 4						
				-		0.		D.OM	/43			ARLA IMILI	1601			36.00		
				***	• • •	P) 177	v., 1	14.4	4/5	447.1	157.0	. (*:	40/5	1.4	2.4	/:		
••••	*****	*****	•••••	****	•••••	*****	*****	*****	*****	******	********	**********	*****	*******	*****	*****	********	
								-										
**					1/1	OCTAN		WOOF F	311	# 17F 0	MIA IS.	CULDAS 110				U .**	- Incles	
404											164113							**
	••	10	+6	**	100	110	150	130	1-0	1>0	100							
	4.4																	
			4.6		6.0	6.0			6.6	6.6	6.6							
-						-:-	6.6				0.0							
100		**.1	80.6		07.7						101.5							- 3
143			****		****					**:;								••
100	44.2			.7.6	***		67.5			101.1								,,
	44.4	85.7				*1.*		****										
256		.1.		w				*7.4										• • •
*10	**.*	49.6		41.4				100.0										
-	91.1							104.1										
•		**.1						105.0										••
.30								107.7										••
000	***	***			***	Lad. A	14.3.3	107.0		****								:
	***	.7.1		40.4	144.5	161.1	144-4	104.1	111.0		117.							
				**.*	101.5		144.4	100.6	1.1.1	1.4.	114.4							
	**.*	*4.1		w	161.4	101.4	104.1	100.7	111.9	115.4	4.4.							
.06		+7.4	40.3	****	161.5	161.4	164.5	109.0	110.7	111.4	1.4.4							1.
. 20	93.7	97.2						140.4										
-13	**.*	97.1			161.5	144.2	144.9	100.7	164.1	114.7	112.5							
.00		.7.3	98.7	**.*	102.1	144.6	107.1	100.5	100.5	110.3	110.5							1
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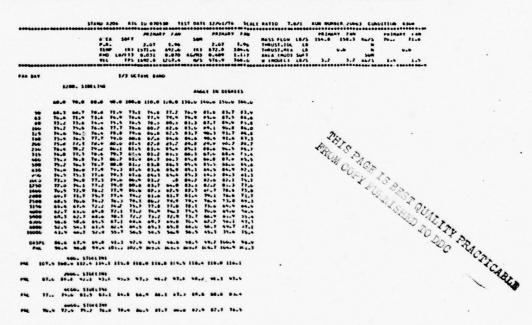
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

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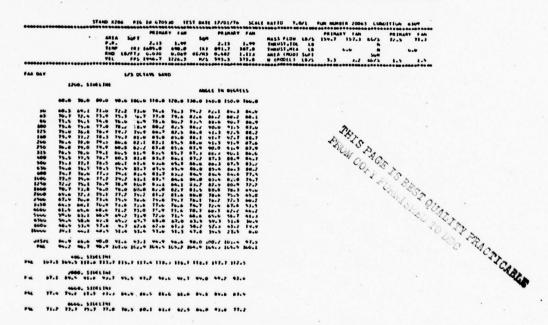
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

### 147/13 GOOD	#### \$40 50 50 50 50 50 50 50	 :!:
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91.2 95.2 95.2 97.4 10.2 107.3 100.4 109.3 100.4 102.3 100.4 102.3 10.4 102.3 10.4 102.3 10.4 102.3 10.4 102.3 10.4 102.3 10.4 102.3 10.4 102.3 10.4 102.3 10.4 102.3 10.4 102.3 10.4 102.3 10.4 102.3 10.4 102.3 10.4 102.3 102		
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그리고 그렇게 가장 그리고 그리고 살아 있다면 이 작업이 없었다. 하게 그렇게 그렇게 되었다고 있다면 하는데 이번 그리고 있다.		
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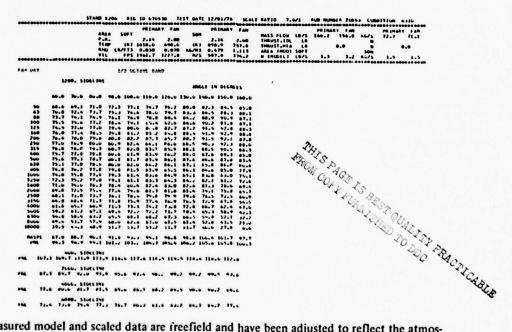
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

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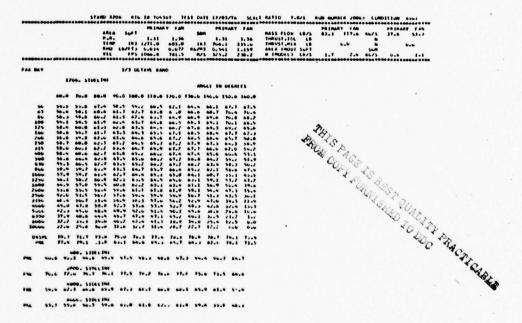
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							IMARY			P#	-				Palmer				
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-		71.0	75.0	D.4	P	75.0	70.0	80.8	04.0	85.4									
45	71.3	11.0	74.4	D.4	76.6	77.0	17.6		04.3	67.1									
	11.5	79.7	N	n.1	19.4	75.9	17.0	79.0			91. 1								
-	77.0	73.0	N.0	77.0	78.1	79.2	41.3	83.2	87.6		**								
10	74.1	74.1	77.4	10.4	19.3	84.4	47.5	83.3		*1.7	**.2								
13	17.6	11.0	70.0	14.0		45.3	84.6		*1.3	*>.1	98.3								
-30	79.1	70.0	82.0	83.0	84.1		87.5	+0.2	*3.*	.7.8									
-						85.5	41.4	10.5		***	**.6								••
36		•1.4	87.4	43.0	85.2		89.2		*4.5	•7.1									
-	41.10			M.2	03.0			*1. *	94.7	**.	**.*								
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	86.4	41.4	83.7		4.4	47.7		47.0	45.7	47.7	70.1								16
	80.7	40.1	47.5	43.0	85.7		4.1		84.3		70.1								
	70.1	77.4		42.2		43.3				70.7	73.2								10
	74.7	70.0	80.1	41.3	02.9	43.4	84.4	07.4	00.5	74.4	71.7								10
	14.0	73.0	m.;	70.5		42.2		80.7	70.0	74.5	49.4								10
	72.0	73.0	70.3	77.3	79.1	80.1	00.1	79.1	75.3	72.1	47.7								10
	70.0	20.0	73.4	M.2	70.0	74.9	74.7	79.4	77.0	49.4									
	44.4	47.4	PO.0	71.1	72.7	73.0	73.3	72.0	49.2	47.3	47.3								
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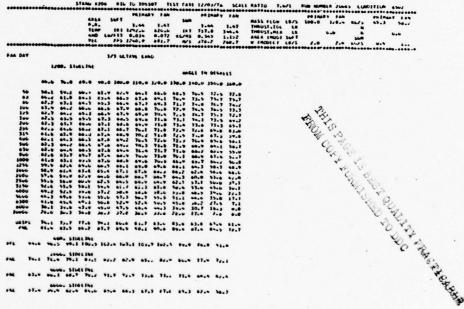
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

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100	72.0				74.0	34.4													
140	14.6	10.3			10.1					*1.2	¥1.0							•	1-
100	74.7	19.4			78.4	19.5													
.00	14.0	****								••.,									
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	43.3		47.1			*1.*	***				100.7								
	45.1		47.4	w	10.7				**.7										**
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- 94	11.4					*1.0	**.		***	****									***
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			87.4		90.8					67.6									
		43.0			84.3	+6.6	*1.1			**.*	86.4								
											74.2								
1.5	70.0		03.5		67.2	87.5					74.1								
•.0	70.7		81.2		84.8					76.4									1-1
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9.0	76.6	****	74.0	D.0	77.0	74.1	70.			77.6	71.7								••
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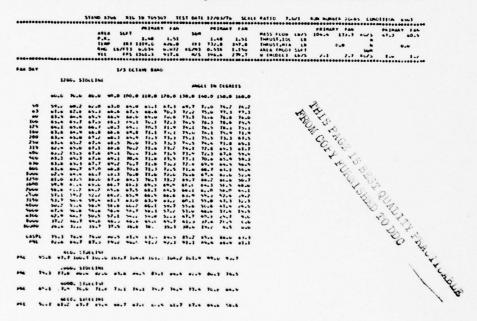
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

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-0	0.0									1.4									
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>0	70.0		.1.6						**.*										
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15				91.4	4		47.1		97.2	99.1	*1.1								
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					*1.0	**.1	**.7	47.1	**.2	* 1.5	1								
			90.0	91.5	41.3	**.0	**.7	44.7		*1.1									
	84.4				41.4			**.*	**.1										
				*1.0	42.6	**.*	₩.0		**.*										
						****		** *		*1.*									
						•	*				65.7								
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	•1.*		****			w.,		**. 7											
.,	19.5			٠.٠	**.	••••			****	*1.7	70.3								
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		17.0		77.6	70.0	77.6	***	19.7	74.7		13.0								- 1
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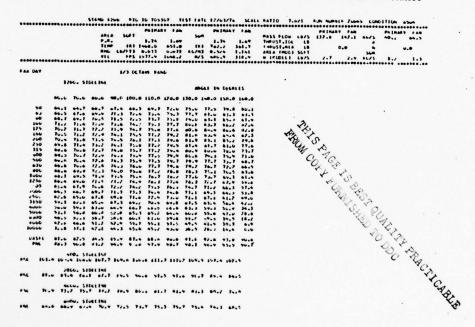
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

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b.) Model Data Scaled to Predict JT3D Engine Jet Noise At 1200 Ft. Linear Distance



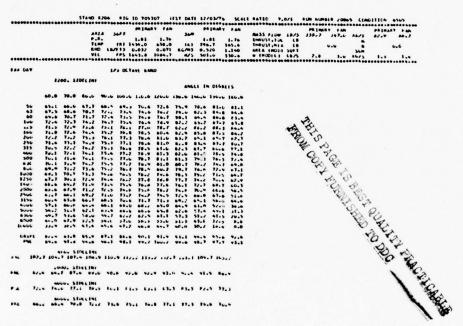
Flow Simulation; Configuration 5A; Condition 6505

a.) Model Data Measured At 15 Ft. Radius

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b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

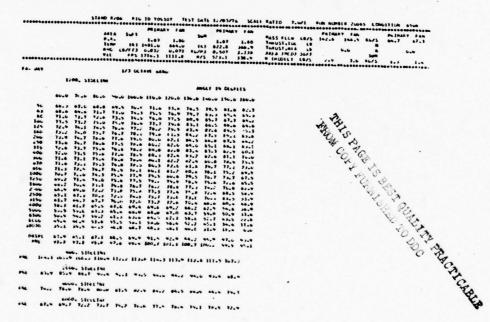


Flow Simulation; Configuration 5A; Condition 6506

a.) Model Data Measured At 15 Ft. Radius

93.5 93.5 93.1 93.1	0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 10.0 10.0 10.0 10.0 10.0 10.0 1	100 100 100 100 100 100 100 100 100 100	## 14- ## 14-	# PAME #	#LDE L	######################################	1.1 0.7 0.2 0.2 0.2 0.2 15 10 15 10 0.0 0.0 10.0 100.0 100.0 100.0 100.0 100.0 100.0 100.0 100.0	Gata	13.00	MASS PILE Jeffinsspill Hemissign Andrew Hemisesign Hemi	1 201					1.00 1.00 1.00 1.00 1.00 1.00 1.00 1.00
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87.4 87.4 87.4 87.4 87.4 97.5 97.1 97.1	#5.0 #5.7 #5.7 #6.7 *6.0 *6.5 *6.5	****	40.5 40.5 40.5 70.7 70.7 70.7 70.7	***	##. 1 # 1 1	90.0 90.0 90.0 90.0 100.1	02.6 03.2 04.6 101.7 100.7 100.5	08.2 100.4 100.5 100.5 100.2	101.5								11
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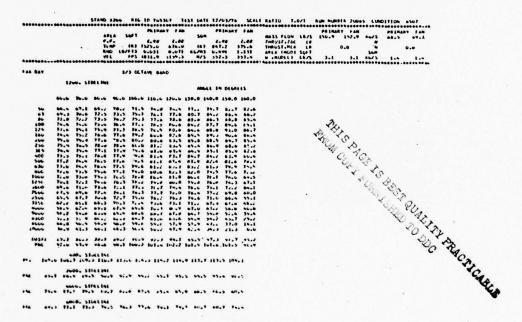
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

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10	Week Vic. Week Week Using		.7.4	97.4		-	141.4	1.1.			*****								1.1
15	Sec.	*		97.0		**.	141.9	141.4	16.7.6	104. 3	*****		113.1						10
	Sec.	15	**.	****		95.7	171-0	103.0	107-0	104.7	100.0	110.4	****						
10	\$\frac{\pi_1}{\pi_2}\$\$ \$\frac{\pi_1}{\pi_2}\$\$ \$\frac{\pi_1}{\pi_2}\$\$ \$\frac{\pi_2}{\pi_2}\$\$ \$\frac{\pi_1}{\pi_2}\$\$ \$\frac{\pi_2}{\pi_2}\$\$	u	96.0		**.*	**.*	10/-1	144.7	107.1	lue. 1	166.2	Buv. A	164.7						
10	\$ \ \partial \p	*	**.*	4.0	W	**.	141.9		144.	167.	107.0	104.0	107.5						1.0
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10	90.7 No.1 No.1 No.1 No.1 No.2 No.2 No.2 No.2 No.2 No.2 No.2 No.2	•	**.1	***		w.v	102.0	14.1	140.6	100.5	165.7	160.5	105-1						
**	\$\\ \frac{4}{1} \\ \frac{4}{1} \\ \frac{4}{1} \\ \frac{4}{1} \\ \frac{1}{1} \\ \frac{1} \\ \frac{1}{1} \\ \frac{1} \\ \frac{1} \\ \frac{1} \\ \frac{1}{1} \\ \frac{1} \\ \frac		94.7		**.1		141.0	163.0	105.5	105. 7	105.2	104.0	140.4						
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	91.0 91.4 95.4 95.6 00.0 101.4 102.0 103.0 102.4 102.0 00.5 92.4 102.0 00.5 92.5 92.5 92.5 92.5 92.5 92.5 92.5 92		41.7	90.7	.7.1	w	101.3	141.7	104.5	144.7	144-6	104.4	141.1						
10 Mar 9cc 93.1 9c.9 4v.1 10v.5 101.1 101.1 101.2 100.7 9c.5 1 88.7 9c.6 13.6 13.6 13.6 13.6 10.2 1v.6 9v.0 12.2 10.2 1 88.7 9c.6 13.6 13.6 13.6 13.6 10.2 1v.6 9c.6 12.2 1 10 81.7 8c.6 12.2 13.2 9c.6 8c.6 8c.9 9c.5 97.9 97.0 97.0 97.0 97.0 97.0 97.0 97.0	### \$\text{\$\\ \frac{1}{2}\$ \$\\ \frac{1}			*1.4	45.4	*7.0	100.0	101.4	102.0	103.0	104.4	102.0	**. >						
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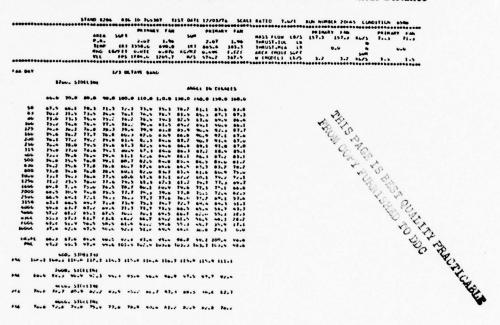
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

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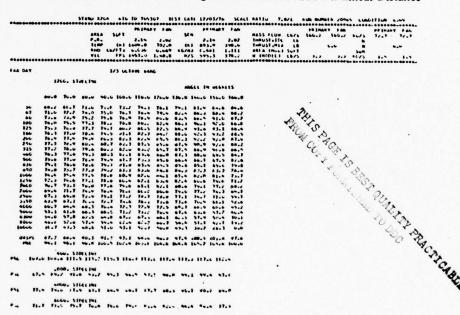
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

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740		84.1	W.4	47.0	*7.0	41.0	*5.4	**.*	104.3	109.4	14							
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-	10.0	**.7	**.2	100.5	101.7	102.4	143.7	110.1	117.6	126.7	114.6							
	****	**.1	100.0	111.4	162.7		107.6	411.0	7.0	141.1	144.0							
.19	**.7	luu. b	101.5	162.6	163.0	165.0	140.	112.2	417.7	144.7	119.0							
	166.2	101.1	102.1	102.5	100.6	103.7	100.3	114.0	110.0	1.0.7	1,0.0							
	100.4	100.0	101.4	100.1	11	143.0		111.0	115.0	110.0	114.6							1.
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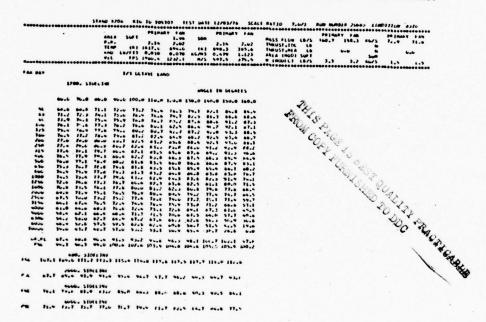
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

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v							40.4		**	10	14.7							
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					****	****	****	****	160.0	104.5	111.7							
•			***		****	****	**	104.1	107.3	*14.2	14:.7							
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				-			100.1	167.7	111.0	111.0	110.1							
-		***	•••	146.0			**	110.7	111.0	110.7	110.3							- 1
		**	140.0	1.1.1			107.0	111.6	117.4	170.0	114.7							- 1
	**.*	114-1	1-1-4	167.7		103.0		117.1	110.0	141.3	1/0							
	Na	141.7	14.7.2	14.,				117.	111.0	170.4	114.6							- 11
•	104.4	1.0.5	101.7	100.7	10- 1	101 -			****	171.0	170.0							
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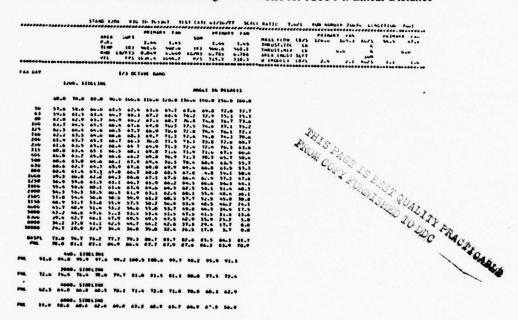
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

					0.5		-			PR 1	MAY PAR			PRIMA	* **		PR IAME	
				AREA					340			MASS FLOW		0.6	6.0	46/5		
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	-		-	-	100		170		140	130	100							De -17
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43			0.0	0.0	6.0					0.0	'0.0							•
-		0.6	0.0	0.6						0.6	4.6							
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143	14.2	70.0	70.5	77.8	70.4													100
100	76.6	79.3	77.6	78.4	70. 3	70.0				*1.7	**.*							1 ve
744	70.1	77.0	16.7	14.0		.7.7	00.0			**.	96.2							410
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115	₩.1		41.4	67.6	*3.0	****					163.5							110
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-23	99.7			M.2						**.*	90.2							447.
-00		83.7	84.0	44.4	89.9		-			***	***							110.
	84.4	85.8	87.1	4.1		92.2	***			94.1	97.5							110
- 20	44.1	45.7	87.1	40.5	90.3	92.3				***	91.3							110
-13	84.2		84.9	44.2	96.2	94.2	*1.	+5.2	**	92.7	84.2							11-
.00	4.4	85.7		84.3	94.1	97.6	93.1		*3.7	91.2	01.5							117
.00		84.0	80.2	87.7			92.1	* *1.3	92.3	89.5	85.7							11.
-30	84.0			07.5	87.3	**.*	w2.	42.5	*1.1		84.7							112.
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	77.4	74.1	.7	87.2		43.2	85.			14.1	73.4							lus.
1.5	75.7	77.0	79.5			****				17.>	11.0							
4.0	74.3	73.7	70.0		01.4		04.			14.1	71-5							Lu.
3.0	17.0	13.4	75.5	70.5	74.1	70.0	77.0			74.4	11.0							Iw.
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~			-	800 1	10:-0	101.0	100-	167.2	140.4	112.0	111.7							

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

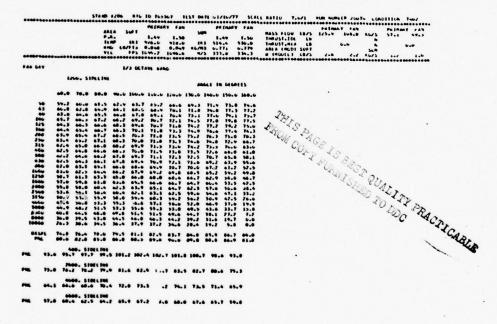


a.) Model Data Measured At 15 Ft. Radius

							11	-		PAI	MANY F	-	***********		PR 184			-	
				-					544				RASS FLOW	18/1	0.6		***	-	:
				14.0			.49	1.50		1.4		90	THEUST . SO		-		:	•••	
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						S 100	•••	1000.6	N/S	333.				10/	2.0	,	14/1	1.2	1.6
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		0.0			0.0	0.0	6.				6.0								
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lue .	17.0	15.7	10.3	N.1	77.0	00.1													
125	73.4	14.0	70.0	N.1	41.1	64.3	•1.				*3.0								
144	70.2		77.3	N.4	74.7	***	*1.			*1.2	**.*								101
-	27.4	70.2	04.2	41.4	87.4		85.												100
750	70.0		81.7		84.2	****													
345	41.10			84.4	45.1				**.7										443
				07.4	49.4	90.7	•2.		97.1										
100					89.4		***		100.0										111
-		00.2		****	96.7	97.4	**.		105-5										144
-				49.7	91.1	92.4	-		107.4										121
	87.4				*1.7				102.4	100.0	100.2								140
.29				10.4		*1.*	**:		101.0										4.0
-44					91.7		**.		100.>										110
					1.	*1.*	*		99.4										
				40.2	92.1	w	w.		***	**.*									447
.45					92.0	***	*												
				10.2		*1.*	*			*>	*1								1.4
	00.2									41.4	47.4								443
. 30	3.40				41.2		**			94.5	****								110
					10.4		**			44.4	44.7								
				01.0	96.6	*1.7	97.			67.3	02.0								44.
			65.7		84.6	*1.0	***		40.5	84.4									11.
		80.1	84.7				90.				10.7								
				03.3					44.0	47.4	70.4								
***	P		04.0			67.1				41.2	75.0								144
1.5	27.4	19.0		4.10	45.5				47.4	79.0	14.4								101
	75.7	11.5	10.0							70.1	73.7								100
	79.0	75.1	17.4	70.5	61.2		07.0		10.0	74.4	74.1								100
	N	11.1	74.5	n	11.0	10.0	PO. 2		10.7	75.0	75.4								14
			PO.0	78.0	74.3	79.0	75.0		73.3	14	74.5								*
w.					4.4	4.0				0.0	0.6								**
							2000												

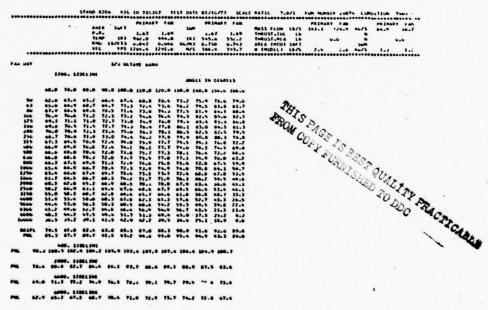
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							IMARY	PAR		-81	MAY (-				De I MAR			PRIMARY	
				-	540				100				MASS .	LOW	LB/S		0.0	aws.	4.4	
				P			47	1.69		1.6			THRUST	1,104	LO				1000	on man
				TOM	- (8			••••	(41				THRUST				6.0			
				RHO		. 0.0			ECW)				AREA (10001	241			-		
virginal.			esa merusian	AFF		1 1244		4- 141	WI	384.		1.7		DELI	18/3	2.0	2.0	ML/S	1.3	1.2
*****	*****	****	*****	*****	*****	*****				*****	****	****	*******	•		••••••	• • • • • • •	*****	***********	••••
					W	05. TAW	-	MODEL	411	10114	474	15.6		20					- (muse	
-							-		DAME OF			Desire Co								••
calta									& AMEL		111001									POW
ami I	••	~	*	**	100	110	126	1 30	140	150	100									14-1
				0.0	6.6		6.6	0.0	6.0		4.0									
. 663	0.0	0.0		0.0	0.0	0.0	0.0		0.0	0.0	0.0									
					6.0		0.6	6.0	6.0	4.0	6.0									ī
140	70.0	19.9	87.6	4.3	03.0				44.2	43.0										100
125	20.5		82.7	4.50		45.0	84.6			94.7	98.4									. 10
.100							85.2			*7.1	100.6									117
200		45.1		45.7																440
./20			84.0		47.3		. 7		**.*											417
. 389				.1.7		40.4	*7.*		101.1											120
			10.1	.1.0	.2.7	94.2			164.7											140
100	•	***	₩.0	*1.0		**.			107.1											153
30	*1.1		42.7	*1.7	· . 7	**.5			107.4											1.0
		***	*2.2	2:1	****				160.0											Les
	****	****	***	-:-	**.*				104.9											140
	****	*4.4	*3.4	****	**.1				104.5											10
	94.3			90.2	**.6				163.4											17.
	****	.1.5		94.3	94.2				104.3											14
1.12			94.7	94.1	94.1				103.0											121
			92.0		94.7				101.0											170
	96.5		42.3		**				100.3		*5.2									110
	***1			*1.4	95.0	.7.7	**.	140.1	**.6	**.*	*1.)									
				*1.1	45.2		**.*	**.1	97.7	95.0	91.5									440
16.6		87.6	40.0	92.4	94.7															
14.5	47.3			45.0	*3.0	**	.7.0													
10.0			44.4	*1.5	93.3	**.*	95.9				84									115
. 0.0	**.7			₩.1		94.0														
***	.1.0		.7.3	04.3	*1.1	45.5	*1.0			67.5										111
14.5							.1.7			6>.7	.1.5									4.4
			****				**.				70.0									100
30.0	10.0	14.4	79.1	01.4		***														101
	12.1	75.1	77.0	77.1	79.4															300
146.	11.4		7.0		4.0	6.0	0.0													101
	•••			0	•••	0.0		•••	0.0	•.•	6.0									

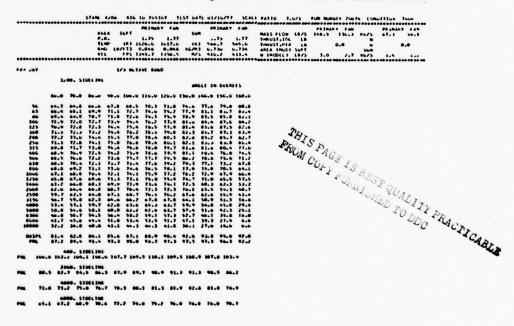
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

						m	IMARY	-		PRI	-			PRIMARY	-		PRAMARY	
				-					Soft			MASS FLOW	10/5	0.0		444	6.0	
							*	1.17	Carlot at	1.7		THE UST, ICL						
				1100		1 1030		117.0	(4)			THEUSE, MEA		•			•	
				Red)		> 0.0		.044	EEW?			MEA (400)		191120		Sum		
				VIL		5 1305		336.3	N/S			4 (MODEL)	10/3	3.0	1.7	19/1		1.4
*****		******					•			•		*************		********	*****	*****	•••••	
					1/1	GL 1446	-	-		MO1 54	DATA 15.	OFT RADIUS		-			- 48404	
~					10.00		-		-									•
								-			SE CAFES							fue
1241		70		••	100	110	150	130	140	130	100							84 -8.
636		4.0				6.6					0.0							
	0.6		6.0	0.0	6.0	0.0	6.0			0.6	0.0							- 2
		0.0	0.4	0.0	0.0	0.0			0.0		0.0							•
14	70.4		83.4			05.0	04.3	89.1	90.8	95.0	**.3							
129			44.7	04.1	85.9	87.4		89.1	93.2	***	99.7							112
100					84.4	45.2	84.7	88.7	93.1	98.7	102.1							
446		01.4	85.0	84.2	87.6	84.3	10.6	¥2.6	97.9	161.2	105.5							
256										104.0								440
315				89.1		94.1				101.5								140
-			*1.3		w.1	**				111.1								1.3
200		*1.7	.5.1	93.1	**.0	90.1				113.2								4.0
. 10	**.*	*3.*		**.3	**.*					114.6								120
-00	*3.6	**.7	*1.0	97.4						115.1								1.0
	****	****		97.0						111.0								100
	.0	**	*5.0	****						111.0								120
	92.0	**.1	*>	***						100.0								10
. 34	92.2	****								100.0								12
		*5.4	**.>	95.7						104.6								10
	94.3	*).4	94.4	94.1						102.9								123
	97.7	.2.4		9.7	97.5	**.*	162.3	103.3	167.5	100.9	**.*							47.
. 30			43.7	*5.3		99.7	101.9	105.1	101.1	**.2	47.2							4.1
	w	*1.*	*1.4		***	**.3	161.1	101.4	**.*	*1.1	**.*							120
10.6	**. >		97.0		***			100.7										419
10.5	87.4	40.1	42.1	*1.0	95.7	****	**. 3											
10.0	47.0			43.2	**.6	*1.5			*5.4									
14.0				*1.*	*1.0	90.1												4.0
13.0			**.0	*1.0		**.5	**.											114
11.5		****	40.0		92.1	*3.4	**.2											***
			***	9.00	90.1	*/.1	94.2											1.1
	10.5	10.4	83.6	65.0	44.1	****	***											12
0.0	73.1		77.0	70.5	80.9	42.7												14.
100.	0.0		6.0	0.0	0.6	0.0	0.0											
																	DAPOL .	1 36
				100.2														

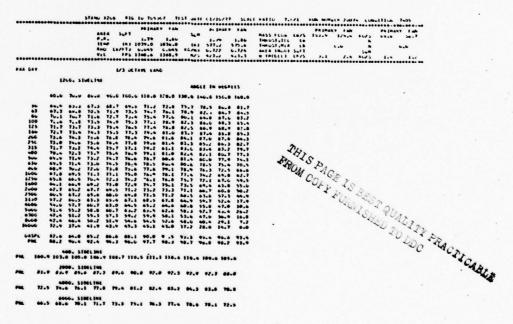
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							41				MARY FAM	***************************************		PRIMARY	-		PRIMARY	
				-		**		40.00	540			MASS FLOW	10/5	0.0		86/5	•.•	
							. 70	1.00		1.7		THRUST, TOL						
				**	10/	10		4.06.0	(4.)			THRUST . MEA						
				YEL				0.045	46/41			AREA (MOD)				-		
***	****			****		PS 1 M	••••	1,68.9	~!	47).	2 475.3	. (40041)	10/5	3.4	7.0	46/5	4.4	
										*****		***********	*****		****	*****	*******	••••
					1/1							OFT RADIUS						
-					•••					-		OFT MADIOS		-		D PL	- (#1111	
	-						-	-	-									
-		76	86	**	100	116	176	134	100	110	100							PLO
										•••								** - 1
-30	0.6	0.0	0.0	0.0	0.0			. 0.0	6.6		0.0							
••	0.0	0.0	6.0								6.0							
100	4.0	0.0	0.0	0.0	***						6.6							- 1
100	70.0		84.2								100.0							111
153				7							100.7							11
200	63.0	.1.2		65.4	45.1				****		165.9							114
434		***	***	.7.1	***				****	103.9	100.5							
114	6,.6	12.0				**		• ••	161.1	100.0	160.4							12
***	67.4	97.1	**.7			•1.		1 99.7	143.1	100.5	115.5							
=	**	*/.5		93.2	**.*		•••	102.5 5 103.7	107.6	115.1	115. 1							8.0
31.0	***				****			· 100.1	110.0	110.4	114.0							140
144	***	***			***	***	101.	100.1		113.0	110.0							
	.*	**.*			*	1	161	# lu7.		110.5	115.0							
.23					164-6	161.	100-	. 100.0	11.5.	110.7	110.0							
	>.				**.	101.	100.	1 107.4	110.		****							4 34
	**.*	45.6	**.*	47.5	99.1	141.0	100.	1 107.4	100.		112.2							
. 24	93.7		95.4	97.3	**.0	101.0	104.	1 104.9	100.	105.5	.10.2							10
		94.7	*5.5		90.7	101.5	100.	2 100.7	100.0	167.1	147.4							1.3
	*1.1	**.*	*5.>	46.7	***	161.5	100.	2 105.7	105.4	143.1	144.7							
		43.0	****		10.5	101.0	.01.	1 104.4	103.0	147.9	101.9							
.30	*4.5	*1.1			**.3	100.	162.	. 103.5	142.4	1.1.1	**. ,							
	.1.6		**.2		*1.6	160.	162.	. 162.4	161.0	**.2								1.1
	***			*>.1				4 161.4										
::	**.	*1.1																
0.0	60.7	90.2		7:1	**.1		***											
	65.7	67.4	***			***												114
1.5	47.4	44.0				**.			*1.1									445
	14.3	10.4			41.1													
	06.2					90.												11/
3.0	77.5	74.6			95.3				***									
	73.7		70.4			•1.0												100
w.	6.6	6.6																
								-	***	•••								•
																	-	111
	144.4		147.4	14.2	116.4	-11 3.4	115-	. 117.4	121.2		134.1							

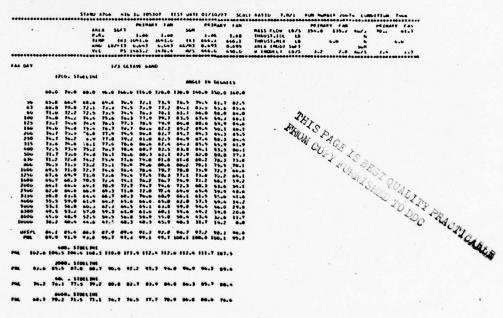
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							-	-						PR IMAP	*		PE IRALY	-
				-					500			MASS FLOW		0.6	6.6	44/5	0.0	6.6
							. 84	1.00				THEUST-10						
				11-		1 100		0.140		***		THRUST					•.	
				AHO		3		6.43				AREA INCL				144		
				ver		1 100			v	****	420.0	# (MCD(L)	10/5	3.2	4.0	4/5	1.4	1.3
****	******									•••••		••••••	******	•••••••	*****		******	••••
					1/1		-	MODE 1	-	-		CUEDAR THE					- INCOL	
-							-										-	
							410	ROPHUM	-		PIGALES							-
	••	70		90	100	110	120	1,00	100	134	140							44 -4
- 30	6.0		•.•	•.0		•.•	•.•				0.0							•
143			0.0			6.0	0.0				4.0							
100		0.0	0.0	•.•	0.0	•.•	•.0				4.0							•
10	97.3	****	45.5	99.5	87.5	47.1	47.0				101.0							114
100	****									166.6								114
	45.4	83.3				****	92.0			104.9								1.3
/20	****	****			40.3	*1.4				107.6								121
115		80.7	90.0	*1.1						100.								14
-	*1.3		*1.5							111.0								177
200	***	**	*1.*	**.						115.4								17
610	90.1	****	*	47.0						110.0								1.0
-	**	**.>								117.5								1.1
	**.	**.*	+1.0	90.2	100.3	102.7	100.0	164.1	114.9	110.0	117.0							1 14
		41.4								117.2								1 >1
	47.4									110.4								8 .1
	**.*	*1.7								114.5								121
. 26		**.*								115.7								140
-15	43.1									110.0								427
	***	*>.0								100.0								
·w	**.7		44.1							101.7								133
										104.0								174
	71.0	****		17.4						100.3								122
5.5	*1.1			w.0						**.0								1.1
	40.2		*1.2	****				100.0			***							1
	***		*7.0	w		**.1		47.4										***
	17.7					****	•7.											447
1.5	****	.7.4			***	**.*					44.4							114
4.6	***	***		w.,	*2.*	**.*	**.											110
4.0		44.5	**. *		91.4	92.5	92.0	*1.0	90.4		47.3							1.2
3.0			84.6															
	77.0	14.2				47.7												100
10.	0.0	0.0	0.0	0.0	0.0						6.6							
																	-	
							110.											

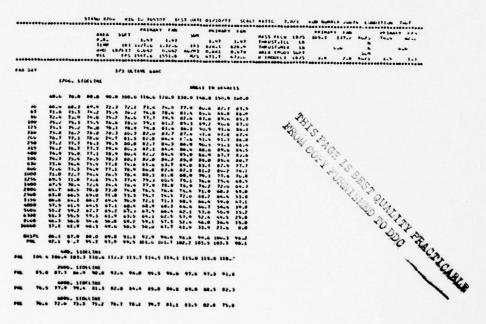
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

				444			LAMA						**********		P4 1 548			PR 184	
				7.4			. 97		Son				-455 FLOW	10/5	6.6			6.6	
				11.5				1.97		1.0			Desust.100						
				810		3 0.0		0.04	4/1				IMUST. mt			6.6			4.0
				WIL											6.60		244		
	*****		*****		*****	*****	*****	*****			7 473.	.0	# (mcDet)	10/5	1.4	1.0	46/5	1.5	1.,
														****	••••••	******	*****		
					1/3	OL TAVE	-	-	41	-	-					222 1025			
-	200														RAW C		0 251	- INC	411
							410	AGPHON	-										
11	40	70		**	100	1.0	120	134	14	176	104								PU
		401740							70.00										46-1
***	•.•	6.0			0.0		•.0		0.0		0.0								
*	•.•	6.0			6.0	0.0	0.0			6.0	0.0								
00	4.0	0.0			0.0		0.0												
7	**.1	87.1			87.4	64.5	**.				101.0								4.1
-	***										107.7								- 55
~			87.3				**.		97.0	W1.3	M>.2								
×			84.5			.5-1	•).•	9.7	101.7	160.0	160.7								
			91.3		****	*2.0	*>.1	*1.*	103.4	100-1	110.0								12.
	.1		93.4			**	•1.	101.5	105.0	110.1	114.3								
=	*4.		****		*7.0	**.*	100.0	105.0	110.5	114.2	117.7								42
100		*1.7			•7.4	****	102.0	104.7	117.0	117.6	110.1								1.
					100.4		103.4	104.1	114.6	110.2	110.2								11
-		**.1			161.4	101.	104.3	104.4	113.4	114.5	117. 5								
0		**.*	**.4	141.2	112.0		100.7	*****		110.7	110.4								
	144.1		144.6	161.1	102.5		16.7. 2	*****		114.4	116.5								
	.*	**.1	**.1	141.4	164.3	144.5	107.4	116.0	**		110.7								41
×	70.4	**.)	**.2	146.0	164-1	11.4.7	14.7 7												
15	*7.7	40.2	**.*	100.4	107.0	104.4	167.4	109	111-4										8 41
~	**.	****	***	100.2	100.1	104.4	107.	109.1	110.0										4 34
•	***			**.*	101.0	164.7	140.0	107.4	100.7	110.4									
*	*5.7		*1.7	**.7	101.7	100.1	100.5	107.0	107.3	100.0	101 4								10
•	**.7	70.1	**.*	**.1	101.2	101-4	105.7	105 .	104.4										17
	*1.0	**.*		**.7	166.4	103.6	100.4	1-0-4	104.1	104 .									1.
	****	**.*	**.1	** . 1	166.1	162.4	103.4	169-1	107.1	101 4									14
	**.6	43.5	*>.,	.,.	**	101.7	BUC. T	102.1	101-4	102.0									**
	W 3		**.1		40.5	100.4	101.0	100.3	**.5	100.2	*1.4								1:
.5	***	91.7			*1.1	**.1	100.0	90.5											111
				.*	**	**. 5	**.*	*1.4											***
	84.1		*1.7	*1.5					45.2		*7.3								***
	*1.7		84.0	10.0				**	*3.4		90.4								
			.7.7	44.5	07.7			*1.2	*6.4	41.0									
Ď.		0.0		0.0				67.1											100
-	2.0		3.0	3.0	0	0.0	0.0												

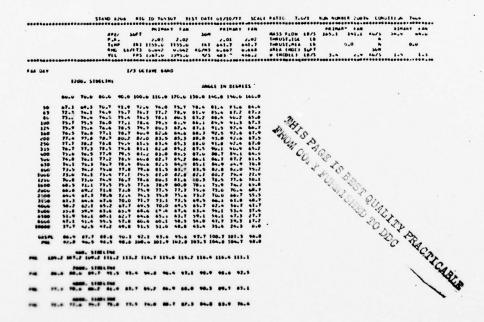
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							18481	*		~ (A	MY FAR			0.0	6.0		PA JAMAY	0.0
				****	500	٠.		2.02	348	2.01	2.02	THEUST . I DE			0	:		0.0
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				-			- 4 19		P/L	441.7	484.2	w (MODEL)	4/5	3.4	2.0	RWS.	1.5	1.1
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000	•.0	0.0	47.3	0.0	47.4	4.0	40.6	92.4	14.5		102.7							111
125	43.4	01.5	47.1	00.2	90.6	90.2	40.0			**								111
140				44.4	47.0	44.4				162-1								11
200		47.4			91.5	92.5				160.5								12
750	88.3	84.1		*1.4	92.4	*1.5				100.7								37
212				43.2						::1.0								10
w		**.*		97.6	90.1	**.6	161.4	105.5	110.6	110 0	110.1							13
340	93.7		**.*		97.7		102.0	104.8	113.0	117.0	110.0							.,
434		.1.0		94.5														
-	***	97.9	96.1	**.*	101.7	102.2	105.1	110.9	114.9	120.0	117.9							13
	**.3	98.9	90.7	100.2	107.5	104.7	107.4	1115.2	110	126.5	119.4							**
	14.2			101.7														12
	100.4	100.4	100.5	101-2	103-5	105.6	100.5	115.1	147.5	150.>	140.0							15
.00	**.*	**.1	100.3	101.4	103.1	103.2	100.4	111.0	110.2	110.5	117.7							
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.45		**.*	**.	401.0	103-1	103.3	108.0	.10.4	113-3		113.0							
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6.6	*1.6			*7.3														10
				96.7						166.7								31
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W-0	1	87.6		41.0			***			*6.2								- ::
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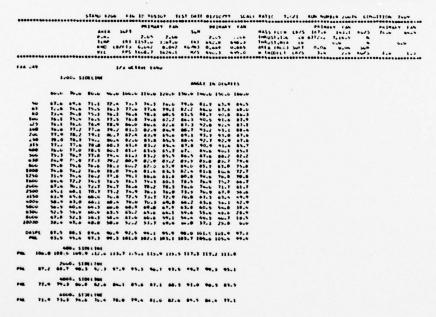
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

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20 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1				44.5			90.1	92.1	97.9	102.0	105.8								**
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10 TO 100					****	****	101.6	105.0	.11.2	111.5	110.0								13
16 TO			**	**.*	100 1	1	103.0	107.1	117.4	110.7	117.1								13
1 166		.7.0	**.*	100.3	161.5	100.	104.				110.1								
40 lou	•.•	**	99.2	166.7	1W. 9	10 .L	167.7	112.4		121 0									1>
. 100		160.3	100.7	103.3	100.0	104 7	100			1.4 4									1)
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u iu		100.0		102.1	101-4	105.4	104.4	112.2											
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06 97	-		100.2	101.2	102.4	102.5	160.0	100.5	110.0	111.4	100.3								
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		**.7	94.4	w	161 -	104.4	104.2	106.2	107.6	100.0	107.0								120
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.0 .1			PG. U	4.1	45.5		**	*1.1	99.1										***
		.1.0		M.,		90.5	46.0	40.0	91.5	*3.4									11:
٠. ٠.				4.0	6.0	0.0		•.0	0.0	0.0	•.•								•••
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PL 111.																			

b.) McJel Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

## All a Subst 1-32 1-19 1-32 1-19 1-32 1-19 1-32 1-39 1-32 1-39 1-32 1-39 1-32 1-39 1-32 1-39 1-32 1-39 1-32 1-39 1-32 1-39 1-32 1-39 1-32 1-39 1-32 1-39 1-32 1-39 1-32 1-39 1-32 1-39 1-32 1-39 1-32 1-39 1-32 1-39 1-32	
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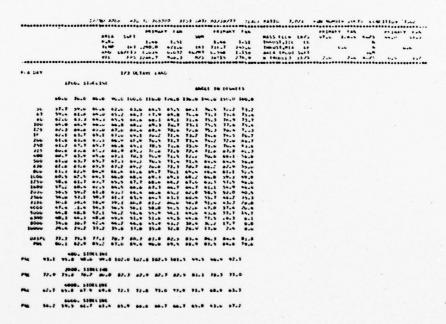
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

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		70.0		~	100.0	110.0	120.6	1	100.0	140.0	160.0							
50	53.7	>>.2	34.1	St	54.6		.1.3			47.2	47.4							
• •	***	>1.1	>4.4	.1.1		41.4												
•0	37.4	70.4	>*.>			4 7		44. 1		7								
114	****	,,	,	45.7		4>.4				70.5	40.1							
100			••••	62.4			47.4											
200		77.1		•	••••	• • • •	***			67.3								
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466	10.1	**.*		3.0		****	••)		• • • •									
300	34.0	50.4	.1.5		44.4		****		• * * * *		31.0							
630	37.3	74.4	.1.3			45.5		-1-1		70.0	71.0							
-	37.1	>0.>	•••	62.1	4		44.7											
1000	30.0	30.0	****		41.4	44.1				\$3.1	41.4							
1520		>1.4	>4.1				4.4	41.7	.7.	21.4								
4000	37.0	>>	57.0	>1.1	*4.5			30.1	**.*	44.0	37.3							
4340	3	34.0	37.0	34.3	04.1		40.2	17.0		41.1								
1170	**.*	3/./	****	50.7	**.*	30.7	>7.4	**.*	40.0	41.4	17.8							
****	****	****	31.6	20.0		>>.7	34.7	W.7	****		16.0							
1000	****	****	****	31.7	**	***	·	44.5	••. 7	30.4								
4106	****		****	****	>4.6	36.2	41.0		37.3	41.4	4.1							
800w		11.7	17.1		****	****		30.6	,,,,	10.7	0.0							
10000	11.0	****	10.1	10.1	11.0	10.4	27.2	41.4	11.1	11.1	0.0							
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-	15.4	**.:			**.				41.7	77.3	11.5							
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a.) Model Data Measured At 15 Ft. Radius

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103	0.0			0.0	0.0	6.0														
w	vet	4.0		6.0	U.0	0.6	6.6													
144	71.0	14.0	17.4	70.0	10.1	19.0	.0.													100
143	14.5		10.0	M.1	74.0					**.										100
144	74.0		74.5	70.2	70.3			01.0												100
200	14.0	11.7	19.1	1.00						****										44.
313			62.0	*1.0						**										
			*>	10.5	20.1					.03.0										115
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b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

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444	••	10		**	100	1.0	450	134	100	136	100									41 -4.
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		0.4		6.6	6.0	6.0	0.6			0.0	4.0									•
	6.0	0.6		0.3	0.0			0.0		0.0	0.0									:
100	74.6	13.0	70.1	N.2		DO.4	•1			****										
125	77.4						41.7			¥1.0	**.1									1.7
100	77.0		79.7	.0.0		.1.1														
	10.4	76.5	86.1	66.8	84.1	45.2	65.6		*3.5	94.4	140.4									44.
- 34	70.0		4.1	.1.0	04.1				*>.2											114
.13			84.0			.1.4	.,		***	101.0										
100		83.4							1-1-6											
***	****					.1.5			163.1											1:0
+10	4		67.4	40.2					163.0											1
100				w					101.5											
			44.0	40.1					102.0											1.0
-23				90.2					141.0											
-60	80.7	41.5		W.3		04.5			14.0											110
.50		67.6		41.0		****			**.7	¥7.1										:";
-15				*1.2		***	-			*7.4										147
		44.2			*;;;						¥1.									117
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		40.1				*>-1		***			***									***
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	49.3		***	w	9:.5	**.!		**.	97.3	90.1										
				w	**.*	*1.>	*1.5		*1.1											
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		47.0		.7.7					.1.2											44.
1.5		.1.6		n.,																
		70.7		.1.2		***	**.*				75.1									
6.6			10.7	M. 5			w.,				14.0									****
3.6		71.0		70.3				17.4												
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w.	0.0	0.0	4.0	0.0	0.0	•.0		0.0	r.0	0.0										
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b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

******				STAND	1766	* 14	ati Tus		151 14	1. 41/	10/17	SCAL	+ 44f10 7.	410	-	14			1.0.
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						140			34				MASS FLOW	11/2		1 33.0		****	· · · · ·
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					mp	14) 1			-	1 7,0			IMPUST , MI &						
						# t3				3 0.1	30	1.15	AREA IMCLI	34.1			**		
								+15.4		3 341		205.3	. INCOLL!	10/>	***	1.7	44/5	U. Y	
••••••		•••••	•••••	******	*****	•••••	•••••	******	•••••	*****	*****	• •• • • • •	• • • • • • • • • • • • • • • • • • • •	*****	******	******	*****	******	*******
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			N	80.0	46.6	100.0	110.4	170-0	130.0	14.0	154.6	166.0							
100		anarona.	782 10																
*	1	****		.7.0		44.0	**.*	47.7	70.1	17.9	14.4	75.2							
60		• • • •	****	•>.1		40.4		10.4	72.0	75.6									
104				****	•/.3	***	**.	71.7	****	70.1	****	75.6							
423						70.3	11	12.0											
100				.7.0		10.7		12.1	74-0		77.1								
244						71.4		***											
290				.7.)	40.4	24.4	2/-	73.4		75.2		44.4							
313					40.7	70.7	70.0	71.1				41.0							
***					44.1	11.1	72.1	71.0				41.1							
364		1.50	42.1	47.3			74.5	71.2	73.5	71.4	44.2								
630					49.0		12.5	73.2	72.7	70.5		97.0							
Bre				4.4		74.0		72.7				\$4.9							
1000		w.,	•••					71.5				92.3							
1250				•3.7	67.6	44.0		70.0				50.0							
2000				***	***	****			.1.7			47.3							
2500				.1.7		****	***	67.5	• • • • •			*1.*							
1156			34.2	***	41.7	4		47.7	****	***		30.9							
4000		49.4	33.2	50.3	20.7	44.4		40.4	44.4		40.3								
5000			14.4	34.6	30.1	30.4	>4.0	20.7	41.7	49.7	30.6								
6300	•	****	43.7	44.6	51.0	>1.0	>>-	31.0	44.0	30. 1	20.5	1.0							
8000	. 1	3>.4	4.6	43.8	45.5	40.3	47.4	44. T	10.0	31.0	17.7	4.0							
10000		15.7	34.5	30.5	30.0	20.4	37.4		20.0	10.1	4.2								
0450					-														
Prot						*/	• > • •			.,		64.1							
		••••		• • • •	••	*1.5	*1.*	*1.7	*1.1										
		•		OF L 101															
-	47.2				. 164														
				W. 194															
~	**.*	77.	·								1 79.								
											000								
		***	× 11	Det Int			an Service												
~	••••			71.	7 73.	74.	. 11.	1 75.	74.	, ,1.	* **.	,							
				DE L 140															
-																			
			•••		• • • • • • • • • • • • • • • • • • • •	- 00.				. 45.0		•							

a.) Model Data Measured At 15 Ft. Radius

							a laws	-			MARY FAR	***************************************			***		P. 1848Y	
				P.A.			-		Son	-		MASS FLOW				-	•.•	0.0
				1100			. Pe	1.70		1.7								
				***	40/11	1 145		4>7.6	141									
				***		1 134		6.071	KEM)							544		
***	****			*****	-		****	1022.7	×	40).	2 ,20.0	# (MCDEL)	10/2	2.0	7.0	44/3	1.7	1.5
								-				************	****	********	*****	*****	******	****
					1/1	CE TAV	-	-		-		OFT RADIUS						
-									•••					AAH CU	ant Cit	9 74 F	- 1844	4.1
							810	-			DICALLS							
and I	*	70	-	**	100	110	120	1.04	100	134	104							PL.
									-									10-4
-	6.6				0.0				0.0		0.6							
Les	•.•		0.3		0.0		0.0		4.0	6.6	4.6							•
•		0.6	6.0		6.6		0.0		0.0	6.0	0.0							•
100	11.5	80.2	82.0						91.0	94.2	**-3							
143				m.1	12.4				*3.5	**.*	**.5							110
-											161.7							400
200	41.0	.2.7			.1.1					104.5								44.
276						44.4		* *5.4										112
115																		17
-								161.0	160.0	114.3	113.4							100
**		.1.0			***	**.	**.	101.3	109.4	114.5	111.6							1.0
• >•	•••	**				**.1	100.	165.7	110.4	114.4	115.2							170
•	****	**.1				7	100.0	165.1	111.5	110.5	114.0							1
.25	• 3 >	*1.1				100.0	101.4	144.3	111.0	114.3	11							1.0
	****	••.				100.5	102.7	100.4	116. 3	117.4	143.3							4. 7.
		****			****	14.5	164.1	100.1	100.7	110.7	111.1							
	****	*3.2						100.0										1.3
	41.5	94.7			**.1	100.4	105.4	165.7	100.5	105.9	100.1							
	***			****	****	100.0	101.1	105.0	105.1	101.7	103.5							
		*7.0			****	101.6	101.7	14.0	104.1	165.0	101.1							111
- 10	**.0	94.7						103.7									•	4
	W. 1	****						103-5										
0.0	90.4	***			****		101.	107.5	101.6	**	**.*							1.1.
	99.1	91.7		***			101.3	101.4	100.7	****								
•		*1.1			****		100.0	14.6	**. !									
					****			*7.4										11.
>		***			****													
1.3	44.2	**.>		11.5		****	*1.7		97.1		• 7. L							
				10.0	****			71.1		***	61.0							*1
						.	**.1			**.	**							
	13.3	11.4	70.0	****	89.1			****										
	71.1	7/-4	72.4			*1.7				14.0	01.0							
w.			4.0			6.6			7.6		0.0							
		-																
				14.4														

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

*****	*******	*****	•••••	•••••	*****	*****	*****	•		*****	W.A.	7.	wı	-			-111	
					5401	** 1 **				-							******	*******
					3601	1. 70			4.0	-		MASS ***	10/3	1/4.5	111.1		20.1	** ***
				-						. 74	1.10	f# usf . 1 (4						
					A1) .	*****	457.			1.1	304.6	Immust.mt			4.0			W-1101
							1037				1.434		54.1			>		6.0
******	******	*****	*****	*****	*****	*****			/> ••	1.4	14.0	. INCUIL!	10/5		4.9			
									•••••	•••••	*****	. IMCUIL 1	*****	-	******			
** ***																		
		· u. s	14 : 14															
		10.0	~		100.0	110.0	170.0			154.6								
-																		
••						10.5	71.0	15.6	70.7	79.5	80.>							
			70.1			74.1	75.2											
100	10.1		20.7			70.7	15.0		0		01.7							
	14.7		11.1															
400	10.1			14 4	10.1	10.0												
244	71.7		*11.0	17.0	17.2													
250	14.4				10.0	76.,	****											
115		10.0	****	74.1	****	****				67.6	14.							
			17.5		14.0	10.4				74.0								
100	67.4	20.1		10.1		10.0	74.0			77.1	*1.7							
	40.1	49.0	14.4	10.1			Pv. 4	w.?	14.5	14.5								
***			*4.5	75.0		17.0	10.3				47.5							
1000			**.2		79.0		17.0	10.7	****	10.4	****							
1436			74.7	77.7	79.4	70.1		??	74.0		41.2							
1000		.7.4		71.4	10.2	75.7		***	4.1									
2000					73.7				10.0		**.*							
1500				44.2	74.2													
1134								47.0		>0.0	****							
4000		10.4		44.7				41.4	\$7.7	***	30.0							
1000	31.7	35.7			****					**. *								
		****	**.							***	14.3							
-	*1.1	45.1								17.0	1.7							
10000	10.4	17.0	19.0		****			12.4	23.4	17.5								
-			***															
~					****	*7.0	***	****	**:	****	*1							
	.» led.	4 100.	. 107.	100.	110.	, 11v.	. 110.	100.	. 167.	103.								
	20	11																
	. 3 67.			44.				***										
	71.	1 79.	3 77.1	n.				92.										
		. si																
			1 71.0															
								70.0	****		,							

a.) Model Data Measured At 15 Ft. Radius

							IRMA	-		PRI	MARY F	**		PRIMARY			PRIBARY	
				484					100			MASS PLOI	10/5	0.0	6.0	aws		
				P.A.			74	1.74		1.7		TA THRUST, 10	4 10					
				Item		1 1434			141					6.			6.	
				MMU				0.670	KLA)				34.			Sum		
				Aff		3 1035	.7 1		N/S	490.	. ,,,	- B U (MODEL	10/3	2.7		46/5	1.7	4.3
					******	******	••••		****	•		***********	******	*********		•••••		****
							***					13.00T RADIUS						
-							-	-	**			13.001 KADIUS		*** (0	MECAL	. 201	- (mube o	
											DEGat ts							
and I	-	10		96	140	116		100	100	136	100							•
					•••	•••		•••	•	• • • •	•••							44-1
***			0.0	0.0	4.0	6.6	0.6	0.0		0.6	0.0							
663	0.0	0.0		0.0	0.0	0.0		6.0		6.6								÷
000				0.0	4.0		0.6	0.0	0.0	0.0	0.0							ī
100	74.3	60.9	81.4	W.2				w		45.7	**.5							111
125	.1.5						86.7				100.5							,
100			84.0	85.7		86.2	67.6		**.2									114
300				41.2					**.*									
4>0									101.0									1
315	**.1			₩.2					104.1									
***		***		*1.0					107.0								7	10
430				**.2					116-3									
800			***						117.0									
					**.1	**.*	101.7	104.0	117.0	115.4	113.0							6.0
.25	***	*>			100.7													1 11
-44			10.7		**.*													
			**		**.*													1.0
		94.1			**													
-13			**		****													11
					**.*	107.0	100.2	1	100.5	100.	100.0							1
	93.1		92.5		**	101.0	101.4	104.1	10- 4	103.0	104.1							10
. 30					100.0	101.9	101.4	104.4	101.4	167.3	had . 1							
					**.*													:::
6.6	91.3	97.5	**.*		**.													
		+1.0			99.5													
4.4		.2.5		**.5					106.7									
1.0			94.7		97.3	**	w.,	**. 1	***	*>	40.7							
3.6							.7.0	.7.4		94.6								
1.5			w	42.3		44.2	***	45.4	**.5	91.0								11.
0.0				₩		**	*1.7		*									
0.0		04.0			64.7		₩.4											
1.0	10		.1.1															
0.0		10.0			.1.>		***				41.4							
w.	4.0																	

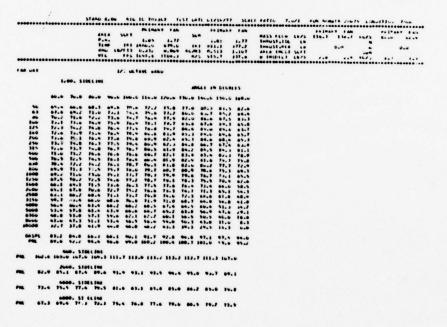
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

						~1~												
				1				*				-	10/5	114.1			24.1	
						4. 79	1.74			. 79	1.70	1-0-051-114						
			-	~			444.6		1 14			Imau51,#14			6.0			4.4
			-		***	0.01.	u			117	1-445							
			**		1	1.00			5		17.0		10/3			44/4		
******	*****	*****	******	*****		•••••	******	*****	*****	*****	• • • • • • •	**********	*****	******	******	*****		
		ee. si	-															
									-									
		N	**.0		14.0	.10.0	170.0	130.0	14.0	110.0	100.0							
50						21.0	14.1		70.0									
			10.0	72		75.1	10.0											
		N		1					85.1									
100		12.0		19.0		11.2				00.1								
145	71.0		71.1	13.3				41.0										
100	74.0		14.5	77.0	71.7	P												
300	17.5		15.1	10.0	70.0	74.0	.1.1											
130	14.1		15.0	70.2						05.0								
	70.4		73.0	75.4	77.5	74.3					10.0							
+00			7	15.5			41.0			14.7	70.1							
100			73.3	75.4					77.9	77.1	71.0							
410	69.4		13.6	17.6		PO. 1			10.7	73.5								
1000			17.0						10.0	77.0								
1336		N. 1	77.1	14.7					15.2	71.0								
leau				****		77.0	****	****		**								
2000				71.7		71.4		74.1	17.0	*1.1	37.7							
1500				Pe		13.4			47.7									
1136								47.3										
-		**.*				47.				**.								
1006		10.0						40.1			40.1							
4 10u	41.1		34.4					19.1		10.4	10.1							
8000			10.2			35.1			41.1		0.1							
10,100	31.0					43.6		10.1	11.0	10.1								
04501	4.1			47.1	44.0		*1.4											
-		*1.7	*1.4	45.7		**.1	10.2		**.2		***							
101.						. 111			. 100.	. 105								
***				1 00.			05.	1		1 70								

a.) Model Data Measured At 15 Ft. Radius

									427777		MARY FAM						** ***	
				***					500			MASS FLLW						
				11.0				1.77		1.0								
									(0)						6.0			
				-				4.60	CO/M)							-		
				VAL		-> 100		1104.3	4/5	115.	1 ,,7.6	. (4011)	10/3	2.0	7.0	/>	1.1	
				•••••		•••••	*****		*****	•	*****		*****	******	******	*****		
						~						1 041 105						
-					.,,			- Tue			DATA 15	CULIAN LAN.			CHARLE		· Inun	
	••	10	-	••	144	110	176		100		DIGATES							-
			••		•••		170	1 30	140	100	100							44.
				0.0														
043	0.0	0.0	0.0	0.0	0.0	0.0												
-		6.0	6.0			6.6												•
1-0	14.1			43.1	45.2						100-0							
143			62.9		87.3				*>		161.4							
1			85.7							144.								
200								* **. >										
124								* *7.5	10.1.1	1								
113					97.4	94.4	•		107.1	10.0	111 4							1.0
***	+0.1		¥2.4	**.*	***			163.0	1000		*****							./-
		1.			**-1	30.7	1.4	. 107.0		114.5	*****							1.
-	95.0	**.		97. 1		Made . de	16.2	107.6										1.0
		***				100.7	Au.	100.2	****	117	*****							4 21
		**	¥7.0	Vd .4	146.3	142.7	140-	1 100.7	114.5	111.5	1117							
.25		97.7	90.4	**.1	101.4	101.0	105	109.4	114.1	114.								
-00	**.*	*7.6			101.1	101.1	105.	100.9	117.4	113.5	111							4 74.
	99.4	14.1	**.	w.5	14.1	107.0	103.	1 100.0	110.4	112.4	111.4							
. 24	94.3	95.7		94.5	100.9	101.7	105.	100.3	100.4	111.1	44							
.15		**		90.4	100.7	103.4	100.	101.1	100.2	100.4	100.1							
-	****	****	**.7	50.5	161.2	101.1	103.0	107.1	167.1	107.4	101-1							
		94.7	94.4	90.4	100.9	101.1	100.	. 100.0	104-1	145.7	144.6							
. 10	*1.*	**.	w. 1	**.3	lul.u	101.1	104.6	. 145.7	144-9	144.4	162.1							***
		**.5		46.2	141.6	101.0	100.	105.0	104.4	101.1	164.6							1:3
			90.0		144.0	164.6	163.	7 100.1	140.5		44.4							
				41.4	100.5	142.5	ius.		107.4	100.4	47.4							4
			**.1	*1.i	166.6	141.4	102.		161.0	**.4	**.*							1,
				45.4	**.*	14.6		1 144.3	**.*	*1.4	42.1							in
					.1.5		**.	1 90.0			96.1							
1.5			*1.3		**.>			* **.*	40.1	**.*	07.0							110
٠.٠		***		*1.4		*>			*3.3		45.3							111
									46.7									41/
		1																
		70.0	74								*/. 3							1
w.		v.6	6.0	0.0		6.6			6.6	4.6	0.0							
																		100

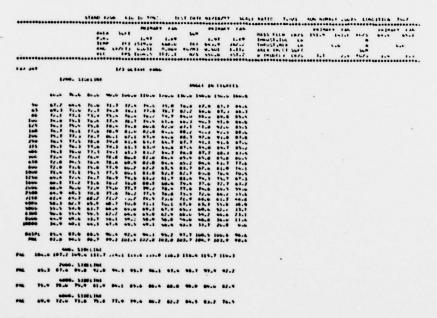
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft, Radius

				481	1 10		-		344		-			PR IMARY		46/5	0.0	0.0
						1		1.00		1.0		THRUST, IDE		•.•	•.•		0.0	0.0
				-		. 434			(0)			THRUST . MEA						
				440		\$ 100		0.000	46/4)			AREA (MOU)				SUM		
	****					2 100			~~	150.	357.2	# (MUDEL)	10/3	1.1	7.9	86/3	1.4	1.3
								-								•••••	••••••	••••
					1/3	UL TAN	-	-		-	DATA 85.	OFT RADIUS		-	RECTE		- IMULA	
-																		
	fate										DICALLS							-
and I	••		••	••	100	110	150	12	144	134	100							41 -1
					6.6		0.6			6.6	0.0							
.00)		0.0	0.0	0.0	0.0				6.6		6.6							
000	0.0	6.0			0.0													
bue	tu. 1				***		84.1				162.3							114
. 25					**.		84.1			14.1								
100	07.1									162.6								447
300		60.2								100.5								1.1
100	94.7									100.0								1
		**								111.1								
-										117.7								10
				-	166. 5		102.			114.7	110-7							101
-	*7.0		•1.		101.1	102.4	104		****	117.6	114.7							433
		**-		144.4	107-4	104.4	144.			170.4	115. 1							111
			146.	441.4	16 7	164.0	147.	117.7	114.1	119.7	110.1							434
	**									110.7								ix
										110.3								11
				161.6	163.4	165.7	100.0	111.3	113.9	117.1	110.3							450
1.15	17.6		**.2	140.0	103.1	105.0	100.0	110.1	114.4	111.0	114.0							
				101.0	101.4	100.0	100.2	110.1	111.3	111.0	111.0							1
			10.1	166.0	101.1	105.0	107.	100.7	110.1	111.4	104							
	w	***								100.7								4
		**.								101.4								1.0
	***	***								100.7								
	1.	**.								100.0								1.0
4.0	****									101.0								1. :
		****								100.0								
1.5										70.1								
										****								:::
								***										i
1.6	10.5					40.7				vu.5								***
6.0	13										43.7							
											4.6							•
					-													
																	-	1

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							I I MAR!	-		. PO 1	MARY F	-			PRIMA			PRIMA	
									500			MASS		10/5		6.0	-	0.0	
						2.		1.45		2.6	1.	Y5 THE	101.TC						
				117		11 134		443.4	(4)	839.		THAT	418.12			6.6			
				AMU	40/11	1 0.0	146	6.071	ELAS.	0.40	1 1.1	SO ARE	48001	Sett		0.0	244	W-004	·
						S 107.	4.0	110).0	4/5	370.0			WHI.	10/3	2.1	2.1	84/3	1.0	4.9
*****	*****	*****	*****	*****	*****	*****	*****	-	*****	*****		*******	-		******	****	*****	******	*****
-					1/3	CL 1AW	-	-	41		DATA	S.OFT RA	102		RAU C	CARLLIE	D 760	- (mu	411
											21 18410								-
LINE)	-	74		**	166		170	1.4	100	136									PLO
-			••	•		•••	***			. 30	100								41 -1
-			6.0	0.6	0.0	0.0			0.6		0.4								
100	0.0	0.0	2.0	0.0		0.0													:
000	0.0	0.6	0.0	0.0		6.6				0.6	4.4								:
146	62.1	49."	47.5		47.9	84.6	89.				102.0								
125			00.0		64.4	*1.0				144.5									
100	.7.0	89.7	44.3	44.7	44.2			91.6	**. ,										
200			84.3	10.2			**.		107.4	107.6	10v.7								1.1
· >	87.4			42.1	41.0		4201		100.0	100.0	111.3								. 1.
115			92.1	91.7		94. 3		. 107.0	167.7	111.5	115.9								
-	42.7			90.0		100.1	101.	160.1	111.7	115.7	110.0								
×		**.7						100.0											
			90.7					110.0											4:
-	.,.							1111.4											
								1112.0											1 1
.25								114.9											. 21
								114.3											
								1 117.1											
	**.7							1.4.1											
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	*7.7	**. 1																	1 1
. 30								110.4											
	**.7							1 100.5											4
	****							100.4											1
	.	**						107.											• • •
	1.							100.5											•••
	94.4	**.1						100.0											100
								167.9											
1.5		*1.5	***					141.											**
	66.7	44.6	94.4			**.*		* **.1											
						**.			*5.4										:::
	6v.				40.5	*1.*													:::
	75.0	70.2									****								
	4.0		v.0		6.6	4.0	0.0	0.0	0.0	0.0	0.0								•
										-									

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

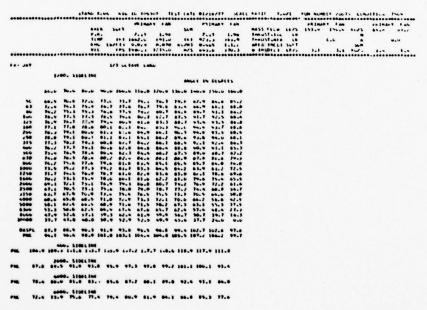
							De 1 40				-				**1*			** *** * *
				44		1406			*				MASS +	11/1	100.0	151.0		67.0
							4.00	1.45			.00	4. 93	14051,10					
				11			47.0					.70.4	1me 151 .m			1.0	•	6.0
							0.031	0.07				1.136						
		••••	•••••				677.0	1141.5		3 37		****	* INCL!!	16/1	1.1	1.1		
						,, w.		-										
			wo. 51	M (14)								***						
	1		70.0		**.	100.0	110.0	110.0		140.0	130.0	100.0						
				16.7				15.0	10.									
		14.2		11.1		70.0		10.6				.1.7						

10			75.0	77.2	70.2				07.0									
15			10.0	16.6														
		12.0	14.0	70.4	10.				**.*		****							
20		10.7	17.4	14.2														
*			70.0	70.4					47.4	97.6	**.							
**			10.0	74.1	77.0			***		***	90.1							
-			73.4	11.1	14.4				***		**	41.7						
•			72.3	17.4						67.4		74.1						
			74.4	70.5	70.5						02.1			*				
100		72.2	74.0	70.0								71.7						
123			73.2	75.0	77.4						77.1	44.2						
100			11.0	74.6						70.1	73.9							
500		.7.0	Pu. 7	73.5	77.0					75.0								
530		45.7		71.0	74.4				74.0	73.0	47.3							
400		10.6	***	**	21.0				17.4	**.*	***							
100		34.3			47.7				**.1	41.5		11.1						
430				w.,	67.4													
				14.7					11.0									
1000		35.6																
							****			161.7	101.4							
**								101.7										
_																		
•	165.4	167.	• 11.	. 1 . 2 .	> 114	110	110	117.	.1 117.	. 110	.5 110	••						
			w. 51	. 191				.,										
							_					-						
	24.2			Of t Int				. 1										
_					•													
			. 11	DE L 140														
	74.7	72.	. 74.	. h.	. 70			02.		2 03	74							

a.) Model Data Measured At 15 Ft. Radius

							INMI				-			SE IMPEA		POIRARY	
				-					-			MASS FLOW			 •		0.0
				P.R.			1)	1.00		2.11		IMRUST . I DL	LB		 •		
				11.00		1 1003		101.6		.517.1		maust.44			 *	•	
				AND		1 1980		15.0		0.401		# (MOU!)		4.1		1.4	4.0
••••	*****		*****									***************************************			 		•••••
					1/3	CL TAVE	-	-	411	101 M	ATA 15.	ert RADIUS		-	 	- INLO	
-											460645						-
	-410							116	140	156	100						41-4
~	••	10		••	14	110	126			. >0							••••
***					6.0		6.0	6.0	•.•	6.0	0.0						:
00)	•.0	0.0	0.0	0.0			0.0	0.0	6.0	6.6	6.0						÷
•••	•.•	43.3	47.0	0.0	47.4	44.3	w	*1.4			101. 4						115
140	67.0	47.1			70.3					166.7							
		90.7			***		91.0			163.2							
200	.7.4		w. 2		92.1					107.2							1.
~	80.>			*2.4			90.1			100.3							141
315		94.4				**.*	**.	160.0	160.1	112.6	110.6						101
-										113.5							
*					**.*	101.0	103.5	108.0	110-4	110.5	117.0						
4 10		**.2			101-1	102.0	100.0	111	117.1	120.1	114.6						:5
-	****	***	**.	100.0	10.4	163.7	160.0	117.3	116.0	121-1	110.0						12
		.*	100.2	101.0	163.7	100.1	100.1	***		171.4	170.1						12
	101.1	101.4		1w	10			****	110.1	144.1	110.0						
		101.3	101.0		14- 7	1-4.4	14-		114.7	171.0	114.4						3 95
		100-6	141 1	14.2.1	144.0	147.3	14.0. 1	141.1	117.1	110.0	1.0						. 3
			100.0	14.5	100.0	107.4	100.0	112.3	1120	111.9	112.7						
		**. 1	14.0	10/-0	140.0	1-1.0	100.5	11/	114.5	110.1	113.7						
			100.5	107.3	104.7	400.9	100.3	111.0	41 3.2	114.4	1.1.3						
	***		100.0	. 104	105.0	101.1	100.4	1.0.1	117.1	114.5	100.1						
	***		:00.7	102.1	14.0	107.7	140.0	104.1	111.0	110.6	101.0						
10.0		**.*	100.4	107.3	100.0	100.7	104.1	100.	104.1	.04.2	103.0						14
1	***	40.0	100.0	Iw.e	104.0	104.7	101.1	101.	100.0	107.0	104.1						
10.0	***		**.*	101.	144.1	100.0	100.	107.1	101.4	100.0	102.						
	***									10							
***										101.2							
11.5	₩.	*1.7								**.							
30.0	****									**							
. 1.6						**.1		****			40.7						
11.0	10.3							****			***						
NE PORT		-	-				-										
		. wante &			417.0				1								

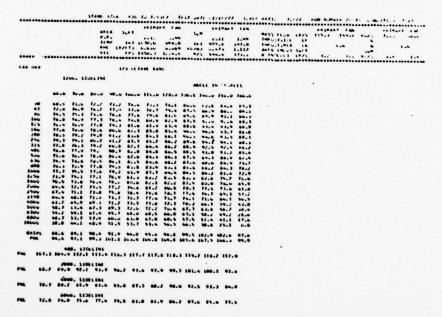
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

		•••••	•••••			1 103	.11	1.00	100			MASS #4	DM		0.0		2012		0.0
MD		****		110		I LON	0.0												
MD		****	•••••	440	LOVE				200	2.4			IOL						
MD		****	••••					4.00.4		· .									
MD		*****	•••••			1 103				6.47			001	2001			-		
MD		-0.500				3 14×		224.4		304.	• 171.		.,	10/3	1.7	1.1	44/3	1.4	1.4
49.00						***							•••				****		****
49.00					1/3	OL TANK	-	-			-	5.047 Reptus							
					•••				•••						(0	MICIE	0 301	- 1846	
1							MIL	-	-		DICALLS								
	-	70	80	••	160	110	120	1.4	144	100	144								-
										-5355	1000								44 -1.
***			0.0	0.0	6.0	0.0	0.0			0.6	0.0								
16 3		•.•	•.0	0.0	•.•														::
000	•.•			0.0		6.6	4.6												::
	1.50						w. 1				103.0								115
	***		44.4			*1.7			70.1	166.9	104.7								. 7.
						**.*	₩.			16 5	100.7								
			91.0	*1.4		**.*		*7.5	103.2	167.3	110.1								1
			****		****	**.*	**.2	**.	103.5	100.6	111.0								140
			*1.4	7.1	****		****	107.1	108.0	114.6	110.1								./1
			.7.0	****	****	100.5	107.1	107.1	115.1	115.4	110								
	••••		-	**.			103.7	100.0	117-1	110.5	117.1								1 14 .
		.		100.7			104.6	***		170.7	114.5								1.4.
		100.1	144.	141.5	101.6	144.5				171.0	110-1								
		164	161.0	107.7	145.2	144.4	100.0		114.4	121.0	170.5								1 14.
	07.4	101.1	101.4	104.0	100.0	107.4	104-1												4 34 .
		luv. T	101.2	107.6	14.7	Ive-8	164.4	111. 1											1 m.
. ••	**.*	160.4	101.1	107.0	14.9	147.1	109.8	111.6											
.15	**	**. 7	100.0	164.5	104.4	107-1	104.4	114.7											4 10.
·u	**.7	**	101.0	102.5	141.4	167.4	114-4	1: 2.1											
	**.	.,	160.7	107.7	104.7	101.3	109.1	141-4											1 44
. 20	***	**. 1	100.4	4. 104	10>-4	107.5	104-1	110.0	11.4.4		1								
		**.*	100.0	107.5	103.1	101.5	100.1	110.0	111.5	111.4	100.0								1
		**.*	100.0	107.5	105.1	101-5	100.3	100.1	110.1	101.4	404.2								1.5.
::	****	****	1.00.	10 1	100.0	107.1	101.0	100.4	100.0	100.4	164.5								1
	****	**.1	100.1	101.4	100.1	lue.e	104.4	161.5	161.7	100.4	14 1.:								1. 7.
			****	101.0	103.3	100.	10	105.4	104.1	162.4	14.0								
		****	****	100.0	104.0	100.0	100.0	100.1	1.4.6	10:.4	**. *								
						101.0	102.0	107.6	167.0	10:-1	*1								0
	-::							***	111.6		•								1
			.			**		****	**.*	**	****								.1.
			43.7		****		-11.1	*1.2											
								6.6											11

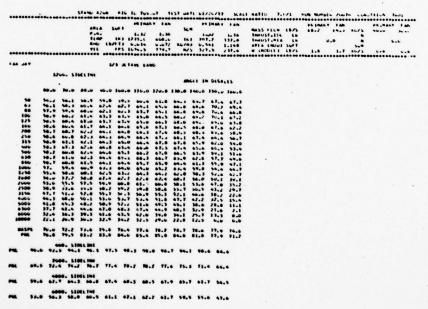
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

				-			I MARY							PRIMAR			PR IRANT	
					500				SUM			MASS FLOW		0.0	4.6	46/3	1.0	
				11.00				1.14		1.3		THAUST . 10						
				416	10/01			0.017	44/83			THEUST . ME			·.·	•		
				711		LIUM		179.1	N/S	247.5		AREA INCO				>		
*****			*****							22 1.1	2,7.6		10.2	1.0	1.7	44/3		0.
															•••••	*****		••••
					1/1	Well)	-	-		C134 C	ATA 13-	UFT RADIUS					- 18444	
440																		
							#1C	UPICIA	-	5 In 6	LGREES							
land I	••	lo	-	•	100	110	126		14	136	100							
.000		6.0	0.0	0.6	6.0		0.6		0.0	0.0								
.063	0.0	0.0	0.6	0.0	0.0	0.0	6.0											
.000			0.0		6.6					0.0	4.6							
123	•1.5	11.0	11.0	75.0	15.2	15.1	165	84.5	.1.0	65.7	40.5							141
100	74.1	10.4	14.1	70.1	75.4	11.0	11.0		****	07.0	**							
266	12.0			17.0	70.1	70.1	17.0	74. 0										10.
.01	73.7	14.1	74.4	***	P	14.0				****	*1.0							
319		11.0	A															
•40	11.0		41.0	1.64		67.2	****	***										
***						**.		****		****								
				14.7					**	*1.0								
	41.7			n.1		**.*			**.*		**. *							
				***	***	• 7.3			**.1		95.0							
				**				*/		****	30.							
							10.6		*1.1									
	04.7			04.1			w		47.0									
. 20					41.5			*1.*			40.3							
.15									***		45.1							
-									44.7									
.00																		***
	04.2																	
	84.4																	10.
0.0		04.5		1.00		.1.0		47.4			P							100
						07.5					11.4							1-1
4.6	Pv. 1				43.5						75.7							1.
0.0	77.4	19				63.6	17.6		41.7	10.0	72.0							44
	70.0	70.2	14.4						06.1	16.7								
11.5	14. 5	10.1	70.1	4.5					77.6	1								in
		1	70.3	m			74.4	76.1	15.7	14.5								
	14.2	11.0	11.7	17.0	11		10.0		77.0	14.7								**
			16.5	12.0	10.0	10.1	71.0				.7.0							
					h.1				44.5									*
w.					4.6					6.6	6.6							
																	LAPPL .	

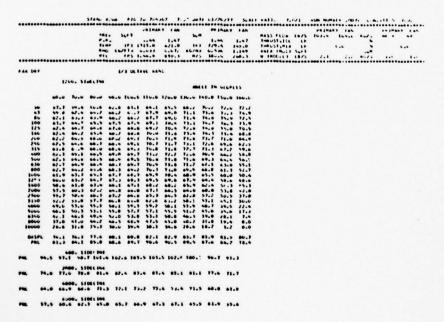
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

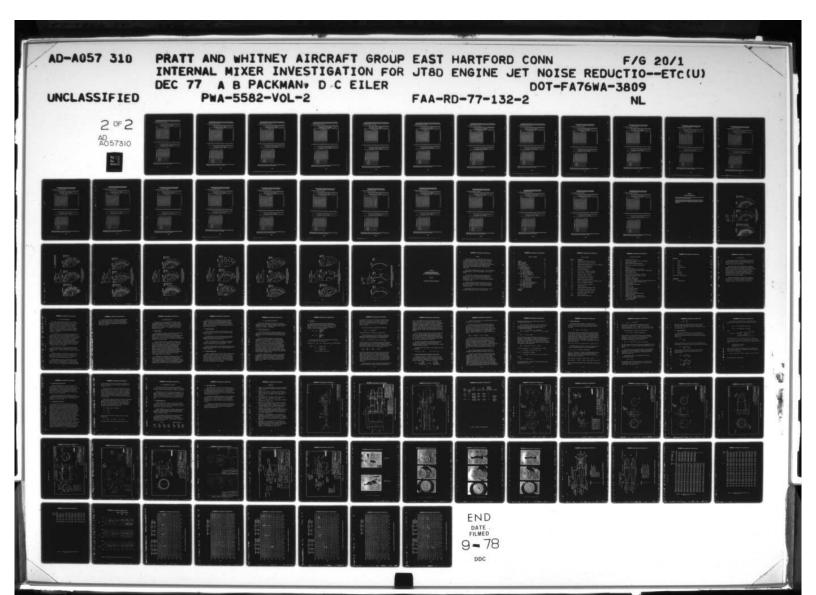


a.) Model Data Measured At 15 Ft. Radius

				-			IMARY		-					**!**		100000000	-	
						٠.,		1.47	200	1.4	1.47	THRUST-114		0.0	0.6	46/5		
				11.00		,;;		21.0		10.		Indust .nt						
												AREA INULI				>		•
				***		1 1700		80.5	W					2.4	7.7	44/3	1.0	4
****	****	•••••	*****	•	****		••••		*****	*****			*****	*******	*****	*****	*******	
												of 1 RADIUS						
440					.,,	W1.4		-			**** 1>.	Cri wanios		(******	0 500	- 19624	
	.at.						mic.	-	-	5 10	-t-REES							
• 140	••	70	••	~	-	110	150	1 .	100	120	1 ~							
-								0.0	4.4		6.4							
	0.0	0.0	0.0		0.0													
200	0.0							6.6										
luc	78.0	14.4	70.4	M-3	7	14.4	*1.0		45.7									1. :
.45	14.0	14.0	74.0	10.)		41.5			67.6									
100	10.1	14.6	70.1	P. 7	70.0				67.7									••
		16.0	14.4		•1.•	**			*1.7									
113		41.2		62.3					**.*									**
106		••••							****									***
						**.												- ::
						*1.0					403-1							
-			84.3								101.7							
				44.7	w.+	4/.1	**.*	+1.1	**.	161.3	100.1							
.25						47.4					***							
				₩.		46.4												
						*1												
. >0		.1.6		·).•		7					-1.7							
				*1.0					•5.•									
				.1.1		****												••
.00				•		**:					07.4							**
		*1.5		w		****			****									::
			***			****			*4.5									- ;;
					****						99.7							::
						****			****									- ;;
								****										11
								**										
1.5		.1.1																
			41.3															
		70.0							14.0									
1.6		1,.0	15.5			,			75.4									•
				73.1							0.0							
			*						0.0		•••							

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance





a.) Model Data Measured At 15 Ft. Radius

						-	-	-		Pal	-	-			PRIMARY	-		-	
				444	341				340				MASS -: CH	10/5	4.6	4.6	46/1		
							•	1.01		2.0		41	THEUS .IM	40					
				110		1 10			tas	744.	1 1%		THEUST , MIA						
				-	44.41				16/M)	4.60	. 1.1	***	MILA INCCI	5441			244		
				***		10%		190.0		0/5.			. (40041)	14/5			/5	1	1.5
•	*****	*****	******	******	******	*****		****	*****	*****	*****	****	**********	*****	*******	******	*****	********	****
					2 10/2013	-322	LOS ANA		9 5 5										
-					1/3 (CIANE	-	-		PCIN I	DATA	15.0	IT RADIUS			*****	r 201	- INC.44	
-11.							-												
1741		10																	
	•0		••	40	100		1.00	130	100	1 36	100								
40	0.0			0.0	v. e				0.0										
-																			٠.
20		0.0	0.0	0.0	0.0	0.0				0.0	0.0								•
100	73.1	70.4	PO. 1	80.4		b0.0	81.4												
125	10.6	.1.1		41.1			11.				** 1								
140	70.1	41.7	19.7	.1.1															1.4
/90	****	70.0	4.00						*3.2		94.7								
174	10.0		41.7			45.2					144.7								117
345		07.0	.1.4							141.3									
-00			80.7			41.6			100.5	104.6	167.								
500	60.1					41.0	45.0	****	161.6	165.4	105.5								1
630	.7.0			10.4			****	90.	102.0	144.4	107.2								10
-	84.7		.7.0		91.2				162.1										117
				+1.1			99.1	14.1	101.5	163.0	101.0								111
.2>		.1.0	84.0	****		**.1	.,		101.6										110
				*1.4					166.0										140
		10.7		91.7	94.7		90.1		**.1										
	87.6			45.4		w.,	*1.2	****	**.*										
		89.6		47.0	*3.3		97.7												
	••			4.4	*3.7		**.												
		80.6	44.4	45.0	*1.5			* *7.7											110
. 20		****		•2.0		*>. ;	**.												
.00		**.*	***	*2.4		***	**.				44.7								440
	.3		07.2	22.5	*3.3		**				40.0								11:
	83.5			****		**.:	**:			95.7	•7.4								
	*1.5			*1.2		77	77.				47.4								
3.0	****					*1.0				44.>									
1.5	80.5	0/.7		47.3															
0.0	10.6			47.3							74.4								***
0.0	14.4	10.1	40.1	04.3							12.0								16.
1.0	1/.0	14.7	17.1	79.1		61.2					74.0								***
6.6	44.3	74.7		74.4		11.5	11.												**
	0.0	v.0		0.0															*
and a		-		-		101010					***								
																		-	

b.) Mod 1 Data Scaled to Predict 17 3D Engine let Noice At 1200 Ft. Linear Distance

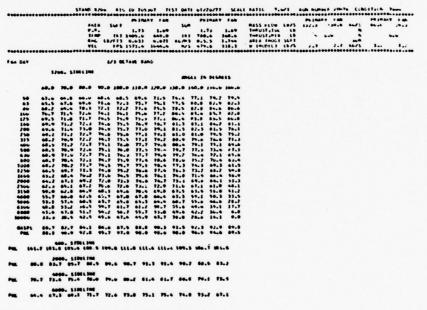
•••••				31400	1700	•16	16 705	30)	1657 00	11 01	//0/11	SLA	I MATTE 7.	41	-	H4 .W			1003
							** (**			**	Inati	f 44		****				P+ 134	
						1 404			14				MASS FLOW	10/		103.	44/5	./	·
							2.00	2.41			.04	2.41	THRUST . ICL		a recent				
					-		١. د مد	4 . 1 . 4		1 70		134.6	THAUST . MEA			6.0			
				::		***		4290.0	46/			1./**	AREA INCUI			2000			_
*****	****	••••					*****	4/4.0			•••	***	# (ACD) L 1		1.1	1.5	44/3	14	15
DA																			
				141.14															
		••								KL1 11	-	***							
								lows											
		**			44.1		45.7			71.7	71.0	74.4							
:		•1.1	• •	•>.5	.7.0		69.2	70.4	12.1	75.5	17.0	74.0							
10	~	•••	***	• • • • •	47.7			74.5	73.6	74.0	70. 5	74.7							
12						••••	70.0	72.1	74.7	76.6	70.0	76.5							
10				***			7	73.0	****	70.0	73.7								
20								73.5	73.0	76.0									
25	•		41.0	47.4	49.0	* > . 7		11.0											
				47. 1				145	74.4	77.0									
40	•	63			70.5	71.2	72.0	71.0	74.4	77.6	44. 4	40.0							
			44.1	41.7	PO.5	71.0	77.4	*1.0	73.5	71.2	44.3	10.7							
**			44.7	47.7	10.2	75.7	17.3	F3.5	72.0	10.0	.5.1	\$7.1							
66				66.6		70.6		72.7	71.T	44.6	65.6	14.6							
.00		•3.5	**.1	4.0			71.5	11.2	70.0	47.4	42.3	17.4							
1530		•1.7		44.2	4.0	47.0	71.7	71.5			41.6								
200		 :	41.7	4.0	.7.0	4.0	**.					47.7							
230											\$4.4								
313				37.4	****		47.1	43.7	***		53.1								
-00				****		****		30.0		33.3	40.6	32.3							
500		4.0		***			10.6	17.0	***	***	****	24.0							
410	ō	41.7	41.7	\$1.7	41.7	**-*	** .	52.0	***	41.5	34.8	13:4							
800					44.1	10.4	49.7	47.0	47.1	****	21.0								
1000		20.2	31.0	31.1	39.4	41.2	39.6	34.7		.0.7	3.4	3-6							
											-								
MAS	PL	75.0	77.7	79.0		92.3	03.6		85.5	95.7	83.9	91.9							
PH			85.6	.7.4	W	91.4	97.5	42.4	11.4	89.5		01.1							
*	1						2 100.	4 100.	4 102.	1 10.	2	, .							
				OE (1812															
•	79.6	10.	• •	.) 0).	1 14.	1 05	.1 65.		• •3.	· .	3 74								
		**		DE L 184															
•	**.			1 77.	. 73.	1 10	. D.	4 75.	3 73.	· · · ·									
				DE L 140															
	19.1	62.	1 63			3 60													

a.) Model Data Measured At 15 Ft. Radius

						*	-	-		-	-			PA IRARY				
				-					100			PASS FLOW	LO/S	6.6		46/3		
				P.A.		1.		1.40		1.71		neust. In						
				110		1000			**			THRUST . MLA						
				-	(MIII					6.541		MAIL (400)			27.76	544		
i i		anners en en		411		LIND.				479.6				1.1	4.1	**	1.4	1.4
*****	*****		••••••	•	*****	****	***	*****	****	•	*******	***********	***	**********	*****	*****	*********	****
												MI BADAUS					- 100441	
-					.,,											• ••		
	-						-14		-									-
1	••	70		**	100	1.9	120	114	14	134	100							44 -1

-									0.0		•.•							
			•.•								0.0							
-00	6.6	0.0	•.•	•.•	0.0	•.•					0.0							
	70.3		02.0	m.1		80.2					94.7							
152	80.2		M.1															
100						****					W1.0							
	64.4									142.4								4.1
*		84.7	****			4				104.7								
115		44.4	41.4	77.0		•1.•				144.6								
-	4.0	*1.7								117.0								100
		****	***	9.5						.13								123
-	****		****							414.1								17
	¥7.7		***	4.1						117.1								1
	•2.4		****	4.4						107.4								10
	*2.1									147.4								124
			94.2							145.2								100
		*1.7			¥9.1 1													124
	92.1	91.7		91.7		1.00.	161.6	160. 3	144-1	107.0	100.0							4/1
	92.7		41.2	47.5	90.0	100.7	101.4	100. 1	103.3	101.0	**.*							1.1
		*1.0	**.	97.4	90.0		167.0	141.5	167.7	160.0	70.7							100
	97.9	90.0	85.6	*1.7	99.0	100.9	107.1	103.4	102.3	100.0	****							
.00	91.7		**.1	97.4		100.9	104.0	103.4	104.1	144.4								4
	*1.3			41.1						140.2								1
7.5	.1.7									**.>								
		44.1	•1.5	m.7					100.1									4. 4
•.•	84.7			w					****									***
	67.4	90.2	*1.*	••.7							**.*							
1.5	87.0	***	~	*)	95.7	••.1												
***	43.7		30.0	*1.3	*3.4	**.2	***				****							111
•.•	•1		• • • • •	M - 2		:::	*1.0											***
2.0	****		74.4	m.1			**.											
	73.4	10.1	7		***	• 3	****				10.7							***
•••	4.6	•••			•.•			•••										•

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

						~	-	,		PR 1/	-	**			PE 1848 T				
				-					Son					10/1			**		
								1.10		1.1		74	THEUST. IDE	4.0					
				-		1 100			141				1						*
					10/1			0.471	86/43								-		
				44.		10>		om.e	~1				# (P@4()		1.6	7.1	4/1		1.4
					W1	RION	-	-	41 0		-	11.00	1 PAP LUS		-			- 19464	
•																			
		-	200000	-	SBU	12500					-	•							Pu
41	*	70	••	**		110	1/4	-	100	140	100								4 - 6
	6.6	6.6	4.6																
									0.0										
	20.1	41.7		**.*	M.7	45.7	87.6												.,
•	.1.0		**.*																
•						45.4					m1.2								
•	• 1. 1		***	61.2			*1.2		**.*										•
•		61.1																	
•	****	87.0	****				••••		141.2										4.
•		**.*		*1.4	**.*			101.0											
:	***		9/.1					w											••
								405. 3											
:	***	**.1			*1.1														
	****	=:			**.*														
:	*1.7		****		**.*														
-			**-1		**.*														::
		94.7			**.7														::
7			91.0		****														
		***			**.7														17
-					99.7														
~					**.*														- ::
					100.0														i.
	4.10																		
. 5	94.0	94.0	95.5	97.3	**		142.0	1-1-0	102.1	W1.3	**. 1								
-		*1.7						107.7											i.
				97.2				14.3	****	***	*1.1								
		.1.1																	-
			41.3	** . *		41.2													• •
		67.4		1	w.1														
	4.1	**.		67.3															
••	10.4	41.7					bv. 1												٠.
.•	10.3	**		To at															
••																			

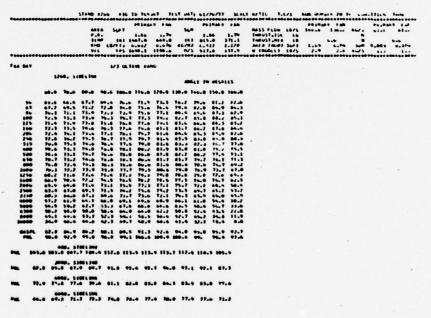
b.) Model D. ta Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

						PR I MAR		•		-							Pe	** **
			7. 11	•.	 	٠,١٠	3,70 001.0				1.74	10051,114 10051,114 10051,114 10051	::	115.1	***	275	-1.0	•
			**		175 10		1074.0	•	1 44		2.1			4.4	7.1			:
						-												
	12	w. 11																
		N.0		**	100.0	****	176.6	130.0	106.6	114.0	140.0							
10	44.1		44.0			A.,	74.1	**.*	70.0	86.0	*							
• >			10.5	77.7	1).7	10.5	13.0	70.7										
200	71.0	7/.4	73.0	72.5	70.0	70.4	70.4	Pa. ,		87.0	***							
123	N.4	12.0	12.0	14.4	75.0		70.0	91.3	****									
100	71.1	72.5	10.4	11.0	10.0	10.0	**.*	07.5										
100	11.0	14.0	14.2	17.0	11.0	70.4	80.1	44.7			***							
110	N. 1	17.7	14.0	15.0	10.1	70.0	00.7	04.7	04.5		**.0							
315		14.0	***	19.1	70.0	10.0	84.7	4.6		10. 4	14.0							
***		**.*	73.0	13.0	**.1	m				10.0	77.4							
100	**.5	70.0	75.5	n.,	10.0	70.0		00.00	10.0	79-3								
• 10	44.1	71.0	73.4	77.4	77.7	10.0	10.0	P. 4	17.0	71.1	14.1							
1000	**.	71.7	****	Po. 0	N	70.1	19.4	10.5	76.7	11.7								
1/16	61.4	Po. 6	14.4	N. 5	14.1	***	70.4	**.*	M	Page 4								
1000			****	**.*	Pa	10.7	77.1	74.0	14.0	40.1								
2000		44.1	10.7	12.4	M		72.0	14.1	P0.0		11.0							
1700			44.2	10.0	11.4	Po.2	14.0	**.*	44.7	41.0	14.0							
1110		41.4			P.,	71.0	-	44.7	***	**.*								
1000	****	14.1	****	***	4.0		**.*	***	4.7	17.1	M.1							
	44.4	m.1	37.3	***		***	***	41.7		43.0	77.7							
***			11.1			17.0			***	37.3	4.4							
	14.4	P.0	43.4	44.4		47.0	41.1	m. 1	10.7	10.0								
-									45.4	***								
-	**.*	*1.*	*1.6	9.7		**.1	.,,			***								
101	.1 100-	. 100.	. 100.	4 110.	. 117			. 1 110.	1 100	4 101	• •							
			-															
	.7 04.	• ••.	,		• •	• •	• •	• •		3 03								
. 79	, , ,		-															
- "		-		-				• • • • • • • • • • • • • • • • • • • •	-									
			-															

a.) Model Data Measured At 15 Ft. Radius

						-	-	-		PR 10	-	-				***			-
				4884					100				8415 FLOW	10/1			46/5	4.0	
						1.		1.7		1.0		. 20	THEUS! . 104						-
				110	101	1 1007			161	011.		1.1	THEUST, MEA						
					40711			6.0 P	E6/MI				4814 1400 F				100		
				***		100		100.0		\$17.0			# (MCD44)	10/3	2.0	2.0			
*****	****	*****	*****	•	****	****	****	****	****	*****	*****			••••		****	*****	*******	
					1/1		-	-				15.0	- 1 040145		-			- 1944	
***							-												••
		-		2.				ACPHAN			H441 11								-
W 1	••	10		**	100	110	D.	130	:40	110									84 - 8
990					0.0														
41							•.												
400	0.0		•.•		6.4						•.•								
100	.,	07.7			03.4						100.2								
113			**.*				.7.				141.5								
140	85.1	47.5					• 1.				103.0								414
700		84.7		67.6					****										661
110	***								101.1										10
***					*1.*				104.2										171
•••	**.*				****				100.0										1.0
-	***	*/.*	99.3	2:1					110.7										200
=		9.0	9.3	***					117.6										
=		77.4		***					*****										1:3
.75	****				99.7														1.5
-	***	***		W.7	***														***
-		***			****														
.30		99.0			100.2														1.0
.15	94.4				14.1														***
-66					16.0														
					144.0														
			97.2		101.0														
-00	91.4	**.*	41.2		101.1		105.	103.1	100.1	100.0	163.3								
																			144
4.5					100.0														
	92.0				100.5														10
0.0	90.7		**.1		**.1			104.1	101.4	100.0	****								141
			m	m			1w.	100.1	99.1	90.0	90.1								**
1.5				41.0	*7.7	**.*			****	**.*	*1.0								
-				****	**.*				****										
		85.2			****														
	14.5						*1.0												
6.6	n	70.1		Po.0															1.1
w.		0.0		•.•	6.6		•.0	• • • •	•••	•.•	•.•								

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

100 mm m		**************************************		11 mm	1/3	1 121	11 1.0 I		AT I	1.00 0.30 331.0	1.11	MASS THEU THEY THEY		1007			•••••	- (max)	
100 mm m		n		11 mg	1/3	1 1971 1 0-4 1 1001 0C. PAVE	04AC	107:0 107:7 107:7	# # # # # # # # # # # # # # # # # # #	943.6 931.6 931.6	1.11	7 AALA 2 0 10	18001 6011	1007		·::·	4/1	::	.1::
100 mm m		n		94	1/3	0. PAVI	04AC	165.2 165.2 10001 L	# # # # # # # # # # # # # # # # # # #	531.0 531.0 5000000	1.1	7 414	CD4 []	10/1		·::·	4/1	::	.::
100 mm m		n		**	1/3	0. 1000 0. 14W	04AC	MOCE L	#/5	**********************	133		COLL)	10/5	••••••	*****		::	.::
100 mm m		n		**	1/3	GC 1444	944C	1004 L	AT I			••••••	•••••	19/5	••••••	*****		••••••	••••
100 mm m		n		**	1/3	1.0	-	-	A1 0					•••••		*****	•••••	••••••	••••
100 mm	**	::	::	•.•	100	1.0	-16	-	-		-		avs.		-			- (max	
100 mm	**	::	::	•.•	100	1.0	-16	-	-			3.00 T BAD	W.			PARCIE		- (mas	
100 mm	**	::	::	•.•	•.•														
2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	**	::	::	•.•	•.•														
**	1.0 1.1		::		•.•					120	144								*
**	1.0 1.1		::							•									14-1
	4.4	***																	
17	11.4				•.•		0.6		6.0	0.6									
129 0					6.0		•.•												
					84.5						101.4								
					44.2		**.1				107.0								
			**.2	67.6															
	1	**.2	• 7. 1	***	90.5	*1.5	•1.1			100.0									4.64
	M. !	***		**.	*1.5					107.0									
	1.2	**.		92.1	*1.1					14.2									
		****	***	W.4						114.5									
			~		2.5		101.0	103.0	*****	417.7	*****								
	3.4	*1.4	***		100.0	141.	1000		112.0	*****	*****								
	M.4				101.2														12
.20 9		90.4			141.7														1.7
	4.50	90.0			141.7														12
	1.01	90.7	96.0	100.2	101.7	103.0	100.0	147.0	111.0	111.7	110-6								10
	N.4			14.2	100.0	w.,	107.0	104.4	114.3	111.4	111.0								14
.15 .	W.5			100.0	162.2	104.6	167.4	104.1	169.2	110.5	111.2								4/1
	m.s	**.1		100.0	14.0	100.2	107.4	101.0	100.5	109.7	110.3								
					144.7														4.1
					165.6														101
	n.,				103.4														1.1
		*1.4	***	100.4	103.1	10>.0	100.1	167.0	167.1	166.6	100.7								
					IVA														
	1.1	****			104.5														
		***			161.4														100
		1.	w.,		****														
	7.1	87.4	77.0	~~		20.1				70.3									• **
			.		*				**.*										***
						•3.1	*1.1												11:
	77.4			W.4		***				47.7									111
			4.0																";

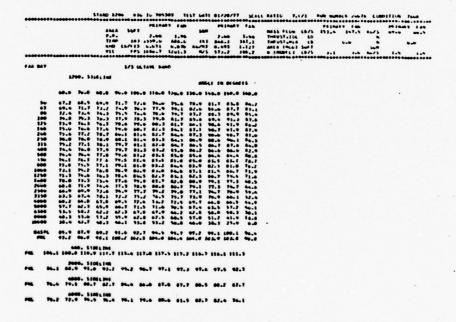
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

•••••	****		*****	******	*****	*****	****	-	-	****	*****	24	**********	*****	******	****			
							-					* ***			** 1~			P-184-7	
				P .					34				MASS FLLE	1017	149.0	1.1.1	46/1		***
				140		41 12		1.01			. **	1.00	THEOUST. S			200	•		
									467			1-147	AALA IRUS				. :	•••	•
				44.0				145.2		* **		***.7			3.4			1.0	1.1
*****	****	*****	*****		*****		-			*****	****	*****	**********		******				
						414													
-						~		-											
		6.2	w. 10	WE 198															
									-	-	- 44								
			10.0				110.0	40.0	110.0	100.0	134.0	100.0							
	•	:	11.4	72.3					****	-		****							
		71.4	17.0	11.2	14.4		***	70.7	****										
16				75.1	77.4	11.7	70.0		****										
17			79.4	****															
- 10																			
				17.5	20.4			140				65.1							
21	•			11.0															
		14.5	77.0	10.0	PO.4	10.7													
•				10.0	78.6	14.4	.50												
*		71.9	m.e	70.4	76.4	**.*				04.0		17.0							
•1		77.7	15.1	70.4	70.4							10.1							
				75.6	70.1	14.0			07.0		70.4	****							
100			10.1	77.6	11.1	P9. 1		4.10		P		71.0							
127					17.9	19.4				10.1	Pa. 1								
100				14.1		70.6			10.7	77.0	74.4								
204				****				79.3											
274					73.5	10.2	17.7	77.4	75.6	77.0		17.1							
911			4.2					14.7											
***	•							71.1				41.5							
100						**.*	m.1					11.5							
			10.0	40.7								11.1							
	•				M.4							1.00							
3000	•		4.3				**.*	**.1	••.)	30.0	12.4	•••							
941	~							-				-							
					H			M1.0	101.1	101.0	101.4								
				-															
-	104.6				110.0	119.0	110.	. 115.	. 114.										
		-																	
~				7 91.4															
-	-			OC 100			-												
-																			
		-		-															
~		11.		1 74		74-1													
-								-		-		••							

a.) Model Data Measured At 15 Ft. Radius

							13451	***		2011	w	•			P0 144			PRIMAR	
				4014	14	1		1100000	544	0				10/3	0.0	. 0.0		0.0	
							.00	1.94		4.0	1.1	. 1	UST.104						
				**		1 159			(81				UST						
				4140	INF		• • • •	4.0N	84/87	0.49	1.12	7 44		5041			244		
				447		\$ 100		4-1-1	W				OCDIL!		3.1	3.0	14:	1.0	
****	***	****		*****	****	*****	*****	*****	****	*****	******	*******	• •• •• ••	*****	******	•••••	******	*****	
-					1/1			-		WIN (-	5.00 T BA	DIOZ			LOSICE	to ser	- 144	
							-												-
-		~		**		110	120	10		110	140								41 -
										• • •	•••								
-						6.0				6.0	6.6								
*			0.0				•.0	0.0	0.0	0.0	0.0								
-	•.•		6.0																
100	84.4			87.4			w. (
123					****	*1.1				100.5									
100	D U			w.7						102.3									
***	84.7									100.0									.,
730				71.4	97.7					100.3									
115	.,7		*1.1							111.2									
**			**.*							649.4									
100	**.*	****								117.3									
**	*7.5	*1.2								110.0									
	**.*	**.7								111.5									
	***									110.0									
										110.0									
	***									110.2									
										110.0									•
::5										11									
	***									117.0									
=	10.4									117.2									1:
	97.0									117.6									
-										111.0									::
	90.1									111.1									
	4.0									110.1									.,
	93.1									101									
	*1.4	95.7								No 1									17
	.1.0	99.6								100.1									.,
1.5	94.2		77.4							101.1									ï.
		*1.0		*1.0						101.1									
	84.1									***	****								44
3.0								95.1	****										
	14.0					*1.4													
w.	0.0				•.•		•.0	• • •	0.0										
																		18	
																		LANDE	

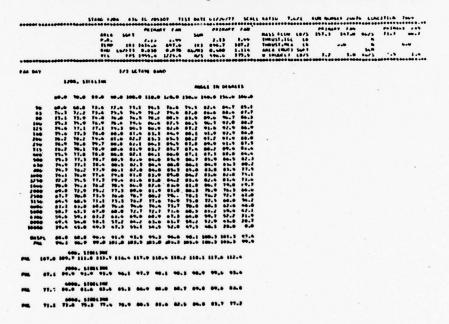
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

							-	-		***				PEIMAT	-		-	-
				-	14				100			MALA FLOW	48/1	0.0		4/1		
						1.		1.00		2-13	1.00	THE UST. IDE						
				TOP				47.4	141	800.7	307.2	THRUST . ME A						
				-		1 0.0		0.0 PO	44.70				-			100		
				44.		. 100		215.4		100.0		M INCOAL!		1.1		46/5	1.5	1.4
****	****		-	*****	****	*****	****	•••••	*****	*****		************	****	****	*****	****		••••
					1/3	C 14W	-	-	AT		414 15.	AUI BAR I TO		-	Ricke		- (84,000	
-																		
	***	70		**	100	110	176	130		130	100							4-1
							-		-	-								
100	0.0	:::	::	***	***	•.•	•			:::								•
-	0.0					*:	*				•.•							:
-	42.5	43.7	47.3		44.3	44.1	10.1				101.1							
75	84.7		47.9		91.1	*1.*	10.4			101.7								- 11
	40.0	91.0		07.2			*1.0			103.1								**
-	47.2				97.1	43.2	73.6			107.5								1.
*	80.1	89.2		42.2	*1.2	***	99.1		104.0	109.4	111.7							11
**	W	90.0		41.7	**.5	90.1	96.1	103.1	107.4	112.1	113.9							
-	91.0				90.3	100.0	104.1	100.4	111.0	115.0	110.5							
*	**.*	**.	**.2							110-1								
	**.*									114.0								
*	**.)									He-2								
	90.4									170.0								
·n										114.0								
**										110.0								1:
										110.0								
										113.7								11
										115.0								- 12
										114-3								
										114.1								
										113.0								
			1.0.7	102.0	143.3	41.1		140.1	114.0	112.0	110.0							
1.5										111.7								
	**.*									110.1								
•.•	90.4									100.0								
										144.0								.,
	41.7	****								103.3								
	**.	*2.								103.1								
	67.2	**	44.5		****	**.1			97.1	140.7	94.7							::
3.0	P	****	**.7								97.3							ii
···	7.5	****	****			7::					****							•••
	3.0					***	1	•	••••								2000	
																	(waf .	
									120-5									

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

				-			1	, ,,,,,		60 IV	-			PRIMARY			PR I RANT	
				***					140			MASS +LOW				**		
							.10	1.02		7.1		THOUST . I M				•		
				***		11 130		**1.0	(4)			THRUST ,ME			-6	•		
				714		13 6.		2 10.1	40							200		
												# 14(Dt L)	10/3	1.7	3.1	16/5	1.>	
							****								*****	*****		••••
						-	-	-	41	-	1414 A5-0						- (844)	
-									•••									•
							-	-	-									
(Long	*	76	-		100	110	126	1 10	100	150	100							84 -1
440			0.4				•.				٠.							
•		•.•	•.4				•.0				•.•							
-		4.6					•.											
-	07.7						••,				103.3							
10	****									101.5								
-	***									101.2								
POL										100.0								
700								****										
***	w.,							101.1										1.1
-	*7.*		**.1					106.4										
								107.5										4 .1
-								110.5										
=		**.;						1111.1										
75								117.5										1.1
								112.2										
=								112.6										
								111.0										1-
								111.>										1 1
								111.4										
=								116.7										**
:=								110.0										1 11
								116.										***
								149.7										12
								1 104.1										1.3
	94.1							100.4										
								100.0										
	*1.6	96.4						105.2										14
		94.1	97.0	97.1	14.0	163.4	14.	161.7	104.5	105.5	101.0							40
0.0	84.7							102.2										
	67.4		94.1	***	***	****	**.	100.0	100.2	101.1	**.*							14
	84.1		87.		99.4	***			97.1	90.3	94.9							
	84.7		03.							****	94.2							
									6.4		0.6							
	200	October 1			1000	LOSES AN	0.000			an about the	1000							
-				1111-4														

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

							141		44						P4 154			P-144	
						-				THE .			8855 PLG#	10/5	130.0	100.0	46/3	7:	
				•			7.1		•		7.14	1.07	THAUST . ICA						
				**			1300.			-		365.0	IMPUST, nt &			·	•		
				**			4.01			/A) ·		1.174	MAIN INCH				344		
					•	•••	100.	, NX	•	W))	٠.,	17.0	# (milet)	16/5	3.7		44/1	4.,	1.0
									*****					*****	******	******			******
	•					1/3 0	-	-											
			. 11																
										-	016								
			10.0	***	•-	- 100		He.	. 130.		150.								
			44.3		**		. 15												
:		PO		73.0	n.		. 76												
10		73.7	14.7		10.														
12		14.7	17.3		P.		: ::												
100		73.7	****																
20		P1.0	70.1	79.4															
24		76.0	70.0																
34		Pe.5	10.3																
-		P. 1	77.0	70.0															
50		75.0		70.7						7	7 00.								
		75.1	77.2							1 05.	D 01.	1 00.5							
		74.0	70.4																
141		74.3	74.2		74.														
100			1>.7		7							71.1							
700				75.4															
210		47.0		73.4															
3154		4.1	40.4		77.		. 11												
-		41.1	44.4	***	R.														
5000		30.0					. 12												
									s	2 59.	. 11.	1.14							
8000		44.4		50.5	**.				7 90.			21.4							
1000	•	w.,	44.9	**.3	•*•			. 52.	<i>.</i> 47.	7	10.	•.•							
-												****							
-												100							
			****	••••								-							
•	167.4	100.	• 111.		• 41	1	10.0	18.9 11	11	0.3 110	11	1.1							
		-		-															
	07.5						7.0												
	77.4						7.0												
-				-	_														

Severe Cutback Short Flowpath Mixer With Engine Secondary Flow Simulation; Configuration 10A; Condition 7801

a.) Model Data Measured At 15 Ft. Radius

							1	-			-			PRIMARY			PRIMAR	
									Sun		•.•	MASS FLOW	Le/S	0.0	0.0	4-45	0.0	
				7.4.			. 31	1.34		1.3								
				T (NO		1 1200		>47.0						0				
				Yes		1 1632		0.071	rews.							244		
	****	****	****		•••••	2 1034		773.5			235.6	# (MCD4L)	4/3	1.4		*6/3		1.4
												-off BAULUS						
-						~	•••	-		mist f		-0-1 840105				a ser	- 1944	
							-10	UPHON	-									-
-	*	~		**	100	110	110	1 30	146	130	100							14 -4-1
-				4.0	6.6	6.6		0.6		6.6								
663	0.0				0.0		•.•		0.0	6.6	8.0							••
.000		0.0		6.6					0.6		6.6							
100	41.1	71.4	71.5	74.6	74.7	74.0	76.0	70.5										
129		75.0	n	D.1	70.6	11.2	10.5				44.3							1
		13.2	10.)	Pa.0	74.0	73.1	70.0	70.4		\$7.2								Luci
200	71	74.1	79.7	70.0	77.0	70.0	06.7	87.6		*1.7								
230	11.0	10-5	70.0	70.1	70.7	77.0				**.5	****							lut.
115	77.1	70.0	10.7	P. 7	46.7		M-1		90.9	****								lev.
***		41.7	47.4	63.2	04.7						100.7							417.
>60				43.1	•3.•	01.1				**.3								
***		62.1			***		**.6	*1.5		40.2								
-00	61.7		82.5	m.1				*1.0		*7.3								
	67.5	W.0	41.2	4.3	84.0	47.4	**. 7	*2.0		****	**.							114.
	47.5	**:		en.)	47.7		49.7			*3.4	44.4							111.
==		04.3	83.4	M.	80.4					94.1								
	43.4	42.7	41.7		***		47.0			***	65.4							110.
. 13		.7.		85.0			49. >											110.
	84.3	04.0	83.9	m. 1	80.7		44.1											in.
			43.5	84.7	84.2	47.1				****	01.7							100.
. 30	84.1	02.3	41.4		80.6						79.0							sut.
.00	04.4	04.0		84.4			07.5				70.4							100.
	02.5		42.9							41.3	17.6							1-1.
7.5	07.4	.7.0				67.4			02.7	19.7	1242							130.
•.•			82.5		5.00	84.9			01.5	76.5	73.4							103.
v		74.2	80.4	41.0				:2.1	74.6	10.0	11.1							14.
>.0	79.7	10.0	19.9	01.2	02.5	02.5			70.0	74.4	10.1							400
1.5	77.4	70.5	78.5	P.0	.1.9			19.2		*1.5								
•.•	15.7	10.0	70.0	77.8	14.0	74.0	74.6	77. 4		11.6								Iw.
4.0	77.0	17.4	14.3	15.1	77.5	**.	77.1	75.9		70.6								١
2.0	70.3	64.5	712	74.3	73.0	***	74.1	74.0		44.5								
	***		47.4		70.5	76.4	10.4	72.2		.7.0								
				0.0	0.0	0.0	0.0		9.4									
	•.•		-					1000	-		10.00							
	•.•	•••									1044						CAPEL .	

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

*****			*****	****	*16	0 765	167 1		11 01/	20/77	SCA1	1 PATIO 7.	W1		910 rh		.1164	7-01
																	PR 184	
					•1								40/5	19.9	100.8	4-1	30./	**.,
											1. 30	THOUST . ILL				•		
												IMPUST, ALA						
																344		
	***		****		42 N		777.7	~~~	3 3/6	••	· >>.*	· IMODEL)	LOIS	1.0	1.1	***		1.0
				*****									•••••	******	••••••	*****	• • • • • • •	•••••
-				V		-												
		a. 11																
									41 14	-	40							
		N.0			.00.0	110.0	170.0	130.0	100.0	130.0	140.0							
			17.1															
	39.3	50.0																
		\$7.0	44.2					44.7										
	4.1	4.2	61.7	4.50	41.4		44.3	47.7	49.3	70.0								
	41.5					64.7	44.2	.7.0	49.3	49.7	4>-4							
								44.1										
											10.0							
										4.1	34.7							
										.1.7	31.3							
***					41.1													
	11.4	11.1		\$7.0	\$1.0													
	10.4	10.0	91.0	4.5	\$5.5	35.4	11.7	10.1	44.5	10.1	20.6							
		44.7			33.2			40.5	40.7	30.7	14.0							
								44.1	37.6	27.4	10.0							
					47.4			10,0	M.1	21.1	2.5							
									23.7	13.7								
-	25.7	21.0	31.2	N	33.0	32.5	20.5	24.7	171	-	•.•							
							** *											
			41.4							70.1	****							
-			••••			•	• • • •		****	****	****							
		. 11	00 L 190															
•1.	• •/.		,	97.1	• • • • • • • • • • • • • • • • • • • •				* **.		•							
200																		
12.	. 17.	1 M.	. n	17.6	11.	2 17.	1 M.	. 74.			,							
							The services	allo com										
•1.	• •1.			00.1				• ••-		s ss.	•							
		. 11	004 PM															
			. 10.0					7 50.	1 11.									
	100 000 000 000 000 000 000 000 000 000	50 50.0 50.0 50.0 50.0 50.0 50.0 50.0 5	120v. 13 120v. 13 10v. 10v. 10v. 10v. 10v. 10v. 10v. 10v.	1200. 1001 Dec 9.0. 1001 Dec 9	### 1200. 31001 Date #### 1200. 31001 Date ##### 1200. 31001 Date ##### 1200. 31001 Date ##### 1200. 31001 Date ###################################	### 1607 Pade Pade	### 1607 PRINT ### 1607 PRINT ### 1607 PRINT ### 1608 PRINT	### 150 151 15	### AND T 1.31 1.36 P. C. 1.31 1.36 THE 1.36 1.36 THE 1.36 1.36 THE 1.36 1.36 THE 1.	### API A SAFT PRIMARY PARK PRIM	### 100 PRINCE PRINCE PRINCE ### 101 ### 101 ### 101 ### 101 ### 101 ### 101 ### 101 ###	### 1007 1.31 1.36 1.36	### 1581 1.56 1.51 1.51 1.56 1.51	### AND	### 1847 1.51 1.56 1.51 1.51 1.56 1.51	### 100 Parker Pa	### AND 1.31 1.36 1.31 1.30 1.30	### 3467 1 3467 1 346 346 347 3 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1

Severe Cutback Short Flowpath Mixer With Engine Secondary Flow Simulation; Configuration 10A; Condition 7802

a.) Model Data Measured At 15 Ft. Radius

					4 10						MAR Y		*******			PRIMA			04 In		
				7.4					Sun				MALS FL	.00	10/5		0.4				:::
				140		41 12	.43	1.45		1.0		. 45	THRUST,	14		400	-				•.•
				RMO		73 0.		417.0	(A)			2.0	THRUST.		LD						
								0.072	REMI			193	AREA IM	140	1441			34.4		•.•	
***					*****		•••	845.2	W.S	374.	. 24	3.7	w tacot		10/5	2.0	2.4				
							*****		*****		****	****	# (#Ø!		*****	*****	*****	*****			
																					•••
-								-	311	BD176	DATA	15.0					Mette		- 14		
tates											and the second										
1 SHA	-	70		**	100	110	120	NO HOL	a Amer		-										-
			7.7			•••	170	1 30	146	150	100										-1
010	6.0			0.0																	-
		0.0								0.0											
100	44.1	14.4			77.0	70.4															- :
25	70.4	79.1			19.7																
	12.0	74.1			70.7	70.4															2
00	14.0	17.5	P	80.2	41.2																
10	77.4	14.4			42.5				.2.0												
115	86.1	01.7	42.2		80.3																111
66	60.4	65.6	66.5		47.4	44.9	VI.1		**.*		103.3										
**	87.4	85.4				49.7	****		****	M4.3	103.4										117
30	84.8	87.2	07.5			**.7	****	****	100.4	100.0	103.7										110
-		84.7				**.*	*3.3	**.	100.7	100.7	103.7										
	80.2				90.7	*1.0	****	****	106.4	14.5	163.4										14
23	.7.0	84.3	87.4		90.3	41.7	**.		**.5	167.4	107.3										14
		86.2			w.,		**.		70.5												.,
	47.4		87.4			92.0	**.*		97.5												
30	87.3		87.7	47.1		*7.7	-:		****	**.*											12
43	87.6	86.6	87.4			**.2	m. 1		****	**.3											
•	14.2	84.7	87.0	4.70	90.9	97.7	94.0			*7.0											
-		84.5			90.4	*1.7	*3.3		****	\$1.2											
30	84.3	84.5		0.40	W.7	11.0	12.0	****	*1.*	**.	84.4										
~					90.9	91.7	97.3		90.0												
	84.5			64.7	90.1	**.	*1.0			17.7											
	87.5			w.7	42.1		*1.4			**.1											1.
		84.0		47.4	49.1		90.1		67.4	****											1.
	m.5				47.9					11.4											84
				80.2	47.2	.1.5		00.7	44.4	91.0	11.1										w.
	87.1				00.9			84.7	02.7	79.7	73.9									1.	
		**.4		04.0	**.7	05.0	84.8		84.7	70.1	14.0										
	10.2	70.0	P.5		42.7		.2.5	81.1	70.0	74	75.4									44	ws.
	74.7	13.7	10.2	77.5	70.0	79.4	79.4	70.)	75.0	72.4	75.7										
	70.7	70.1	72.7	73.3	75.4	15.5	D.9	75.4	70.0	7,,,	74.5										
4.				0.6	0.0		6.0														
							,			3.0											

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

			(00	200		MIM				I MAN Y	****				47 144			********
					1			34					10/	*7.0	117.0		4.0	
				••		1.4	1.41				1.45	IMBUST-10		****	117.0		•	24.5
				-		20.0	417.4		11 710	7	14.6	THOUST . THE			6.0	- :		
				. 10	***		6.672		13 4.5	***	1-199	ME & INCL	1441			34.0		
****			**			130.1	****		15 374		203.7			1.0		44/3		
					•••••	•	•••••	*****	*****	****		# INDIL 1	*****	*****	*******	*****		
- 047					/3 CC		NO											
		. 1	104 L 194															
									-									
	40.0	70.6		~	100.0	116.6	170.0	10.0	144.6	134.6	146.6							
14	10.7				42.9	45.4	**.*	47.0	49,7	71.3	71.5							
**	41.4				44.7	47.1		71.1			74.0							
100	44.0				64.1		44.	71.5	75.2	70.0	72.0							
12>		**.3				40.0	70.>		77.1	77. 1	74.7							
140	45.4		45.3	• * * *		44.6	70.0	11.6	75.4	10.7	77.0							
700		2.7		47.4		44.7			74.0		70.0							
250			47.7		44.7	44.6	71.7		P4.0	72.4	44.1							
113			67.4	.7.0				17.0		76.3	43.4							
400			45.4		40.5	40.6	71.2		71.4		\$4.4							
104						40.7	71.2		10.,		>0.2							
	44.4	43.7	45.4		44.4	40.7	70.7		44.6		**.*							
					.1.7	4.5		**.*	67.3		33.7							
1000					47.4	47.4			43.1		10.0							
1/24					44.7	4.0	47.1		41.7	37.5	40.2							
1404	44.4	44.6	67.0	4.4			45.7		39.7		44.0							
2000	40.5			45.4	44.4	44.2		41.1	\$7.4		19.7							
2100	\$7.7	\$7.0		41.0	62.0	42.9		50.2	54.4	47.3	24.2							
3150				50.3		40.2		14.0		47.6	27.4							
4000		32.7			\$1.0	\$7.5	50.1	52.1	44.7	37.2	21.2							
1400		30.2		>>.3	\$7.0	>>.0	34.0	40.4	43.5		17.1							
4100		4.1		\$1.7	52.0	\$2.0	49.9	49.4	37.0	27.2								
8000		41.5		44.3	44. 5			P. 7	11.7	19.7	0.0							
	~	34.1	34.2	×	34.0	37.5	34.0	29.0	10.0	3.2								
-																		
-	43.9		??.?		**.1	4.1	***	23.7	17.1	84.1	20.7							
										•								
			6 LOO.	101.	. 101.	7 101.	. 100.	**.			•							
			-															
1 76	. 76.			. 02.	• • • • • • • • • • • • • • • • • • • •	, 11.	2 07.0		70.0	17.	•							
	-	. SI	ML PH															
	.2			71.	1 71.	7 17.	\$ 12.5	71-1										
											-							
		w. 11	DET THE															
				•				45-4		-	•							

a.) Model Data Measured At 15 Ft. Radius

							IMARY			PAI	MARY FAI							M 1447	
				441					344			MAS	5 +LL.	10/3			**/>		
							••	1.51		4.4			UST . ILL						
				11.00		1 1344		.31.0		7 33.0			USI .MEA		٠.				
				440				6.671	ac/m3				A IMULI				**		
				***		1 100		*10.0	4/5	>**.	177.0		414-11	467			14/3		
•••••	*****	******	******	******	******	•••••	*****	•••••	*****	•••••	*******	******	•••••	*****	*********	*****	*****	*******	
*					1/1	MINN	6440	MINE C	M1 .	WIN !	1	3.CF 4A	Lius				: : **		
	-	lu	~	46	100	110	126	130											
	-		-	**	100	***	176	1 30	140	1×	110								
-			0.6				6.6	0.0	1.0	0.0	0.0								
	0.6			0.0	0.0					0.0									
-	4.6						4.0				U.I.								
14	.5.3	15.7	10.0	74.0	70.9	19.1				**.7									
13	71.7			14.0		61.4	84.7				*1.7								
*	10.0		77.4	N.0	74.4	77.7	11.1			;									
00	14.1	10.0	80.4	41.3				.1.		*4.7									
10	10.7			47						****									
1>	41.4		43.7	1	85.0				**.5										:
4-0		.7.	.7.0			***	****	***	Itt.I	144.4	1.7.4								•
~		.7.1			64.2				142.5										
. 10	*1.5				41.1	4:.2			162.7										:
-00	.1.0				41.5	44.3			11.000										
			7	w.4	41.4	47			162.5										:
	44.5		84.0	w.2	v	*3.5	****	**.1	141.6	164.4	161.1								•
	19.4			w	*1.1	****			44.7										:
		87.4		w.,					**. 6										:
20	7		84.3	w.1	41.4	**.1			**.*										:
1>				w.7	42.4	**.1		47.6											
-	69.7		84.3	***	4	44.4	**.*	Vo	94.1	**.1									:
••				W.7	44.3	73.7	*:	40.1	14.4	41.0	****								:
×			14.6	***	47.3	*3.6			+1.0	*1.7	.1.1								·
u			.4.7	***		*3.7		94.7											- 1
		• * • •	40.7	*1.1		V													
••		17.0	**.L	*1.0	47.7	***		47.1		***									
				**.	40. 1	47.0		*1.*			7								
				•		40.6		**.*											
	41		***	•7.•	**.6	***	****				14.1								
	* ***		• • • •		**.7			**.*			17.0								
								45.5			70.0								
			***			**.	***			10.7	10.1								
٠.		17.1		14.0		11./					****								
		74.3	****			**	**		14.0	14.6	13.3								
••			6.0			***													

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Listance

			46					: 4			0.50	#455 FLEW	100					
				٠.		1.44	1.51				1.51	100 021.1:1						•••
			10	~			£31.4		1 71		* . 6	Tm. (51 . 4)				, .		
					+1,		4 .4 7	44/1			.144	A-14 14: (
			*1		*** 1	1.4.1	-10.0		> 14	.1 2	N						****	
*******	****	*****	*****	*****	*****	*****	•••••	*****	*****	*****	*****	*********		******	******			
4 044																		
				•			••											
	•••	u. sı	141 141						1									
			60.4															
									100.0	150.0	166.0							
**	34.3	***	• • • •	47.5	****	****		• • • •		73.6								
•	-::					**	70.4			71.6								
144								13.4	??-!	25.7	7.01							
123					67.6	10.0		74.4		70.7								
100			.7.1		70.1	*1.0		73.1	77.0									
200						71		13.7		75.00								
424		15.7			10.1	71.5		74.7	17.0	7/.0								
315		17.4		44.7			7 ,											
***			.1.7		14.1	71.0		****										
500			.7.1		10.1	11		1: .		**. 7								
						71.4	1											
-						10. 1		14.0										
Inc	64.4	4.1	65.0	.7.							30.5							
1250						40.4				20.4								
1300										35.0								
400-		.1.7			47.4	67.0			54. 5									
1746	>4.1	\$9.1	.1.>			45.0		.1.)	20.7	****	10.7							
1150	>0.2	34.7	>4.2	.1.0	.1.7													
****		50.1	37.4	w	57.1	34.4	101	34.1	40.7	30.5	21.6							
5000	50.7	>1.0	35.2	>0.4	50	>7.7	20.0	>1.3			10.0							
6 500			11.5	40.6	**	34.6	*1.0	47.5	10.9	79.7	4.5							
0.000	41.7		****		****		40.0		11.0	21.1								
looco	31.0	33.5	3	34.5	···	37.0	h	14.0										
CASPL	70.1	17.5	10.0	***					**.1	***	01.>							
-			•1.,	••••						. /	,							
			WL 191															
****			. 101.	03.			10/.	. 101.			•							
			141 141															
1 77.4			• 4.	2	, .,,					7 74.	•							
			4.14															
		• ••.	, ,,.	• 7	• "	. 74.	5 74.	, ,,	7 74.									

a.) Model Data Measured At 15 Ft. Radius

							I PART			PRI				PRIPARY				
				44.4	34				364			8435 FLUE		0.0	0.0	Au/:		
						1.	. 23	1.00		1.7	1.64		10	•••		Nu/ -	u	
				Itas		I Lave		44.0	141	777.								
				-				v.u17	46/43	4.57								
				***		\$ 1564	1	v4.7	M/S	470.	111.4		164	1.0		A./:	1.,	
	*****	*****	*****	******	*****	******	*****	•••••	*****	*****	*******	***********						: :
	40																	
480					1/3	CTAVE	BANK	MICH	#1 1	MC131 1	Ala 15	.ut 1 BAUMS					- 144	
AML I					20.0						144115							
ww.		76	**	**	In	110	14	130	1-	130	100							1
100																		
		4.0			6.0	6.0			6.6		6.0							
400	4.0		6.0	v	4.0	v.4					6.6							
100	0.0	0.0				·					4.6							
123	10.7			41.2			64.2				+7							
	10.0	64.5					1				****							i
			.,.,	**.*	65.3		6: . *		44.5	40.0	100.0							11.
. >6									47.6	10.1.4	103.0							
342		::::	****			** - *	****	**. 3	**.5	143.0	105.0							
44				10.7			* 5.0	*7.4	101.6	101.5	104.7							4.4
>46		****		44.7	*:	**.7	****	100.	10:.5	104.3	117.6							
634			****		****	****	****	104.0	104.1	11	111							
+44			****	:	****	**.	**	164.4	10.5.5	111.3	111.6							1. 1
-64				9.7	****	****	**	105.1	110.7	11.00	114.7							
.25		****		**.1	****	****	101.0	105.7	110.0	113.7	113.4							
	- 6	****		****		****	102.0	103.6	104.1	117	111.4							1.1
.00				****		****		10.3.3	100.0	F 14	104.7							
. >	****			**.;		****	101.	103.0	107.0	107.2	100.1							
-15		*1.0		.,		****				107.1	104.3							
.00				*>.1		****			104. 5	102.4								
	- 1			****				10,	107.0	101.6	46.4							
. 30			*	*: .*						****	40.4							1.1
			-1	••••	47.4				***		/							
		4.	**.1	*>.1			144.4	1.0.		40.4								
1.5	****	9441		**						40.4								11.
	7	*1.4	****	*>.1	44.4	97.4	***	***										
	*1.1	****			****													
5.6	•	**.*	4	*3.7	*:	****	**.	***			b7.,							
1.5				42.5	44.7	**.												411
4.6				w.1	****		*1.0		w(1. 4	87.0								
			14.3			*w.1	*1.4	****			****							
		7	+	**.7	. 1.4			67.3	***	45.4								416.
4.4	77. 3	77.4	74.0	+1.1					A 0									
w.							4											
								3.0										

b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

						PFIME			-						47 FA'.			
					111			34				PASS File			1.7.1			
				٠.		1.73	1.64		,	.73	1.07	Instist . Itt	11				20.00	
							4-0.0		1 77	1.2	6.000	Impil, MIA	10					
							1.612	16/6	. 6.	177	1-1-0							•
******			**				1047		5 47		111.6	w thities !	10 /	4.0		11/2	-	
	•••••	•••••	•••••	*****	*****	•••••	******	*****	*****	*****		**********	*****	******				
041																		
							MI.											
	1.	w. 11	JE 1 14															
								*			115							
		10.0	30.0	w.c	14.1		1	14.6										
		****	***		****	****	10.5	75.6	10.0	71.1	10.0							
		71	2		:"	***	75.7			•1.0	***							
100	73.0	71.0	7	1	1	17.0	13.7		•/.•	**.1								
125	11.7					75.0	17.1	-1.		• • • •	• • • • •							
100	71.5		14.1	14.2	15.1	74.0		*1.*	***									
2110	11.1		71.3		11.4	74 . 8	74.6	41.7		14.1								
336	11.1	74.9	17.7	14.0	17.5	11.0	14.0											
313	71.7	10.0	7:	71.5	19.2	70.0	70.7			70.0								
***	70.7		7	73.0	73.3	10.0	74.9	B9	74.5		71.4							
300	70.4	10.2	71.7	13.0	75.1		70.4	14.5	77.7	73.7	47.0							
6 tu	70.7	****	11.1			76.3		10.1	75.4	711.4								
100		**.*					77.3	10.0	70.0	44.7	41.1							
1250		**.1		17.1			10.7											
1000	.7.0				73.0		75. 1	74.1	74.5									
200				71.5	71.0	72	77.4											
1200	4>		30.7			70.0		46.1										
150								****			**.*							
400	30.3	59.7				44.4					37.4							
000	50.1	31.5				44.0												
300	31.4	31.3	\$7.7	50.1	.0. 4		10.4	44.7	47.7									
000	** . *	****	>7.1	30.0	24.7	33.2	30.0		.1	b4. 4.								
~~~	14.4	31	43.3		47.6	41.0	41.7	10.6	10.5	15.1	4.0							
			100															
asec	• • • •			7.	****		44.5			93.7	70.0							
PML			****	.F.,	*	••.•	*7.1	•7.1	***	***	**.7							
101.1			01.144	1 100.	7 100.	1 110.	1 100.											
		00. 51					,											
•	• • • • •	• • • • • • • • • • • • • • • • • • • •	• • • • • • • • • • • • • • • • • • • •	••••	• •••			****	•		•							
-	40	00. 11	DE L 194															
73-1	* ***	1 1.	0 70.1	70.	· 10.	6 80.	5 61.			1 14.								
		F																

a.) Model Data Measured At 15 Ft, Radius

																**   ** ** T	
				P.A.				1.15	200	1.00	1.75	MASS FLOW	0.0		AL/S		
				TIMP		1 1000		470.0	***			Intustina	4.		7		
				-				71	/83			Anta INCLI		••			•
				***	**	\$ 10?5	. 1			***		# IN'	4.7		/:		
••••	******	*****		******	*****	*****	****	*****	*****	*****		***********			*****		
190					1/,	CE TAVE		MULE !	**	M.174 C	DATA 13.4	PAULUS	DAD LL		. **		
1244	~	74		46	114	110		170	100		PICALLS						
			••	**		•••				.50	104						
vou						4.6					4.4						
			4.6			6.4											:
wu			6.6			w.f.				6.6	0.0						
100	70.7			+1.4						*1.*	****						
1/2	77.4		84.4	4	P6.0	*6.7	+>-6										
100				****	04.1	**.1				V#.1							
							M	***	**.1	104.1	104.4						
*			**				47.0	****	160.6	100.0	160.7						
-11		1		**.1		****	****	****	10	100.0	110.5						
***	****		93.3		****	****	***	Ic.	100.0	110.0	114-1						
36	**.		**.1	45.5		****	****	101.1	104.7	1	114.5						1.
		**.*						107.0	112 3	115.7	*****						••
		****		41.6		lute. 1		100.	111.1	115.4	117.1						
		93.7		47.0	**.	144.4	10	4.7.		.44.2							
		*>.7		41.1	44.7	Iuv.s	141.1	144.4	111	112.6							**
		****		***	41.7	1.0.0	163.4	144.7	164.1	114.1	lux. i						:
. >4		**.>	****	45.4	*1.1	101.6	81 Jes	146.	14 7.7	lut-7	14 7-1						
.15	*5.5	**.,	**.*	47.0	46.4	144.6	163.4	115.7	16e . 1	165.3	464.4						
		**		47.0	**.0	1.0.0	163. 1	115.6	164.0	101.0	10						
**		****		41.0	41.4	160.6	107.6		107.3	101.5	**						
. *		**		47.1	**.6	106.4		101.1	107.4	100.2	41.7						
		**		**.1	**.*		100.0	104.0	101.4	****	Vo. 4						
		*3.7		****	****	:00.0		101.5	100.4	****	45.3						
	****	****	95.0	17.0	****			*****	****	*1.5	****						
		-1								****							
		*1.0								****							
	47.7		*1.1			****		*>.			47.0						•
					***					**							::
1.5							***		** **								
						. 5.5	15.7	es. 1									
		0.6		4.4	4.4					1.1							٠.

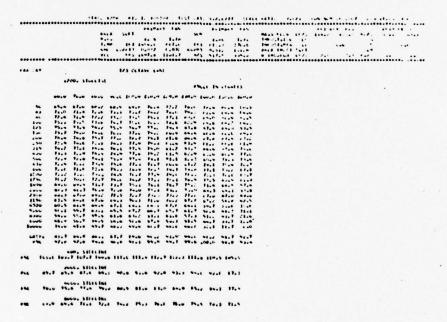
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

			310'e 3	/0¢ +	16 16 76	267 1	151 10	11 01/	///	: (4)	411. 7.0			 	
					***			•••••	1 ***		•••••	•••••	PR 1 14 1 1	 	
			404	4 34 5			34						133.0 135.0	44 . /	
					1	1.15		1.	<b></b>	1.1>	1 = 0:1,111		1,,,,,	 	
			1.0			A >t . t				41.0	Interitant &		1.0		
			-	10/11	2 0.077	0.074				-134	APER INCH				
			Vel		5 40 11	lunt.4	~						7. 1.	 	
	******	******	******	*****	*******	*****	*****			*****	**********	*****	************		
4 047					LA TAVE LA										
				•,,	W 1241 17	~ (									
	17	ne. :1	- t t 14t					41 14							
		N.0			r.o 11m.	4	130.0	1.0.		into to					
- 10									55.00						
•1	****	21.2	11.0	1	·· · · · · · · · · · · · · · · · · · ·	1	24								
• ;	21.1	****	****	***	1.1 10.			*1							
100	74.4		11.2					**.*		*1.7					
1/2				15.3 1			04.5								
teu					1.0 70.			67.							
***	15.0		75.0			Pu.7		***		17.0					
420					7.4 70		44.1								
113	17.4			.5.1 7		10.1				70.0					
400	74.3	71.0	77.1	75.0 1	7		41.4		70.4	74.7					
300	71.0	71.5	75.1	74.4 1	17.1	14		14.5		11.0					
410	71.0	71.3		74.0 7	17.0	74.5	74.7								
***					17.			15.0	11.1						
1000	71.6					11.7		73.4							
1500	44.5						15.7	1		50.4					
Bout							73.0	74. 1							
*000		47.4	•••			73.7		4.1		\$1.5					
2500		*>			1.5 17.0			es. 7		****					
4000		60.0			v.1 N.					40.2					
Meter			01.0				67.6	34.3		33.0					
-	33.4				··· 67.4					74.4					
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a.) Model Data Measured At 15 Ft. Radius

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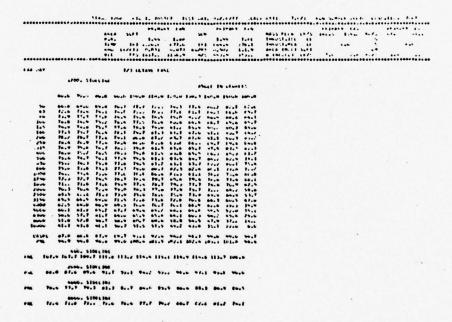
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

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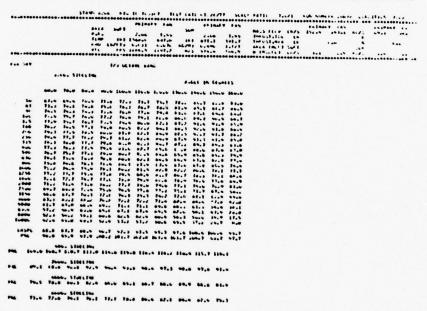
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



#### a.) Model Data Measured At 15 Ft. Radius

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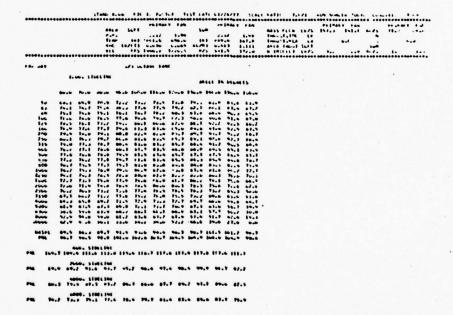
### b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

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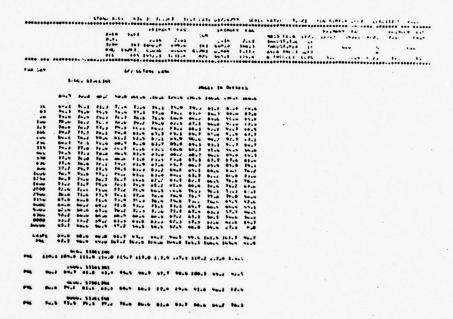
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

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b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

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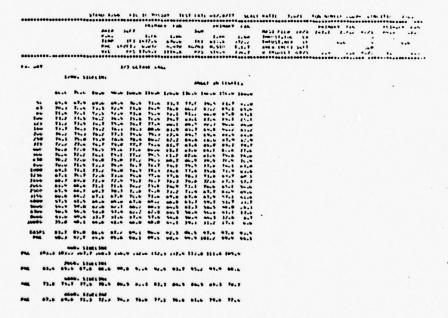
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

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a.) Model Data Measured At 15 Ft. Radius

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b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

### 100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   100   10
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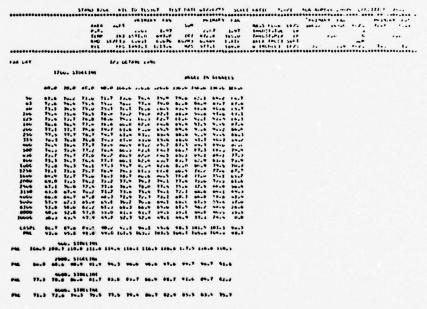
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

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a.) Model Data Measured At 15 Ft. Radius

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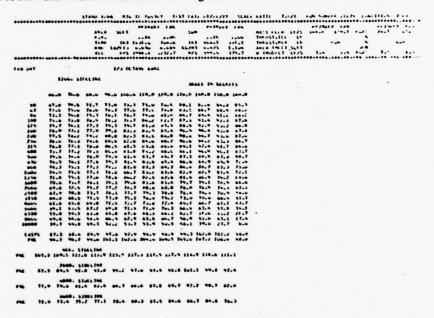
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

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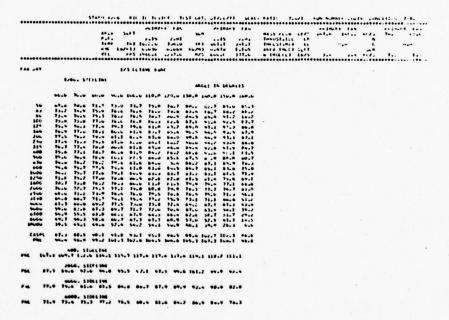
b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance



a.) Model Data Measured At 15 Ft. Radius

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b.) Model Data Scaled to Predict JT8D Engine Jet Noise At 1200 Ft. Linear Distance

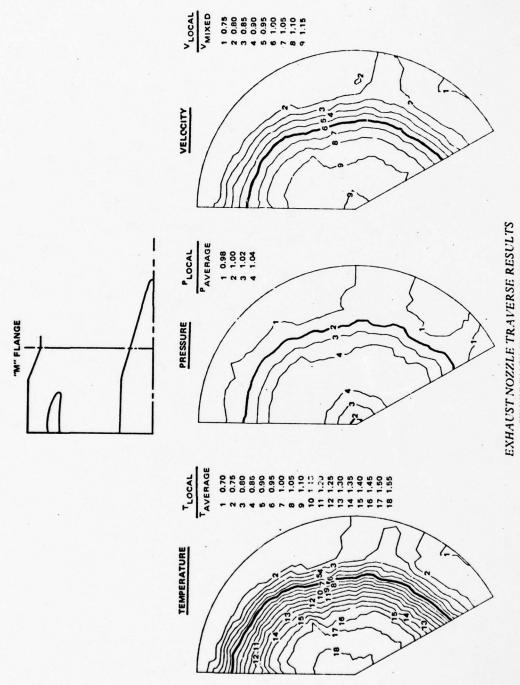


### APPENDIX C

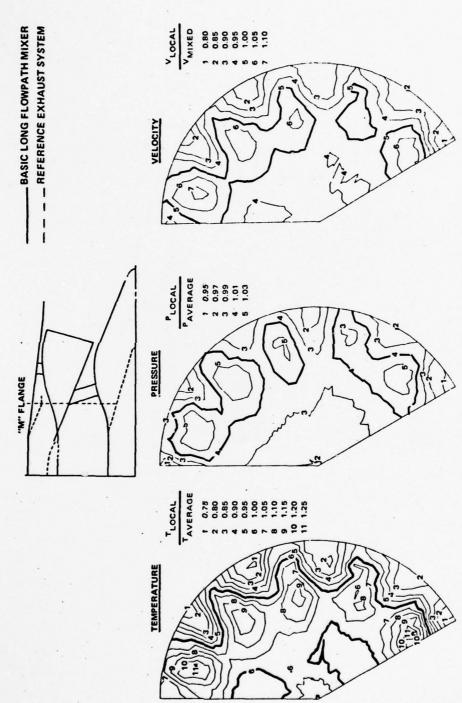
### **NOZZLE TRAVERSE RESULTS**

The figures contained in this section present the results of nozzle traverse testing conducted under the FAA contract as described in Section 3.

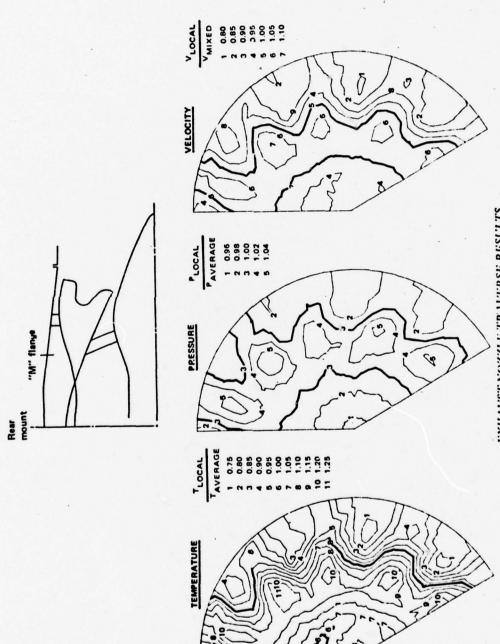
Contour plots of total temperature, total pressure and calculated fully expanded velocity are presented with contour line values normalized to average temperature or pressure and the calculated ideally mixed velocity. Test results are based on testing at takeoff nozzle pressure ratio.



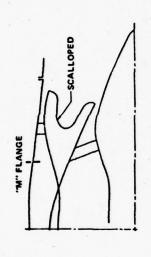
EXHAUST NOZZLE TRAVERSE RESULTS
REFERENCE EXHAUST SYSTEM
CONFIGURATION 244

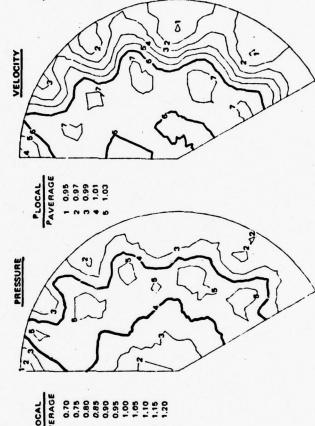


EXHAUST NOZZLE TRAVERSE RESULTS
BASIC LONG FLOWPATH MIXER
CONFIGURATION 1A



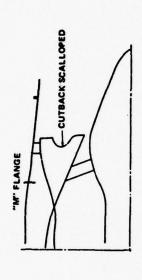
EXHAUST NOZZLE TRAUERSE RESULTS
SHALLOWED SCALLOPED LONG FLOWPATH MIXER
CONFIGURATION 3A

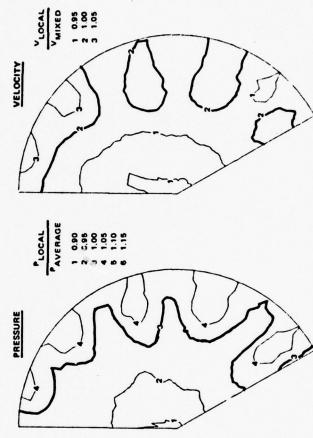




EXHAUST NOZZLE TRAVERSE RESULTS
SCALLOPED LONG FLOWPATH MIXER
CONFIGURATION 4A

TEMPERATURE



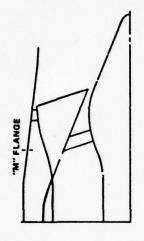


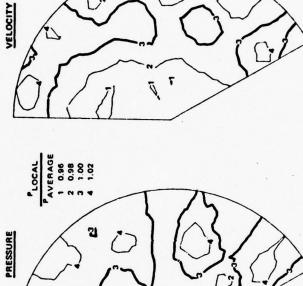
EXHAUST NOZZLE TRAVERSE RESULTS CUTBACK SCALLOPED LONG FLOWPATH MIXER CONFIGURATION 5A

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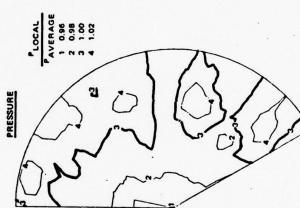
TEMPERATURE

1 0.96 2 0.98 1.00 1.02





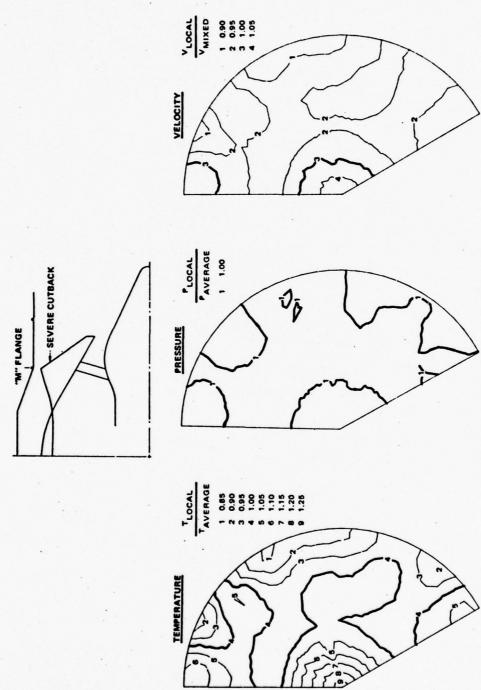
VLOCAL VMIXED 1 0.90 2 0.95 3 1.00 4 1.05



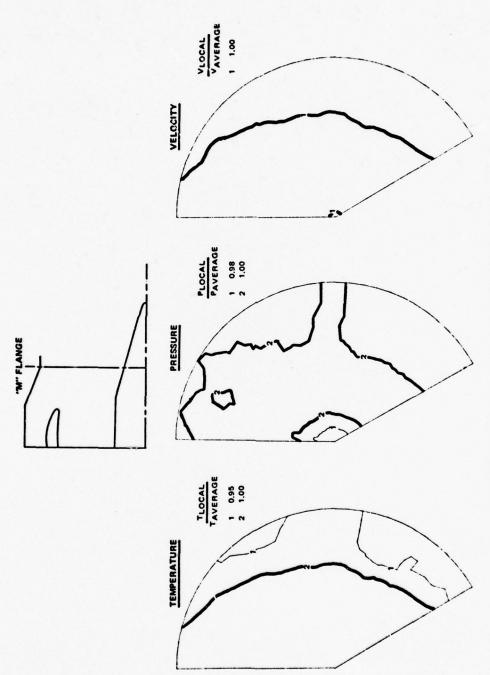
EXHAUST NOZZLE TRAVERSE RESULTS
SEVERE CUTBACK LONG FLOWPATH MIXER
CONFIGURATION 6A

TLOCAL AVERAGE

TEMPERATURE



SEVERE CUTBACK SHORT FLOWPATH MIXER WITH SECONDARY FLOW SIMULATION CONFIGURATION 10A EXHAUST NOZZLE TRAVERSE RESULTS



EXHAUST NOZZLE TRAVERSE RESULTS REFERENCE EXHAUST NOZZLE SYSTEM TESTED AT IDEALLY MIXED CONDITIONS CONFIGURATION 24m

### APPENDIX D

HOT/COLD FLOW MODEL TESTS TO DETERMINE STATIC PERFORMANCE OF 1/7-SCALE JT8D MIXER EXHAUST NOZZLES

Prepared by

FLUIDYNE ENGINEERING CORPORATION

#### SUMMARY

This report presents the results of hot/cold flow, 1/7-scale model tests conducted to determine static performance of two mixer exhaust nuzzles for a JT8D engine. The test program was conducted by FluiDyne Engineering Corporation for Pratt & Whitney Division of United Technologies Corporation. The model tests were performed in the Channel 11 static thrust stand at the FluiDyne Medicine Lake Aerodynamic Laboratory.

Three model configurations were tested. Static performance was determined for a free mixer (Reference) and two 12-lobe mixers.

Test conditions included core nozzle pressure ratios from  $\lambda_8$  = 1.6 to 3.2, and core-to-fan total temperature ratios from 1.0 to 2.41. Fan-to-core total pressure ratios were nominally .88, .93, and 1.00. The test program included 69 performance tests.

Facility checkout tests were made using two standard ASME long-radius metering nozzles. These tests demonstrated facility data accuracy at hot/cold flow conditions similar to the mixer tests.

Test results include static thrust coefficients, nozzle discharge coefficients, and effective throat areas.

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7	Run Schedule and Major Test Results
8	Test Results, ASME Checkout Tests
9a-c	Thrust coefficients
10a-c	Nozzle Discharge Coefficients

### DEFINITION OF SYMBOLS

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A	Cross-section area, in. ²
C	Real-gas A/A* correction factor, dimensionless
C	Balance readout, counts
$c_{D}$	Discharge coefficient, dimensionless
$\mathbf{c}_{\mathbf{T}}$	Static thrust coefficient, dimensionless
D	Diameter
F	Steam thrust, 1b
g	Acceleration of gravity, 32.174 ft/sec ²
G	Real-fas stream thrust correction factor, dimensionless
H	Thrust component, 1b
H ₂	Axial balance force, 1b
K	Critical weight flow parameter, °R1/2/sec
K	Balance force calibration factor, lb/count
L	Calibration load, 1b
M	Mach number, dimensionless
m	Mass flow rate, slugs/sec
P	Pressure, static unless otherwise specified by subscript, psia
ΔΡ	Static pressure difference across seal, psi
r	Radial distance, in
$R_{N}$	Reynolds number, dimensionless
T	Temperature, °R
<b>v</b> .	Velocity, ft/sec
V	Vertical balance force, lb
₩ .	Weight flow rate, lb/sec
y	Distance from wall
δ	Boundary layer thickness
Y	Ratio of specific heats, dimensionless
Δ	Incremental quantity
λ	Pressure ratio, P _t /P _a , dimensionless
θ	Meridian angle measured clockwise looking
	upstream, degrees
P	Density, slugs/ft ³

, 1-1-1-1/2/

## Subscripts

- a Ambient
- e Exit
- i Ideal
- r Resultant
- t Total conditions
- w Wall
- x Axial
- y Vertical
- ∞ Freestream
- 1,2,4,5,7,8,9 See Figure 6

### Superscripts

* Sonic condition

I THE THE LITTER LINE

#### 1.0 INTRODUCTION

The present test program was conducted to provide mixer nozzle performance data for a JT8D turbofan engine. The test model was a 1/7-scale simulation of the exhaust system flowpath, including two mixer nozzles.

The mixer nozzles were furnished by Pratt & Whitney.

Additional model components were designed and fabricated by
FluiDyne using contours and instrumentation locations specified
by P&W. The models were attached to existing model-tofacility adapters. The tests were conducted in FluiDyne's
two-temperature-flow static thrust stand (Channel 11).
Technical liaison for Pratt and Whitney was performed by Mr.
Jerrold Blatt and Mr. Maurice Bridge.

This report describes the test facility, test models, data acquisition and analysis procedures, and presents the test results. Test conditions and major test results are tabulated in Figure 7 and are plotted in Figures 8-10.

#### 2.0 FACILITY DESCRIPTION

The tests were performed in Channel 11 at FluiDyne's Medicine Lake Aerodynamic Laboratory. Channel 11 is a two-temperature-flow static thrust stand used to determine performance of exhaust nozzles in which the two exhaust flows are at different temperatures. Nozzle thrust is determined from force measurement with a strain gage force balance. The general arrangement of Channel 11 is shown in Figure 1. Photographs of test model installations are presented in Figure 5a.

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The airflows for both the cold and hot passages of a test nozzle are obtained from the facility high-pressure dry air storage system. Air for the cold passage is throttled, metered through a long-radius ASME nozzle, ducted to the cold passage of the test nozzle, and finally exhausted to atmosphere. Air for the hot passage is throttled, passed through a regenerative storage heater, mixed with unheated bypass flow to achieve a desired temperature, metered through a long-radius ASME nozzle, ducted to the hot passage of the test nozzle, and finally exhausted to atmosphere.

The air heater used for the hot flow contains alumina pebbles which are preheated to approximately 1250°F with a combustion heater. The heater capacity is nominally 40 lbs/sec at 1200°F.

The model assembly is supported by a strain gage force balance and is isolated from the facility piping by two elastic seals; see schematic in Figure 6. Calibration of the balance and seals is described in Section 4.7.

The ASME meter at Station 1 is water-cooled to protect the elastic seal from thermal effects. Since the cooling water is confined to the upstream (i.e., non-metric) hardware only, no tare forces are introduced by the water supply lines.

Facility instrumentation is provided to calculate mass flow rates at Stations 1 and 4, and to calculate the exit thrust produced by the test nozzle; details are described in Section 4.0. The data were recorded with Polaroid cameras and digital printers.

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#### 3.0 MODEL DESCRIPTION

The present tests used existing model adapters from the test programs reported in References 5, 6, 7, and 8, model components provided by P & W, and model components fabricated by FluiDyne for the present tests. Assembly drawings of the models are shown in Figure 2. Photographs of test hardware are shown in Figure 5.

#### 3.1 Model Adapters

The test models attached to common adapting hardware which supplied separately-metered flows to the fan and core nozzles. The fan air flow was nominally at ambient temperature for all tests. The core air flow was nominally at ambient temperature for the "cold" tests, but was heated to approximately 600-800°F for the "hot" tests.

The main support member for the adapters (Spider, 0937-902) had been rebuilt for the Reference 7 tests. Adapters for the core passage consisted of an insulated duct, a jet breaker, a choke plate, two screens, and a common core shroud adapter which supported a common splitter/mixer adapter. Charging station instrumentation in the core passage included three 10-tube area-weighted total pressure (Pt₈) rakes, five area-weighted total temperature (Tt₈) probes, one thermocouple for controlling the flow temperature, three static pressure taps on the inner wall, and four static pressure taps on the outer wall.

The adapters for the annular fan passage included a choke plate, two screens, a common bellmouth contraction, and a common fan cowl adapter. Charging station instrumentation in the fan passage included four 12-tube, area-weighted, Pt7 rakes; one 4-probe Tt7 thermocouple rake, one thermocouple for control purposes; and four static taps on the inner and outer walls.

_ 4 _

The single control thermocouples at the two charging stations were used to set the desired temperature ratio,  ${}^{T}t_{8}/{}^{T}t_{7}$ . Outputs from these thermocouples were amplified, divided, and displayed on a digital panel meter to provide the facility valve operators with a visual indication of the actual temperature ratio.

#### 3.2 Nozzle Components

Nozzle components included a reference splitter (free mixer), two 12-lobed mixers, three plugs, two fan duct cases, and one tailpipe.

The mixers were of the same basic design; one had deep scallops (Configuration 4A) and another had scallops and shortened mixing lobes (Configuration 5A).

Three test configurations were assembled, as summarized in Figure 3. Photographs of the test hardware, including partial assemblies, are shown in Figures 5a-d.

The only static pressure taps on the nozzle components were added for the thrust reverser tests; details are shown in the drawings (Figure 4).

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### 3.3 Standard ASME Nozzles

Facility demonstration and checkout tests were performed using two standard ASME long-radius flow metering nozzles (Figures 2a and 5a). The upper nozzle (cold flow) simulated the fan passage of the mixer nozzle, while the lower nozzle (hot flow) simulated the core passage. The two ASME nozzles were tested separately and simultaneously (only simultaneous test results are presented in this report).

#### 4.0 DATA ANALYSIS PROCEDURES

The following subsections describe the data analysis procedures used in the present test program. Station notations are defined in Figure 6 Computer programs written in BASIC language are included in the Appendix.

# 4.1 Pt7 and Pt8 Definitions

Pt7 was defined as the area-weighted average of four 12-probe rakes. The area-weighting method was adequate for the present tests since only non-distorted pressure profiles were involved.

COMMENT OF THE STATE OF THE STA

For all configurations except the reference model (Configuration 2A),  Pt_8  was defined as the area-weighted average of three 10-probe rakes. For the reference model tests,  Pt_8  was defined as 1.0075 times the mass-flow-derived  Pt_8 . This correlation was used because of disagreement between the three  Pt_8  rakes. The correlation was obtained by comparing the mass-flow-derived  Pt_8  to the area-weighted  Pt_8  on the configuration 2A tests. After the configuration 2A tests were completed, a second screen was added to the core passage upstream of the core charging station, resulting in good agreement between the three  Pt_8  rakes; the correlation was not used for subsequent tests.

The above paragraphs define  P t₇ and  P t₈ at the charging station rakes. These total pressures were then adjusted to a hypothetical charging station located 1.2 inches further downstream. The calculation procedure, specified by P&W, is detailed in the Appendix. The P_t corrections were on the order of 0.1%.

### 4.2 Flow Rates

The mass flow rates through the test nozzles were determined using choked ASME long-radius metering nozzles. The core nozzle flow rate was calculated at Station 1, (see Figure 6) and the fan nozzle flow rate was calculated at Station 4, using the following equations.

$$w_1 = w_8 = \frac{K_1 C_{D_1} A_1^{P_{t_1}}}{\sqrt{T_{t_1}}}$$

$$W_4 = W_7 = \frac{K_4 C_{D_4}^{A_4} P_{t_4}}{\sqrt{T_{t_4}}}$$

The critical flow factor, K, was calculated as a function of total pressure and total temperature.

$$K=0.52820 + aT_t + bT_t^2 + cT_t^3 + 0.186 \times 10^{-4} \times P_t \times e^{-.0067(T_t-5000)}$$

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where: 
$$a = 0.1654 \times 10^{-4}$$

$$b = -0.2119 \times 10^{-7}$$

$$c = 0.6008 \times 10^{-11}$$

 $\mathbf{T_t}$  is in  ${}^{\circ}\mathbf{R}$  and  $\mathbf{P_t}$  is in psia.

This equation was obtained by curve-fitting tabulated values in Reference 1; the curve-fit is accurate to within  $\pm$  0.03% for 0 < P_t < 30 atmospheres and 460 < T_t < 700°R, and is accurate to within  $\pm$  0.1% for 0 < P_t < 40 atmospheres and 460 < T_t < 1800°R.

 $^{\mathbf{C}}\mathbf{D_{4}}$  was calculated using a semi-empirical equation

$$C_{D_4} = 1 - 0.184 R_{N_4}^{-0.2}$$

and varied from 0.991 to 0.993 for the present tests.

^CD₁ was calculated from a similar equation, modified to account for a thermal boundary layer. This thermal boundary layer results from water-cooling of the Station 1 meter.

$$C_{D_1} = 1 - (0.184 R_{N_1}^{-0.2}) (1.574 - 0.574 T_{t_1/T_w})$$

The above equation was derived assuming constant static pressure in the boundary layer, a 1/7 power velocity profile, thermal boundary layer thickness equal to velocity boundary layer thickness, and a density distribution in the boundary layer defined by

$$\frac{\rho}{\rho_{\infty}} = \frac{T_{\infty}}{T_{W}} - (\frac{T_{\infty}}{T_{W}} - 1) (\frac{y}{\delta})^{1/7}$$

 $T_{\rm w}$ , the wall temperature at the nozzle throat, was estimated from heat-balance calculations of heat transfer from the air stream to the cooling water.  $T_{\rm w}$  values calculated for the present tests varied from 105° to 165°F.  $^{\rm R}{\rm N}_1$  was calculated using a mean temperature  $(T_{\rm w} + ^{\rm T}{\rm t}_1)/2$ .  $^{\rm C}{\rm D}_1$ , calculated using the above equation, varied from 0.991 to 0.997 for the present tests. Given sufficient wall-cooling,  $^{\rm C}{\rm D}_1$  may exceed unity (Reference 2).

The above equation for  $^{C}D_{1}$  is believed to be correct within  $\pm$  0.002, on the basis of results from facility demonstration

tests. These demonstration tests included test series with either a 2.5-inch or a 4-inch diameter ASME nozzle located downstream of the water-cooled Station 1 meter. The downstream nozzle was essentially at adiabatic conditions (thin-wall construction, backside insulated). Flow rates calculated at Station 1 (using the above  $^{\rm C}{\rm D}_1$  equation) agreed within  $\pm$  0.25% with flow rates calculated at the downstream nozzle (using adiabatic wall  $^{\rm C}{\rm D}_1$ ), thereby indicating the adequacy of the  $^{\rm C}{\rm D}_1$  equation.

 $A_4$ , the geometric throat area of the Station 4 meter, was 2.5475 in².  $A_1$ , the geometric throat area of the Station 1 meter, was calculated assuming thermal expansion from 70°F to  $T_{\rm w}$ . The largest value of  $A_1$  calculated for the present tests was 1.3490 in², representing a thermal expansion (area change) of 0.16% from the nominal area of 1.3468 in².

 P t₁ and  P t₄ were measured on Heise bourdon-tube gages.  T t₁ and  T t₄ were measured with shiedled chromel/alumc1 thermocouples and recorded on the facility Vidar system (analog to digital converter, printer).

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Calculated flow rates (lbm/sec) for the present tests were in the ranges

$$2.6 < W_1 < 10.2$$
  $2.2 < W_A < 9.1$ 

#### 4.3 Discharge Coefficients and Effective Throat Areas

Discharge coefficient is defined as the ratio of actual flow rate through a nozzle to the ideal isentropic flow rate at the overall nozzle pressure ratio. Overall nozzle pressure ratios are defined as  $\lambda_7 = {}^P t_7/P_a$  and  $\lambda_8 = {}^P t_8/P_a$ .

$$c_{D_7} = \frac{W_4 \sqrt{T_{t_7}}}{K_7 A_7 P_{t_7} (A^*/A)_7}$$
 and  $c_{D_8} = \frac{W_1 \sqrt{T_{t_8}}}{K_8 A_8 P_{t_8} (A^*/A)_8}$ 

 $^{\rm K_7}$  and  $^{\rm K_8}$  were evaluated using a previous equation, as functions of  $^{\rm P}{\rm t_7}, \ ^{\rm T}{\rm t_7}$  and  $^{\rm P}{\rm t_8}, \ ^{\rm T}{\rm t_8}.$ 

The throat area of the 3.75-inch ASME nozzle was  $A_7 = 11.0447 \text{ in}^2$ . The throat area of the 4.00-inch ASME nozzle, when measured at room temperature, was  $A_8 = 12.5664 \text{ in}^2$ . The actual throat area of the 4.00-inch nozzle, when tested with hot flow, was calculated assuming thermal expansion from 70°F to a recovery temperature,  $T_w$ .  $T_w$  was calculated assuming isentropic expansion and a recovery factor of 0.89, i.e.,

$$T_w/^T t_8 = (T_8/^T t_8) + 0.89 (1 - T_0/^T t_8)$$

 $A_8$  values calculated for the ASME tests ranged from 12.710 to 12.714 in². For the mixer nozzle tests, reference areas  $A_7$  and  $A_8$  were not defined, and therefore, fan and core nozzle discharge coefficients could not be calculated. However, the effective throat areas  $(^{\rm C}{\rm D}_7 A_7 + ^{\rm C}{\rm D}_8 A_8)$  were calculated. In addition, an overall nozzle discharge coefficient was calculated as  $^{\rm C}{\rm D}_9 = (^{\rm C}{\rm D}_7 A_7 + ^{\rm C}{\rm D}_8 A_8)/A_9$  where  $A_9$  is the exit area of the tailpipe (see Figure 6a) and equals 14.21 in².

Pt7 and Pt8 were measured on multiple-tube mercury manometers, and were defined as described in Section 4.1.

Tt₇ and Tt₈ were measured with shielded chromel/alumel thermocouples. However, since the thermocouple rakes were further upstream than the pressure rakes, the measured temperatures were adjusted to account for heat transfer (between the air and the model walls) occurring between the temperature and pressure rake stations. This adjustment was based on analytic estimates which indicated that Tt₈ should decrease 5.7° when the core and fan temperatures differed by 760°F. An energy balance required that the fan flow temperature would increase by approximately the same amount. The temperature

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adjustments calculated for the present tests ranged up to -6° for  ${}^{\rm T}{\rm t}_8$  and +6° for  ${}^{\rm T}{\rm t}_7$ .

 $A^*/A$ , the isentropic area ratio, is used to correct the ideal flow rate when the nozzle is unchoked.  $A^*/A$  for the fan nozzle was calculated using equations valid for  $\gamma = 1.4$ , obtained from Reference 3.

$$A^*/A = 3.86393\lambda^{-0.71429} \sqrt{1 - \lambda^{-0.28571}}$$
 for  $\lambda \le 1.8929$ 

and

$$A*/A = 1 \text{ for } \lambda \ge 1.8929$$

A*/A for the core nozzle was obtained by correcting the  $\gamma=1.4$  value for "real gas effects," to account for  $\gamma_8$  being significantly less than 1.4. The correction was derived by curve-fitting tabulated values from Reference 4; no corrections were indicated for  $T_t < 900\,^{\circ}R$ . First, the critical pressure ratio was expressed as a function of total temperature:

$$1/\lambda^* = 9.667 \times 10^{-6} \times T_t (^{\circ}R) + 0.5196$$

If  $\lambda \geq \lambda^{\star}$ , then  $A^{\star}/A$  = 1. If  $\lambda < \lambda^{\star}$  and 900 <  $T_{t}$  < 1260°R,

$$c = 1 + (\frac{1}{\lambda} - \frac{1}{\lambda})$$
 5.728 x 10⁻⁵ (T₊ - 900)

If  $\lambda$  <  $\lambda^*$  and 1260 < T_t < 1800°R,

$$c = 1 + (\frac{1}{\lambda} - \frac{1}{\lambda_*})$$
 [2.615 x 10⁻⁵ (T_t - 1260) + 0.020621]

Finally,

$$A^*/A = c \times (A^*/A)_{\gamma} = 1.4$$

For the present tests, c (denoted c* on computer output sheets) varied from 1.000 to 1.002.

#### 4.4 Thrust Measurement

The net static axial thrust of an exhaust nozzle is defined as the axial exit momentum of the exhaust flow, plus the excess of exit pressure over ambient pressure times the exit area.

$$H_x = mv_{e_x} + (P_e - P_a) A_{e_x}$$

The net static thrust of an exhaust nozzle model was determined in the present test program by applying the momentum equation to the control volume shown in Figure 6. The analysis of axial forces applied to the control volume includes entering stream thrusts ( $F_1$  and  $F_4$ ), a balance force ( $H_2$ ), various pressure-area terms and the axial exit stream thrust, ( $H_X + P_a A_e$ ). The axial balance force,  $H_2$ , as used here, included seal tare forces. Summing axial forces,

$$H_x = F_1 + F_4 + P_2 (A_2 - A_1) + P_5 (A_5 - A_4) - P_a (A_2 + A_5) - H_2$$

The stream thrust at Station 4 is the exit stream thrust of a choked long-radius ASME nozzle, and was calculated as:

$$F_4 = G_4 (1 + 1.4 C_{D_4} C_{T_4}) .52828 P_{t_4} A_4$$

Use of  $\gamma$  = 1.4 and P*/P_t = .52828 in the above equation imply an ideal gas. The factor G, derived from tabulated values in References 1 and 4, corrects the stream thrust from that of an ideal gas to that of a real gas.

If 
$$T_t < 560$$
°R,  $G = 1.00012 + 6.8338 \times 10^{-6} \times P_t$  (psia)  
If  $T_t > 560$ °R,  $G = 1.0044 - (4.196 - .0059 P_t) ( $T_t + 460$ )  $\times 10^{-6}$$ 

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 $^{\text{C}}_{\text{D}_4}$  has already been discussed;  $^{\text{C}}_{\text{T}_4}$  was calculated in an analogous manner,

$$C_{T_4} = 1 - 0.109 R_{N_4}^{-0.2}$$

This equation is a semi-empirical expression of the thrust coefficient of an ASME nozzle at a pressure ratio of  $\lambda$  = 1.8929 (corresponding to  $P^*/P_t$  = .52828). For the present tests,  $G_A$  varied from 1.000 to 1.0013, and  $G_{T_A}$  varied from 0.995 to 0.997.

The stream thrust at Station 1 was calculated as:

$$F_1 = G_1 (1 + 1.4 C_{D_1}C_{T_1}) .52828 P_{t_1}A_1$$

Each variable in this equation has been previously described, except  $^{C}T_{1}$ .  $^{C}T_{1}$  was calculated in a similar manner as  $^{C}T_{4}$ , but was modified to account for the thermal boundary layer described in the discussion of  $^{C}D_{1}$  in Section 4.1:

$$c_{T_1} = 1 - (0.109 \, {}^{R_N}_1) (0.828 + 0.172 \, {}^{T_{t_1}T_w})$$

The above equation was derived using the same assumptions as in the derivation of  ${}^{C}D_{1}$ .  ${}^{C}T_{1}$  for the present tests varied from 0.992 to 0.996.

Static pressures  $p_2$  and  $p_5$  were measured on mercury manometer. Ambient pressure  $(p_a)$  was measured on a Haas mercury manometer (barometer).  $A_5$  and  $A_2$ , the geometric reference areas for the seals, were both 3.5 in².

The vertical thrust,  $H_{\underline{Y}}$ , was calculated from the two vertical balance forces:

$$H_{y} = V_{1} + V_{3}$$

The resultant thrust,  $H_r$  was calculated as the vector sum of the axial thrust  $H_x$ , and vertical thrust  $H_y$ . Resultant thrust vector angle,  $\alpha$ , was determined as:

$$\alpha = \tan^{-1} \frac{\frac{H_{y}}{Y}}{H_{x}}$$

The sign convention for positive values of thrust components and vector angle is defined in Figure 6a. Note that for the present tests, the nozzle is mounted horizontally and there is no vertical thrust.

#### 4.5 Thrust Coefficient

The static thrust coefficient of an exhaust nozzle is defined as the ratio of the measured nozzle net thrust to the ideal thrust of the actual mass flow when expanded isentropically from  $P_{\mathbf{t}}$  to  $P_{\mathbf{a}}$ .

$$C_{\mathbf{T}} = \underbrace{H}_{\mathbf{m} \ \mathbf{v}_{\mathbf{i}}}$$

For the dual-flow tests, the ideal thrust was calculated as the sum of the fan nozzle ideal thrust and the core nozzle ideal thrust:

$$C_{\mathbf{T}} = \frac{H}{m_7 v_{i_7} + m_8 v_{i_8}}$$

For the present tests, thrust coefficients were calculated for the axial and vertical thrust components and for the resultant vector.

$$C_{\mathbf{T}_{\mathbf{X}}} = \frac{H_{\mathbf{X}}}{m_{7}v_{i} + m_{8}v_{i_{8}}}$$

$$c_{T_y} = \frac{H_y}{m_7 v_{i_7} + m_8 v_{i_8}}$$

$$C_{\mathbf{T_r}} = \sqrt{C_{\mathbf{T_x}}^2 + C_{\mathbf{T_y}}^2}$$

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Ideal thrust was calculated using a dimensionless ideal thrust function,  $m_i v_i/P_t \Lambda^*$ , which is a function of both  $\lambda$  and  $\gamma$ .

$$m_i v_{i_7} = (A^*/A)_7 C_{D_7 A_7}^P t_7 (m_i v_i/P_t A^*)_7$$
 $m_i v_{i_8} = (A^*/A)_8 C_{D_8 A_8}^P t_8 (m_i v_i; P_t A^*)_8$ 

where

$$(m_i v_i/P_t \Lambda^*) = \gamma \left[\frac{2}{\gamma + 1}\right]^{\frac{\gamma}{\gamma - 1}} \sqrt{\frac{\gamma + 1}{\gamma - 1}} \sqrt{1 - \gamma}$$
  
= 1.81162  $\sqrt{1 - \lambda^{-0.28571}}$ , for  $\gamma = 1.4$ .

For the present tests,  $\gamma_7$  was taken to be 1.400. However,  $\gamma_8 \neq$  1.4 and, therefore,  $(m_i v_i/P_t A^*)_8$  obtained from the above equation was corrected to account for "real gas effects" by multiplying by the ratio

$$\frac{(m_i v_i/P_t A^*) \text{ for real gas}}{(m_i v_i/P_t A^*) \text{ for } \gamma = 1.4}$$

This ratio was calculated from tabulated values in Reference 4; for the present range of test conditions this factor was obtained from a curve-fit expression

.9957 - 5.81 x 
$$10^{-5}$$
 x ( $^{\text{T}}$ t₈, °R - 1000) + 1.25 x  $10^{-3}$  x ( $\lambda_8^{-1}$ )

and varied between .9950 and .9972.

#### 4.6 Pressure and Temperature Data

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Pressure instrumentation for facility pressures and charging station pressures were described previously. All other pressures in the model were measured using multiple-tube mercury manometers. Model pressure data were reduced to absolute pressures (psia) and dimensionless ratios  $(P/P_{\tau_0}, P/P_{\tau_0})$ . The results are tabulated on computer output sheets, contained in the Appendix.

Facility and charging station temperature data were obtained using shielded chromel/alumel thermocouples, and were recorded on the facility Vidar system. Temperatures were expressed in °F or °R, or both.

#### 4.7 Force Balance Calibration

The force balance calibration determines the output characteristics of both the force balance flexures and the elastic seals which provide pressure-tight expansion joints between the metric model assembly and the non-metric facility structure. The outputs of the strain-gage flexures are very linear with applied load, but the seals provide an additional axial force which is a function of both axial load and seal pressure. Most of this force carryover results from radial seal deflections required to support the static pressure differentials across the seals when the ducts are pressurized. Consequently, the seal and balance assembly is calibrated under simulated operating conditions of loads and seal differential pressures. The calibration for this mixed flow facility is further complicated by the fact that the location of the applied load during a test is a function of the hot/cold flow split and nozzle pressure ratios; the calibration must, therefore, duplicate both the magnitude and location of the net force which was experienced during a test. As a result of these requirements, it has been found expedient to calibrate "on-point," that is, to determine the balance

output characteristics while simultaneously reproducing the forces, force location, and seal pressures experienced at a specific test point.

The forces and force location for each test point are not known exactly until the on-point calibration is completed. The initial test data are, therefore, reduced (by computer) using a preliminary calibration. The computer is programmed so that, as it reduces the initial test data, it also prints out the required calibration information (calibration load and location), such that an accurate on-point calibration can then be made.

The on-point calibration is made with the seals pressurized to the pressures measured during the test, and the loads applied at the locations obtained from the preliminary data. The loads are then varied slightly to obtain approximately the balance output measured during the test. The on-point calibration factors are then calculated as:

$$K_2 = \left[ (L_x + \Delta P_2 A_2 + \Delta P_5 A_5)/C_2 \right]$$
 $K_1 = V_1/C_1$ 
 $K_3 = V_3/C_3$ 

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The final balance forces for reducing test data were then calculated as

$$H_2 = K_2C_2$$
,  $V_1 = K_1C_1$ ,  $V_3 = K_3C_3$ ,

using the actual balance outputs measured during the test.

#### 5.0 PRESENTATION OF RESULTS

Test conditions and major test results are tabulated in Figure 7 and are plotted in Figures 8 through 10. Detailed data and calculations are contained in a separate Data Appendix.

The tabulation in Figure 7 includes: configuration, run number, actual values of the independent test variables  $(\lambda_7, \lambda_8, {}^Tt_8/{}^Tt_7)$ , the pressure ratio split  $(\lambda_7/\lambda_8)$ , and the major test results  $({}^Ct_1, {}^Ct_2, {}^Ct_3, {}^Ct_4, {}^Ct_5, {}^Ct_5, {}^Ct_6, {}^Ct_7, {}^Ct_8, {}^Ct_8, {}^Ct_9, {}^Ct_7, {}^Ct_8, {}^Ct_8, {}^Ct_9, {}^Ct_7, {}^Ct_8, {}^Ct_8, {}^Ct_8, {}^Ct_9, {}^Ct_7, {}^Ct_8, {}^Ct_8,$ 

#### 5.1 ASME Checkout Tests

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Standard ASME nozzles (Figures 2a and 5a) were tested to demonstrate facility data accuracy in determining  $C_{\rm D}$  and  $C_{\rm T}$  of test nozzles. Results of these tests are tabulated in Figure 7 (sheet 1) and are plotted in Figure 8.

Target performance curves for the ASME nozzles are shown in the figures. These predictions are based on semi-empirical equations, and were obtained by analysis of ASME nozzle exit surveys conducted by FluiDyne.

The test results were statistically analyzed in terms of bias (average difference between actual and predicted values) and scatter (standard deviation of the individual differences from their average). This analysis is summarized in the following table:

Runs	CD7	CD8	C _T	c _{D7}	c _{D8}	C _T
1 - 5	.0002	0010	0006	.0009	.0012	.0006
6 - 10	.0003	0008	.0004	.0014	.0018	.0017

Standard Deviation:

#### 5.2 Mixer Nozzle Tests

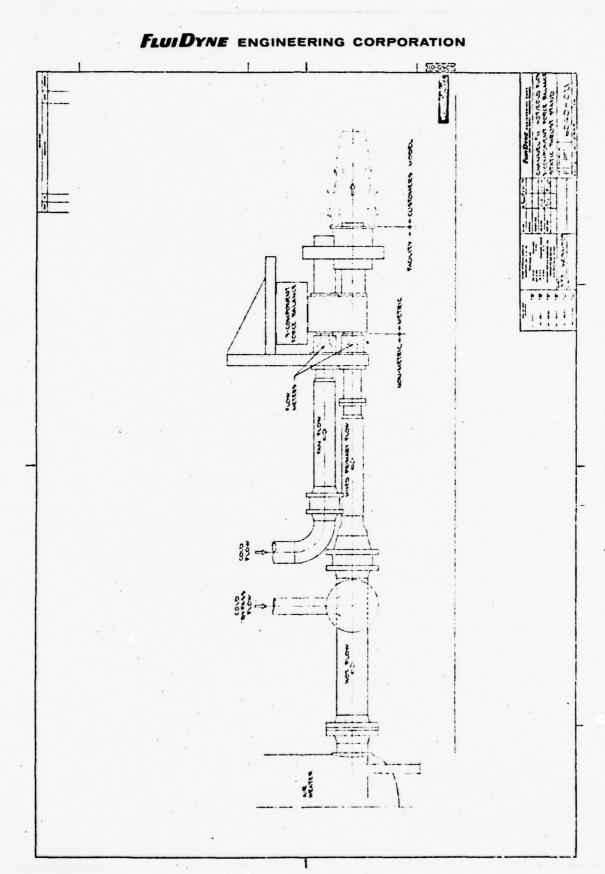
Major results from the mixer nozzle performance tests are tabulated in Figure 7, Sheets 2 and 3. Thrust Coefficients are plotted versus core nozzle pressure ratio in Figures 9a - 9c. Justification of high  $C_{\rm T}$  values ( $C_{\rm T} > 1.00$ ) for mixer nozzles was discussed in Reference 5; similar reasoning applies for the present test results.

Overall nozzle discharge coefficients,  $^{C}D_{9}$ , are plotted in Figures 10a - 10c. The abscissa is the mass-flow-weighted pressure ratio,  $\lambda = (W_{7}\lambda_{7} + W_{8}\lambda_{8})/(W_{7} + W_{8}\lambda_{8})$ . The ordering of the figures is the same as for the thrust coefficients in Figure 9.

In addition to the specified test program, Configuration 2A was tested at three other times (twice near middle and near the end of program). These additional tests demonstrated facility data repeatability.

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- 8. Kirschbaum, R. A., and Brasket, R. G., "Hot/Cold Flow Model TEsts to Determine Static Performance and Jet-Wing Surface Temperatures for 1/7-Scale JT8D-217 Mixer Exhaust Nozzles." FluiDyne Report 1067, Pratt & Whitney Agreement of 30 December 1975. March 1976.



FIGURE' 1. CHANNEL 11 FACILITY LAYOUT

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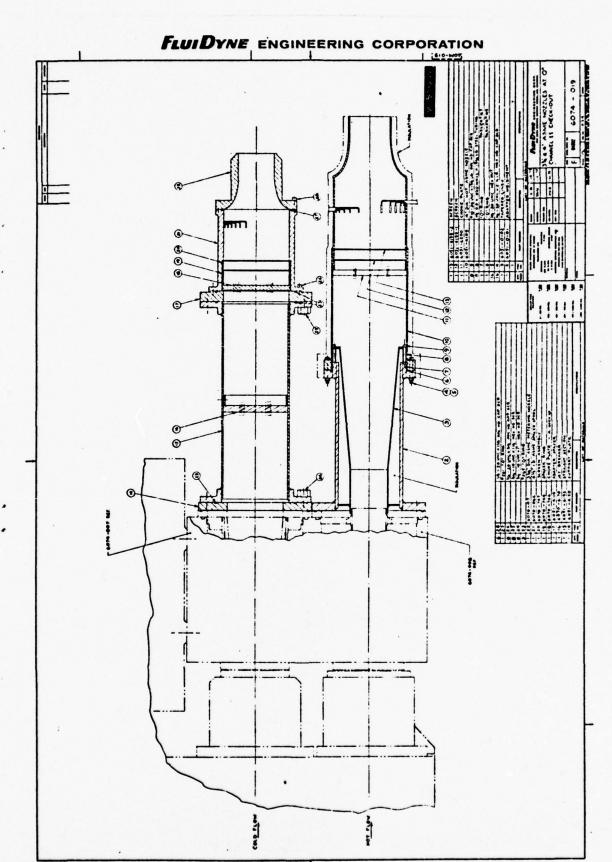


FIGURE 2a. ASME NOZZLE TEST ASSEMBLY DRAWING

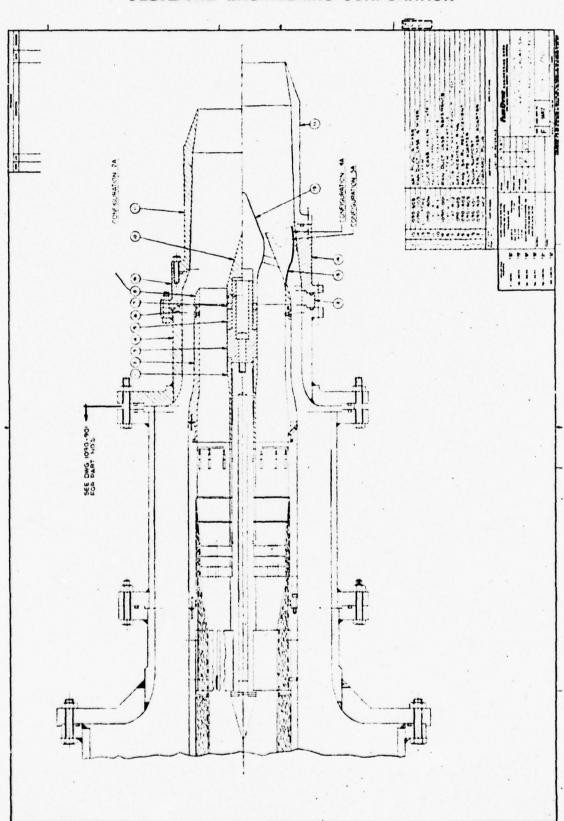
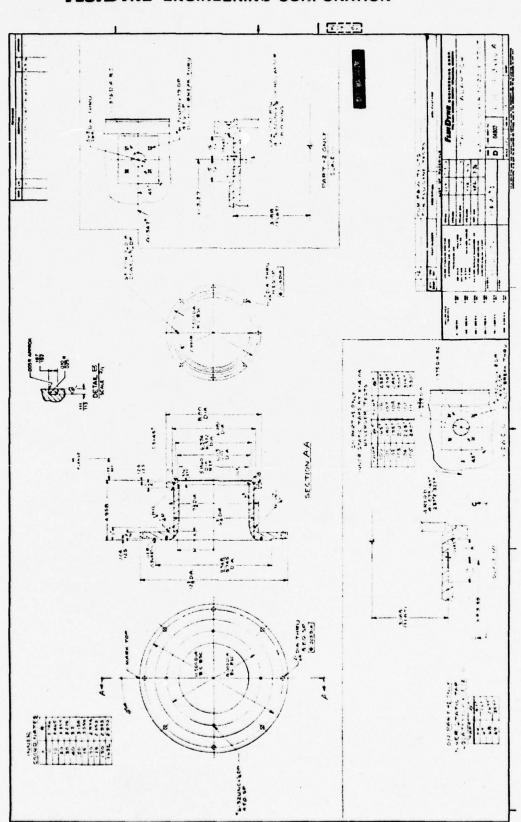


FIGURE 2b. MODEL ASSEMBLY - CONFICURATIONS 2A, 4A, 5A

Run Numbers	Config.	Splitter or Mixer	Tailpipe	Fan Duct Case	Swirl, TEGV, Tangential Struts, Fan Pt Distortion	Other
1-10	Dual ASME					
11-38, 214	2A	Reference Splitter	Common (401)	Reference (001)	No ·	
114-123	5A	<pre>16" Mixer, Cutback, Scalloped</pre>	Common (401)	16" Mixer (003)	No	1090-406 duct case spacer
124-153	4A	16" Mixer, Scalloped	Common (401)	16" Mixer (003)	No	1090-406 duct case spacer

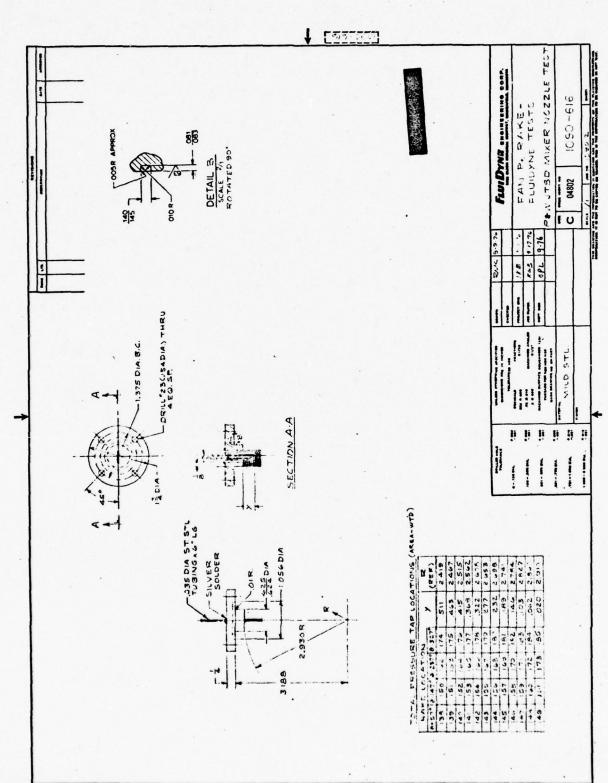
FIGURE 3. SUMMARY OF TEST CONFIGURATIONS

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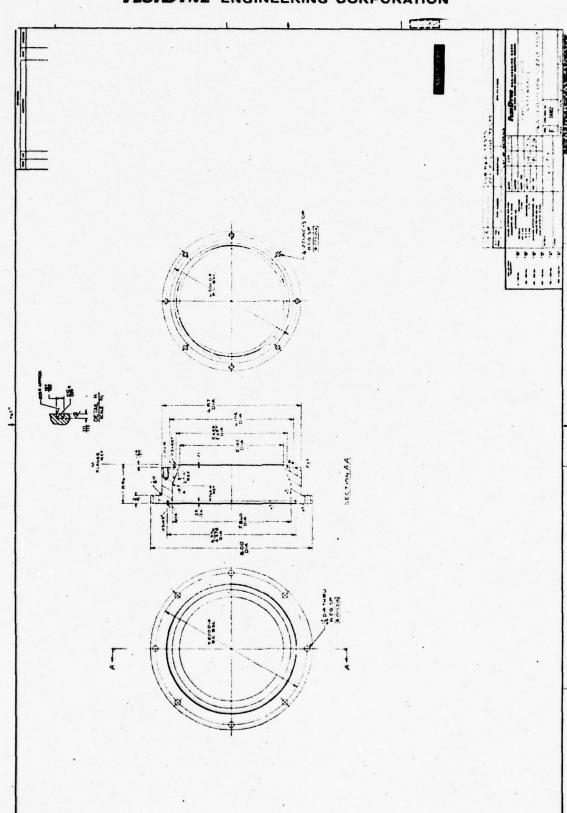
FIGURE 4a. TAILPIPE ADAPTER



FAN Pt RAKE FIGURE 4b.

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FIGURE 4c. FAN DUCT CASE-REFERENCE

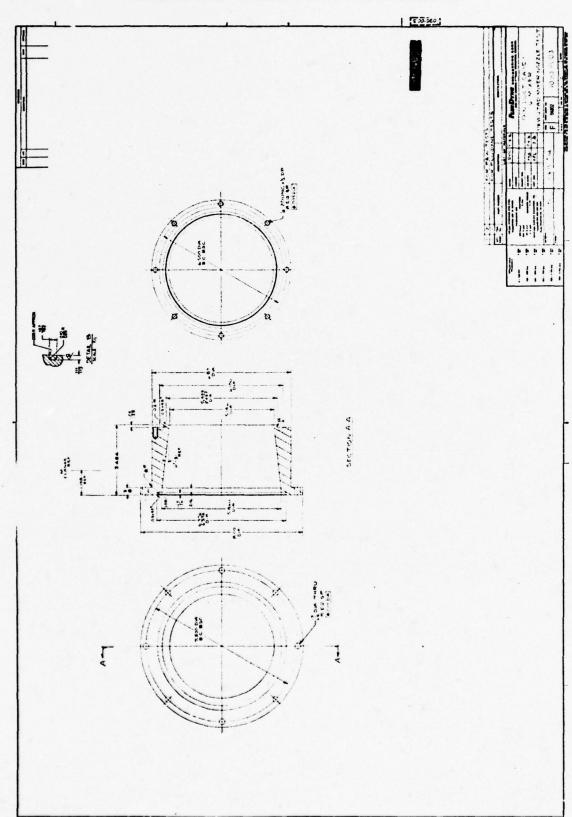


FIGURE 4d. FAN DUCT CASE - 16" MIXER

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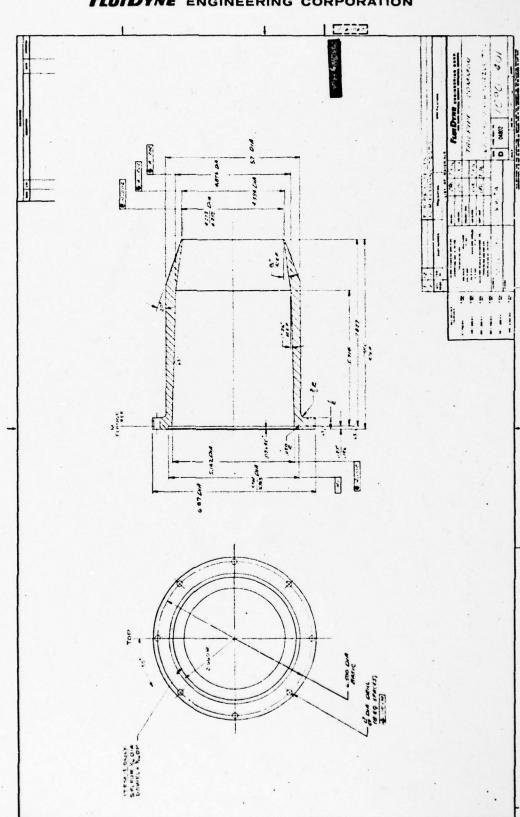


FIGURE 4e. TAILPIPE - COMMON

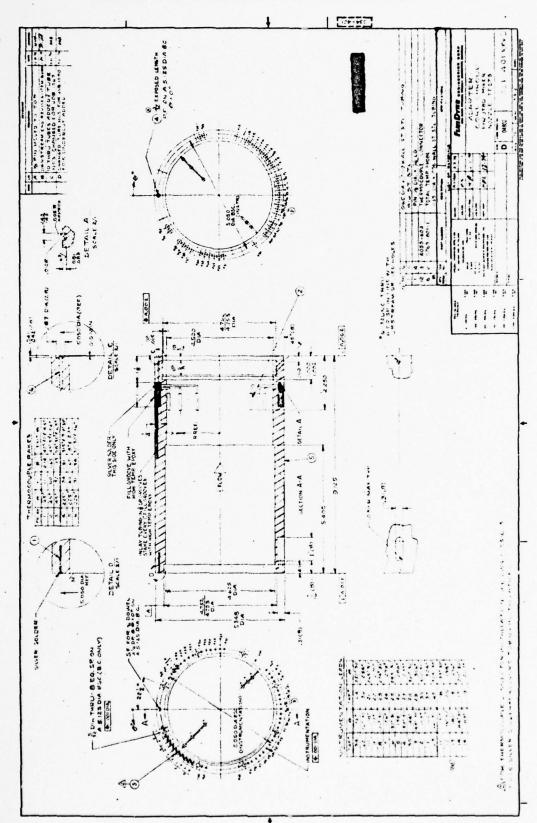
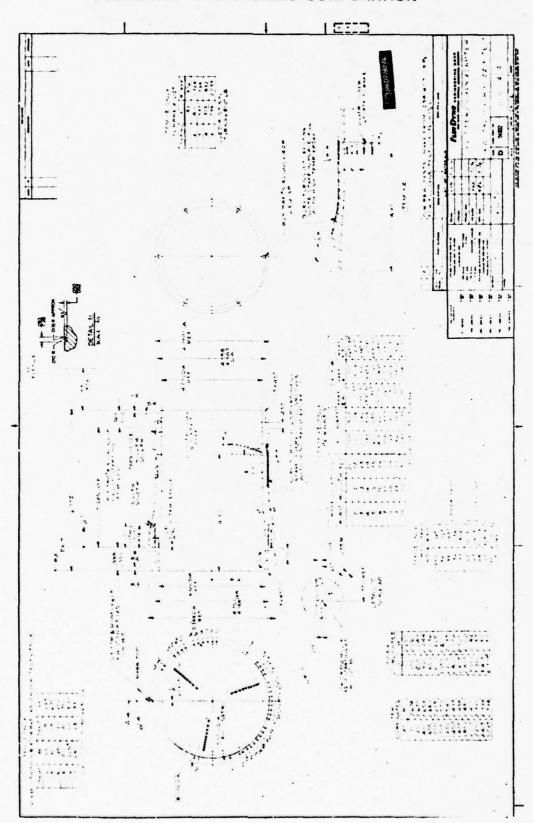


FIGURE 4f. CORE SHROUD ADAPTER



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FIGURE 49. SPLITTER/MIXER ADAPTER

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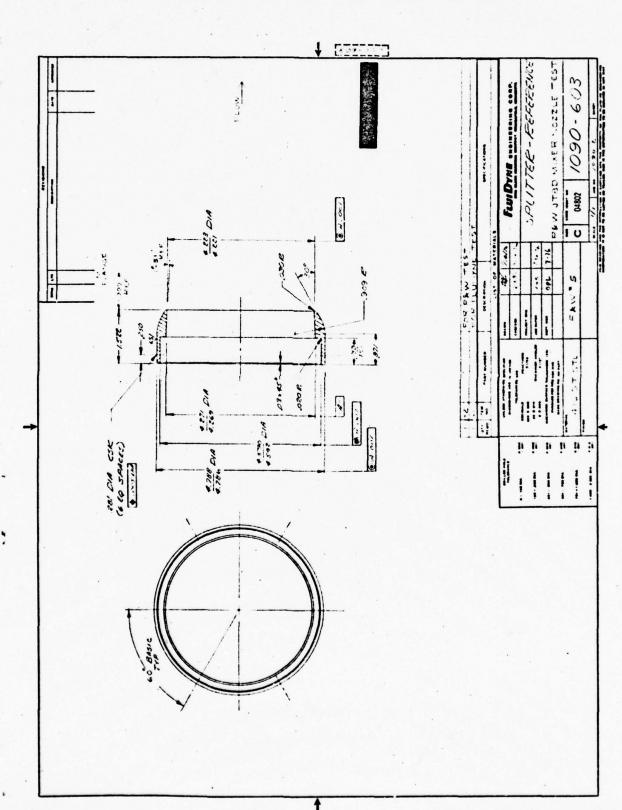


FIGURE 4h. SPLITTER - REFERENCE

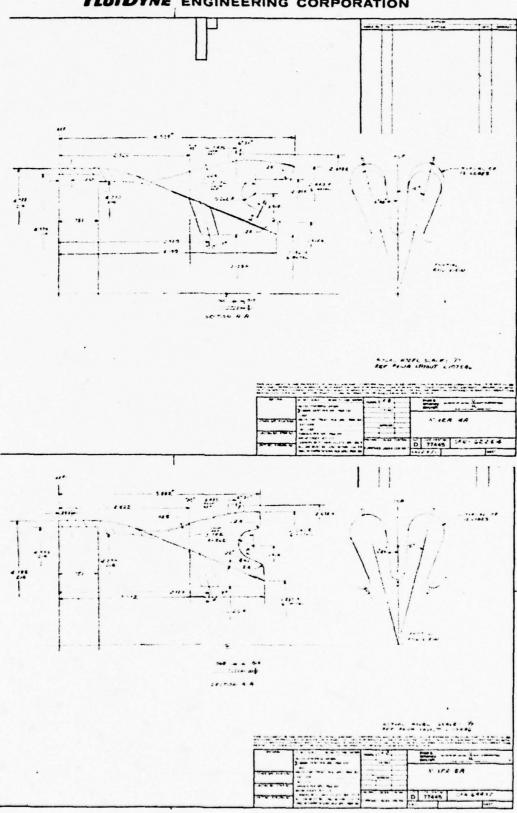


FIGURE 41. MIXER - CONFIGURATION 4A and 5A

11.1

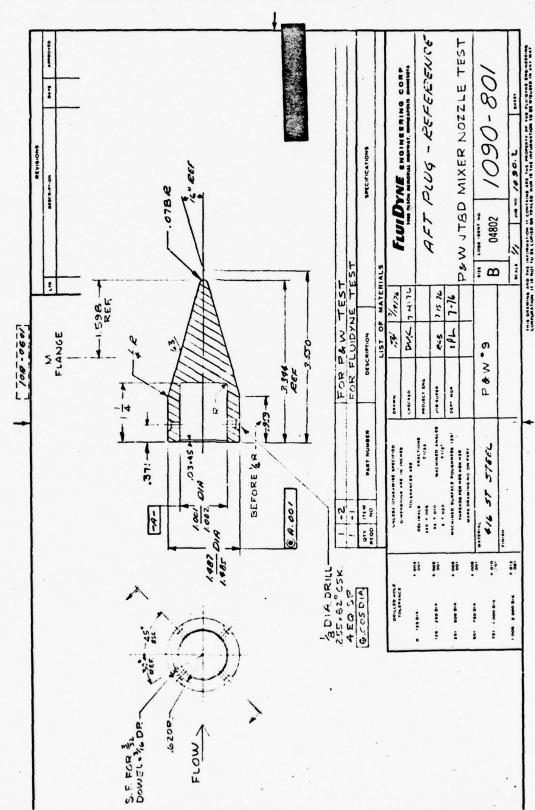


FIGURE 4j. AFT PLUG - REFERENCE

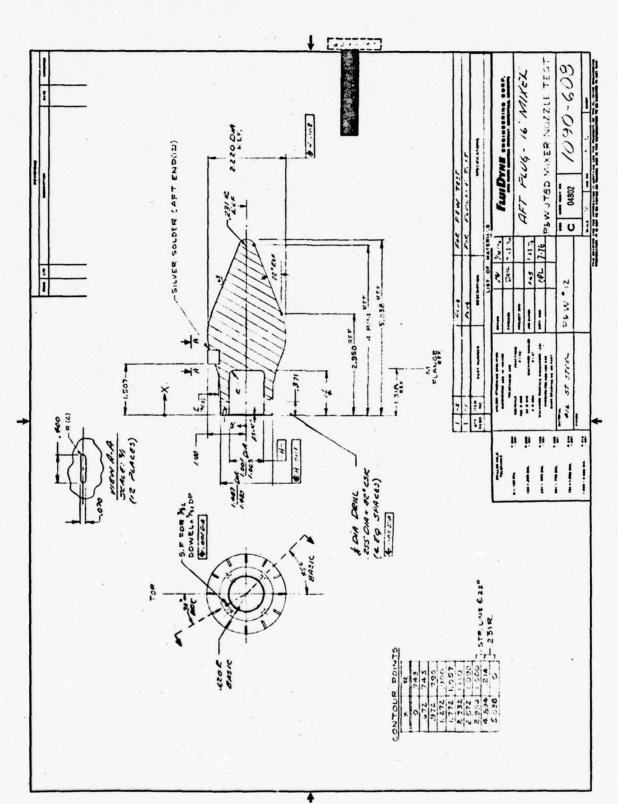
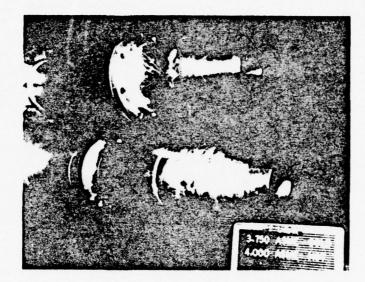
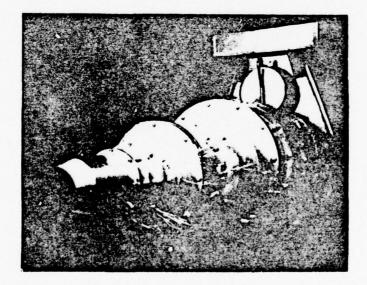


FIGURE 4k. AFT PLUG - 16" MIXER



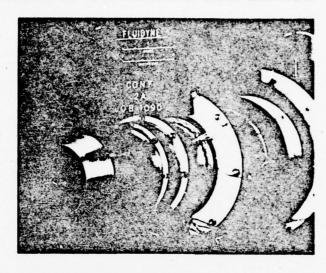
ASME nozzle installation

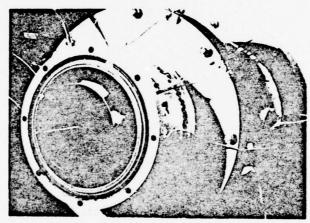


typical mixer nozzle installation

FIGURE 5a. MODEL PHOTOGRAPHS

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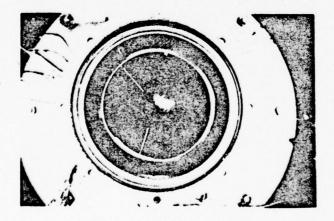
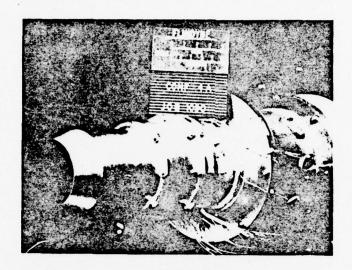
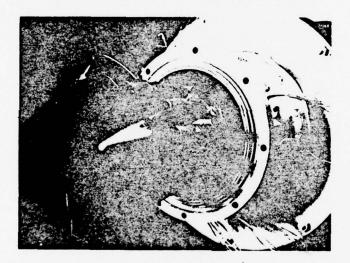


FIGURE 5b. MODEL PHOTOGRAPHS

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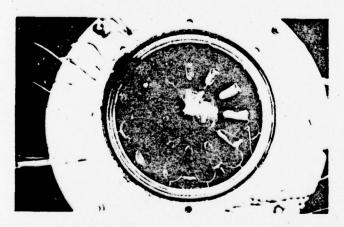
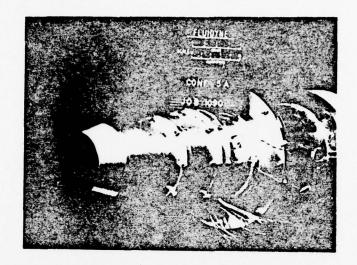
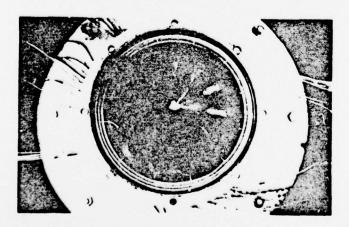


FIGURE 5c. MODEL PHOTOGRAPHS

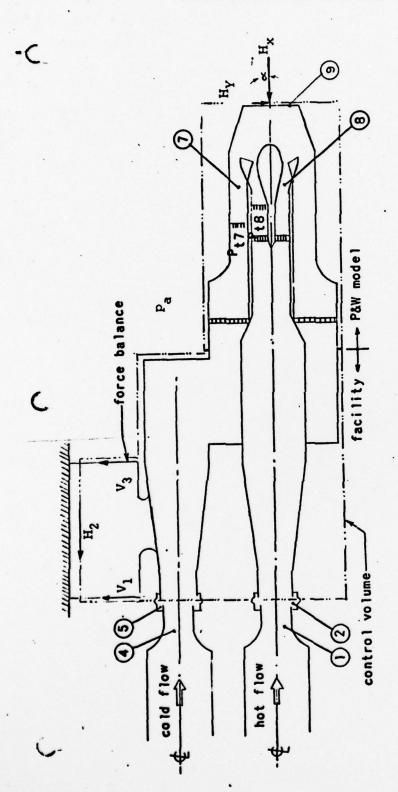
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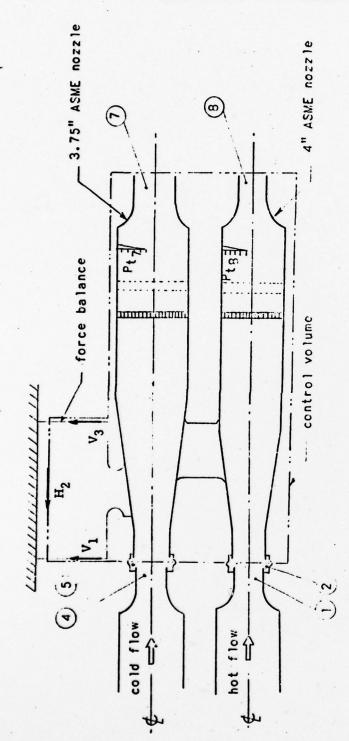
PIGURE 5d. MODEL PHOTOGRAPHS



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Description	ASME Meter Throat (core flow)	Flexible Seal (core flow)	ASME Meter Throat (fan flow)	Flexible Seal (fan flow)	Fan Nozzle	Core Nozzle	Nozzle Exit
Station	1	2	4	S	7	<b>6</b> 0	6

FIGURE 6a. STATION NOTATION, MIXER MODEL TESTS



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Description	ASME Meter Throat (core flow)	Flexible Seal (core flow)	ASME Mater Throat (fan flow)	Flexible Seal (fan flow)	3.75" Cold ASME Nozzle Throat	4" Hot ASME Nozzle Throat
Station	-	2	4	5	. 7	8

FIGURE 6b. STATION NOTATION, ASME CHECKOUT TESTS

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Config.	Run No.	λ ₇	λ ₈	Tt8/Tt7	, c _T	CD7	c ^{D8}			
	1.0	1.506	1.489	1.008	.9919	.9894	.9882			
3.75"	2.0	1.761	1.745	1.007	.9932	.9907	.9885			
ASME	3.1	2.010	1.985	1.006	.9944	.9904	.9893			
cold,	4.1	2.258	2.242	1.004	.9950	.9904	.9893			
upper	5.0	2.510	2.490	1.003	.9954	.9927	.9921			
&		1 504	1 500	2 400	0016	0000	0000			
4"	6.2	1.504	1.529	2.489	.9916	.9883	.9908			
ASME	7.2	1.762	1.752	2.483	.9914	.9895	.9890			
hot,	8.2	2.011	1.993	2.472	.9959	.9926	.9902			
lower	9.2	2.264	2.257	2.455	.9965	.9917	.9876			
	10.2	2.518	2.485	2.446	.9966	.9919	.9879			
Config.	Run No.	λ,	λ _s	λ _{7/} λ ₈	Tt8/Tt7	C _T	CD7A7	CD8y8	c _{D9}	W7/W8
2A	11.4	1.495	1.609	.929	.998	.9884	4.993	8.525	.951	.530
	12.3	2.065	2.214	.932	.997	.9929	5.279	8.636	.979	.569
	13.2	2.533	2.717	.932	.997	.9929	5.286	8.665	.982	.568
	14.2	3.011	3.236	.930	.997	.9886	5.245	8.695	.981	.560
	15.0	1.601	1.609	.995	1.000	.9888	5.889	7.622	.951	.767
	16.0						6.047			
	10.0	2.210	2.221	.995	.999	.9933	6.047	7.825	.976	.768
	17.1	2.403	2.714	.885	.998	.9932	4.681	9.304	.984	.445
	18.1	2.833	3.223	.879	.996	.9894	4.563	9.376	.981	.427
	19.1	1.490	1.617	.921	2.117	.9923	4.759	8.646	.943	.721
	20.1	2.057	2.195	.937	2.125	.9973	5.149	8.670	.973	.817
	21.0	2.481	2.698	.919	2.119	.9974	4.952	8.887	.974	.751
	122.0	2.934	3.204	.916	2.117	,9931	4,927	8.950	.977	. 739
	23.0	1.600	1.594	1.004	2,128	.9935	5.774	7.575	.940	1.124
	24.0	2.204	2.206	.999	2.130	.9967	5.885	7,891	.970	1.095
		2.204	2.200	. 333	2.130	. 3301	3.003	7,031	. 570	1.055
	25.1	2.402	2.712	.886	2.113	.9959	4.503	9.363	.976	.624
	26.1	2.832	3.209	.882	2.111	.9936	4.481	9.372	.975	.618
	27.1	1.493	1.620	.921	2.271	.9932	4.804	8.540	.939	.764
							to a contract			
	28.2	2.051	2.209	.928	2,265	.9977	5.024	8.752	.969	.809
	29.0	2.529	2.713	.932	2.257	.9966	5.066	8.749	.972	.818
	30.3	2.983	3,215	.928	2,264	.9949	5,016	8,827	.974	.800
	31.0	1.604	1.605	1.000	2.275	.9939	5,748	7.614	.940	1.146
	32.0	2.212	2.208	1.001	2.272	,9976	5,924	7.815	.967	1.154
	33.0	2.404	2.717	.885	2,256	.9986	4.467	9.395	.976	.638
	34.0	2.843	3.224	.882	2.269	.9932	4.453	9.428	.977	.633
	35.1	1.489	1,612	.924	2,387	.9949	4.747	8.652	.943	.766
	36.0	2.062	2,208	.934	2,404	.9998	5.089	8.729	.973	.852
							5.053	8,777	.973	.839
	37.0	2,533	2.719	.932	2,402	,9976	March 1981 1981 1981		.975	.853
	38.0	3.002	3,210	.935	2,392	,9960	5,106	8,741		
	214.0	1.668	1.823	.915	2.269	.9966	4.772	8.819	.954	.744

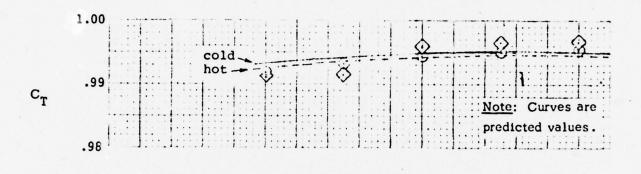
FIGURE 7. RUN SCHEDULE AND MAJOR TEST RESULTS (Sheet 1 of 3)

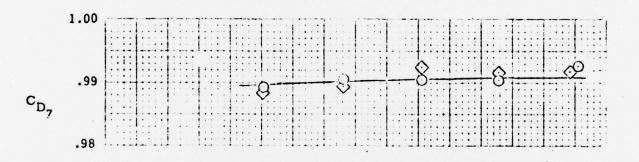
	Run										
Config.	No.	λ ₇	λ8	$\lambda_7/\lambda_8$	t ₈ / ^T t	7 ^C r	CD7A7	CD8y8	C _{D9}	W7/W8	
5A	114.0	1.492	1.606	.929	.992	.9767	5.535	7.713	.932	.646	
	115.0	1.676	1.790	.932	.992	.9789	5.633	7.756	.942	.668	
	116.0	2.056	2.199	.935	.993	.9851	5.788	7.851	.960	.687	
	117.0	2.512	2.685	.936	.993	.9875	5.829	7.894	.966	.688	
	118.0	2.982	3.185	.936	.989	.9871	5.824	7.890	.965	.687	
	119.0	1.488	1.599	.931	2.286	.9819	5.252	7.625	.906	.950	
	120.0	1.675	1.802	.930	2.280	.9881	5.339	7.693	.917	.973	
	121.0	2.052	2.197	.934	2.272	.9964	5.508	7.825	.938	.999	
	122.0	2.516	2.696	.933	2.275	1.0009	5.529	7.856	.942	.999	
	123.0	2.988	3.192	.936	2.282	1.0016	5.551	7.832	.942	1.011	
4A	124.0	1.496	1.607	.931	1.000	.9764	5.473	8.059	.947	.616	
	125.0	2.043	2.194	.931	1.000	.9871	5.602	8.251	.969	.632	
	126.0	2.517	2.687	.937	1.000	.9881	5.641	8.220	.970	.643	
	127.0	2.990	3.181	.940	1.000	.9862	5.669	8.218	.971	.649	
	128.0	1.603	1.605	.999	.998	.9753	5.869	7.650	.946	.765	
	129.0	2.200	2.190	1.004	.997	.9864	6.010	7.793	.966	.773	
	130.0	2.383	2.691	.885	.993	.9868	5.355	8.513	.970	,555	
	131.0	2.818	3.181	.886	.993	.9846	5.385	8.537	.974	.557	
	132.0	1.499	1.608	.933	2,159	.9798	5.290	7.900	.923	.900	
	133.0	2.056	2.198	.935	2.160	.9926	5.481	8.087	.949	.938	
	134.0	2.512	2.698	.931	2.134	.9972	5.458	8.137	.951	.919	
	135.0	2.992	3.190	.938	2.135	.9968	5.482	8.053	.947	.940	
	136.0	1.617	1.614	1.002	2.139	.9784	5.724	7,567	.930	1.115	
	137.0	2.201	2.189	1.006	2.146	.9919	5.902	7.696	.951	1.138	
	138.0	1.499	1.602	.935	2.325	.9792	5.300	7.838	.919	.948	
	139.0	2.051	2.186	.938	2.285	.9940	5.494	8.039		.977	
	140.0	2.513	2.689	.935	2.288	1.0008	5.475	8.071	.948	.968	
	141.0	2.988	3.177	.941	2.294	1.0004	5,492	8.097	.951	.975	
	142.0	1.605	1.605	1.000	2.296	.9808	5.690	7.486	,922	1.159	
	143.0	2.190	2.190	1.000	2.285	.9945	5.845	7.681	.946	1.160	
	144.0	2.375	2.697	.881	2.281	1.0005	5.107	8.367	.943	.819	
	145.0	2.820	3.178	.837	2,277	1.0010	5.154	8.398	.948	.829	

FIGURE 7. RUN SCHEDULE AND MAJOR TEST RESULTS (Cont.) (Sheet 2 of 3)

Config.	Run No.	λ ₇	λ ₈	λ ₇ /λ ₈	Ttg/Tt	C _T r	CD7A7	CD848	с _{Б9}	w ₇ /w ₈
4A(cont.)	146.0	1.485	1.597	.930	2.411	.9828	5.248	7.867	.917	.943
	147.1	2.052	2.211	.928	2.386	.9988	5.315	8.067	.940	.954
	148.1	2.509	2.714	.924	2.382	1.0024	5.332	8.117	.945	.946
	149.0	2.983	3.187	.936	2.400	1.0056	5.467	8.078	.948	.991
	150.0	2.380	2.692	.884	2.420	1.0003	5.105	8.351	.941	.849
	151.0	2.816	3.185	.884	2.422	1.0015	5.105	8.390	.944	.845
	152.0	2.695	2.698	.999	2.283	1.0006	5.847	7.705	.948	1.155
	153.0	2.695	2.686	1.003	2.145	.9963	5.880	7.700	.950	1.130

Symbol	T _{t8} /T _{t7}	Run Nos.		
0	1.000	1-5		
$\Diamond$	2.470	6-10		





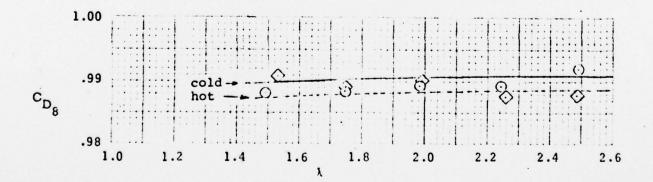
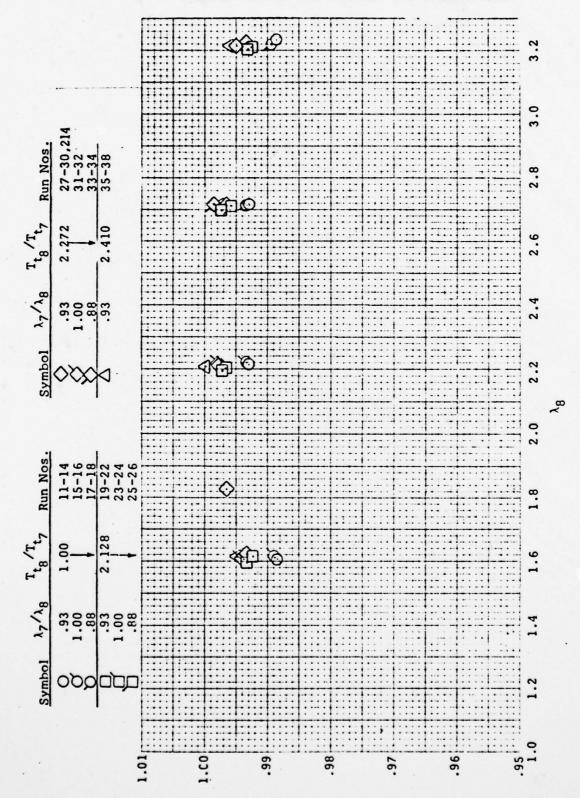


FIGURE 8. THRUST AND DISCHARGE COEFFICIENTS, ASME CHECKOUT TESTS



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FIGURE 9a. THRUST COEFFICIENTS, CONFIGURATION 2A

Tt /Tt Run Nos.

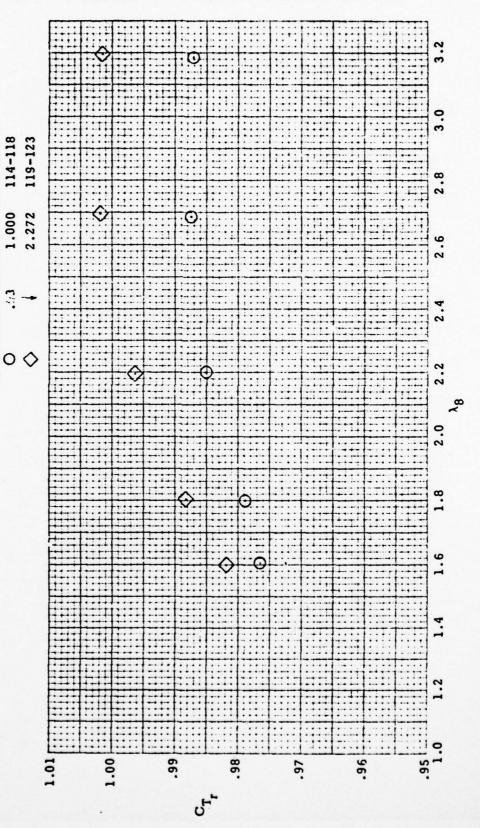


FIGURE 9b. THRUST COEFFICIENTS, CONFIGURATION SA

1. 1

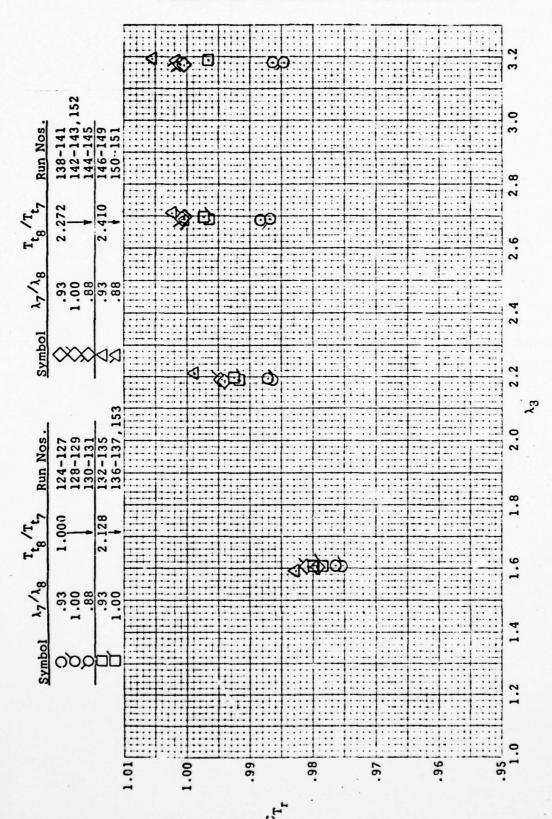
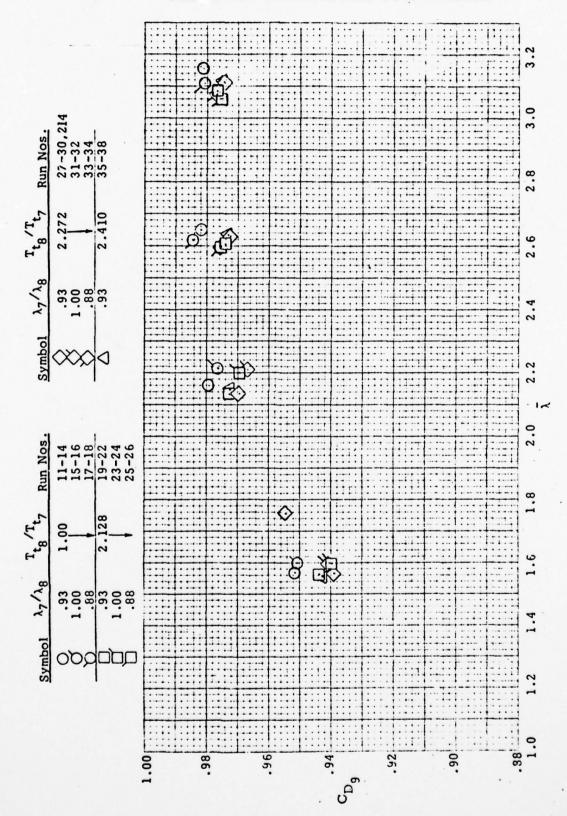


FIGURE 9c. THRUST COEFFICIENTS, CONFIGURATION 4A

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FIGURE 10a. NOZZLE DISCHARGE COFFICIENTS, CONFIGURATION 2A

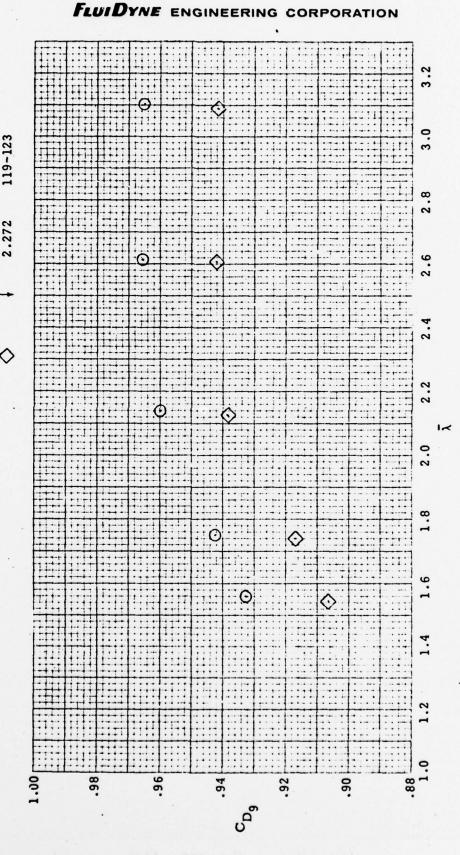
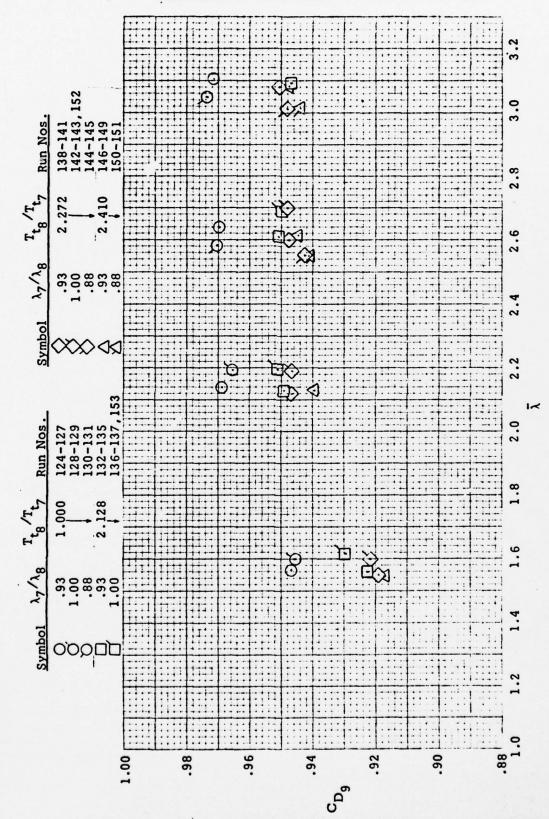


FIGURE 10b. NOZZLE DISCHARGE COEFFICIENTS, CONFIGURATION SA

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FIGURE 10c. NOZZLE DISCHARGE COEFFICIENTS, CONFIGURATION 4A